# FATIGUE EVALUATION OF COMPOSITE REINFORCED, INTEGRALLY STIFFENED METAL PANELS

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#### FATIGUE EVALUATION OF COMPOSITE - REINFORCED, INTEGRALLY STIFFENED METAL PANELS

bу

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#### SUMMARY

This report covers the investigation of the fatigue behavior of composite-reinforced, integrally stiffened metal panels. The structural behavior resulting from combining metal and composite materials and subjecting them to fatigue loading was examined. The detail program results are presented in this report and the program summary and conclusion (ref. 1) give a brief description of the program and its findings.

This work made use of existing materials and processes. The material systems investigated are listed below.

- . Aluminum-Graphite/Epoxy
- . Aluminum-S Glass/Epoxy

Two adhesives were studied to determine the effects of a room temperature cure adhesive and an elevated temperature cure adhesive on the aluminum-graphite system.

The concept of unidirectional composite material sandwiched between aluminum sheet material was examined. Basic composite-metal coupon static and fatigue data was generated and evaluated to determine system fail-safeness and possible weight savings over conventional metal systems. Fatigue metal crack growth in the composite-metal systems was investigated and compared to crack growth in all metal systems. Also the metal crack growth of the aluminum-graphite system was compared to the aluminum-glass system.

After a significant amount of fatigue loading and metal cracking, the panels were pulled statically to failure to determine the residual strength. The results showed that sufficient strength remained to effectively transfer the load from the cracked metal into the composite material and retain a high percentage of its original strength.

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#### INTRODUCTION

An effective fail-safe structural concept which has application for present and future aircraft systems, such as the space shuttle and advanced technology transports, has been evaluated in this research effort. The all-metal integrally-formed panel concept has shown considerable potential for improving aerospace structures in terms of low manufacturing cost, high-strength capability, low weight, and inherent fail-safe compression failure characteristics (ref. 2). The inclusion of advanced unidirectional composite material between the two sheets of the integrally-formed panel provides for further improvements in reduced metal fatigue crack growth, and increased residual strength at no increase in weight.

The integrally-formed structure consists of two sheets of material, an inner stiffener sheet bonded to an outer face sheet. This panel configuration offers a significant increase in the strength-weight index as compared to the conventional riveted Z panels of the same material and design conditions. By inserting advanced unidirectional composite material between the inner and outer sheet materials before bonding, the composite-reinforced, integrally-stiffened metal panel concept is obtained. This structural concept is illustrated in Figure 1. A panel fabricated in this manner has several advantages over the all-metal and all-composite panels.

This program investigated this unique structural concept using the all-metal integrally-formed panel as a base. Coupon specimens were fabricated and tested in static tension and fatigue to characterize the features of the composite-metal system. Fifteen composite-reinforced, integrally-stiffened metal panels were fabricated and tested in tension-tension fatigue to investigate the metal crack growth rates and panel residual strengths. Two composite-metal systems were examined, 7075-T6 aluminum-graphite/epoxy and 7075-T6 aluminum-S glass/epoxy. The material design limit stresses were taken as 2/3's of the ultimate design allow-ables. Two adhesives were used to bond the aluminum-graphite system together to evaluate the effects of an elevated cure temperature adhesive, since an elevated curing temperature induces residual thermal stresses in the aluminum. The adhesives used in this investigation were AF-126 (394°K cure (250°F)) and EA-927R (room temperature cure).

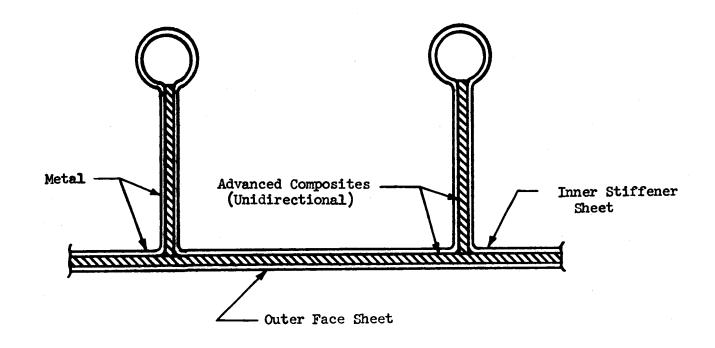


Figure 1 Composite-Reinforced, Integrally Stiffened Metal Panel Structural Concept

#### SYMBOLS

The physical quantities in this report are given in both the International System of Units (SI) and in the U.S. Customary Units. The SI units are stated first and the Customary units afterwards, in parenthesis. All principal measurements and calculations were made in the U.S. Customary Units. Appendix A presents factors relating these two systems of units.

```
total crack length, mm (in.)
2a
             total gross cross-sectional area, mm<sup>2</sup> (in<sup>2</sup>)
Α
             modulus of elasticity, MN/m<sup>2</sup> (ksi)
Ε
             gross stress, MN/m<sup>2</sup> (ksi)
f
             gross allowable stress. MN/m<sup>2</sup> (ksi)
F
             model stress fringe value for photostress plastic, MN/m2n (ksi/n)
F_{\sigma}
             model strain fringe value for photostress plastic,
\mathbf{F}_{\boldsymbol{\epsilon}}
                 mm/mmn ( in/in n)
             stress concentration
K<sub>+</sub>
N
             number of cycles of load
             photoelastic fringe order
n
             load, N (lbf)
Ρ
             ratio of minimum to maximum values of cyclic load
R
             thickness, mm (in.)
t
             weight, kg (lbm)
W
\epsilon
              strain
μ
             ratio of composite stiffness to total stiffness
             density, kg/m<sup>3</sup> (lbm/in.<sup>3</sup>)
Subscripts
             all metal
am
             composite
```

composite-metal

c-m

m metal

max maximum

t total

tl tension limit

tu tension ultimate

#### DESIGN CRITERIA

In order to have a degree of fail-safety and not suffer a weight penalty, a design criteria is necessary to provide the sufficient amount of unidirectional composite in the composite-reinforced metal concept. Too much metal will impose a weight penalty and not provide an efficient load-carrying structure. On the other hand, too much unidirectional composite in the system will not work the metal sufficiently to justify its presence as structure. Therefore, establishment of a satisfactory design rationale is a significant part of the composite-reinforced metal structure concept.

The design criteria established in this investigation was that:

- (1) The composite material would support the total load at limit stress after the metal had completely failed.
- (2) The weight of the composite-metal system would be equal to the weight of an all-metal system which would carry the same total load at limit stress.

Throughout the investigation, it was assumed that the strains associated with the two materials bonded together were equal, i.e., the two materials elongate the same amount under load. Therefore, the strain in each material is

$$\epsilon = \frac{P_{m}}{A_{m}E_{m}} = \frac{P_{c}}{A_{c}E_{c}}$$
 (1)

If we define a stiffness ratio " $\mu$ " for the composite-metal system as

$$\mu = \frac{A_c E_c}{A_c E_c + A_m E_m}$$
 (2)

then the total load on the composite-metal system would distribute as

$$P_{c} = \mu P_{t}$$
,  $P_{m} = (1-\mu)P_{t}$  (3)

Consider the weight of a composite-metal structure, Wc-m,

$$W_{c-m} = \rho_c A_c + \rho_m A_m \tag{4}$$

and the weight of an all-metal structure, Wam,

$$W_{am} = \rho_{mam} \tag{5}$$

for a unit length. Setting the weight of the composite-metal system equal to the weight of the all-metal structure we obtain

$$\rho_{c}A_{c} + \rho_{m}A_{m} = \rho_{m}A_{am} \tag{6}$$

or

$$\frac{A_{m}}{A_{c}} = \frac{A_{am}}{A_{c}} - \frac{\rho_{c}}{\rho_{m}}$$
 (7)

Substituting (7) into (2),

$$\mu = \frac{1}{1 + \frac{E_m}{E_c} (\frac{A_{am}}{A_c} - \frac{\rho_c}{\rho_m})}$$
(8)

To satisfy condition (1) of the design criteria, the composite must carry the total load at limit stress if the metal fails,

$$A_{c} = \frac{P_{t}}{F_{t\ell_{c}}}$$
 (9)

To satisfy condition (2) of the design criteria, the all-metal system must carry the total load at limit stress, therefore,

$$A_{am} = \frac{P_t}{F_{t\ell_m}}$$
 (10)

Substituting (9) and (10) into (8), the stiffness ratio becomes

$$\mu = \frac{\frac{1}{1 + (\frac{E_{m}}{E_{c}})(\frac{F_{tl_{c}}}{F_{tl_{m}}} - \frac{\rho_{c}}{\rho_{m}})}$$
(11)

Therefore, knowing the density, modulus of elasticity, and allowable limit stress of a particular composite material and metal, the correct amount of composite and metal can be determined which would satisfy the design criteria conditions (1) and (2) above.

Using the material design properties given in Table I, the stiffness ratio  $\mu$  for the aluminum-graphite and aluminum-glass systems can be determined using (11). Thus, for the aluminum-graphite system  $\mu$  = 0.57 and for the aluminum-glass system  $\mu$  = 0.29.

TABLE I

DESIGN PROPERTIES OF
METAL AND COMPOSITE MATERIALS

MATERIAL	LIMIT S	TRESS	TENSION	MODULUS	DENSITY		
	MN/m <sup>2</sup>	(ksi)	MN/m <sup>2</sup>	(ksi)	kg/m <sup>3</sup>	(lbm/in <sup>3</sup> )	
7075-T6 Aluminum	359	52	71,000	10,300	2768	0.100	
Graphite/Epoxy (HT/S)	696	101	131,000	19,000	1550	0.056	
S-Glass Epoxy (1009-268)	986	143	61,000	8,900	1882	0.068	

#### ADHESIVE STRENGTH

#### Objective

The objective of this portion of the program was to qualify the double lap shear strength of the bonding adhesives to be used.

#### Approach

Two types of adhesives were evaluated, an elevated temperature and a room temperature cure adhesive. The elevated temperature cure adhesive produces thermal stresses since the coefficient of thermal expansion of the metal is greater than that of the composite material. The adhesive shear strengths were evaluated in double lap shear tests for the same material combinations as used for the coupon and panel specimens.

#### Test Specimens

The specimens used in this evaluation consisted of standard double lap shear specimens as shown in Figure 2. Material combinations of aluminum-aluminum, aluminum-graphite, and aluminum-glass were examined. The adhesives tested were AF-126, 394°K (250°F) cure, EA-927R, room temperature cure, and EC-1614, room temperature cure.

#### Testing and Results

The double lap shear specimens were tested in static tension until failure. The failure load of each specimen was recorded and the lap length and width were measured. The shear strength was calculated using the failure load and measured shear area of the lap joint. A summary of the test data in terms of shear strength is presented in Table II.

#### Discussion

Three adhesives were selected for evaluation for bonding aluminum-aluminum, aluminum-graphite/epoxy and aluminum-glass/epoxy systems: AF-126, EA-927R, and EC-1614. (See Appendix B). These adhesives were selected because of their good shear strength and contain both elevated and room temperature cure characteristics. Because of the large differences in the thermal expansion between the aluminum and graphite/epoxy, high residual thermal stresses may be induced into the system and reduce the bond strength between the composite and metal. The elevated temperature cure adhesive evaluated was the AF-126 and the room temperature cure adhesives evaluated were the EA-927R (film adhesive) and EC-1614 (paste adhesive).

The shear strength of the AF-126 adhesive was evaluated with aluminum-aluminum adherends, aluminum-graphite adherends, and aluminum-glass adherends. The shear strength of the EA-927R and EC-1614 were evaluated with aluminum-graphite adherends.

Table II shows the results of the shear strengths. The AF-126 values are much higher than the room temperature cure adhesives and are equivalent to strengths obtained in present aircraft designs. The aluminum-aluminum adherends gave higher shear strength values than the aluminum-composite adherends. The two room temperature cure adhesives gave the same shear strength values for the aluminum-graphite adherends.

#### Conclusions

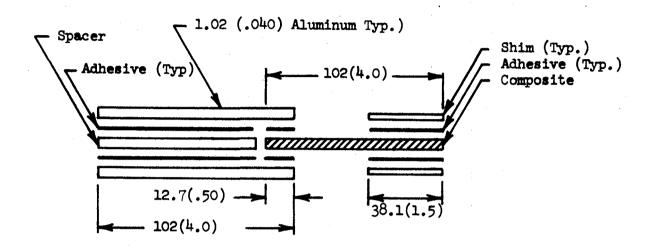
The shear strength of the adhesives tested to bond the aluminum-composite coupons and integrally stiffened panels was equivalent to present values used in aircraft design. The AF-126 (elevated temperature cure) adhesive would be used to bond the aluminum-aluminum, aluminum-graphite, and aluminum-glass systems.

The EC-1614 paste adhesive could not be used to bond the composite-reinforced, integrally stiffened metal panels because of the complexity of the panel configuration. Therefore, the EA-927R (room temperature cure) adhesive would be used to bond the aluminum-graphite system which would be compared to the aluminum-graphite system with the elevated temperature cure adhesive. The EC-1614 (room temperature cure) adhesive would be used to bond the panel end fittings to the panel.

TABLE II

ADHESIVE SHEAR STRENGTH
(Double Lap Specimens)

MATERIAL COMBINATION	ADHESIVE	LAP LENGTH		WIDTH		AREA		2 X AREA		FAILURE LOAD		SHEAR STRESS	
COMPLICATION		mm	(IN.)	mm	(IN.)	mm <sup>2</sup>	(IN. <sup>2</sup> )	mm <sup>2</sup>	(IN. <sup>2</sup> )	N	(lbf)	MN/ <sub>m</sub> 2	(ksi)
		13.08	.515	25.17	•991	329.2	.510	658.4	1.02	26,466	5,950	40.22	5.833
AlumAlum.	AF-126	<b>1</b> 2.95	.510	25.45	1.002	329.6	.511	659.2	1.02	26,510	5,960	40.29	5,843
		12.83	•505	25.15	•990	322.7	•504	645.3	1.01	27,533	6,190	42.26	6.129
											AVG.	40.92	5•935
		12.70	.500	25.65	1.01	325.8	•505	651.5	1.01	19,171	4,310	29.42	4.267
AlumGraphite	AF-126	12.95	-510	25.40	1.00	328.9	.510	657.9	1.02	21,528	4,840	32.72	4.745
		12.70	•500	25.65	1.01	325.8	.505	651.5	1.01	19,660	4,420	30.17	4.376
											AVG.	30.77	4.463
		12.70	•500	25.40	1.00	322.6	.500	645.2	1.00	20,906	4,700	32.41	4.700
AlumGlass	AF-126	12.70	•500	25.40	1.00	322.6	•500	645.2	1.00	20,238	4,550	31.37	4.550
		12.95	.510	25.65	1.01	332.2	.515	664.3	1.03	22,151	4,980	33.34	4.835
											AVG.	32.37	4.695
		12.45	.490	25.65	1.01	319.3	•495	638.7	0.99	12,365	2,780	19.36	2.808
AlumGraphite	EA-927R	12.32	•485	25.65	1.01	316.0	<b>.</b> 490	632.0	0.98	12,855	2,890	20.33	2.949
		12.19	<b>.</b> 480	25.65	1.01	312.7	<b>.</b> 485	625.3	0.97	11,787	2,650	18.84	2.732
											AVG.	19.51	2.830
		13.21	•520	25.65	1.01	338.8	•525	677.7	1.05	13,833	3,110	20.42	2.962
AlumGraphite	EC-1614	13.34	•525	25.40	1.00	338.8	.525	677.7	1.05	13,789	3,100	20.35	2.952
		12.95	•510	25.65	1.01	332.2	.515	664.3	1.03	12,499	2,810	18.81	2.728
											AVG.	19.86	2.881



- NOTE: 1) All dimensions are millimeters and inches, respectively.
  - 2) Graphite Composite 6 plies thick
  - 3) Glass Composite 15 plies thick
  - 4) Specimen width is 25.4 mm(1:0 in)

FIGURE 2 DOUBLE LAP SHEAR SPECIMEN

#### COUPON STATIC TENSILE TESTS

#### Objective

The objective of this task was to conduct static tensile tests to obtain basic coupon data for composite-reinforced metal structure.

#### Approach

Combinations of graphite reinforced aluminum and glass reinforced aluminum coupon specimens bonded with elevated and room temperature cure adhesives were tested in static tension and compared to all aluminum coupon results. Strain gages were mounted on the aluminum portion of the specimens so that the load-strain data could be compared directly to the load-strain data for the all aluminum coupon specimens. Smooth specimens ( $K_t = 1.0$ ) and specimens with a hole ( $K_t > 1.0$ ) were tested and compared to evaluate the effects of holes on composite-reinforced metal structure.

#### Test Specimens

The specimens used to conduct the static tensile tests were as shown in Figure 3. The smooth tensile coupons were instrumented with one axial strain gage located on the longitudinal centerline at the minimum cross section. A stress concentration factor ( $K_t > 1.0$ ) was obtained by drilling and reaming a 3.2 mm (.125 in.) diameter hole on the longitudinal centerline at the minimum width. These specimens were instrumented with a strain gage on either side of the hole at the minimum cross section on one aluminum surface only.

Sixteen static tensile coupon tests were conducted involving all aluminum, aluminum-graphite/epoxy, and aluminum-glass/epoxy with adhesives AF-126 and EA-927R. Table III presents the identification of the test specimens.

#### Testing and Results

The dimensions of each composite and metal thickness of each coupon specimen were measured at the minimum section and recorded before bonding the composite material to the aluminum and after bonding and machining. After recording the dimensions a tensile load was applied at a cross head travel rate of 1.27 mm/min. (0.05 in./min.) to obtain a load-strain curve through ultimate failure for each specimen. The plot of total load versus strain for each of the specimens, 1A through 16A, is presented in Figures 4 through 19. Each specimen critical cross section dimensions and failure load is presented in Table IV.

#### Discussion

The ultimate tensile strengths in Table IV were calculated using the total coupon area  $(A_C + A_M)$  minus the area of the hole. The strengths of the all-aluminum coupons were essentially the same. The strengths of the aluminum-graphite coupons with holes were about 7 percent less for the EA-927R adhesive and about 19 percent less for the AF-126 adhesive than coupons without holes. The strengths of the aluminum-glass coupons with holes were about 21 percent less than coupons without holes. The larger reduction in strengths for the AF-126 adhesive in the aluminum-graphite coupons could be due to residual thermal stresses.

#### Conclusions

The composite-metal structure shows to be more sensitive to static stress concentrations than all metal structure; however, a higher average ultimate tensile strength is obtained.

TABLE III
STATIC TENSILE COUPON IDENTIFICATION

SPEC.	MATERIAL COMBINATION	NO. OF COMPOSITE PLIES	ADHESIVE	К <sub>t</sub>
1A	All Aluminum			$K_{t} = 1.0$
2A	All Aluminum			$K_{t} = 1.0$
3A	AlGraphite	4	AF-126	K <sub>t.</sub> = 1.0
4A	AlGraphite	4	AF-126	$K_t = 1.0$
5A	AlGlass	3	AF-126	K <sub>t.</sub> = 1.0
6A	AlGlass	3	AF-126	K <sub>t.</sub> = 1.0
7A	AlGraphite	4	<b>EA-</b> 92 <b>7</b> R	$K_t = 1.0$
8A	AlGraphite	4	EA-927R	$K_t = 1.0$
9A	All Aluminum			K <sub>t</sub> > 1.0
10A	All Aluminum			$K_t > 1.0$
11A	AlGraphite	4	AF-126	K <sub>t</sub> > 1.0
12A	AlGraphite	14	AF-126	K <sub>t</sub> > 1.0
13A	AlGlass	3	AF-126	$K_{t} > 1.0$
14A	AlGlass	3	AF-126	$\kappa_{t} > 1.0$
15A	AlGraphite	14	EA-927R	к <sub>t</sub> > 1.0
16A	AlGraphite	14	EA-927R	κ <sub>t</sub> > 1.0

TABLE IV
STATIC TENSILE COUPON TEST DATA

			· · · · · · · · · · · · · · · · · · ·		'		Thickness					
Coupon Number		Material Combination	Coupon	Coupon Width Hole Dia.		Tota	Total		Composite		Metal	
Mumber		COMBINACION	mm	(in.)	mm	(in.)	<b>380</b> 0	(in.)	mm	(in.)	mm	(in.)
1A	1.0	All Aluminum	12.80	•504		*-÷	•973	<b>.0</b> 383				
2A	1.0	All Aluminum	12.67	•499			•983	.0387	*****			
3A	1.0	AlumGraphite	12.78	•503			2.090	.0823	.813	.032	•5 <b>0</b> 8	.020
4A	1.0	AlumGraphite	12.78	•503			2.042	.0804	.787	.031	<b>.</b> 508	.020
5A	1.0	AlumGlass	12.40	.488			1.798	.0708	.584	.023	-495	.0195
6a	1.0	AlumGlass	12.78	•503			1.791	.0705	<b>.</b> 584	.023	-495	.0195
7A	1.0	AlumGraphite	12.62	-497			2.276	.0896	.813	.032	<b>.</b> 508	.020
84	1.0	AlumGraphite	12.67	•499			2.205	.0868	.762	.030	•5 <b>0</b> 8	.020
9A	>1.0	All Aluminum	12.70	•500	3.165	.1246	.970	.0382				
10A	>1.0	All Aluminum	12.65	.498	3.172	.1249	•973	.0383				
11A	>1.0	AlumGraphite	12.78	•503	3.274	.1289	1.971	.0776	•737	.029	<b>.</b> 508	.020
12A	>1.0	AlumGraphite	12.78	•503	3.266	.1286	2.009	.0791	.787	.031	<b>•50</b> 8	.020
13A	>1.0	AlumGlass	12.73	.501	3.233	.1273	1.808	.0712	.610	.024	<b>.</b> 508	.020
14A	>1.0	AlumGlass	12.73	.501	3.203	.1263	1.819	.0716	•584	.023	<b>.</b> 508	.020
15A	>1.0	AlumGraphite	12.73	.501	3.307	.1302	2.187	.0861	.813	.032	<b>.50</b> 8	.020
16A	>1.0	AlumGraphite	12.55	.494	3.302	.1300	2.258	.0889	.813	.032	.508	.020

TABLE IV
STATIC TENSILE COUPON TEST DATA (CONT.)

Coupon K <sub>t</sub>		Material	Total Y	ield Load	Tota Failing		Ultimate Tensile Strength		
Number	·	Combination	N	(lbf.)	N	(lbf.)	MN/m <sup>2</sup>	(ksi)	
1A	1.0	All Aluminum	7,117	1600	7,277	1636	584.7	84.8	
2A	1.0	All Aluminum	6,761	1520	7,295	1640	581.2	84.3	
3A	1.0	AlumGraphite			22,685	5100	849.5	123.2	
4A	1.0	AlumGraphite			21,261	4780	815.0	118.2	
5A	1.0	AlumGlass	10,008	2250	22,262	5005	999.1	144.9	
6 <b>A</b>	1.0	AlumGlass	10 <b>,0</b> 52	2260	22,685	5100	991.5	143.8	
7A	1.0	AlumGraphite	440 GEO GEO		22,240	5000	774.3	112.3	
8a	1.0	AlumGraphite			20,194	4540	722.6	104.8	
9 <b>A</b>	>1.0	All Aluminum	5,284	1188	5,444	1224	588.8	85.4	
10A	>1.0	All Aluminum	4,750	1068	5 <b>,231</b>	1176	567.5	82.3	
11A	>1.0	AlumGraphite			12,365	2780	660.5	95.8	
12A	>1.0	AlumGraphite	12,788	2875	13,077	2940	684.7	99•3	
13A	<b>&gt;</b> 1.0	AlumGlass	7,473	1680	13,411	3015	781.2	113.3	
14A	>1.0	AlumGlass	7,562	1700	13,566	3050	784.0	113.7	
15A	>1.0	AlumGraphite			13,273	2984	644.7	93•5	
16A	>1.0	AlumGraphite	15,390	3460	15,657	3520	750.2	108.8	

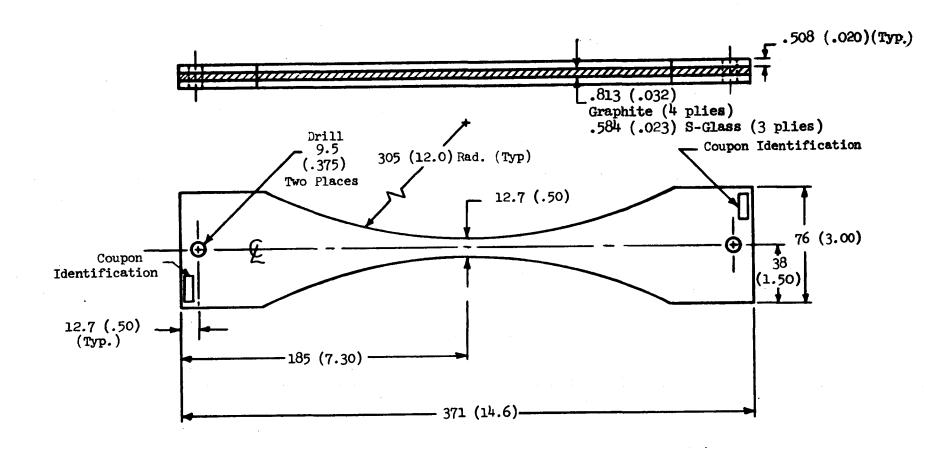


FIGURE 3 STATIC TENSILE COUPON SPECIMEN
(Dimensions in Millimeters and Inches Respectively)

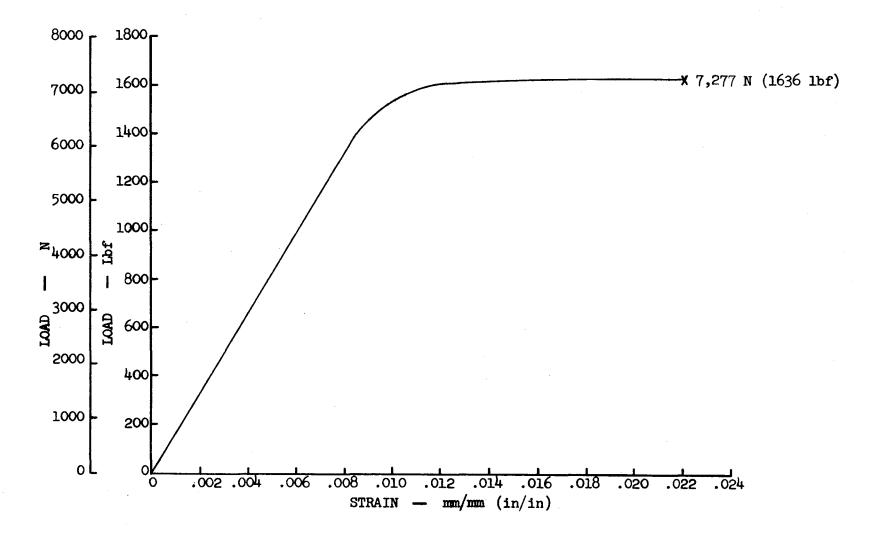


Figure 4 Load-Deflection Plot, Specimen 1A, All Aluminum  $K_t = 1.0$ 

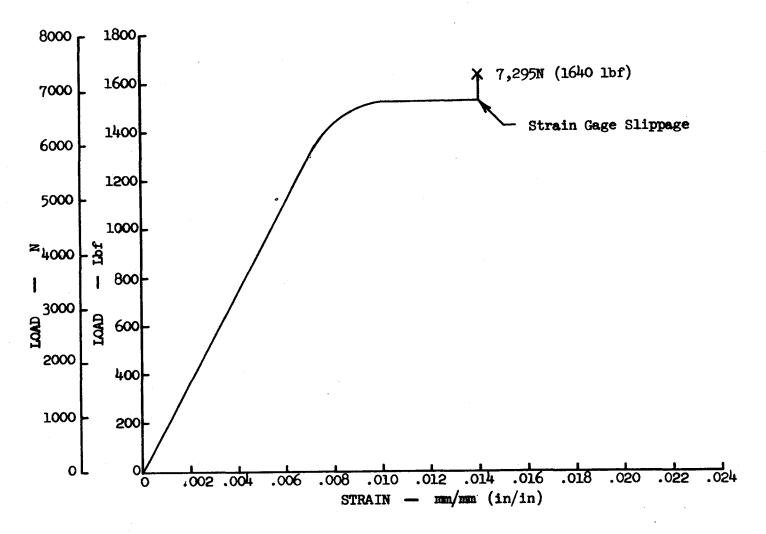


Figure 5 Load - Deflection Plot, Specimen 2A All Aluminum  $K_t = 1.0$ 

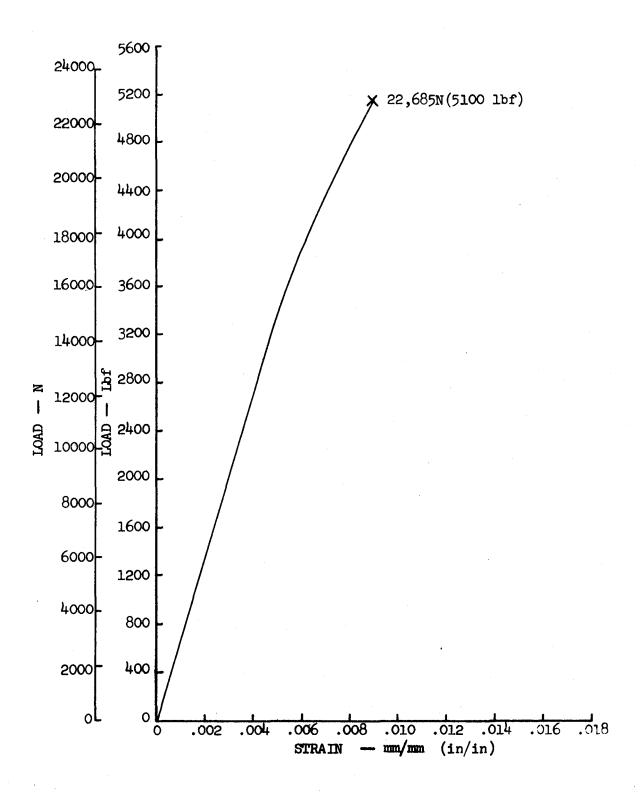


Figure 6 Load - Deflection Plot, Specimen 3A Aluminum-Graphite,  $K_{\mbox{\scriptsize t}}$  = 1.0

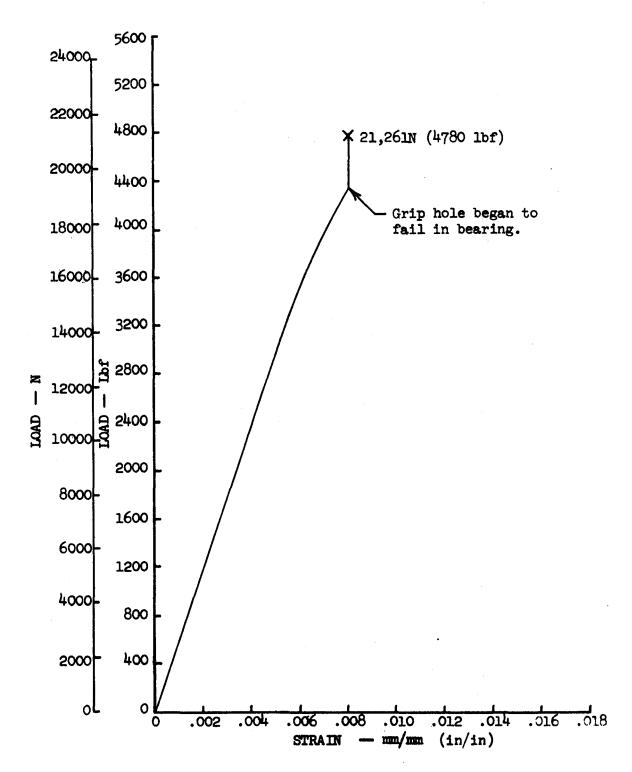


Figure 7 Load-Deflection Plot, Specimen  $^{14}$ A Aluminum Graphite,  $K_{t} = 1.0$ 

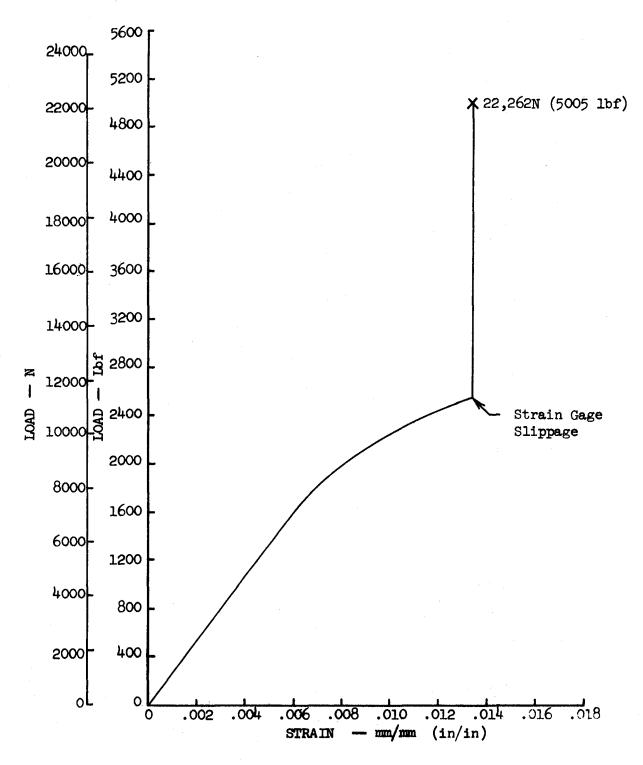


Figure 8 Load - Deflection Plot, Specimen 5A Aluminum-Glass,  $K_{t}$  = 1.0

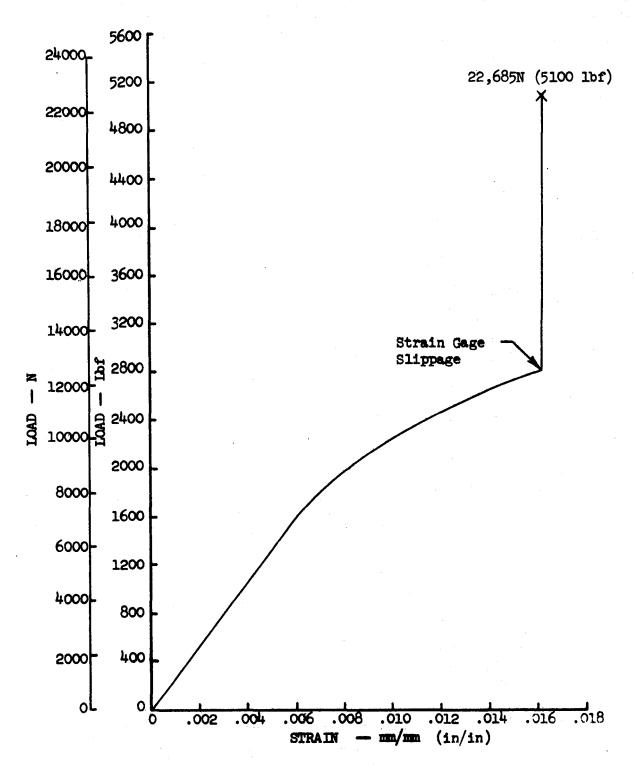


Figure 9 Load - Deflection Plot - Specimen 6A Aluminum-Glass K<sub>t</sub> = 1.0

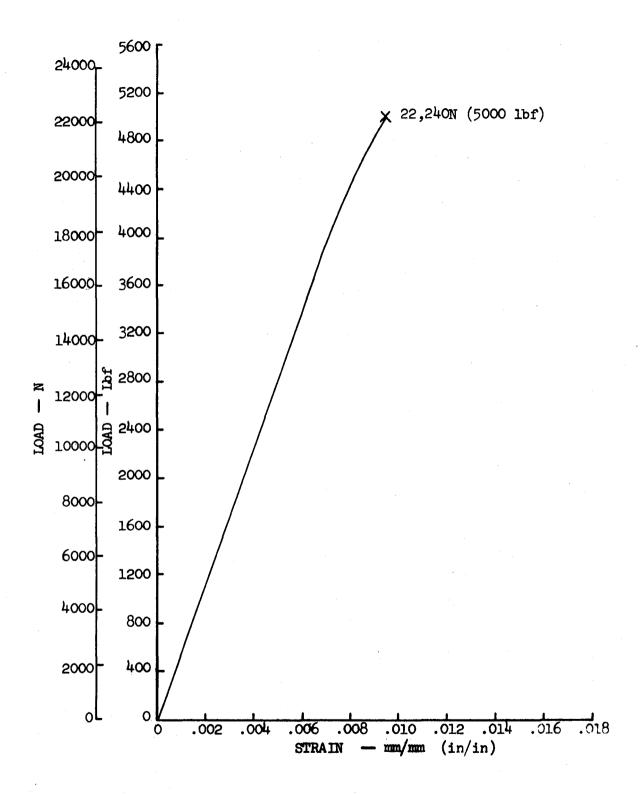


Figure 10 Load - Deflection, Specimen 7A Aluminum-Graphite,  $K_{t} = 1.0$ 

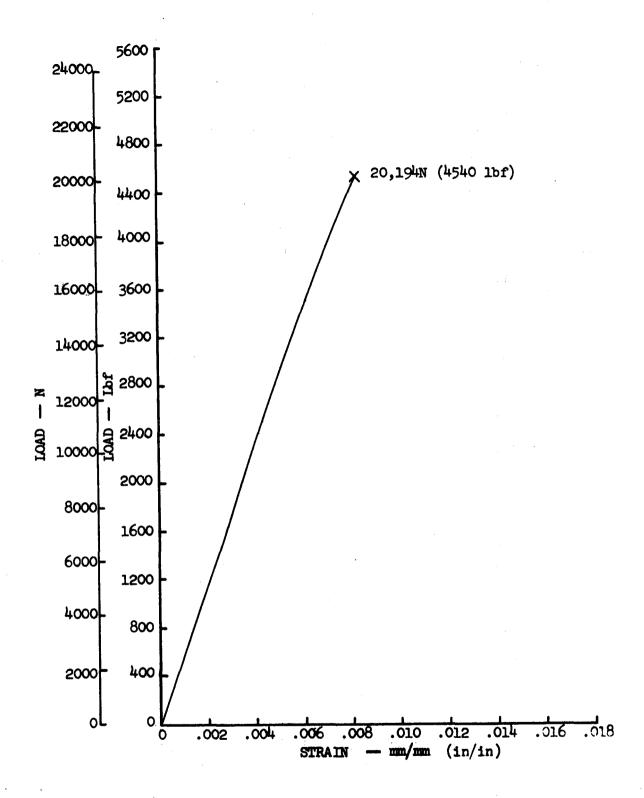


Figure 11 Load - Deflection Plot, Specimen 8A Aluminum-Graphite, K<sub>t</sub> = 1.0

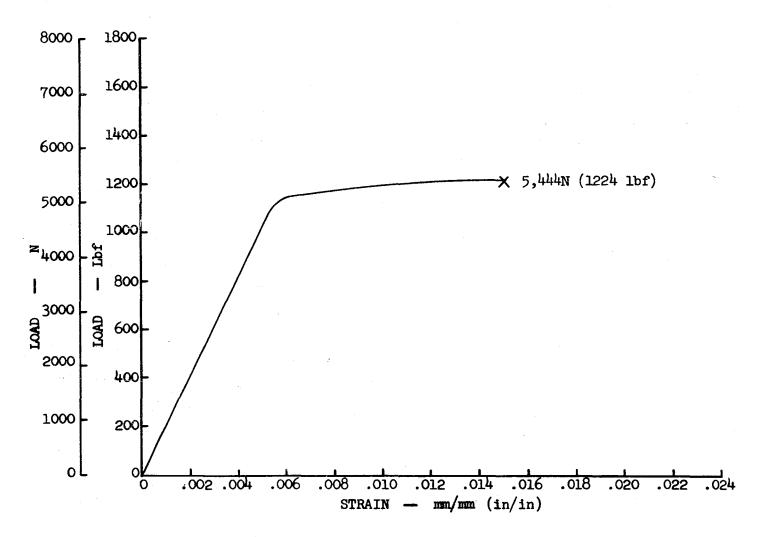


Figure 12 Load - Deflection Plot, Specimen 9A All Aluminum,  $K_{t} > 1.0$ 

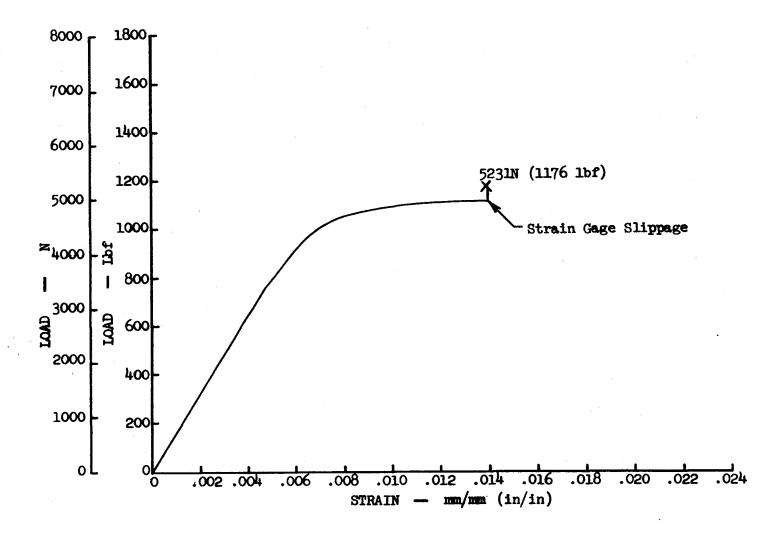


Figure 13 Load - Deflection Plot, Specimen 10A All Aluminum,  $K_t > 1.0$ 

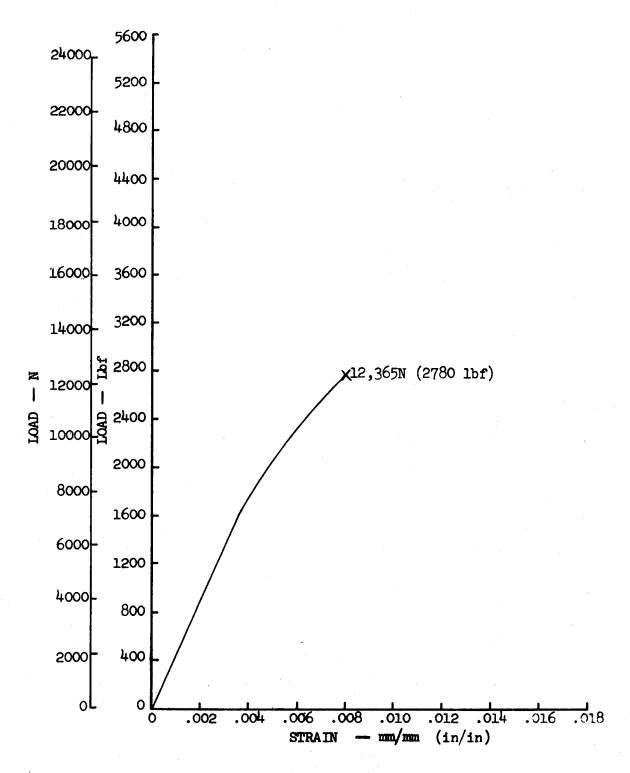


Figure 14 Load - Deflection Plot - Specimen 11A Aluminum-Graphite,  $K_{\rm t}$  > 1.0

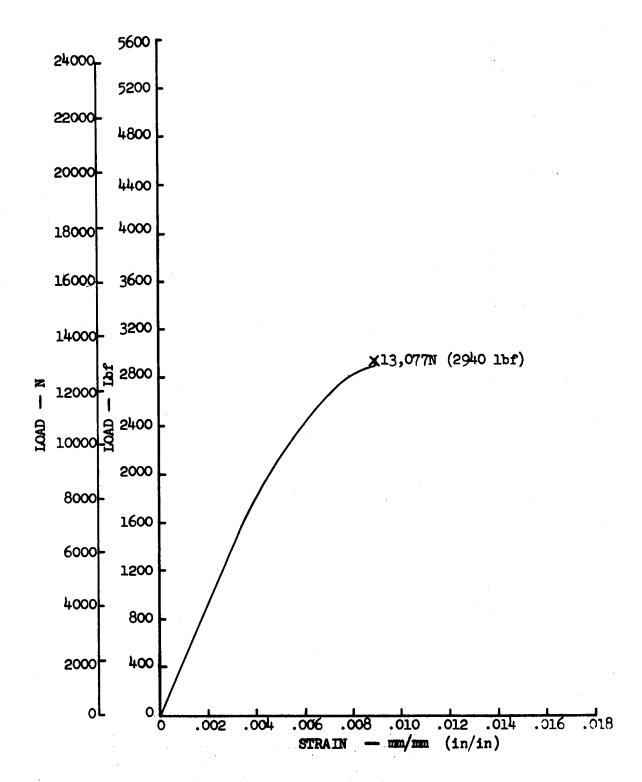


Figure 15 Load - Deflection Plot, Specimen 12A Aluminum-Graphite,  $K_{t} > 1.0$ 

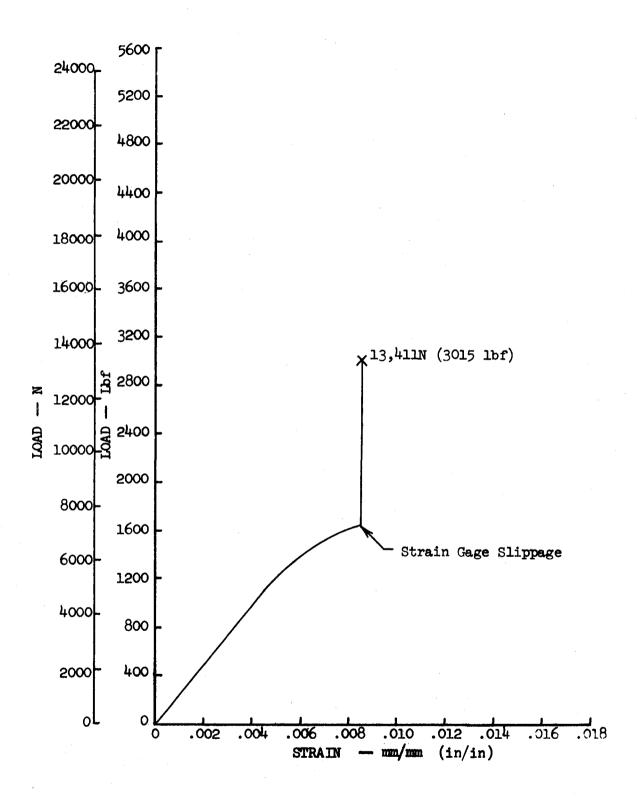


Figure 16 Load - Deflection Plot - Specimen 13A Aluminum-Glass,  $K_{\rm t} > 1.0$ 

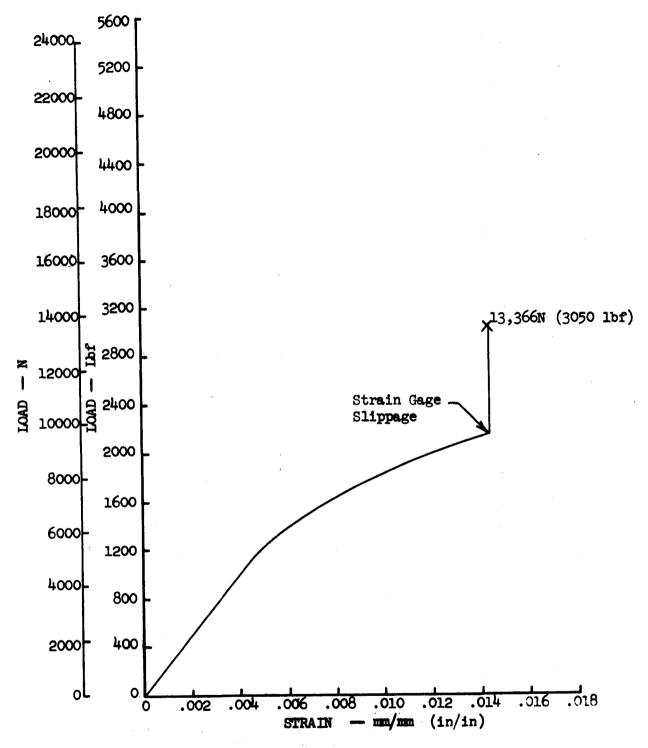


Figure 17 Load - Deflection Plot, Specimen 14A Aluminum-Glass,  $K_t > 1.0$ 

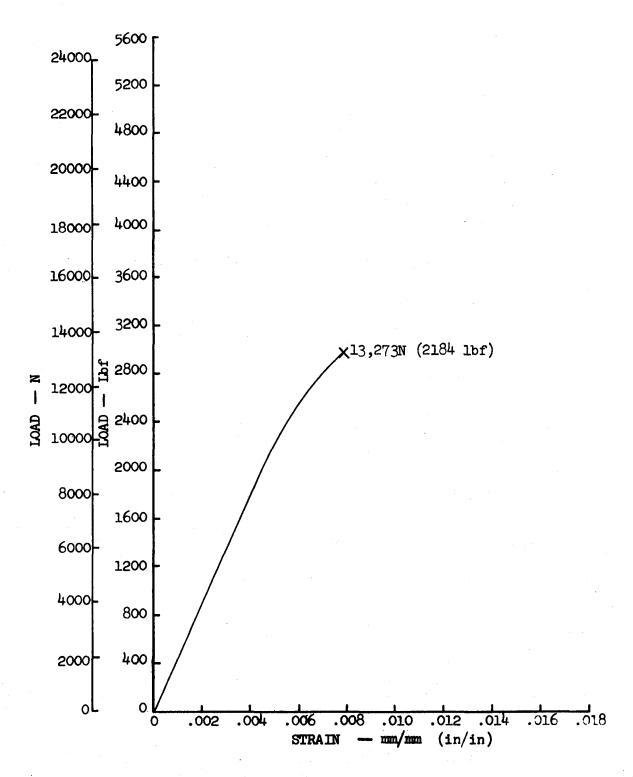


Figure 18 Load - Deflection Plot - Specimen 15A Aluminum-Graphite,  $K_{\rm t} > 1.0$ 

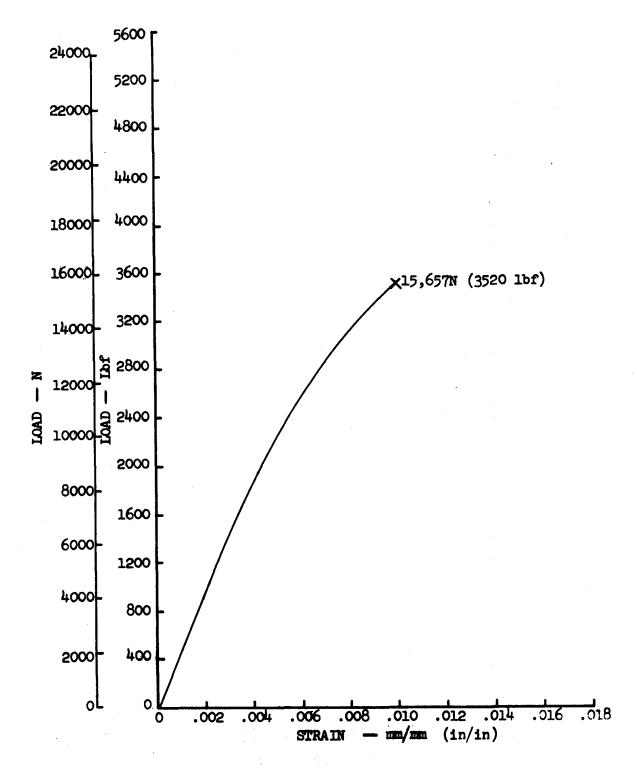


Figure 19 Load - Deflection Plot, Specimen 16A Aluminum-Graphite, Kt > 1.0

#### COUPON FATIGUE TESTS

## Objective

The objective of this task was to conduct coupon fatigue tests to obtain basic fatigue behavior data and to provide data which would show the fail-safe characteristics of composite-reinforced metal structure.

## Approach

Combinations of aluminum-graphite/epoxy and aluminum-glass/epoxy coupon fatigue specimens bonded with elevated and room temperature cure adhesives were tested in tension-tension fatigue and compared to all aluminum coupon fatigue results. Smooth specimens ( $K_t = 1.0$ ) and specimens with a hole ( $K_t > 1.0$ ) were tested and compared to evaluate the effects of holes on composite reinforced metal structure in fatigue loading.

## Test Specimens

The specimens used in this evaluation were as shown in Figure 20. A stress concentration was obtained by drilling and reaming a 3.2 mm (.125 in.) diameter hole on the longitudinal centerline at the minimum width.

Sixty-six (66) fatigue coupon tests were conducted involving all aluminum, aluminum-graphite/epoxy, and aluminum-glass/epoxy with adhesives AF-126 and EA-927R. Table V presents the identification of the test specimens.

The control specimens were fabricated from 1.02 mm (0.040 in) thick 7075-T6 aluminum. The stiffness ratio was  $\mu$  = 0.57 for the aluminum-graphite system and  $\mu$  = 0.29 for the aluminum-glass system.

## Testing and Results

The dimensions of each composite and metal thickness of each coupon specimen were measured at the minimum section and recorded before bonding the composite materials to the aluminum and after bonding and machining. After recording the dimensions the specimens were fatigue cycled in axial tension-tension loading at the load ratio of R = 0.10 and a frequency of 30 Hz (1800 cpm) until metal and composite failure. Each specimen cross section dimensions, cyclic load, cycles at time of metal failure and cycles at time of composite failure after metal failure, is given in Table VI. The gross stress in the metal and composite materials was calculated in terms of the total applied

load using equations (2) and (3). Figures 21, 22, and 23 show  $f_{max}$  - N curves for the various composite-metal-adhesive systems examined. Each figure gives the results for the three particular composite-metal-adhesive system examined for  $K_t = 1.0$  and  $K_t > 1.0$ . Figure 21 gives the results for the aluminum-graphite system with the AF-126 adhesive. Figure 22 gives the results for the aluminum-graphite system with the EA-927R adhesive, and Figure 23 gives the results for the aluminum-glass system with the AF-126 adhesive.

#### Discussion

The fatigue coupon tests were conducted to indicate the stress concentration effects and demonstrate the fail-safe characteristics of composite-reinforced metal structure. The solid and dashed lines at the bottom of Figures 20, 21 and 22 are conventional fatigue curves for 7075-T6 aluminum (Kt = 1.0 and Kt = 2.4, respectively). The solid and dashed lines at the top of the figures indicate trend lines of the composite material failures (Kt = 1.0 and Kt >1.0, respectively) following metal failures.

The value of stress concentration for all metal coupon specimens may be determined by geometry (ref. 3). For a 3.2 mm (0.125 in.) diameter hole in a 12.7 mm (0.50 in.) wide cross section the stress concentration value ( $K_t$ ) is 2.4. Figures 21, 22 and 23 indicate that this  $K_t$  value also exists for the composite-metal coupons since the metal failures of the composite-metal coupons occur along the conventional all metal fatigue curves.

A particular total load applied to the composite-metal system is distributed to each material according to the stiffness ratio ( $\mu$ ) of the system. The maximum cyclic stress applied to the metal is shown in Figures 21, 22, and 23 until the metal fails. At that time the total load is carried by the composite alone; therefore, the composite stress level is increased significantly. At this stress level the composite continues to cycle until failure occurs. Thus a particular load can be carried by the system after the metal has failed demonstrating a fail-safe concept.

The extended life and weight savings of a composite-metal system over the all metal system can be obtained from the coupon fatigue test data. The stress in the metal of a composite-metal system can be determined by:

$$f_{m} = \frac{P_{m}}{A_{m}} = \frac{(1-\mu) P_{t}}{A_{m}}$$
 (12)

The amount of metal in an all metal system needed to carry the same total load at the same stress level is

$$A_{am} = \frac{P_{t}}{f_{m}}$$
 (13)

Substituting (12) into (13) yields the amount of metal saved in a composite-metal system over an all metal system, i.e.,

$$A_{m} = (1-\mu)A_{am} \tag{14}$$

Adding the area of the composite, the percent weight savings can be determined by

% Weight Saved = 
$$\frac{\text{Wt}_{am} - \text{Wt}_{c-m}}{\text{Wt}_{am}}$$
 X 100 (15)

The percent of life extension of the composite-metal system over the all metal system can be determined by comparing the additional cycles of the composite after metal failure to the number of cycles of the initial failure of the metal, i.e.,

% Life Extension = 
$$\frac{N_c}{N_m}$$
 x 100 (16)

Using the data in Table VI and equations (15) and (16), the percent weight savings and minimum life extension for the aluminum-graphite and aluminum-glass systems can be determined. Table VII presents a summary of these results.

## Conclusions

The results of the test data of the composite-metal coupons indicate that the composite-reinforced metal structure is an effective fail-safe concept in terms of extended life and reduced weight. Table VII shows that the composite-metal system provides extended life for the same or less weight as the all metal system.

The fatigue coupon data also indicates that the stress concentration in the metal of the composite-metal system is approximately the same as in an all metal system. The stress concentration effects show a lesser degree of change in the fail-safeness in the aluminum-graphite system than in the aluminum-glass system. Also the use of the AF-126 adhesive (elevated temperature cure) or the EA-927R adhesive (room temperature cure) to bond the aluminum-graphite materials together produced no significant difference in the fatigue characteristics of the system.

TABLE V
FATIGUE COUPON IDENTIFICATION

Spec. No.	No. of Composite Plies	Material Combination	Adhesiv <b>e</b>	Kt
1B 2B 3B		All Aluminum All Aluminum All Aluminum		1.0 1.0 1.0
4B 5B 6B 7B 8B 9B 10B 11B 12B 13B	4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4	AlumGraphite	AF-126 AF-126 AF-126 AF-126 AF-126 AF-126 AF-126 AF-126 AF-126	1.0 1.0 1.0 1.0 1.0 1.0 1.0
14B 15B 16B 17B 18B 19B 20B 21B 22B 23B	3 3 3 3 3 3 3 3 3 3	AlumGlass	AF-126 AF-126 AF-126 AF-126 AF-126 AF-126 AF-126 AF-126 AF-126	1.0 1.0 1.0 1.0 1.0 1.0 1.0
24B 25B 26B 27B 28B 29B 30B 31B 32B 33B	7† 7† 7† 7† 7† 7† 7†	AlumGraphite	EA-927R	1.0 1.0 1.0 1.0 1.0 1.0 1.0

TABLE V (Continued)
FATIGUE COUPON IDENTIFICATION

Spec. No.	No. of Composite Plies	Material Combination	Adhesive	Kt
34B 35B 36B		All Aluminum All Aluminum All Aluminum		>1.0 >1.0 >1.0 >1.0
37B 38B 39B 40B 41B 42B 43B 44B 45B 46B	4 4 4 4 4 4 4	AlumGraphite	AF-126 AF-126 AF-126 AF-126 AF-126 AF-126 AF-126 AF-126 AF-126	>1.0 >1.0 >1.0 >1.0 >1.0 >1.0 >1.0 >1.0
478 488 498 508 518 528 538 548 558 568	33333333333333333333333333333333333333	AlumGlass	AF-126 AF-126 AF-126 AF-126 AF-126 AF-126 AF-126 AF-126 AF-126	>1.0 >1.0 >1.0 >1.0 >1.0 >1.0 >1.0 >1.0
57B 58B 59B 60B 61B 62B 63B 64B 65B 66B	ተ	AlumGraphite	EA-927R EA-927R EA-927R EA-927R EA-927R EA-927R EA-927R EA-927R EA-927R	>1.0 >1.0 >1.0 >1.0 >1.0 >1.0 >1.0 >1.0

TABLE VI FATIGUE COUPON TEST DATA

	COU	ON	1	HOLE	E COUFON TEST DATA						
SPEC.	WII			AMETER	************		CHICKNE				
NO.					TOTA		<del></del>	METAL	COMP		
	mm	(in.)	mm	(in.	mm	(in.)	mm	(in.)	mm	(in.)	
1B	12.65	.498	-	-	.980	.0386	-	-			
2B	12.65	.498	-	-	.980	.0386	-	<b>.</b>			
3B	12.83	.505	-	-	.980	.0386				•	
4B	12.80	.504	-	-	1.996	.0786	.508	.0200	.762	.030	
5B	12.70	.500	-	-	1.989	.0783	.508	.0200	.762	.030	
6B	12.75	.502	-	<b>-</b>	2.065	.0813	.508	.0200	.813	.032	
7B	12.70	.500	-	<b>  -</b>	2.057	.0810	.508	.0200	.813	.032	
8B	12.75	.502	-	_	2.022	.0796	.508	.0200	.787	.031	
9 <b>B</b>	12.75	.502	-	-	2.035	.0801	.508	.0200	.813	.032	
10B	12.75	.502	-	-	2.068	.0814	.508	.0200	.813	.032	
11B	12.75	.502	-	-	2.065	.0813	.508	.0200	.787	.031	
12B	12.70	.500	-	-	2.065	.0813	.508	.0200	.787	.031	
13B	12.73	.501	_	-	2.098	.0826	.508	.0200	.813	.032	
14B	<b>1</b> 2.65	.498	_	-	1.801	.0709	.495	.0195	.610	.024	
15B	12.73	•501	-	-	1.798	.0708	.508	.0200	.610	.024	
16B	12.75	.502	-	-	1.803	.0710	.508	.0200	.610	.024	
17B	12.78	.503	-	-	1.793	.0706	.508	.0200	.584	.023	
18B	12.75	.502	-	-	1.798	.0708	.495	.0195	.610	.024	
19B	12.70	.500	-	-	1.803	.0710	.495	.0195	.584	.023	
20B	12.67	.499	-	-	1.788	.0704	.495	.0195	.584	.023	
218	12.65	.498	-	-	1.791	.0705	.508	.0200	.584	.023	
22B	12.65	.498	-	-	1.793	.0706	.508	.0200	.584	.023	
23B	12.55	.494	-	-	1.793	.0706	.508	.0200	.584	.023	
24B	12.73	.501	-	<b>-</b> .	2.276	.0896	.508	.0200	.813	.032	
25B	12.65	.498	-	-	2.268	.0893	.508	.0200	.787	.031	
26B	12.73	.501	-	-	2.266	.0892	•495	.0195	.787	.031	
27B	12.73	.501	-	-	2.319	.0913	.495	.0195	.838	.033	
28B	12.75	.502	-	-	2.288	.0901	.495	.0195	.838	.033	
29B	12.70	.500			2.220	.0874	.508	.0200	.787	.031	
30B	12.65	.498		-	2.200	.0866	.508	.0200	.787	.031	
31B	12.65	.498	-	-	2.189	.0862	.508	.0200	.838	.033	
32B	12.67	.499	-	-	2.268	.0893	.508	.0200	.787	.031	
33B	12.65	.498	-	-	2.301	.0906	.508	.0200	.787	.031	

TABLE VI (Cont)
FATIGUE COUPON TEST DATA

			FA	TIGUE C	COUPON TEST DATA					
	COUP		HO			T	HICKNES	S		
SPEC.	WID	TH	DIAM	ETER	TOTA	<u> </u>	EACH M	ETAL	COMPO	SITE
	mm	(in.)	mm	(in.)	mm	(in.)	mm	(in.)	mm	(in.)
34B	12.75	.502	3.188	.1255	.980	.0386	-	_	_	-
35B	12.73	.501	3.190	.1256	.980	.0386	_	-	-	_
36B	12.70	•500	3.180	.1252	.980	.0386		-		-
37B	12.73	•501	3.279	.1291	2.022	.0796	•508	.0200	•737	.029
38B	12.73	.501	3.185	.1254	2.042	.0804	.495	.0195	.813	.032
**39B	12.75	.502	3.264	.1285	2.035	.0801	.495	.0195	.813	.032
40B	12.78	•503	3.203	.1261	2.002	.0788	.495	.0195	.787	.031
**41B	12.70	•500	3.254	.1281	2.111	.0831	.495	.0195	.838	.033
42B	12.73	.501	3.233	.1273	2.073	.0816	.495	.0195	.838	.033
43B	12.73	.501	3.251	.1280	1.986	.0782	.495	.0195	.762	.030
44B	12.75	.502	3.208	.1263	2.042	.0804	.495	.0195	.838	•033
45B	12.75	.502	3.246	.1278	2.042	.0804	•495	.0195	.787	.031
46B	12.70	.500	3.269	.1287	2.047	.0806	.495	.0195	.787	.031
47B	12.65	.498	3.244	.1277	1.781	.0701	.508	.0200	.610	.024
48B	12.57	.495	3.152	.1241	1.796	.0707	.508	.0200	.610	.024
49B	12.67	.499	3.231	.1272	1.793	.0706	.508	.0200	.584	.023
50B	12.60	.496	3.200	.1260	1.798	.0708	.508	.0200	.610	.024
51B	12.75	.502	3.162	.1245	1.788	.0704	.495	.0195	.572	.0225
52B	12.60	.496	3.170	.1248	1.798	.0708	.495	.0195	.584	.023
53B	12.65	.498	3.208	.1263	1.801	.0709	.495	.0195	.584	.023
54B	12.67	•499	3.165	.1246	1.798	.0708	495	.0195	.584	.023
55B	12.73	.501	3.203	.1261	1.798	.0708	.495	.0195	.584	.023
56B	12.75	.502	3.221	.1268	1.793	.0706	.495	.0195	.584	.023
57B	12.67	.499	3.218	.1267	2.253	.0887	.508	.0200	.813	.032
58B	12.60	.496	3.114	.1226	2.195	.0864	.495	.0195	.813	.032
59B	12.50	.492	3.231	.1272	2.228	.0877	.495	.0195	.787	.031
60B	12.70	.500	3.238	.1275	2.167	.0853	•495	.0195	.787	.031
61B	12.65	.498	3.241	.1276	2.240	.0882	.495	.0195	.813	.032
62B	12.62	•497	3.246	.1278	2.200	.0866	.495	.0195	.813	.032
63B	12.67	.499	3.231	.1272	2.184	.0860	.495	.0195	.813	.032
64B	12.73	.501	3.254	.1281	2.151	.0847	.508	.0200	.813	.032
65B	12.65	.498	3.266	.1286	2.149	.0846	.508	.0200	.813	.032
66B	12.67	.499	3.155	.1242	2.123	.0836	•508	.0200	.787	.031

\*\*R = 0.17

TABLE VI (Cont)
FATIGUE COUPON TEST DATA

			IGUE COOP	ON TEST DAT	<u>n</u>	
		MAXIMUM				COMPOSITE
SPEC.	MATERIAL	TOTAL LO	AD A	METAL	FAILURE	FAILURE
NO.	COMBINATION			lst SIDE	2nd SIDE	CYCLES AFTER \
].		N	(lbf.)	(CYCLES)	(CYCLES)	METAL FAILURE
1B	All Aluminum	4,448	1000	35,000	-	-
2B	All Aluminum	4,448	1000	44,000	-	-
3B	All Aluminum	6,218	1398	12,000		_
4B	Alum-Graphite	12,454	2800	30,000	35,000	22,000
5B	Alum-Graphite	12,454	2800	17,000	17,000	13,000
6B	Alum-Graphite	13,700	3080	18,000	25,000	64,000
7B	Alum-Graphite	10,364	2330	55,000	69,000	6,229,000 *
8B	Alum-Graphite	10,364	2330	35,000	38,000	679,000
9B	Alum-Graphite	11,476	2580	28,000	33,000	19,000
10B	Alum-Graphite	8,896	2000	47,000	49,000	8,061,000 *
11B	Alum-Graphite	11,476	2580	33,000	33,000	141,000
12B	Alum-Graphite	11,476	2580	25,000	25,000	531,000
13B	Alum-Graphite	8,896	2000	35,000	36,000	7,560,000 *
14B	Alum-Glass	7,517	1690	11,000	14,000	11,000
15B	Alum-Glass	7,517	1690	19,000	21,000	9,000
16B	Alum-Glass	7,517	1690	12,000	13,000	9,000
17B	Alum-Glass	6,539	1470	20,000	24,000	16,000
18B	Alum-Glass	6,539	1470	28,000	34,000	18,000
19B	Alum-Glass	6,539	1470	26,000	26,000	12,000
20B	Alum-Glass	5,382	1210	35,000	55,000	46,000
21B	Alum-Glass	5,382	1210	65,000	68,000	34,000
22B	Alum-Glass	5,382	1210	58,000	74,000	40,000
23B	Alum-Glass	4,715	1060	66,000	98,000	86,000
24B	Alum-Graphite	12,454	2800	39,000	39,000	5,000
25B	Alum-Graphite	12,454	2800	37,000	39,000	3,000
26B	Alum-Graphite	11,053	2485	40,000	69,000	52,000
27B	Alum-Graphite	10,364	2330	95,000	95,000	2,101,000 *
28B	Alum-Graphite	11,031	2480	78,000	78,000	2,247,000 *
29B	Alum-Graphite	11,031	2480	49,000	72,000	62,000
30B	Alum-Graphite	8,896	2000	400,000	400,000	2,104,000 *
31B	Alum-Graphite	11,543	2595	24,000	30,000	7,000
32B	Alum-Graphite	10,675	2400	71,000	83,000	1,302,000 *
33B	Alum-Graphite	10,853	5440	41,000	53,000	83,000

\* No Failure

TABLE VI (Cont)
FATIGUE COUPON TEST DATA

		FATIGO	E COUPON	TEST DATA		
		MAXIMUM				COMPOSITE
SPEC.	MATERIAL	TOTAL LO	DAD	METAI	L FAILURE	FAILURE
NO.	COMBINATION			1st SIDE	2nd SIDE	/CYCLES AFTER \
		N	(lbf.)	(CYCLES)	(CYCLES)	METAL FAILURE
34B	All Aluminum	2,068	465	50,000	-	**
35B	All Aluminum	2,068	465	39,000	_	
36B	All Aluminum	1,543	347	1,041,000	_	-
37B	Alum-Graphite	5,782	1300	9,000	11,000	2,291,000 *
38в	Alum-Graphite	5,782	1300	13,000	14,000	2,486,000 *
**39B	Alum-Graphite	7,081	1592	10,000	10,000	28,000
40B	Alum-Graphite	4,804	1080	19,000	19,000	201,000 *
**41B	Alum-Graphite	7,081	1592	12,000	12,000	174,000
42B	Alum-Graphite	7,464	1678	8,000	8,000	147,000
43B	Alum-Graphite	6,561	1475	11,000	11,000	1,019,000
44B	Alum-Graphite	7,695	1730	8,000	8,000	2,482,000 *
45B	Alum-Graphite	7,473	1680	6,000	6,000	2,074,000 *
46B	Alum-Graphite	7,695	1730	6,000	6,000	1,569,000
47B	Alum-Glass	3,514	790	12,000	12,000	28,000
48B	Alum-Glass	3,514	790	11,000	11,000	33,000
49B	Alum-Glass	2,607	586	27,000	27,000	120,000
50B	Alum-Glass	3,002	675	20,000	20,000	53,000
51B	Alum-Glass	3,002	675	17,000	17,000	100,000
52B	Alum-Glass	2,607	586	32,000	33,000	82,000
53B	Alum-Glass	2,157	485	68,000	68,000	490,000
54B	Alum-Glass	2,157	485	50,000	50,000	329,000
55B	Alum-Glass	2,015	453	79,000	120,000	525,000
56B	Alum-Glass	3,941	886	11,000	11,000	15,000
57B	Alum-Graphite	5,782	1300	14,000	20,000	2,024,000 *
58B	Alum-Graphite	7,117	1600	9,000	15,000	2,354,000 *
59B	Alum-Graphite	7,784	1750	7,000	8,000	355,000
60в	Alum-Graphite	7,962	1790	8,000	10,000	74,000
61B	Alum-Graphite	8,162	1835	7,000	8,000	145,000
62B	Alum-Graphite	8,674	1950	4,000	7,000	37,000
63B	Alum-Graphite	8,718	1960	4,000	6,000	15,000
64B	Alum-Graphite	7,695	1730	7,000	11,000	901,000
65B	Alum-Graphite	7,895	1775	8,000	11,000	496,000
66B	Alum-Graphite	7,748	1742	7,000	8,000	246,000

\*\*R = 0.17 \*\* No Failure

TABLE VII
SUMMARY OF COUPON FATIGUE TESTS

GOLIDON	K <sub>t</sub> =	1.0	K <sub>t</sub> >1.0			
COUPON MATERIAL COMBINATIONS	% WEIGHT SAVINGS	% MINIMUM LIFE EXTENSION*	% WEIGHT SAVINGS	%minimum life extension		
AlumGraphite (AF-126 Adhesive)	37	68	37	100		
AlumGlass	0	40 to 50	0	100		
AlumGraphite (EA-927R Adhesive)	37	30	37	100		

<sup>\*</sup> For aluminum stress levels at or below limit stress.

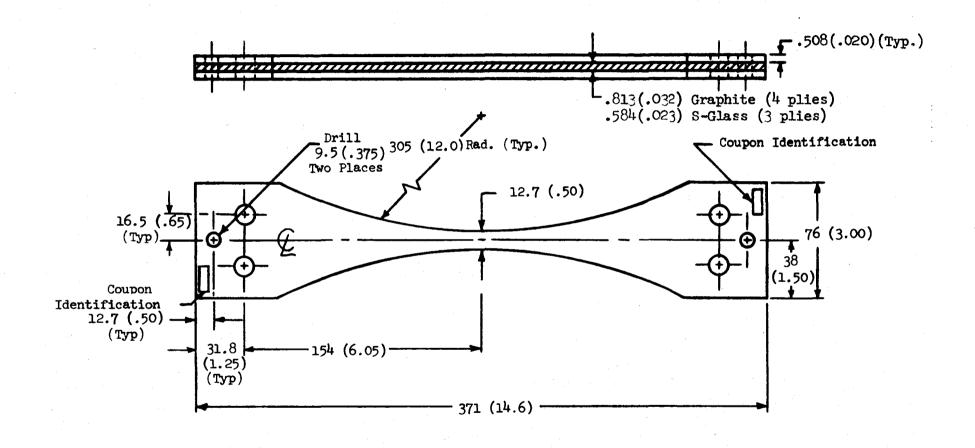


FIGURE 20 FATIGUE COUPON SPECIMEN

(Dimensions in Millimeters and Inches Respectively)

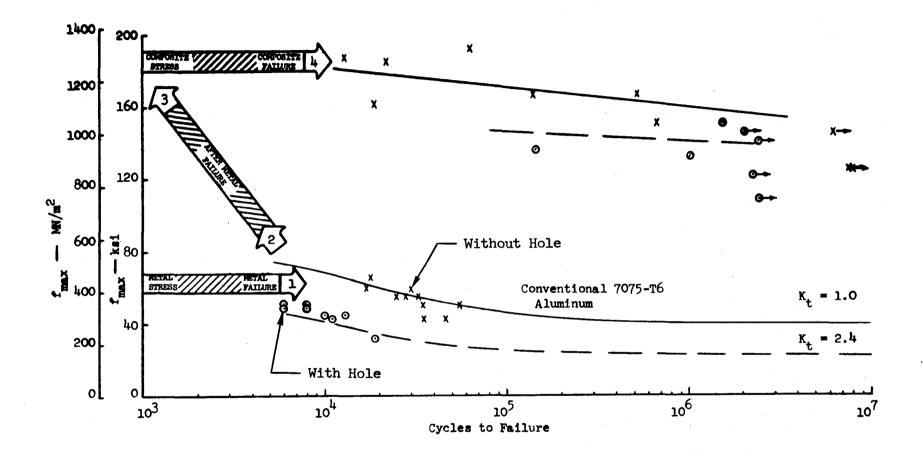


FIGURE 21 FATIGUE LIFE CURVE
ALUMINUM-GRAPHITE
AF-126 Adhesive

µ = 0.57, R = 0.10

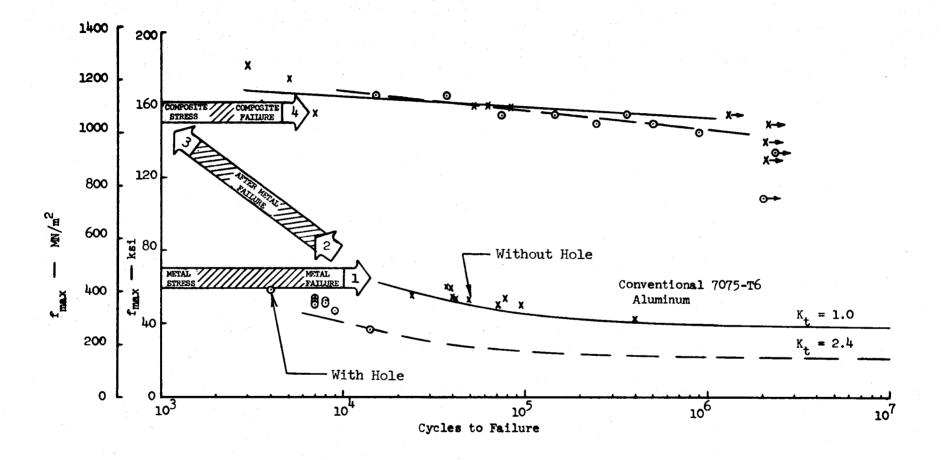


FIGURE 22 FATIGUE LIFE CURVE
ALUMINUM-GRAPHITE
EA-927R Adhesive
y = 0.57, R = 0.10

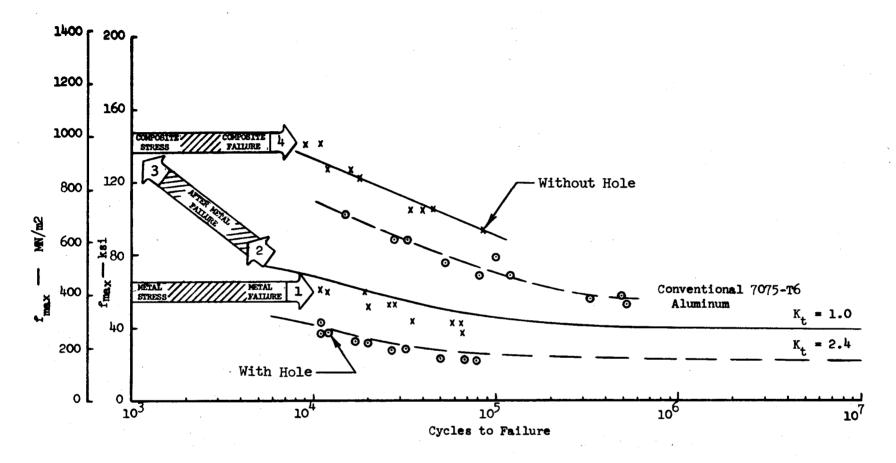


FIGURE 23 FATIGUE LIFE CURVE
ALUMINUM-GLASS
AF-126 Adhesive

µ = 0.29, R = 0.10

## PANEL STATIC LOAD DISTRIBUTION

## Objective

The objective of this part of the program was to verify the load distribution in the all metal and the composite reinforced integrally formed panels.

## Approach

Photostress analysis was conducted on one panel in each of the four groups of panels, all aluminum, aluminum-glass, aluminum-graphite (elevated temperature cure adhesive), and aluminum-graphite (room temperature cure adhesive). Photoelastic coatings were applied on each of the four panels as shown in Figures 24 through 27. The results of the photoelastic analysis were compared to strain gage readings.

Each panel was instrumented with strain gages to obtain panel strain distribution in the test region. Typical strain gage locations for a panel with and without photoelastic coating is shown in Figures 28 through 30.

## Test Specimens

One all-metal, integrally-stiffened panel (1C) was fabricated from 7075-T6 aluminum. The inner and outer sheets were each 0.51 mm (0.020 in.) thick and bonded together with AF-126 adhesive.

Fourteen composite-reinforced, integrally-stiffened metal panels were fabricated. The configuration of these panels is shown in Figure 31 and the panel identification is given in Table VIII. The stiffness ratios for these panels are 0.57 for the aluminum-graphite materials and 0.29 for the aluminum-glass materials.

Each panel was 762 mm (30 in.) long, 305 mm (12 in.) wide. Each panel was necked down to 241 mm (9.5 in.) at the test section with the exception of panels 1C, 3C and 8C. A crack starter hole, 6.4 mm (0.250 in.) diameter, was drilled in the center of the panel. Individual fittings were bonded and bolted to each end of the panels to provide attachment to the testing machine.

## Testing and Results

Photoelastic testing was conducted on the four panels described previously. Figure 24 through 27 show the specific points observed on the photostress coatings, 01 through 03 and II through I3. Observations were made and data recorded at several static load levels. The photostress data was compared to strain gage data at similar locations on the panel. Results of these comparisons are shown in Tables IX through XII.

Strain gage surveys were conducted on all panels. The panel strain distribution was brought to zero at a total load of 11,120 N (2500 lbf) and loaded in increments to the test load or a load in excess of the test load. Strain readings at the maximum load levels for panels 1C, 3C, 8C, and 12C are presented in Tables XIII through XVI. Strain gage readings were used to verify the correct aluminum stress level in each panel and insure equal load distribution across the panel test section.

#### Discussion

The photostress data of Table IX through Table XII provides a method of obtaining the stress concentration at the edge of the quarter inch diameter hole. A value for the stress concentration factor was determined by dividing the average stress at the hole  $(f_{03}+f_{13})/2$ , by the stress at some point on the panel that is not effected by stress risers  $(f_{11})$ . The values of stress concentration factor calculated by this method are listed in Table XVII.

The aluminum stress at the panel test section for fatigue testing was determined both analytically and experimentally. Experimental determination was accomplished by obtaining the average test section strain, at or just above the fatigue test load, from strain gage readings and multiplying by the aluminum elastic modulus value (E) of  $71,020 \text{ MN/m}^2$  (10,300 ksi).

The difference between panel test section maximum strain and minimum strain was held to 100 micromillimeters per millimeter (100 microinches per inch) in 89 kN (20,000 lbf) of total load.

## Conclusions

The stress concentration values, calculated from the photostress data, indicate an increase in static stress concentration factors in the aluminum-composite panels when compared to the all aluminum panel. This increase in stress concentration factors is approximately twice the value of the all aluminum panel.

The panel strain gage data indicates relatively uniform stress distribution across the panel test section since the maximum stress is within 5 percent of the minimum stress.

TABLE VIII
PANEL IDENTIFICATION

Panel No.	Material Combination	Adhesive *	Measu Compo Thick		Test Secti Width	
			mm	(in)	mm	(in)
1C	All Aluminum	AF-126			241.3	9•5
2C	AlumGraphite	AF-126	.787	0.031	241.3	9.5
3c	AlumGraphite	AF-126	.787	0.031	304.8	12.0
4C	AlumGraphite	AF-126	.813	0.032	241.3	9.5
5C	AlumGraphite	AF-126	.787	0.031	241.3	9.5
6c	AlumGraphite	AF-126	.813	0.032	241.3	9.5
7C	AlumGraphite	AF-126	.864	0.034	241.3	9.5
8c	AlumGalss	AF-126	. 584	0.023	304.8	12.0
9C	AlumGlass	AF-126	. 584	0.023	241.3	9.5
10C	AlumGlass	AF-126	.610	0.024	241.3	9.5
11C	AlumGlass	AF-126	.584	0.023	241.3	9.5
12C	AlumGraphite	EA-927R	.787	0.031	241.3	9.5
13C	AlumGrap <b>h</b> ite	EA-927R	.813	0.032	241.3	9.5
14C	AlumGraphite	<b>EA-</b> 927R	.813	0.032	241.3	9.5
15C	AlumGraphite	<b>EA-</b> 927R	.813	0.032	241.3	9.5

<sup>\*</sup> AF-126 is cured at  $389^{\circ}$ K (250°F) and EA-927R is cured at room temperature \*\* All aluminum thicknesses were 0.51 mm (0.020 in.)

TABLE IX

PHOTOSTRESS COMPARISONS AT SPECIFIC POINTS FOR
LOAD LEVEL OF 44.5 kN (10,000 lbf)

(PANEL NO. 1C)

LOCATIONS FRINGE		RELATED		RELATED	STRAIN GAGE COMPARISONS + (IF APPLICABLE)				
LOCATIONS	ORDER		ESS*	STRAIN**	GAGE NO.	STRE		STRAIN	
	n.	$(MN/m^2)$	(ksi)	(mm/mm رر mm/mm رر		$(MN/m^2)$	(ksi)	(mm/mm رر)	
01	1.38	56.7	8.23	1320		·			
02	1.64	67.3	9.76	1570	A0-7	63.4	9.20	1115	
03	2.60	106.5	15.45	2480					
11	1.42	58.3	8.46	1360	AI-2	66.0	9.60	1165	
I2	1.35	55.4	8.04	1290	AI-4	68.0	9.89	1200	
13	2.24	92.0	13.35	2140					

\* Related Stress =  $F_{\sigma}$  Xn

where  $F_{\phi}$  is the model stress fringe value for photostress plastic bonded to aluminum-aluminum composite structure ( $F_{\phi}$  = 41.1 MN/m<sup>2</sup>n (5.96 ksi/n))

\*\* Related Strain = F, Xn

where  $F_{\xi}$  is the model strain fringe value for photostress plastic bonded to aluminum-aluminum composite structure ( $F_{\xi}$  = 955  $\mu$  mm/mm n ( $\mu$  in/in n))

- \*\*\* A0-7 is the closest symmetry gage to the particular photostress location.
- +  $E_T$  = 56,800 MN/m<sup>2</sup> (8,240 ksi) Modulus of Total Structure

TABLE X

PHOTOSTRESS COMPARISONS AT SPECIFIC POINTS FOR LOAD LEVEL OF 44.5 kN (10,000 lbf)

(PANEL NO. 3C)

LOCATIONS	FRINGE	ł .	RELATED RELATED			IN GAGE COMPARISONS + (IF APPLICABLE)		
	ORD <b>E</b> R n	STRE (MN/m <sup>2</sup> )	CSS* (ksi)	STRAIN** (wmm/mm)	GAGE NO.	STI (MN/m <sup>2</sup> )	RESS (ksi)	STRAIN () mm/mm)
01	0.55	38.7	5.61	561				
02	0.55	38.7	5.61	561	A0-7***	41.7	6.06	459
03	2.00	140.7	20.40	2040				
Il	0.52	36.6	5.31	531	AI-2	42.9	6.23	472
I2	0.59	41.5	6.02	602	AI-4	45.8	6.64	504
13	1.77	124.5	18.05	1800				

\* Related Stress = F Xn

where  $F_{c}$  is the model stress fringe value for photostress plastic bonded to aluminum-graphite composite structure  $(F_{c} = 70.3 \text{ MN/m}^2\text{n} (10.2 \text{ ksi/n}))$ 

\*\* Related Strain =  $F_{\epsilon}$  Xn

where  $F_{\epsilon}$  is the model strain fringe value for photostress plastic bonded to aluminum-graphite composite structure ( $F_{\epsilon} = 1,020 \ \mu \, \text{mm/mm} \, \text{n} \, (\mu \, \text{in/in n})$ )

\*\*\*AO-7 is the closest symmetry gage to the particular photostress location.

+  $E_{\rm m}$  = 91,000 MN/m<sup>2</sup> (13,200 ksi) - Modulus of Total Structure (i.e. aluminum plus composite)

TABLE XI
PHOTOSTRESS COMPARISONS AT SPECIFIC POINTS FOR
LOAD LEVEL OF 44.5 kN (10,000 lbf)
(PANEL NO. 8C)

	FRINGE ORDER		ATED ESS**	REIATED STRAIN**						
LOCATIONS	n	(MN/m <sup>2</sup> )	(ksi)	( u mm/mm)	GAGE NO.	STRE (MN/m <sup>2</sup> )		STRAIN (ע mm/mm)		
01	0.85	31.6	4.58	765		·		·		
02	0.72	26.8	3.88	648	A0-7***	43.1	6.25	796		
03	2.64	97•9	14.20	2380				· · · · · · · · · · · · · · · · · · ·		
I1	0.83	30.8	4.47	747	ÀI-2	43.9	6.37	811		
	0.81	30.1	4.36	729	AI-4	45.5	6.60	841		
13	2.55	94.5	13.70	2300						

## \* Related Stress = Fr Xn

where  $F_0$  is the model stress fringe value for photostress plastic bonded to aluminum-glass composite structure ( $F_0$  = 37.1 MN/m<sup>2</sup>n (5.38 ksi/n))

## \*\* Related Strain = $F_{\epsilon}$ Xn

where  $F_{\epsilon}$  is the model strain fringe value for photostress plastic bonded to aluminum-glass composite structure ( $F_{\epsilon} = 900 \, \mu \, \text{mm/mm} \, \text{m} \, \text{m} \, \text{m} \, \text{m}$ )

- \*\*\*A0-7 is the closest symmetry gage to the particular photostress location.
- +  $E_{rp}$  = 54,100 MN/m<sup>2</sup> (7,850 ksi) Modulus of Total Structure (i.e. aluminum plus composite)

TABLE XII

PHOTOSTRESS COMPARISONS AT SPECIFIC POINTS FOR LOAD LEVEL OF 44.5 KN (10,000 1bf)

(PANEL NO. 12C)

LOCATIONS	FRINGE ORDER	1	RELATED RELATED STRESS* STRAIN**					
TOCATIONS	n	(MN/m <sup>2</sup> )	(k <b>si</b> )	STRAIN** (umm/mm)	GAGE NO.	STRI (MN/m <sup>2</sup> )	ESS (ksi)	STRAIN (سmm/mm)
01	0.59	34.3	4.98	521				
02	0.51	29.7	4.31	451				
03	1.70	98.7	14.31	1,500		·	·	
IJ	0.63	36.7	5.32	557	AI-2	44.1	6.40	<b>50</b> 8
IS	0.61	35.6	5.16	539	AI-4	կկ.,կ	6.44	511
13	1.69	98.5	14.28	1,490				

\* Related Stress = F Xn

where  $F_{\sigma}$  is the model stress fringe value for photostress plastic bonded to aluminum-graphite composite structure ( $F_{\sigma} = 58.3 \text{ MN/m}^2 \text{ n } (8.45 \text{ ksi/n})$ )

\*\* Related Strain =  $F_{\epsilon}$  Xn

where  $F_{\epsilon}$  is the model strain fringe value for photostress plastic bonded to aluminum-graphite composite structure ( $F_{\epsilon} = 883 \ \mu \ mm/mm \ n \ (\mu \ in/in \ n)$ )

 $E_{T} = 87,000 \text{ MN/m}^2 \text{ (12,600 ksi)}$  - Modulus of Total Structure (i.e., aluminum plus composite)

TABLE XIII

STRAINS AT  $P_t = 89 \text{ kN (20,000 lbf)}$ (Panel No. 1C)

Gage No.	strain *10 <sup>-6</sup>
A0-1 AI-2 A0-3 AI-4 A0-5 AI-6 A0-7 AI-8 A0-9 AI-10 AI-11 AI-12	2330 2330 2375 2400 2230 2242 2230 2230 2200 2235 2210 2250 2242

TABLE XIV STRAINS AT  $P_t$  = 156 kN (35,000 lbf)

(Panel No. 3C)

Gage No.	STRAIN ×10 <sup>-6</sup>		
A0-1 AI-2 A0-3 AI-4 A0-5 AI-6 A0-7 AI-8 A0-9 AI-10 AI-11 AI-12 AI-13	1568 1653 1706 1764 1603 1653 1605 1584 1603 1722 1654 1673		

TABLE XV STRAINS AT  $P_t = 89 \text{ kN (20,000 lbf)}$  (Panel No. 8C)

Gage No.	Strain x10 <sup>-6</sup>	
A0-1	1668	
AI-2	1622	
A0-3	<b>1</b> 628	
AI-4	<b>1</b> 682	
A0-5	1664	
AI-6	1601	
A0-7	1591	
AI-8	1610	
A0-9	1564	
AI-10	1564	
AI-11	1612	
AI-12	1700	
<b>A</b> I-13	1671	

TABLE XVI STRAINS AT P<sub>t</sub> = 200 kN (45,000 lbf) (Panel No. 12C)

Gage No.	Strain x10 <sup>-6</sup>
AO-1	2301
AI-2	2284
AO-3	2198
AI-4	2295
AO-5	2105
AI-6	2094
AO-7	3062
AI-8	2749
AO-9	1983
AI-10	1926
AI-11	1897
AI-12	2200
AI-13	2232

TABLE XVII
STRESS CONCENTRATION FACTORS DETERMINED FROM PHOTOSTRESS DATA

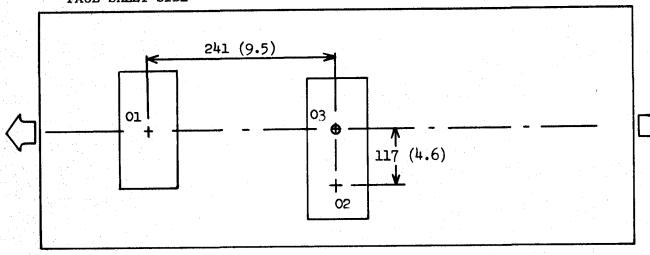
PANEL NUMBER	TYPE OF PANEL		AVERAGE STRESS AT HOLE STRESS AWAY FROM HOLE $(f_{11})$ $[(f_{03} + f_{13})/2]$		STRESS CONCENTRATION FACTOR (K <sub>t</sub> )	
		mn/m <sup>2</sup>	(ksi)	mn/m <sup>2</sup>	(ksi)	
1C	All Aluminum	99•3	14.40	58.3	8.46	1.70
3C	AlumGraphite (Elev. Cure Adhesive)	132.5	19.22	36 <b>.</b> 6	5.31	3.62
8c	AlumGlass	96.2	13.95	30.8	4.47	3.12
12C	AlumGraphite (Room-Temp. Cure Adhesive)	98.6	14.30	36.7	5•32	2.69

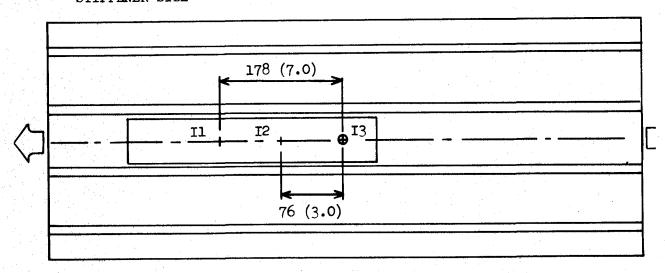
# FIGURE 24

# LOCATIONS OF PHOTOSTRESS COATINGS PANEL NO. 1C

(Dimensions in millimeters and inches respectively)

# FACE SHEET SIDE



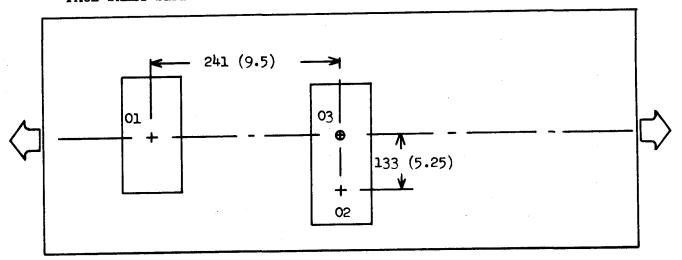


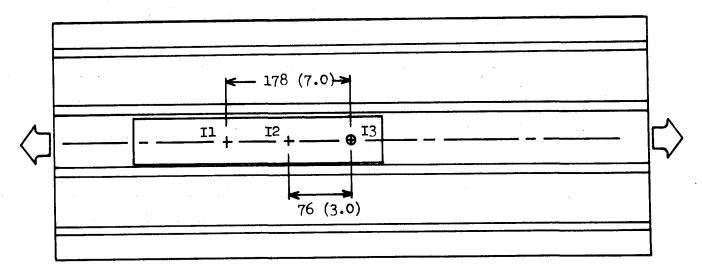
# FIGURE 25

# LOCATIONS OF PHOTOSTRESS COATINGS PANEL NO. 3C

(Dimensions in millimeters and inches respectively)

# FACE SHEET SIDE

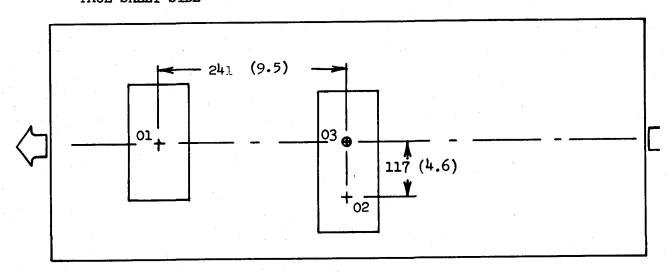


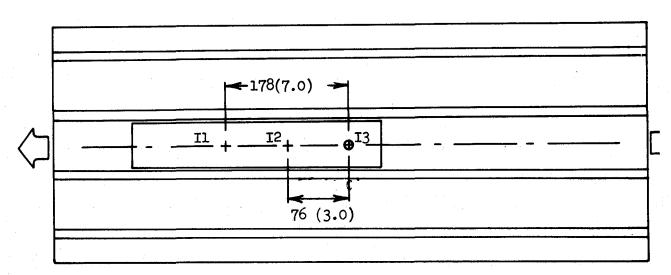


# FIGURE 26 LOCATIONS OF PHOTOSTRESS COATINGS PANEL NO. 8C

(Dimensions in millimeters and inches, respectively)

# FACE SHEET SIDE



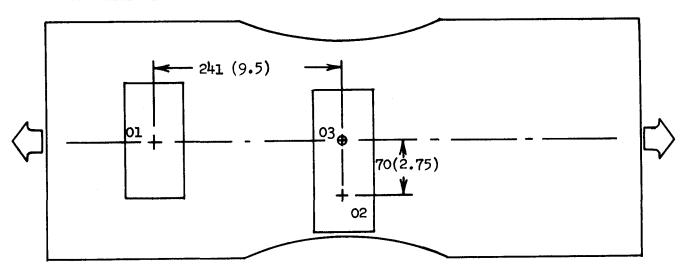


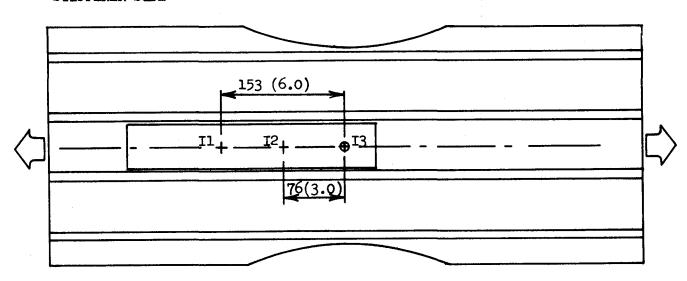
# FIGURE 27 LOCATIONS OF PHOTOSTRESS COATINGS

PANEL NO. 12C

(Dimensions in millimeters and inches, respectively)

# FACE SHEET SIDE





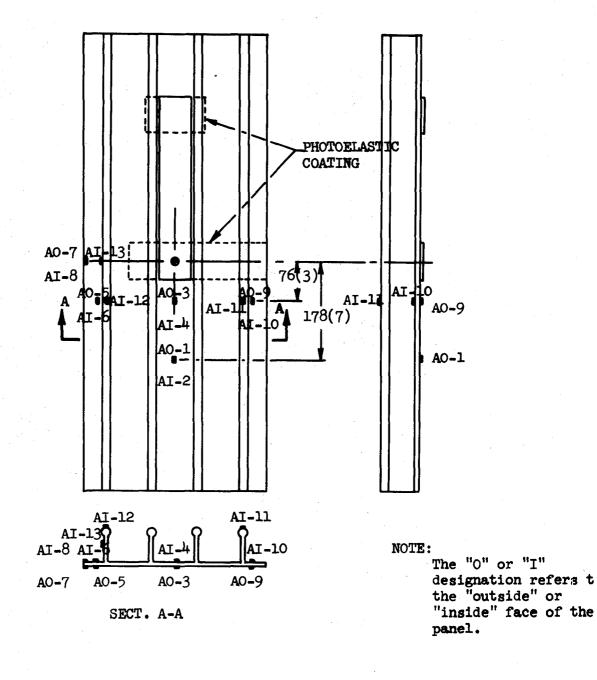


FIGURE 28 STRAIN GAGE AND PHOTOELASTIC COATING LOCATIONS
PANEL # 1C, #3C and #8C
(Dimensions in millimeters and inches respectively)

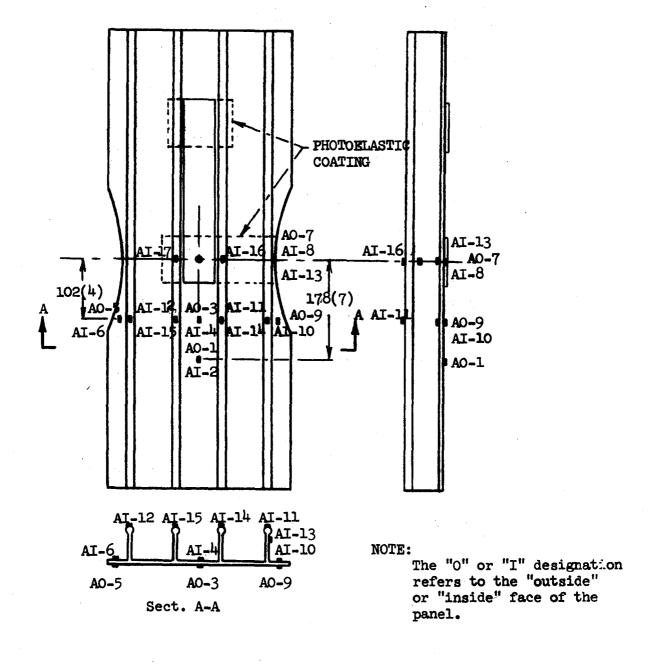


FIGURE 29 STRAIN GAGE AND PHOTOELASTIC COATING LOCATIONS PANEL #12C (Dimensions in millimeters and inches respectively)

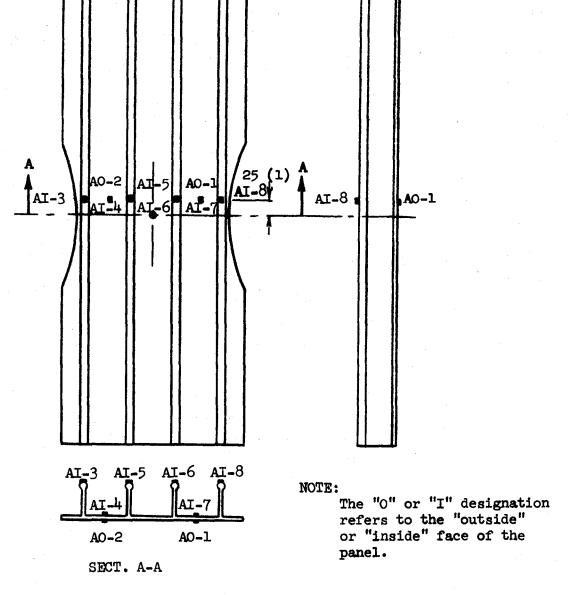


FIGURE 30 STRAIN GAGE LOCATIONS, PANELS WITHOUT PHOTOELASTIC COATING (DIMENSIONS IN MILLIMETERS AND INCHES RESPECTIVELY)

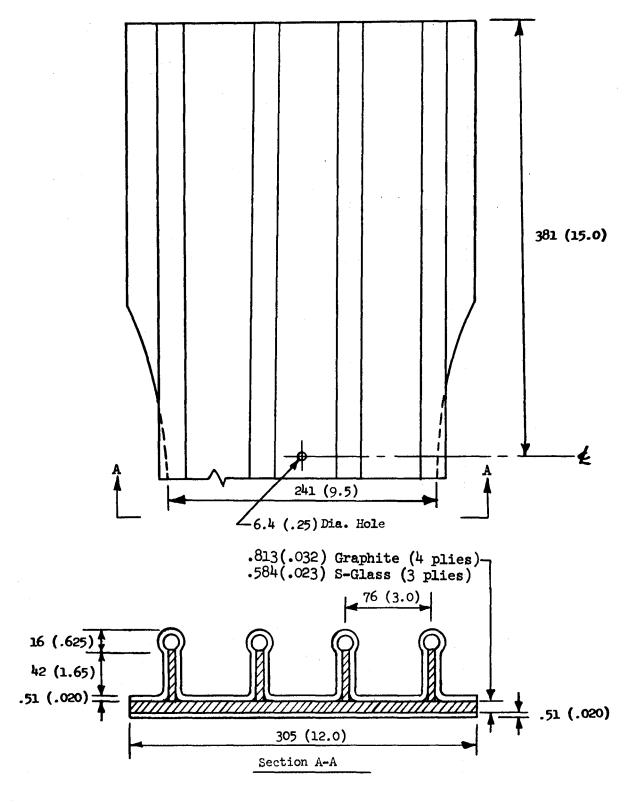


Figure 31 Composite-Reinforced, Integrally Stiffened Metal Panel (Dimensions in Millimeters and Inches, Respectively)

### PANEL FATIGUE CRACK GROWTH

## Objective

The objective of this portion of the program was to fatigue cycle the composite-reinforced, integrally stiffened metal panels and monitor and record the metal crack growth.

## Approach

Composite-reinforced, integrally stiffened metal panels were fabricated and tested in fatigue to examine the metal crack growth characteristics. The metal crack growth for the different compositemetal systems was compared to each other for different levels of metal stress and also compared to the crack growth in an all metal integrally stiffened panel.

# Testing and Results

The composite-reinforced, integrally-stiffened metal panels were cycled in tension-tension fatigue at constant amplitude and load ratio R=0.10. The panels were tested at several levels of aluminum gross stress and the crack length (2a) and number of cycles were recorded. The gross stress in the metal stiffener and face sheets and composite materials was calculated in terms of the total applied load using Equations (2) and (3).

The panels were tested in a load fixture with hydraulic jacks with a load range of 0 to 445 kN (100,000 lbf). The loading frequency ranged from 1-10 Hz (60-600 cmp). The number of cycles required to initiate a crack in the metal at the test section hole was recorded. A paper grid scale with 1.27 mm (0.05 in.) increments was attached to the panel in line with the primary cracks in the panel test section. A 30-powered transit with crossed hair lines and mounted on adjustable stands was used to read crack length increments every 1.27 mm (0.05 in.). There were primarily four cracks to monitor, a crack to each side of the panel centerline on both the stiffener and face sheet sides.

Figure 32 shows the results of the crack growth of the panel face sheet at three different stress levels for three composite-metal-adhesive combinations. The panel crack length (2a) and the number of cycles are plotted starting at a crack length (2a) of 25.4 mm (1.0 in.). Each band shown contains the three different panels examined, aluminum-graphite bonded with AF-126 adhesive, aluminum-graphite bonded with EA-927R adhesive and aluminum-glass bonded with AF-126 adhesive.

Figures 33 through 47 show the results of the fifteen tested panels. These figures show plots of the primary crack length (2a) on both the face and stiffener sheets versus number of cycles. The recorded test data is given in Appendix F.

## Discussion

It can be seen from Figure 32 that, as the gross stress increases, the crack growth in the aluminum also increases, but the crack growth in the composite-metal panels with a stress 2/3's greater than the all-aluminum panel stress is still much less than the all-aluminum panel crack growth. The aluminum-graphite panels with the two different adhesive systems show to have no significant differences in crack growth characteristics. The aluminum-glass panels have a much faster crack growth rate than the aluminum-graphite panels for the same stress levels, e.g., approximately 100% faster at 103 MN/m² (15 ksi), 15% at 138 MN/m² (20 ksi), and 65% at 172 MN/m² (25 ksi).

## Conclusions

The crack growth panel tests show that the metal crack growth rates for the composite-metal systems investigated are essentially constant. The metal crack growth rates in the composite-metal systems are much slower than the all metal crack growth rates. This is true for the same gross metal stress and also for a gross metal stress which is 2/3's higher than the all metal stress (see Figure 32).

For the same gross metal stress, the crack growth rate of the aluminum-glass system is much higher than the aluminum-graphite system. Also, there is generally no significant crack growth rate differences for the two adhesives used in the aluminum-graphite system.

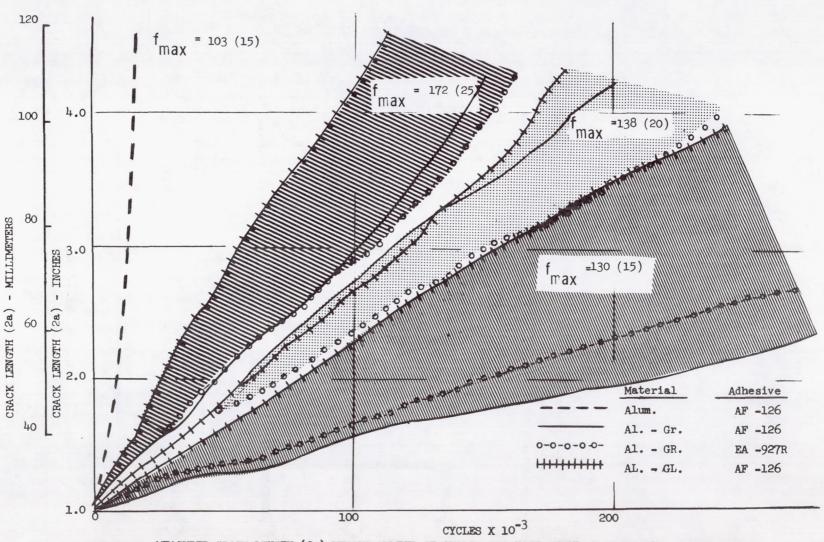


FIGURE 32 MEASURED CRACK LENGTH (2a) VERSUS NUMBER OF CYCLES FOR FACE SHEET OF COMPOSITE - REINFORCED, INTEGRALLY STIFFENED METAL PANELS (MAXIMUM STRESS, f NAX, IS IN MN/m² AND KSI RESPECTIVELY)

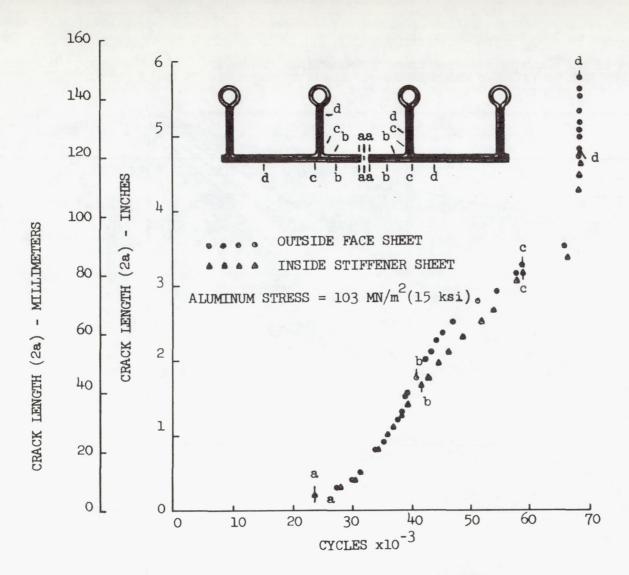


FIGURE 33 MEASURED PRIMARY CRACK LENGTH (2a) VERSUS NUMBER OF CYCLES, PANEL #1C

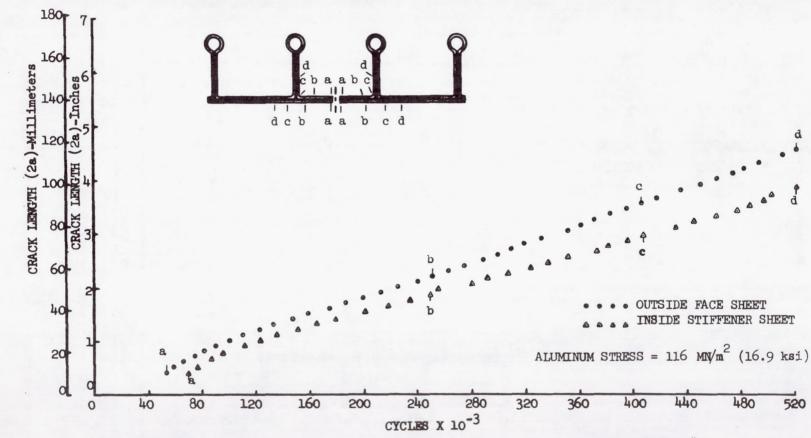


FIGURE 34 MEASURED PRIMARY CRACK LENGTH (2a) VERSUS NUMBER OF CYCLES, PANEL #2C

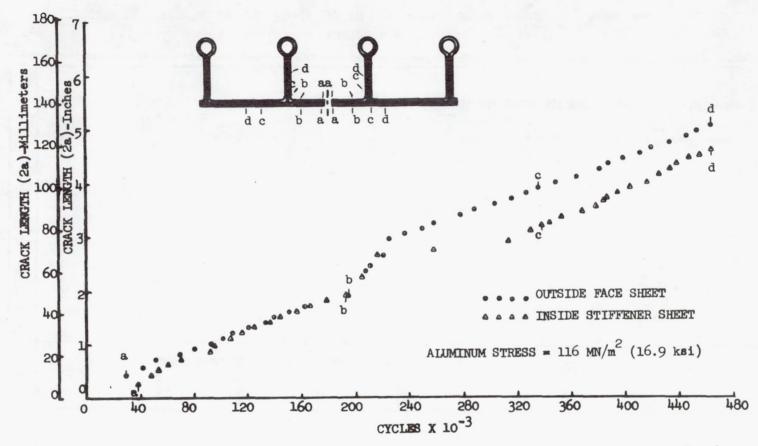


FIGURE 35 MEASURED PRIMARY CRACK LENGTH (2a) VERSUS NUMBER OF CYCLES, PANEL #3C

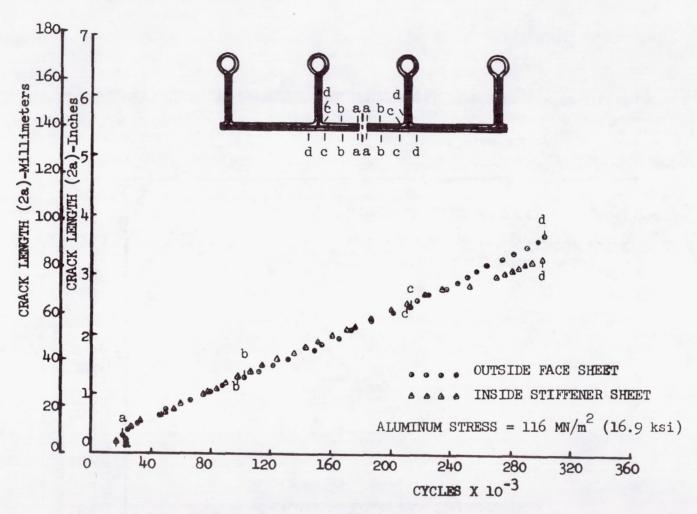


FIGURE 36 MEASURED PRIMARY CRACK LENGTH (2a) VERSUS NUMBER OF CYCLES, PANEL #4C

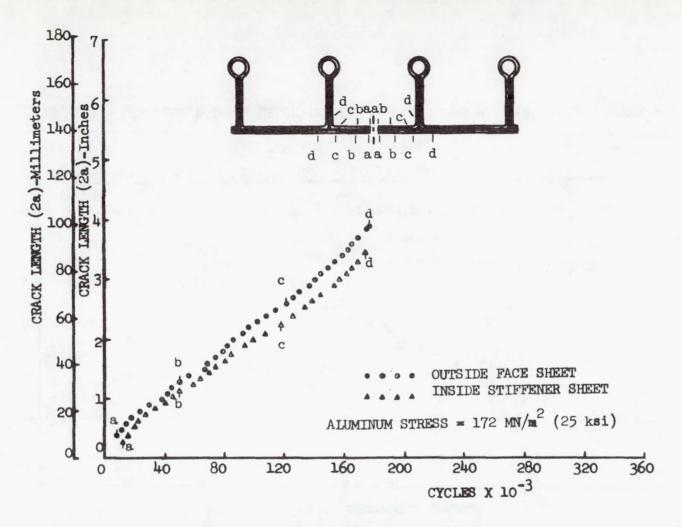


FIGURE 37 MEASURED PRIMARY CRACK LENGTH (2a) VERSUS NUMBER OF CYCLES, PANEL #5C

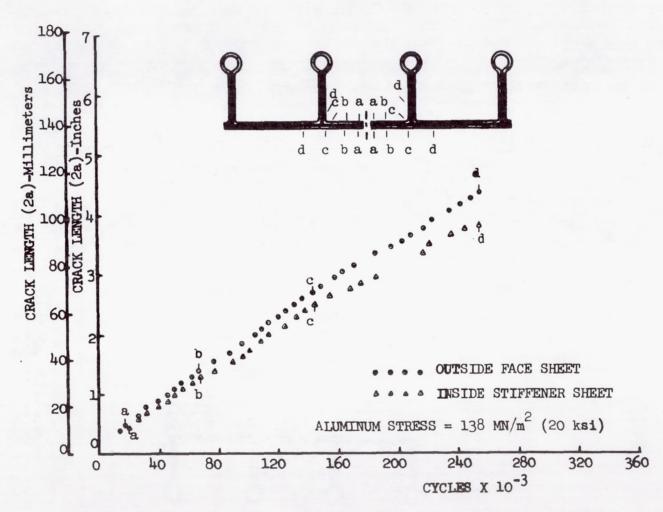


FIGURE 38 MEASURED PRIMARY CRACK LENGTH (2a) VERSUS NUMBER OF CYCLES, PANEL #6C

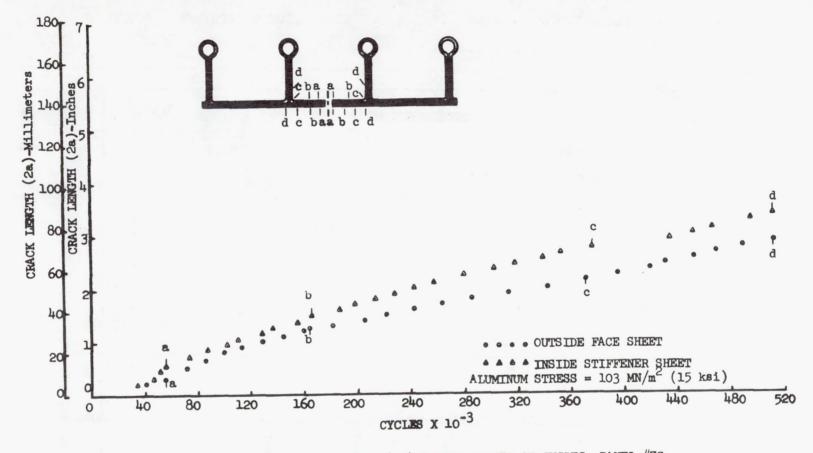
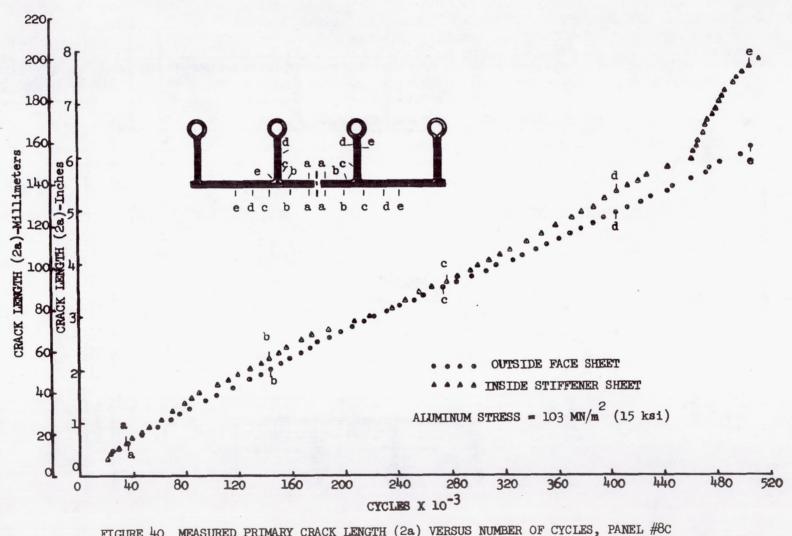


FIGURE 39 MEASURED PRIMARY CRACK LENGTH (2a) VERSUS NUMBER OF CYCLES, PANEL #7C



MEASURED PRIMARY CRACK LENGTH (2a) VERSUS NUMBER OF CYCLES, PANEL #8C

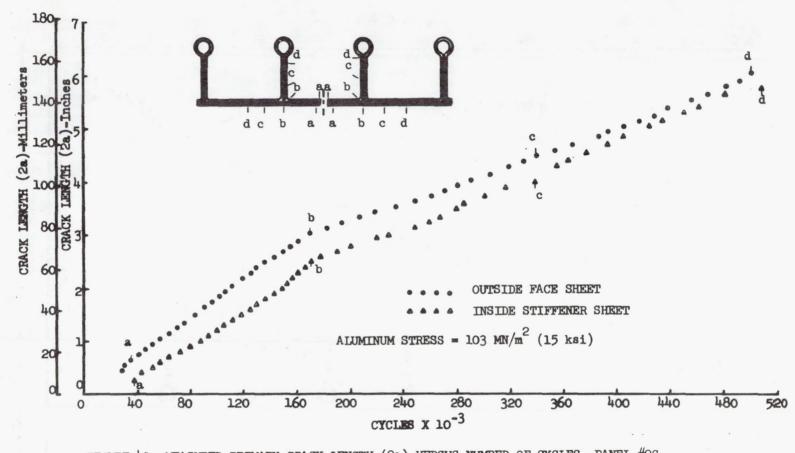


FIGURE 41 MEASURED PRIMARY CRACK LENGTH (2a) VERSUS NUMBER OF CYCLES, PANEL #9C

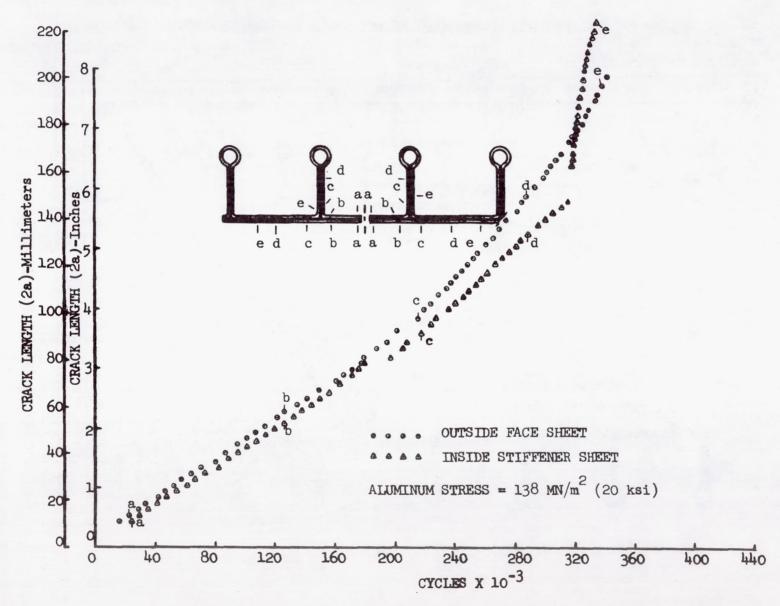


FIGURE 42 MEASURED PRIMARY CRACK LENGTH (2a) VERSUS NUMBER OF CYCLES, PANEL #loc

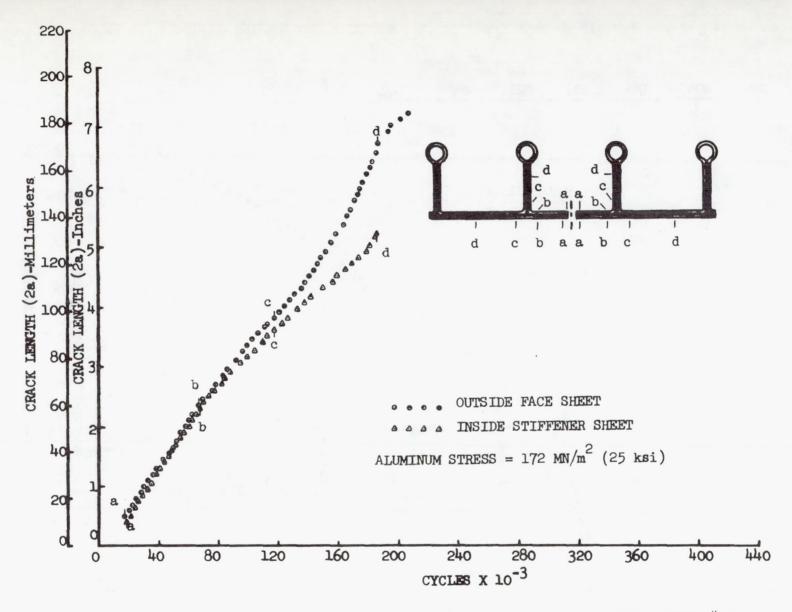


FIGURE 43 MEASURED PRIMARY CRACK LENGTH (2a) VERSUS NUMBER OF CYCLES, PANEL #11C

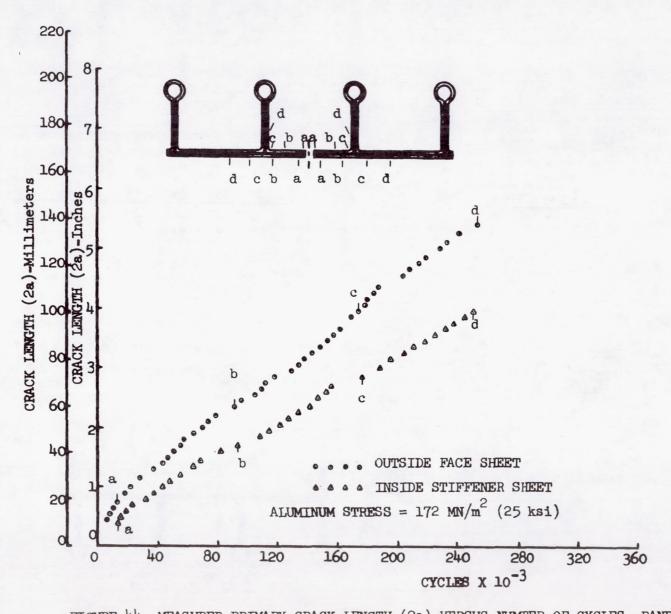


FIGURE 44 MEASURED PRIMARY CRACK LENGTH (2a) VERSUS NUMBER OF CYCLES, PANEL #12C

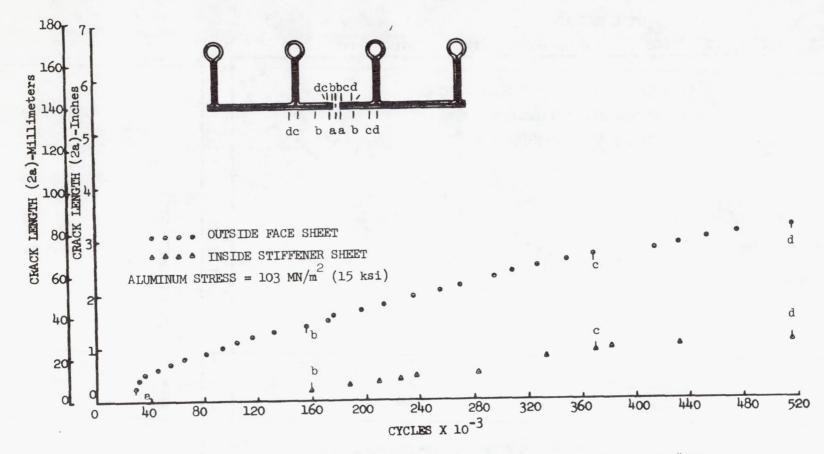


FIGURE 45 MEASURED PRIMARY CRACK LENGTH (2a) VERSUS NUMBER OF CYCLES, PANEL #13C

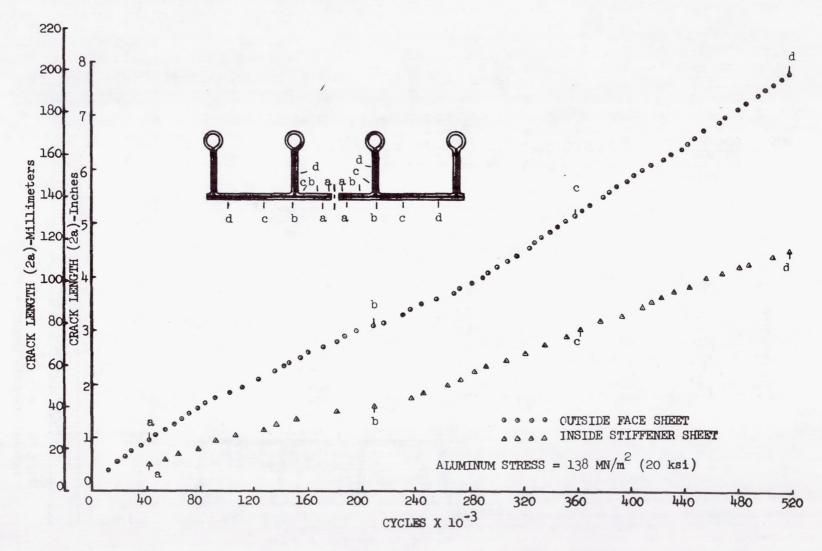


FIGURE 46 MEASURED PRIMARY CRACK LENGTH (2a) VERSUS NUMBER OF CYCLES, PANEL #14C

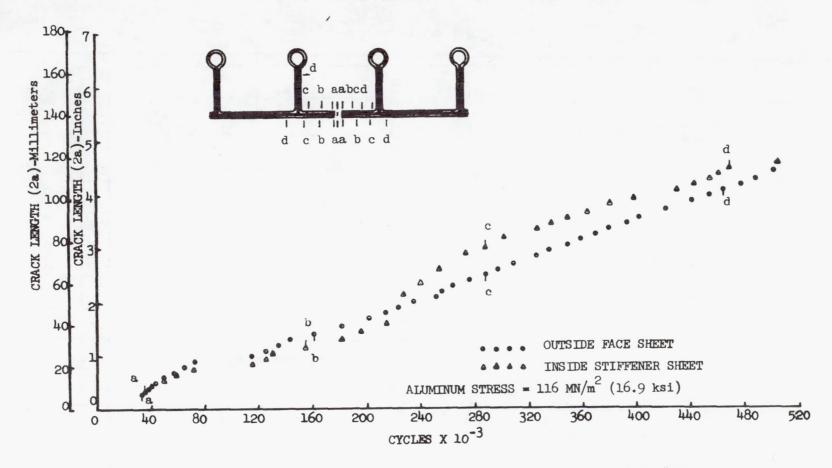


FIGURE 47 MEASURED PRIMARY CRACK LENGTH (2a) VERSUS NUMBER OF CYCLES, PANEL #15C

## PANEL RESIDUAL STRENGTH

## Objective

The objective of this portion of the program was to determine the remaining residual strength of the composite-metal panels after the metal was significantly cracked.

### Approach

After the panels had been cycled and the stiffener and face sheets were significantly cracked, the panels were pulled statically to failure to determine the residual strength. This value of strength was compared to the design limit load to determine if the fatigue cycling had a detrimental effect on the strength of the composite material.

### Results and Discussion

Table XVIII gives the results of the residual strength tests. The design limit load was determined by equation (9).

The amount of metal cracked during fatigue cycling varied from 22-53 percent. Panel 6C residual load was not recorded and panels 12C, 13C, and 14C failed under the end fittings through a row of bolt holes. Figure 48 shows that the residual strength of the composite-metal panels generally exceeded the design limit load and thus satisfied the design criteria. However, some degradation of original composite strength is indicated because the residual strength is generally less than the original ultimate strength (150% limit strength).

#### Conclusions

A composite-reinforced metal panel can be designed to sustain limit load after a significant area of metal has been cracked in fatigue and, at the same time, to be as light as an all-metal panel.

TABLE XVIII
PANEL RESIDUAL STRENGTH

PANEL No.	MAX. FATIGUE STRESS IN METAL		TOTAL	% OF METAL	COMPOSITE AREA		DESIGN LIMIT		RESIDUAL STRENGTH		% of design Limit load
	MN/m <sup>2</sup>	(ksi)		CRACKED	mm <sup>2</sup>	(in <sup>2</sup> )	/kN	(lbf x 10 <sup>-3</sup> )	kN	(lbf x 10 <sup>-3</sup> )	
1C	103	15.0	68,658	47	513.8	.79643	206.8	46.5	52.2	11.7	25
2C	116	16.9	779,842	41	361.6	.5606	251.8	56.6	223.3	50.2	89
3C	116	16.9	465,606	26	418.0	.6480	291.3	65.5	364.7	82.0	125
4C	116	16.9	304,134	22	361.6	.5606	251.8	56.6	263.8	59.3	105
5C	172	25.0	176,452	23	361.6	.5606	251.8	56.6	302.5	68.0	120
6C	138	20.0	254,394	25	361.6	.5606	251.8	56.6	3	3	
7C	103	15.0	1,020,520	36	361.6	.5606	251.8	56.6	539.1	121.0	124
8c	103	15.0	554,529	43	275.0	.4264	271.3	61.0	453.7	102.0	167
9C	103	15.0	631,049	53	237.9	.3689	235.3	52.9	286.9	64.5	122
10C	138	20.0	341,983	53	237.9	.3689	235.3	52.9	276.9	62.3	118
11C	172	25.0	206,476	40	237.9	.3689	235.3	52.9	257.5	57.9	109
100	172	25.0	252,993		418.0	.6480	291.3	65.5	253.5	57.0	87
12C		15.0	1,853,408	A	418.0		291.3	65.5	371.8	83.6	127
13C	103 138	20.0	530,897	<b>A</b>	418.0		291.3	65.5	279.3	62.8	96
14C 15C	116	16.9	506,465	28	361.6		251.8	56.6	351.4		141

Based on equation (9) and limit stresses in Table I.

All aluminum area.

Residual strength not recorded.

Failed at end fittings outside of test area.

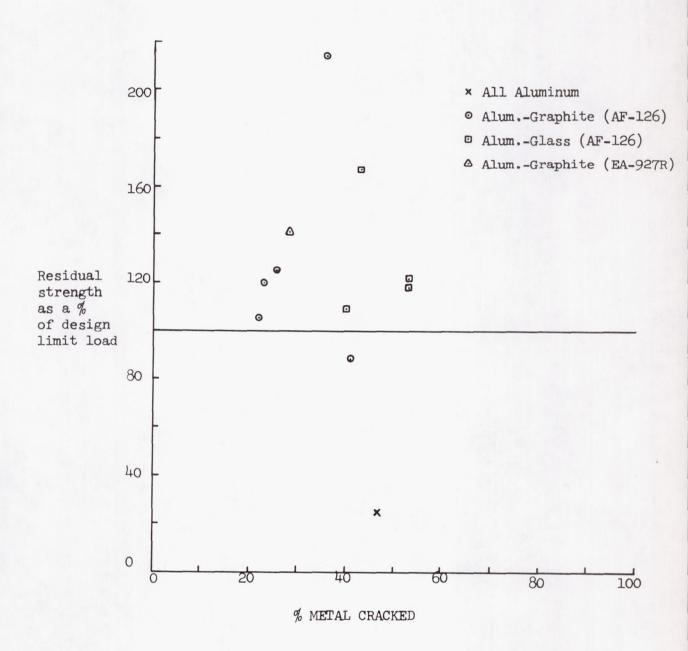


FIGURE 48 PERCENT OF PANEL RESIDUAL STRENGTH vs. PERCENT OF METAL CRACKED IN PANEL

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#### SUMMARY OF CONCLUSIONS AND RECOMMENDATIONS

The fail-safe design criteria established in this investigation was:

- 1. The composite material would support the total load at limit stress after the metal had completely failed.
- 2. The weight of the composite-metal system would be equal to that of an all metal system which would carry the same total load at limit stress.

This criteria produced a stiffness ratio  $\mu$  of 0.57 for the aluminum-graphite system and 0.29 for the aluminum-glass system. If higher values of stress were allowed to be carried by the composite, after metal failure, then the composite-metal system could weigh much less than an all metal system; however, this produces a lesser degree of fail-safety.

The shear strength of the AF-126 and EA-927R adhesives was equivalent to present values used in aircraft design and were used to bond the composite-reinforced, integrally stiffened metal panels. The EC-1614 paste adhesive could not be used because of the complexity of the panels.

The composite-metal structure was sensitive to static stress concentration effects although a higher average ultimate tensile strength was obtained. However, the stress concentration effects in fatigue is about the same as in the all metal, but because of the better fatigue characteristics of the composite material, the composite-metal system provides extended fatigue life for the same or less weight. The coupon fatigue data showed no significant difference in the fatigue characteristics of the aluminum-graphite system when using either the elevated or room temperature cure adhesive.

The composite-reinforced, integrally stiffened metal panels can be loaded uniformly under constant amplitude fatigue loading. The metal crack growth rates produced by fatigue loading are essentially constant and much slower than that in an all metal structure. The crack growth in the aluminum-graphite system is less than that of the aluminum-glass system for the same gross metal stress and for the composite-metal-adhesives used no significant debonding occurred in the vicinity of the crack. Therefore, for the composite-metal-adhesive systems investigated there was sufficient strength in the system to transfer the load from the metal to the composite when the metal was cracked and maintain its effective fail-safeness.

The amount of metal cracked during fatigue cycling varied from 22-53 percent. The residual strength of the composite-metal panels generally exceeded the design limit load. However, some degradation of original composite strength is indicated because the residual strength is generally less than the original ultimate strength (150% limit strength).

# SUMMARY OF CONCLUSIONS (Cont.)

Future studies of composite-reinforced, integrally stiffened metal panels should concentrate on the refinement of the design criteria involving metal-composite thickness relationships required to insure fail-safeness, development of joining techniques suitable for airframe applications, and defining better fracture criteria associated with crack growth and residual strength characteristics.

## APPENDIX A

# CONVERSION OF SI UNITS TO U. S. CUSTOMARY UNITS

The International System of Units (SI) was adopted by the Eleventh General Conference on Weights and Measures held in Paris in 1960. Conversion factors required for units used herein are given in the following table:

Physical Quantity	SI Unit (*)	Conversion factor (**)	U.S. Customary Unit
Density	kilograms per cubic meter (kg/m <sup>3</sup> )	0.3613 X 10 <sup>-4</sup>	lbm/in <sup>3</sup>
Force	newtons (N)	0.2248	lbf
Length	meters (m)	0.3937 x 10 <sup>2</sup>	in.
Mass	kilograms (kg)	2.205	lbm.
Stress, Modulus	newtons per sq. meter (N/m <sup>2</sup> )	0.145 X 10 <sup>-6</sup>	ksi = $10^3  lbf/in^2$
Temperature	degrees Kelvin (K)	<u>9</u> к - 459.67	$\circ_{ m F}$

\*Prefixes to indicate multiple of units are as follows:

Prefix	Multiple		
mega (M) kilo (k) milli (m)	10 <sup>6</sup> 10 <sup>3</sup> 10 <sup>-3</sup>		

\*\*Multiply value given in SI Unit by conversion factor to obtain equivalent in U. S. Customary Unit.

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#### APPENDIX B

#### TEST MATERIALS

The following materials were used to fabricate the test specimens: Aluminum sheet was alloy 7075-T6 per QQ-A-250/13.

Graphite was obtained from Fothgill/Harvey, pre-preg sheets of 305 mm. (12 in.) wide and 1168 mm. (46 in.) long (ERIA 4617 resin with courtalds fiber).

S-glass was "Scotchply" type 1009-26S and was obtained from the Minnesota Mining and Manufacturing Company.

AF-126 adhesive was obtained from the Minnesota Mining and Manufacturing Company. This is a film adhesive of epoxy resin impregnated into a dacron fiber mat and thickness of 0.127mm (0.005 in.). Liquid primer EC-2320 was used on all surfaces.

Epon 927R adhesive was obtained from the Hysol Division of the Dexter Corporation, Pittsburg, California. This is a film adhesive of room temperature curing epoxy resin impregnated into a type 112 glass scrim cloth. The material thickness is 0.127mm (0.005 in.). Epon 927 surface conditioner is used to prime all surfaces.

EC-1614 adhesive was obtained from Minnesota Mining and Manufacturing Company. This is a paste adhesive of room temperature curing epoxy resin.

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#### APPENDIX C

#### TEST SPECIMEN FABRICATION

Two types of test specimens were examined during this program: (1) composite-reinforced metal coupons, and (2) composite-reinforced, integrally stiffened metal panels. The coupon specimens are shown in Figures 3 and 20, and the integrally stiffened test panel is shown in Figure 31.

The fabrication of composite reinforced metal coupons was accomplished as follows: (1) layed-up and cured composite material, (2) bonded aluminum to composite to form sandwich structure, and (3) rough trimmed and machined to final configuration.

The composite reinforced integrally stiffened panels were fabricated as follows: (1) layed-out and formed metal details with the aid of final sizing dies, (2) layed-up and cured composite material, (3) pre-fit composite and aluminum details and bonded, (4) rough trimmed and machined to desired configuration, and (5) attached test fixture to panel to prepare for test. The all metal panels were fabricated as above but bonded without the composite material inserted.

The test specimens used to qualify the shear strength of the bonding adhesives were double lap shear specimens. Several plies of composite were bonded between two pieces of aluminum with a one-half inch over lap. A spacer was inserted between the aluminum for the remaining length and shims were bonded onto the surface of the composite at the other end for griping, see Figure 2. The total specimen width was 25.4 mm. (1.0 in.).

#### Metal Forming

The results of an investigation preformed under an initial program to determine the most applicable methods for producing the integrally stiffened inner sheet at both low and high manufacturing rates have shown that two basic requirements exist. First, each stiffener leg must be preformed incrementally and secondly, final sizing of the stiffener legs must be accomplished either by incremental forming or final sizing full panels in one operation.

Methods of fabricating the preformed parts are separated into two categories representing low and high production rates. Producing the preform parts at low rates and low volume requires only a simple brake bending operation. This type of operation, although relatively slow, produces close tolerance parts utilizing pre-drilled alignment holes. It was found that variations in springback, bend radii and gage had no significant effect on 7075 aluminum preformed parts since these materials were formed at room temperature in an annealed state and subsequently heat treated.

The final sizing of all integrally stiffened panels scheduled during the proposed program were performed utilizing die and rod inserts. The forming sequence is illustrated by the preformed and final formed panels in Figures C-1 and C-2. The only variation consisted of a spacer placed between the stiffener sections to allow room for composite insertion prior to bonding.

## Composite Material Processing

There were two basic composite materials to undergo the processing operation: (1) Courtauld's HT/S graphite pre-impregnated epoxy fiber, and (2) 3 M's 1009-26S glass prepreg. In order to process the integrally stiffened panels and coupons the composite material required two operations: (1) lay-up and (2) autoclave cure.

# Lay-up Procedure

- (1) Remove composite material from 256°K (0°F) storage. Allow to warm to room temperature before unsealing storage container.
- (2) Cut required plies with a sharp knife, taking care not to disorient fibers. Use clean cotton gloves to prevent soiling plies.
- (3) Plies shall be cut such that no tow end-to-end splices shall be incorporated in the laminate assembly.
- (4) Plies shall be cut such that side-to-side joints (joining of sheets or tape within a ply) are staggered a minimum of 25.4 mm. (1.0 in.) within a thickness of five plies.
- (5) Tows shall be kept within  $+1-1/2^{\circ}$  of the intended orientation.
- (6) Side-to-side gaps between plies shall be located no more than 0.38 mm. (015 in.) apart with no overlap.
- (7) Apply nylon peel ply cloth to both sides of laminate.

# Autoclave Cure

- After lay-up operation vacuum bag composite structure with nylon film and apply suitable bleeding and release system. Locate in autoclave.
- (2) Apply vacumm (686 mm. (27 in.) Hg. min). Apply  $.516 \pm .03 \text{ MN/m}^2$  (75 + 5 psi) autoclave pressure and vent to atmosphere.
- (3) Cure component as follows:
  - (a) Courtauld's HT/S graphite 1 hour @ 435°K (325° F),.516 MN/m<sup>2</sup> (75 psi)
    - 1 hour @ 450°K (350° F), .516 MN/m<sup>2</sup> (75 psi)
  - (b) 3 M's 1009-26S glass pre-preg. 1 hour @ 440°K (330° F),.210 MN/m<sup>2</sup> (30 psi) 4 hours @ 445°K (340° F) post cure in air circulating oven

Following the autoclave cure and machining operations, the composite materials were then ready for bonding.

# Ultrasonic Inspection

After the lay-up and cure of the composite material, and before bonding to the metal, the composite was inspected using ultrasonic techniques to check for voids, delaminations, etc. This was accomplished by immersed ultrasonic through transmission techniques. (See Appendix D). This provided an initial record of any defects present in the composite material itself and verified the quality of the initial composite lay-up.

# Joining Procedure

Previous development efforts under contract AF 33(615)-3756 have shown good strength to weight indices for the integrally formed panels joined by adhesive bonding. Adhesive bonding was, therefore, selected as the primary joining method for the integrally stiffened panels in this program. This joining procedure included: (1) surface preparation, (2) adhesive application, and (3) final bonding operation.

# Surface Preparation

Prior to bonding these integrally stiffened composite panels the materials comprising the sandwich structure was subjected to a thorough cleaning procedure. The following method (per CVA 8-21A) was used for the aluminum preparation:

- (1) Vapor degrease in stabilized trichloroethylene.
- (2) Alkaline degrease, 15 minutes at 361°K (190° F)
- (3) Rinse.
- (4) Etch for 12 minutes in a solution of the following composition 339°K (150° F) 30 pbw distilled H<sub>2</sub>O 10 pbw concentrated H<sub>2</sub>SO<sub>14</sub> 4 pbw sodium dichromate
- (5) Rinse.
- (6) Force dry at 339°K (150°F) for 15 minutes.

The composite materials used in this program were prepared for bonding with the aid of a nylon peel ply cloth in accordance with CVA-207-7-430. This nylon peel ply remained in place prior to adhesive application in order to insure maximum cleanliness.

# Adhesive Application

Following surface treatment, a primer coat was applied and allowed to dry. This was then followed with the required adhesive film (AF-126 or EA-927R). The panels were then pre-fitted with the aid of clamps and other holding fixtures. The only difficulty experienced was in placing the composite with adhesive into the stiffener leg area. However, this was easily accomplished with a tool to insure the gap remained in the open position during the composite application.

# Bonding Operation

Bonding operations required for joining the integrally stiffened composite panels are essentially the same regardless of the alloy used, differing only in cleaning procedures, and curing cycles employed for the particular materials in use. Based on previous experience, it was anticipated that bonding of the 7075-T6 aluminum composite panels would present no major problem. The elevated temperature adhesive was autoclave processed as shown below:

(1) Form a conventional vacuum bag over assembly with canvas cloth and nylon film; apply clamping device.

(2) Locate bagged panel in autoclave. Apply vacuum [686 mm. Hg. (27 in. Hg) minimum]. Apply .345 MN/m² (50 psi) autoclave pressure; vent vacuum bag to atmosphere.

(3) Cure as follows: 394 + 6°K (250 + 10° F) for one hour at .345 MN/m<sup>2</sup> (50 psi).

The room temperature film adhesive (EA-927R) required 24 hrs. at .345  $MN/m^2$  (50 psi) at ambient temperature, but required no autoclave cure.

## Machining

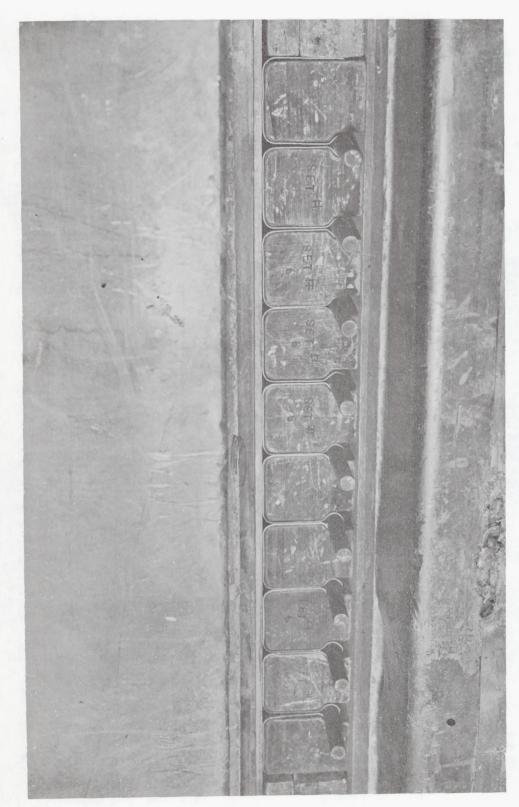
The primary machining requirement in the proposed program was to provide a processing method such that the test articles would sustain a minimum of delamination in the machined area. This was an extremely critical factor in achieving high quality test results. Another consideration was that parallelism be achieved in the test panel ends after test fixture attachment to reduce loading eccentricities which cause premature panel failures.

The principle machining operations encountered in this program included: (1) machining of the composite details, (2) final machining of the test panel periphery, and (3) machining of the coupon radii. All of these processing operations were accomplished on a milling machine utilizing carbide tipped slitting saws, solid carbide end mills and drills while clamped in a holding fixture to provide stabilization of panel details. A template tracing attachment was required for milling the large radii on the coupons. Machining fixtures were also provided to facilitate alignment hole placement in the coupon specimens and in the integrally stiffened composite panel test fixture attachments.

Following FS panel fabrication, aluminum end fittings were machined to size and subsequently adhesively bonded in place to facilitate testing. A tooling jig was prepared to hold the required tolerances while performing the drilling operation. After locating the aluminum test fixture in place, the integrally formed composite panels were then ready for fatigue testing.



FIGURE C-1 TYPICAL INTEGRALLY STIFFENED PREFORM PANEL WITH FINAL SIZING DIES



TYPICAL INTEGRALLY STIFFENED PANEL WITH FINAL SIZING DIES CLOSED FIGURE C-2

#### APPENDIX D

#### ULTRASONIC INSPECTION

Ultrasonic through transmission techniques were used to detect voids and delaminations in composite test specimens used in this program. Immersion tests were conducted following fabrication and contact techniques were used during fatigue testing. C-scan recordings were made where possible to produce a permanent record of the bond quality of test specimens used.

Some bond delamination resulted from drilling and was detected and recorded using immersion techniques. Porosity in the bond line was also detected and was found to be more common in room temperature cure adhesive. Contact techniques were used on specimens during fatigue testing. No delamination was detected at fatigue crack locations. This lack of delamination was confirmed by immersion testing after specimen failure. Considering test sensitivity, no delamination was present greater than one eighth of an inch wide at the fatigue crack locations.

Results of contact ultrasonic through transmission techniques are illustrated in Figures D-1 and D-2. As shown in Figure D-1, six regions of bond delamination were discovered on panel #4C by the contact method before crack growth testing. These regions on the outside stiffeners and the right hand interior stiffener were outlined with grease pencil. The crack growth test was conducted with no apparent affect on primary crack growth (compare Figures 34 and 35 to Figure 36), but an extensive amount of secondary cracking was observed on this particular panel (see Figure F-15).

One large region of bond delamination was detected on panel #5C during crack growth testing as shown in Figure D-2. This region was located between the right hand outside and right hand interior stiffeners above the centerline of the panel, as shown. No apparent affect on primary and secondary cracking was observed as a result of this debond region.



FIGURE D-1 INSIDE STIFFENER SHEET OF PANEL #4C SHOWING REGIONS OF DEBONDING PRIOR TO CRACK GROWTH TEST

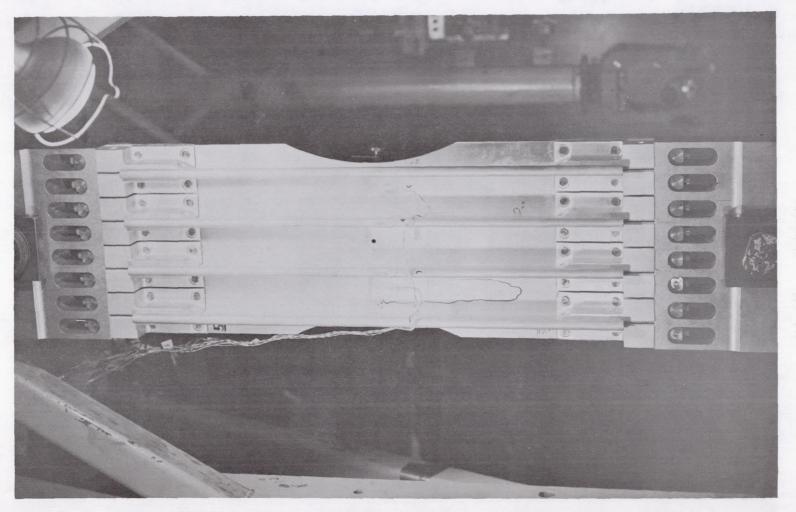


FIGURE D-2 INSIDE STIFFENER SHEET OF PANEL #5C SHOWING A REGION OF DEBONDING THAT WAS DISCOVERED

BY ULTRASONIC THROUGH TRANSMISSION INSPECTION DURING CRACK GROWTH TEST

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#### APPENDIX E

#### TESTING EQUIPMENT

All lap shear and static tensile coupon tests were conducted on the Reihle screw type test machine. Its load range is from 0 - 712 kN (0 - 160,000 lbf.) and has head speeds of 6.4 to 51 mm/min (0.25 to 2.0 in./min.).

All coupon fatigue tests were conducted on the Baldwin Models SF1-U or SF10-U fatigue machines. These machines have load ranges of 0 - 4.45 kN (0 - 1000 lbf.) and 0 - 44.5 kN (0 - 10,000 lbf.) respectively. The cycling rate is 30 Hz (1800 cpm) with a least count of 1000 cycles. The static load is applied by springs and the alternating load is applied by a rotating weight. The time to obtain the desired load level on the specimen is approximately 10 seconds.

All panel fatigue crack growth tests were conducted in the loading fixture shown in Figure E-1. The hydraulic loading jack had a load range of 0 - 222 kN (0 - 50,000 lbf.). The loading frequency ranged from 1 - 10 Hz (60 - 600 cpm).

The loading jack was controlled and monitored by the closed loop electro-hydraulic test equipment as shown in Figures E-2 and E-3. A paper grid scale with 1.27 mm (0.05 in) increments was attached to the panel in line with the primary cracks in the panel test section. A 30X transit (see Figure E-2) with crossed hair lines and mounted on adjustable stands was used to read crack length increments.

The residual strength tests of the panels were conducted on the Reihle test machine previously mentioned or the 0 - 1.33 MN (0 - 300,000 lbf) Baldwin test machine shown in Figure E-4.

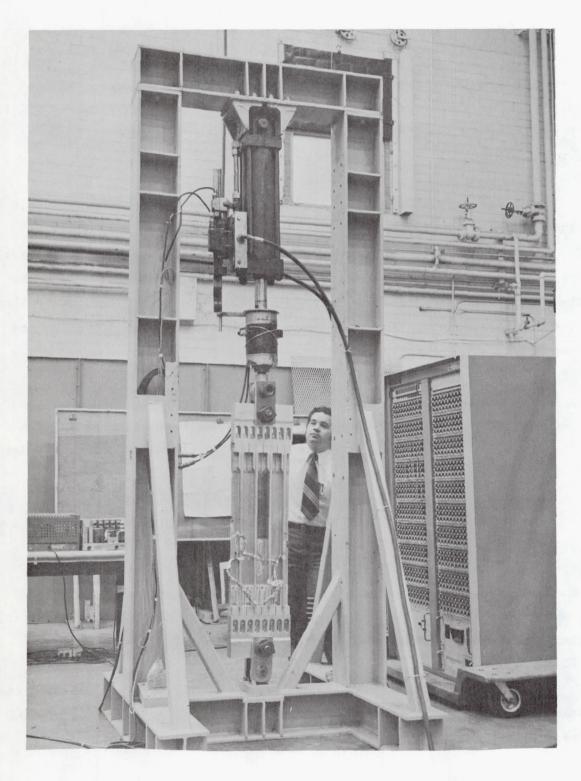
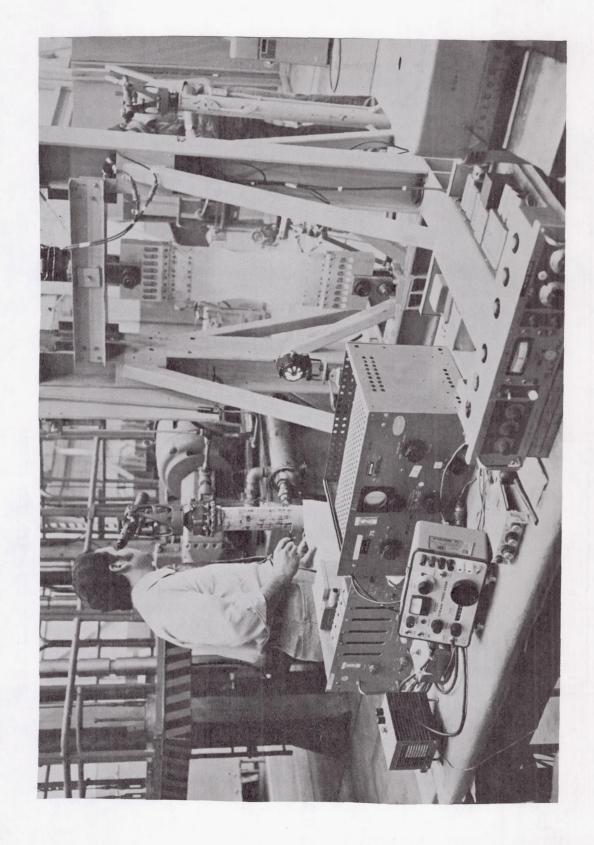


FIGURE E-1 LOADING FIXTURE FOR FATIGUE CRACK GROWTH TESTS



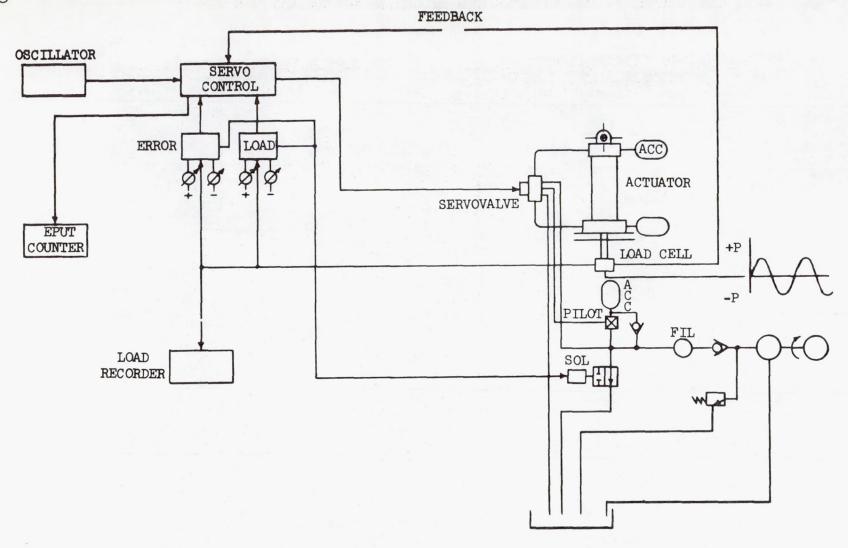


FIGURE E-3 SCHEMATIC DIAGRAM OF LOAD CONTROL AND MONITORING SYSTEM

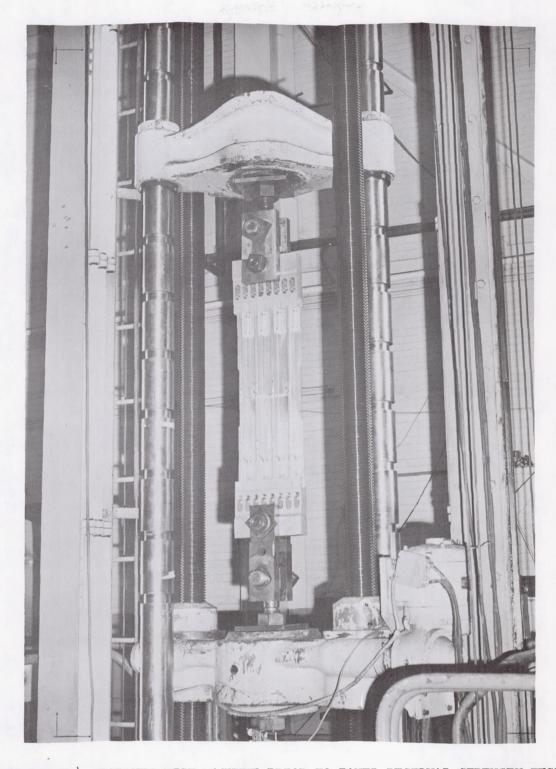


FIGURE E-4 BALDWIN TEST MACHINE PRIOR TO PANEL RESIDUAL STRENGTH TEST

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#### APPENDIX F

# PANEL CRACK GROWTH MEASUREMENT DATA

The purpose of this appendix is to present the individual panel crack growth data. This data includes panel identification, panel photographs, sketches describing secondary crack locations, raw crack growth data, and curves of crack length versus cycles.

Each section of panel crack growth measurement data is identified with panel number, material combination, adhesive, aluminum stress and maximum fatigue load. This data is summarized in Table F-1.

Panel photographs are presented to visually describe the extent of primary and secondary cracking. Primary cracks are defined as those originating at the center quarter inch diameter hole. All other cracks are secondary cracks. (See Figure F-1)

Sketches of secondary crack locations are included to further define secondary cracking. If no secondary cracking occurred then no sketches are included.

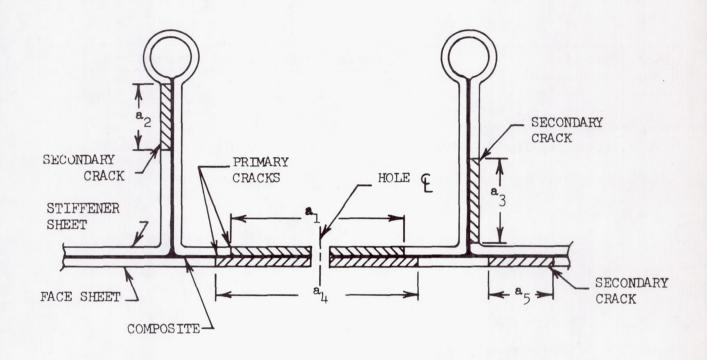
Panel crack growth data is presented to describe the progression of all primary cracks and those secondary cracks that are within the test region.

Crack growth curves are included within this appendix. These curves describe the total crack length, primary crack length plus secondary crack lengths and is measured along the panel surface (see Figure F-1).

TABLE F-1
SUMMARY OF PANEL CRACK GROWTH TEST PARAMETERS

Panel	Material	Maria I sala	Aluminum	Stress	Maximum Fa	tigue Load
No.	Combination	Adhesive*	MN/m <sup>2</sup>	(ksi)	N	(144)
10	All Aluminum	AF-126	103	15	48,930	11,000
2C	Alum-Graphite	AF-126	116	16.9	137,890	31,000
3C	Alum-Graphite	AF-126	116	16.9	155,680	35,000
4C	Alum-Graphite	AF-126	116	16.9	137,890	31,000
5C	Alum-Graphite	AF-126	172	25	200,160	45,000
6c	Alum-Graphite	AF-126	138	20	164,580	37,000
7C	Alum-Graphite	AF-126	103	15	124,540	28,000
8c	AlumGlass	AF-126	103	15	84,510	19,000
9C	AlumGlass	AF-126	103	15	74,730	16,800
10C	AlumGlass	AF-126	138	20	100,080	22,500
11C	AlumGlass	AF-126	172	25	124,540	28,000
12C	AlumGraphite	EA-927R	172	25	200,160	45,000
13C	AlumGraphite	EA-927R	103	15	124,540	28,000
14C	AlumGraphite	EA-927R	138	20	168,690	37,925
15C	AlumGraphite	EA-927R	116	16.9	144,560	32,500

<sup>\*</sup>AF-126 is cured at 389°F (250°F) and EA-927R is cured at room temperature.



STIFFENER SHEET CRACK LENGTH = 
$$2a = a_1 + a_2 + a_3$$
  
FACE SHEET CRACK LENGTH =  $2a = a_1 + a_5$ 

FIGURE F-1
DEFINITION OF CRACK LENGTH MEASUREMENTS

## PANEL #1C

MATERIALS: ALL ALUMINUM

ADHESIVE: AF 126

ALUMINUM STRESS: 103 MN/m<sup>2</sup> (15 ksi)

MAXIMUM FATIGUE LOAD: 48,930N (11,000 lbf)

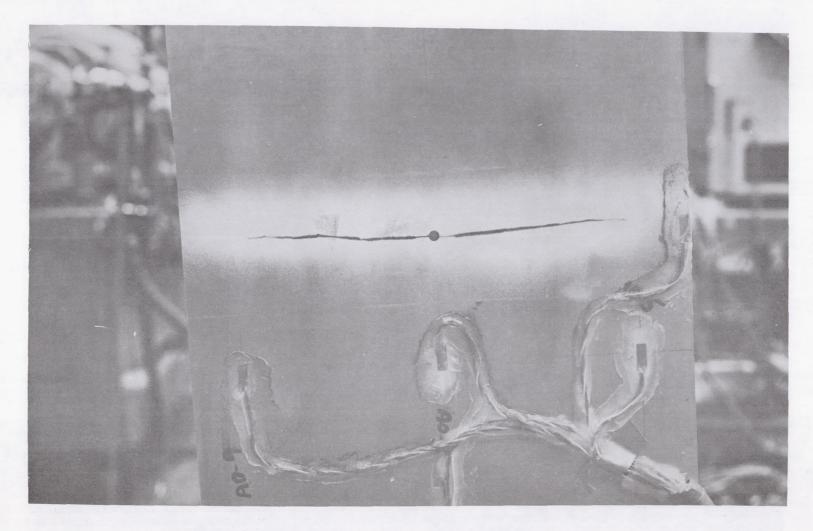


FIGURE F-2 OUTSIDE FACE SHEET AFTER CRACK GROWTH TEST - PANEL #1C



FIGURE F-3 INSIDE STIFFENER SHEET AFTER CRACK GROWTH TEST - PANEL #1C

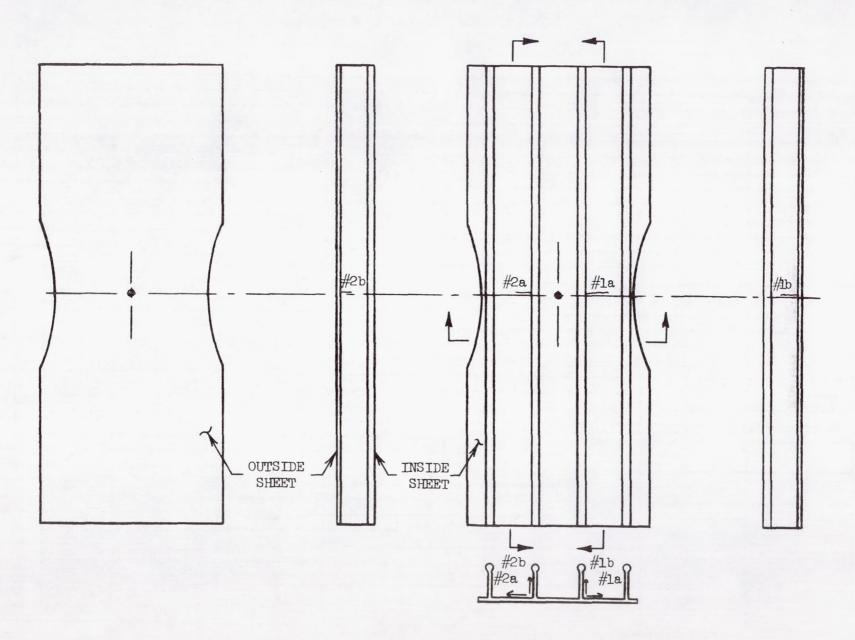


FIGURE F-4 SECONDARY CRACK LOCATIONS ALL ALUMINUM PANEL #1C

Panel	No. 1C		Primary Cra	ack Lengths s (inches)		Secondary Crack Lengths millimeters (inches)	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside		
5 Hz	23,500	5.1 (.20)			5.1 (.20)		
1	26,300	6.4 (.25)			6.4 (.25)		
	27,400						
	28,000				7.6 (.30)		
	29,100	8.9 (.35)					
	29,300				8.9 (.35)		
	30,000	10.2 (.40)					
	30,300				10,2 (.40)		
	30,700				11.4 (.45)		
	31,100						
	1 31,389	12.7 (.50)	-	-	12.7 (.50)		
	33,389	14.0 (.55)		1	14.0 (.55)		
	34,389 34,889		6.4 (.25)	( ) ( 05)	-		
	34,889		( ( )	6.4 (.25)	-		
	35,389	15.2 (.60)	7.6 (.30)		3( = / (=)		
	35,689 36,189 36,789 36,889 37,489		-	8.9 (.35)	16.5 (.65)		
	36,189			10.2 (.40)	+		
	36,789	35 0 ( 50)	170 0 ( 10)	10.2 (.40)	17.8 (.70)		
	36,889	17.8 (.70)	10.2 (.40)	-	17.0 1.101		
	37,489	1200 ( 75)	11.4 (.42)		-		
-	37,889	19.0 (.75)	-	-	19.0 (.75)		
	38,189	-	10 7 ( 50)	12.7 (.50)	19.0		
_	38,289	20 0 ( 00)	12.7 (.50)	12.1 (.)0)	-		
	38,589 38,789	20.3 (.80)	15.2 (.60)				
-	39,189		12).2 (.00)	14.0 (.55)			
-	39,109	22.9 (.90)	1		21.6 (.85)		
-	39,589		16.5 (.65)	1			
	40.389	1	17.8 (.70)				
-	10 100		100000	16.5 (.65)			
-	40,489 40,889	25.4 (1.00)	1				
5 Hz	40.989	1	19.0 (.75)				

Panel	No. 1C		Primary Cra			Secondary Crack Lengths millimeters (inches)		
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	I/H Inside	R/H Inside			
na.ce	-			Inside	Inside			
5 Hz	41,589	26.7 (1.05)						
A	41,689			17.8 (.70)	24.1 (.95)			
	42,389		21.6 (.85)		100			
	42,589				25.4 (1.00)			
	42.789	29.2 (1.15)					-	
	42,989			19.0 (.75)				
	42,989 43,289		22.9 (.90)					
	43.689	30.5 (1.20)						
	44 280		25.4 (1.00)					
	44.389	31.8 (1.25)			27.9 (1.10)			
	44.789	(34,7)		21.6 (.85)				
	45,389		27.9 (1.10)					
	46,189	31.8 (1.25)		22.9 (.90)				
	46,389 46,889				30.5 (1.20)			
	46,889		29.2 (1.15)					
	47,389	34.3 (1.35)						
	48.589			25.4 (1.00)				
	48.689				33.0 (1.30)			
	48,889 51,389		31.8 (1.25)					
	51,389	36.8 (1.45)	33.0 (1.30)	27.9 (1.10)				
	52,389				35.6 (1.40)			
	53,389				36.8 (1.45)			
	54.089			30.5 (1.20)				
	54.389	38.1 (1.50)	35.6 (1.40)					
	54,389 55,389			33.0 (1.30)				
	55,889		36.8 (1.45)					
	57,742	40.6 (1.60)	39.4 (1.55)	38.1 (1.50)	39.4 (1.55)			
	58,489		40.6 (1.60)					
	58,589			39.4 (1.55)				
	59.189			1	40.6 (1.60)			
Y	59,289	41.9 (1.65)						
5 Hz			41.9 (1.65)					

Panel	No. 1C		Primary Cra			Secondary Crack Lengths millimeters (inches)		
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#la Secondary	#1b Secondary	#2a Secondary
5 Hz	60,489			40.6 (1.60)				
5 Hz	65,889	44.4 (1.75)	44.4 (1.75)					
4 Hz	66,389	1.01	1.0.	41.9 (1.65)	43.2 (1.70)			
4 Hz	67,189	45.7 (1.80)			120			
4 Hz	68,437	69.8 (2.75)	49.5 (1.95)	47.0 (1.85)	61.0 (2.40)	31.8 (1.25)	24.1 (.95)	11.4 (.45)
.25 Hz	68,440							
A	68.467							12.7 (.50)
	68,469					34.3 (1.35)		
	68.477	71.1 (2.80)						
Test of	68,484		50.8 (2.00)					
	68,497			48.3 (1.90)				
	68,500	72.4 (2.85)						
	68,510							14.0 (.55)
	68,521	73.7 (2.90)			-			
	68.525						25.4 (1.00)	
	68.527			49.5 (1.95)				
	68,527 68,529		52.1 (2.05)					is considerate the second control of the sec
	68,544					36.8 (1.45)		
	68,545	74.9 (2.95)						
	68,549				63.5 (2.50)			
	68,550				ļ		26.7 (1.05)	
	68,552							15.2 (.60)
	68,556 68,571		53.3 (2.10)					
-	68,571	76.2 (3.00)			-			
	68,579		54.6 (2.15)					
	68,581			50.8 (2.00)	-	-		
	68,590	77.5 (3.05)			-	1001 (5 55)		
	68,593					39.4 (1.55)	07 0 /2 75	
	68,595				-		27.9 (1.10)	
	68.606		55.9 (2.20)	52.1 (2.05)	-			70.0 ( ==)
OF Y	68,607	79 7 /2 101						19.0 (.75)
.25 Hz	68,609	78.7 (3.10)		L				

Panel No. 10		Secondary Crack Lengths millimeters (inches)									
Cycle	No. of	#2b									
Rate	Cycles	Secondary									
5 Hz	60,489 65,389 65,389 67,189 68,437 68,440 68,467 68,469			1.							
5 Hz 5 Hz	65,389										
4 Hz	65,389										
4 Hz	67,189										
4 Hz	68,437	6.4 (.25)									
.25 H	68,440	8.9 (.35) 10.2 (.40)									
	68,467	10.2 (.40)									
-	68 177										
	68,477 68,484 66,497										
	1 68:497										
	68,500 68,510 68,521										
	68,510	11.4 (.45)									
	68,521										
	68,525										
-	63,527										
-	68,529										
	1 60,544				-						
-	1 69 510				-						
_	68 550	12.7 (.50)									
	68.552										
	68.556										
	68,573										
	68.579										
	68,581	14.0 (.55)									
-	68,590										
-	68,527 68,529 68,544 68,545 68,549 68,550 68,556 68,571 68,571 68,571 68,590 68,593 68,593 68,593 68,595 68,606										
-	1 68,595										
	58,000										
.25 H	z 68,609										

Panel :	No. 1C		Primary Cra			Secondary Crack Lengths millimeters (inches)		
Cycle Rate	No. of Cycles	L/H Outside	F/H Outside	I/H Inside	R/H Inside	#la Secondary	#1b Secondary	#2a Secondary
.25 Hz	68,614	80.0 (3.15)						
A	68,618				66.0 (2.60)			
	68,621	81.3 (3.20)						
	68,624						29.2 (1.15)	
	68,627	0 - ( ( )	57.2 (2.25)			11= = (= 00)	-	
	68,632 68,637	82.6 (3.25) 83.8 (3.30)				45.7 (1.80)	-	
	68 630	03.0 (3.30)	58.4 (2.30)					
	68,639		70.4 (2.30)	53.3 (2.10)				
	68.646			1	67.3 (2.65)		The second secon	
	68.647	82.6 (3.35)						
	68,652					48.3 (1.90)		
	68,653							22.9 (.90)
Y	68,656	86.4 (3.40)	(2 25)		-		-	
.25 Hz	68,658	-	59.7 (2.35)			-		
Final (	rack							
Leng	hs	86.4 (3.40)	59.7 (2.35)	53.3 (2.10)	67.3 (2.65)	48.3 (1.90)	29.2 (1.15)	22.9 (.90)
Residua	1 Strengt	h = 52,220N	(11,740 lbf)					
		-				-		

Panel	No. 1C	Secondary Creck Lengths millimeters (inches)								
Cycle Rate	No. of Cycles	#2b Secondary								
.25 Hz	68,614 68,618			- <i>I</i> -,						
	68,621	15.2 (.60)								
	68,614 68,618 68,621 68,624 68,627 68,632 68,637 68,639 68,640 68,640 68,646 68,652 68,653 68,656 68,658									
	68,640									
	68,647	16.5 (.65)								
.25 Hz	68,656 68,658									
Le	Crack engths	16.5 (.65)								
			•		,					
-	-	+								

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### PANEL #2C

MATERIALS: ALUMINUM-GRAPHITE

ADHESIVE: AF 126

ALUMINUM STRESS: 116 MN/m<sup>2</sup> (16.9 ksi)

MAXIMUM FATIGUE LOAD: 137,890N (31,000 lbf)

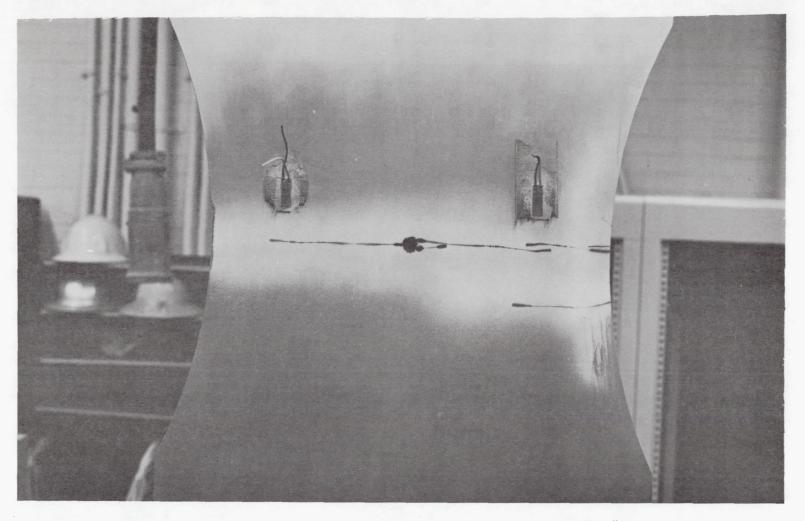


FIGURE F-5 OUTSIDE FACE SHEET AFTER CRACK GROWTH TEST - PANEL #2C

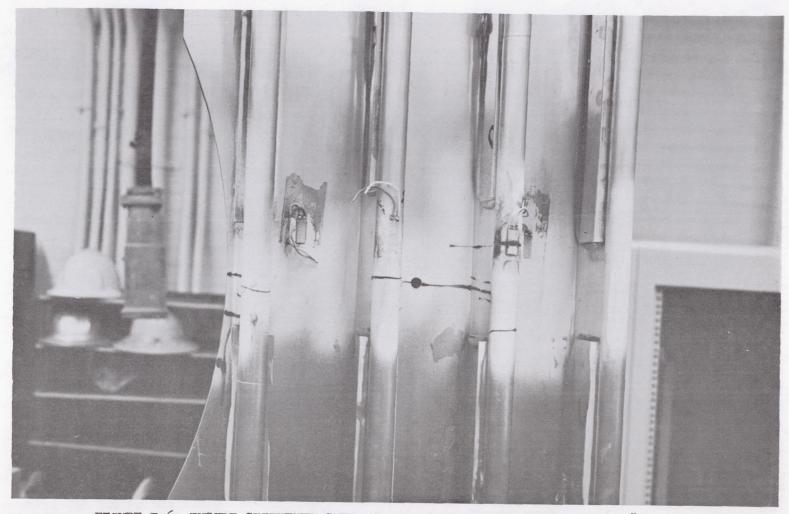


FIGURE F-6 INSIDE STIFFENER SHEET AFTER CRACK GROWTH TEST - PANEL #2C (LEFT OF CENTERLINE)

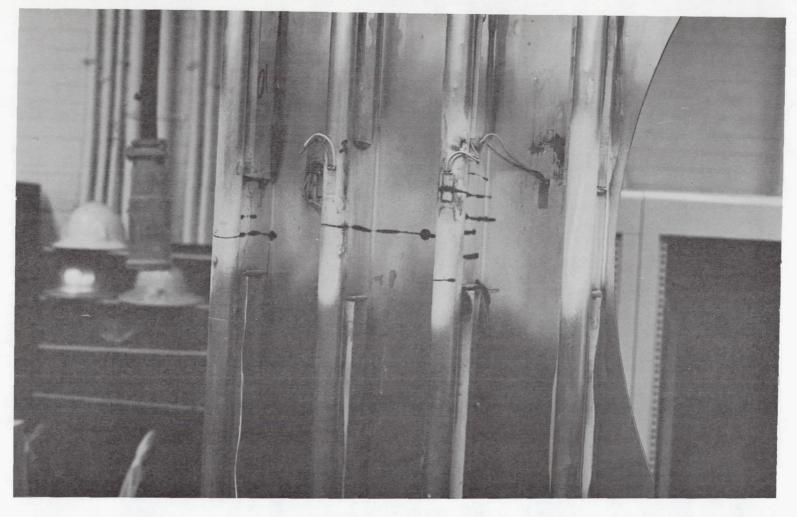


FIGURE F-7 INSIDE STIFFENER SHEET AFTER CRACK GROWTH TEST - PANEL #2C (RIGHT OF CENTERLINE)

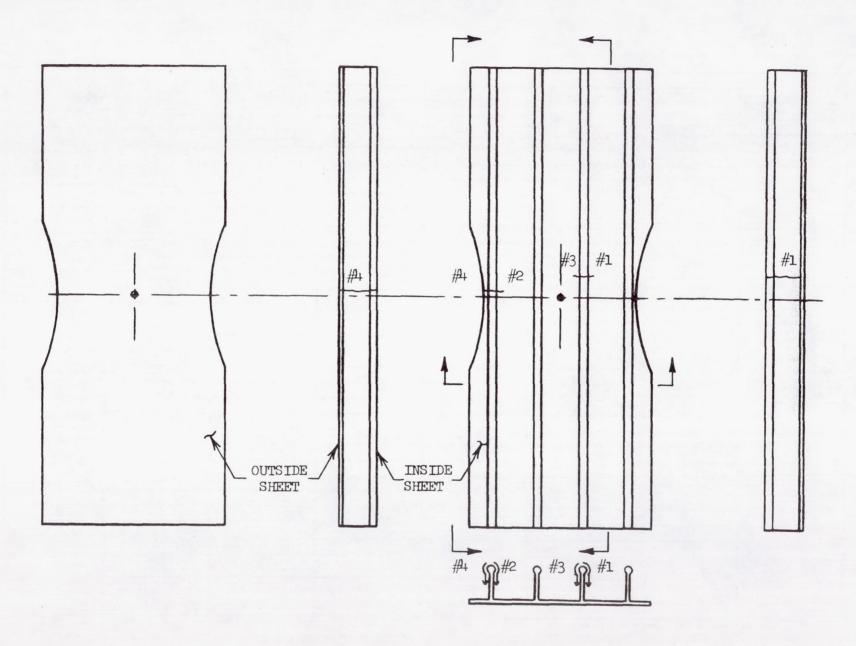


FIGURE F-8 SECONDARY CRACK LOCATIONS ALUMINUM-GRAPHITE PANEL #2C

Panel	No. 2C		Primary Cramillimeters			Secondary Crack Lengths millimeters (inches)
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	
10 Hz	50,350	Initiation	Initiation			
5 Hz	54,166	5.1 ( .20)	5.1 ( .20)			
1	58,266	6.4 ( .25)				
1	60,766	The second secon	6.4 ( .25)	Initiation		
-	65,166	7.6 ( .30)				
	65,666			5.1 ( .20)		
	67.466		7.6 ( .30)			
	68.766				Initiation	
	70,866			6.4 ( .25)		
	73,166	8.9 ( .35)				
	74 066	1			5.1 (.20)	
V	74,966		8.9 ( .35)			
5 Hz	78.366			7.6 ( .30)		
9 Hz	81.066	10.2 (.40)				
	81.866				64 ( 25)	
	81,866 82,966		10.2 ( .40)			
	86,166			8.9 ( .35)		
	88,566				7.6 ( 30)	
	89,666	11.4 ( .45)				
	90,966		11.4 (.45)			
	.93,966			10.2 ( 40)		
	97.066				8.9 ( .35)	
	98.066	12.7 ( .50)				
	102,866			11.4 ( .45)		
	103,066		12.7 ( .50)			
	108,766	14.0 ( .55)				
	109,166				10.2 (.40)	
	112,666		14.0 ( .55)			
	113,766			12.7 ( .50)		
	120 566	15.2 ( 60)				
1	121,366				11.4 ( .45)	
9 Hz	124,866		15.2 ( .60)			

Panel No. 2C			Primary Cra millimeters		Secondary Crack Lengths millimeters (inches)	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	
9 Hz	124,966			14.0 ( .55)		
	131,466	16.5 ( .65)		14.0 ( .)))		
-1-	133,366	100) ( 00))	16.5 ( .65)			
-	134,466		2007		12.7 ( .50)	
	136,366			15.2 ( .60)	1201 (0)0)	
	147,538	17.8 ( .70)	17.8 ( .70)	1705 001		
	150,338	100	100	16.5 ( .65)		
	151,238			1001	14.0 ( .55)	
	158,138	19.0 (75)			TTOU ( COLD)	
	160,438		19.0 ( .75)			
	163,338		1,	17.8 (70)		
	166,338				15.2 ( .60)	
	173.438	20.3 ( .80)				
	173,438 176,238		20.3 ( .80)			
	176,738			19.0 ( .75)		
	182, 238				16.5 (.65)	
	185,238	21.6 (.85)				
	188.438		21.6 ( 85)			
	191,138			20.3 ( .80)		
	191,138				17.8 ( .70)	
	1198.238	22.9 ( .90)				
	201,438		22.9 ( .90)			
	204,938			21.6 ( .85)		
	211,338	24.1 ( .95)				
	214,238		24.1 (.95)			
	215,838				19.0 ( .75)	
	220,538		-	22.9 ( .90)		
		25.4 (1.00)				· · · · · · · · · · · · · · · · · · ·
-	225,738		25.4 (1.00)			
-	233,738			0) = ( ==>	20.3 ( .80)	
9 Hz	235,638	26.7 (1.05)	100 - 10	24.1 ( .95)		

Panel	No. 2C		Primary Cra millimeters			Secondary Crack Lengths millimeters (inches)	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside		
O Hz	249.338				21.6 ( .85)		
7112	249.538			25.4 (1.00)			
T		27.9 (1.10)					
_	251.938	-107 1000	27.9 (1.10)				
_	255,638			26.7 (1.05)	22.9 ( .90)		
_	261,088	29.2 (1.15)					
	265.538		29.2 (1.15)				
	275 338	30.5 (1.20)	30.5 (1.20)	Access of the		:	
-	280,738	1		27.9 (1.10)	24.1 ( .95)		
	289,138	31.8 (1.25)	31.8 (1.25)	(2020)			
	291,938				25.4 (1.00)		
	292.138			29.2 (1.15)	->		
		33.0 (1.30)	33.0 (1.30)				
	305.638	1		30.5 (1.20)			
	305,638 309,538				26.7 (1.05)		
	309.838	34.3 (1.35)	34.3 (1.35)				
	310.838	35.6 (1.40)	35.6 (1.40)				
	323 038				27.9 (1.10)		
	323,038 325,738			31.8 (1.25)	-10) (-100)		
	331.738		36.8 (1.45)				
	333,238	36.8 (1.45)					
	336,738	1			29.2 (1.15)		
	338,538		7	33.0 (1.30)			
_	344.538		38.1 (1.50)				
	351.138	38.1 (1.50)					
	352.038	1		34 3 (1.35)	30.5 (1.20)		
	353,138		39.4 (1.55)	3 10 (-137)			
	361,738	39.4 (1.55)	40.6 (1.60)		31.8 (1.25)		
	371.638	40.6 (1.60)					
	373.838			35.6 (1.40) 36.8 (1.45)			
	380,938			36.8 (1.45)			
9 Hz	382.138				33.0 (1.30)		

Panel No. 2C			Primary Cra		Secondary Crack Lengths millimeters (inches)	
Cycle	No. of	I./H	R/H	1./H	R/H	
Rate	Cycles	Outside	Outside	Inside	Inside	
9 Hz	382,738	41.9 (1.65)				
9112	383,238	71.9 (1.0)	43.2 (1.70)			
T	301 338		+3.2 (1.10)		34.3 (1.35)	
	394,338 396,038 399,838	43.2 (1.70)	44.4 (1.75)		7.07 / 1.07/	
	399,838	13.2 (1.10)	1101 (201)	38.1 (1.50)		
	406,338		45.7 (1.80)			
	407.638	44.4 (1.75)	1201	39.4 (1.55)		
	409.638				35.6 (1.40)	
	409,638 416, <b>7</b> 38		47.0 (1.85)			
	421,298	45.7 (1.80)				
Y	424.318			40.6 (1.60)		
9 Hz	429,218		48.3 (1.90)	()		
5 Hz	429.799					
1					36.8 (1.45)	
1	431,018 434,318	47.0 (1.85)		41.9 (1.65)		
5 Hz	438,718		49.5 (1.95)			
9 Hz	1446,118			43.2 (1.70)	38.1 (1.50)	
-	450,918	48.3 (1.90)	50.8 (2.00)		•	
	462,318			44.4 (1.75)	39.4 (1.55)	
-		49.5 (1.95)	52.1 (2.05)			
-	475,418	50.8 (2.00)	53.3 (2.10)			
	478,518			45.7 (1.80)	40.6 (1.60)	
	482,918 483,818		54.6 (2.15)			
	1483,818	52.1 (2.05)	-	1 - (- 0-)		
	487,418		-	47.0 (1.85)		
-	487,618	-	ł		41.9 (1.65)	
	493,018	-	55.9 (2.20)			
-	496,918	53.3 (2.10)		10 - 12		
-	498,318			48.3 (1.90)	1	
	498,418		F7 0 (0 05)		43.2 (1.70)	
9 Hz	502,918		57.2 (2.25)	49.5 (1.95)		

Panel No. 20			Primary Cra	Secondary Crack Lengths millimeters (inches)			
Cycle	No. of	L/H	R/H	L/H	R/H		
Rate	Cycles	Outside	Outside	Inside	Inside		
9 .Hz	504,918				44.4 (1.75)		
A	512,318	54.6 (2.15)	58.4 (2.30)				
	513,018			50.8 (2.00)			
	523,118	55.9 (2.20)	59.7 (2.35)	52.1 (2.05)	45.7 (1.80)		
	529.618			53.3 (2.10)			
	530,218		61.0 (2.40)				
	534,018				47.0 (1.85)		
	536,018	57.2 (2.25)					
	538.718		62.2 (2.45)				
	545.018	58.4 (2.30)			TOTAL CO.		
	546.018			54.6 (2.15)			
	548,018				48.3 (1.90)		
	548.118		63.5 (2.50)				
	548,118 555,818				49.5 (1.95)		
	557,618	59.7 (2.35)	64.8 (2.55)				
	560,418			55.9 (2.20)			
	564,718				50.8 (2.00)		
	566,418		66.0 (2.60)				
	571,218	61.0 (2.40)					1//
	573,018			61.0 (2.40)			
	574.018				59.1 (2.35)		
	575,618		67.3 (2.65)			1	
	584,618		68.6 (2.70)	62.2 (2.45)	61.0 (2.40)		
	585.018	62.2 (2.45)					
	590.018				62.2 (2.45)		
	593,018		69.8 (2.75)				
Manager		63.5 (2.50)					
	594,618			63.5 (2.50)	63.5 (2.50)		
	599,218		71.1 (2.80)				
	600,118			64.8 (2.55)	64.8 (2.55)		
	604.818	64.8 (2.55)		( , , , , ,			
9 Hz	605,818				66.0 (2.60)		

Fanel No. 2C		Primary Crack Lengths millimeters (inches)				Secondary Crack Lengths millimeters (inches)		
Cycle Rate	No. of Cycles	L/H Cutside	R/H Outside	L/H Inside	R/H Inside	#1	#2	#3
			70 1 (0 95)					
9 Hz	610,818		72.4 (2.85)	66.0 (2.60)				
-	611,018	(( 0 (0 60)	72 7 (0 00)	00.0 (2.00)				
-	616,718	66.0 (2.60)	73.7 (2.90)		67.3 (2.65)			
-	617,818 622,442 622,942			67.3 (2.65)	68.6 (2.70)			
	1622,442		74.9 (2.95)	67.3 (2.65)	60.6 (2.70)			
-	1602 1110	67.3 (2.65)	14.9 (2.92)					
_	623,442	07.3 (2.02)				06 7 (1 05)	:	06 7 /2 05)
	624,442		-		69.8 (2.75)	26.7 (1.05)		26.7 (1.05)
	620,042		· · · · · · · · · · · · · · · · · · ·	68.6 (2.70)	09.0 (2.15)			
-	629,542			00.0 (5.10)		07 0 (2 20)		07 0 (3 30)
-	630,342 633,442 635,542		7( 0 /2 00)			27.9 (1.10)		27.9 (1.10
-	1633,442		76.2 (3.00)			29.2 (1.15)		29.2 (1.15)
	1635,542			69.8 (2.75)		29.2 (1.13)		- yor (101),
-	636,342	68.6 (2.70)		09.0 (2.1)				
	1030,242	00.0 (2.10)	<del> </del>		71.1 (2.80)			
_	639,142 644,042 649,142		PT 5 (2.05)		11-1-12-001			
-	640 142		77-5 (3.05)				5.1 ( .20)	
-	649,942			71.1 (2.80)				
-	650.442			1101 (200)		30.5 (1.20)		30.5 (1.20)
-	650.742	69.8 (2.75)				July (10CV)		- Man Lack
-	651,342	09.0 (2.1)	-				6.4 ( .25)	
-	653.042	<del> </del>	-				7.6 ( .30)	
-	1653.042		-		72 4 (2.85)		(-0 (-0 30)-	
_	654,142	-	-	72.4 (2.85)	12.4 (2.05)			
-	659,942		78.7 (3.10)	12.4 (2.05)				
	662.142	1	1001 (3010)			31.8 (1.25)	12 7 ( 50)	31.8 (1.25
	662.642	71.1 (2.80)	-			01011 (102)		30000
-	663,642	(100 (C.OO)	-				14.0 ( 55)	
-	665.542	-	-				14.0 ( 55)	
-	666.042	<del> </del>			73.7 (2.90)			
0 17-	668.842	-			13-1 (1-91)		16.5 ( .65)	

Panel	No. 2C		Secondar millim	ry Crack Length eters (inches)	ns		
Rate	No. of Cycles	#4					
9 Hz	610,818 611,018 616,718 617,818 622,442 622,442						
1	611,018						
1	616,718				-		
1	617,818						
	622,442						
	622,942						
	623,442						
	624.442						
	623,442 624,442 628,842						
	629,542 630,342 633,442 635,542 636,342						
	630.342						
	633,442						
	635.542						
	636,342						
	1638,242						
	639.142						
	649,142						
	649,142	5.1 ( .20)					
	649.942						
	649.942 650.442 650.742						
	650.742						
	1651.342	6.4 ( .25)					
	653.042	7.6 ( .30)					
	654.742						
	659,942						
	1660.042						
	1662.142	12.7 ( .50)					
	1662 6/12						-
	663.642	14.0 ( .55)					
	665,542	15.2 ( .60)					
Y	663,642 665,542 666,042 668,842						
9 Hz	668.842	16.5 ( .65)				1	

Fanel	No. 2C		Primary Cra millimeters						ry Crack Leng eters (inches	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	I,/} Insi	l lde	R/I		#1	#2	#3
9 Hz	670,042			73.7	(2.90)					
1	675,142				1				19.0 ( .75)	
	677,742								20.3 ( .80)	
1	679.342							33.0 (1.30)	2003 ( 00)	22 0 :/1 201
	680,442 681,442 682,142	72.4 (2.85)	80.0 (3.15)					33.0 (1.30)		33.0 (1.30)
	681 442	1507 (500)	10000	711:0	(2.95)					
	1682,142			14.9	(2092)				21.6 ( .85)	
	683,042					74 0	(2.95)		C1.0 ( .0)	
	684.842					1707	10-091		22.9 ( .90)	
	685.542	73.7 (2.90)							7 ( 90)	
	689,042								21 1 ( 05)	
	696.042								26.7 (1.05)	
	696,342			76.2	(3.00)					
	606 642	4				107		34.3 (1.35)		34.3 (1.35)
	696,642					76.2	(3.00)			
	701,042						ACK		27.9 (1.10)	
	703,642					STOPP	ED IN		27.9 (1.10)	
	705,542	74.9 (2.95)				BULB	RADIUS			
	708.642			BUL	В	76.2	(3.00)		30.5 (1.20)	
	711.942	76.2 (3.00)	81.3 (3.20)	RADI	THE REAL PROPERTY AND ADDRESS OF THE PERSON					
	714,342			RETA	ARTERIOR CONTRACTOR PROPERTY AND ADDRESS OF				31.8 (1.25)	
	777.942			GROW					33.0 (1.30)	
	719,142							35.6 (1.40)		35.6 (1.40)
	720.342	77.5 (3.05)								
	724.342		,						34.3 (1.35)	
	725,942							36.8 (1.45)		36.8 (1.45)
	729.142	78.7 (3.10)								
	731.342								35.6 (1.40)	
	734.842								36.8 (1.45)	
	739,242	80.0 (3.15)							(	
		(3,-,-)	82.6 (3.25)				Y			
9 Hz	739.542			76.2	(3.00)	76.2	(3.00)	38.1 (1.50)		38.1 (1.50)

Panel	No. 2C		Secondary Cre millimeters	ack Lengths (inches)		
	Cycles	#4				
9 Hz	670,042					
A	675,142	19.0 ( .75)				
	677 742	20.3 ( .80)				
	679,342 680,442 681,442 682,142					
	680.442					
	681,442					
	682,142	21.6 ( .85)				
	1683-042					
	684 842 685 542	22.9 ( .90)				
	685.542					
	689,042 696,042 696,342 696,642 697,742 701,042	24.1 ( .95)				-
	696,042	26.7 (1.05)				
	696,342					-
3	696,642					-
	697,742	·				-
	701.042	27.9 (1.10) 29.2 (1.15)			-	-
	703.642	29.2 (1.15)				-
	1705.542				-	
	708,642	30.5 (1.20)			-	<b></b>
	711,942					-
	714,342	31.8 (1.25)				<del> </del>
	717,942	33-0 (1-30)				
	719,142				<del></del>	<del> </del>
-	720,342					
	724,342	34.3 (1.35)				<del> </del>
-	725,942				-	-
	729,142				-	
-	731,342	35.6 (1.40) 36.8 (1.45)			+	<del>                                     </del>
-	734,842	36.8 (1.45)			+	<del>                                     </del>
-	739,242 739,442				+	1
9 Hz	739,442	<del> </del>				1

anel	No. 2C		Primary Crac			·		ry Crack Leng eters (inches	)
Cycle	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Insi		#1	#2	#3
9 Hz	741,642				76.2	(3.00)		38.1 (1.50)	
	746,042 748,442			80.0 (3.15)					
	748,442			81.3 (3.20)					
	749,542			82.6 (3.25)					
	750,142						39.4 (1.55)	39.4 (1.55)	39.4 (1.55)
	751,642			83.8 (3.30)					
	752.842	81.3 (3.20)							
	753.742			85.1 (3.35)					
	755.142			,		-	40.6 (1.60)		40.6 (1.60)
	760.042			86.4 (3.40)				1 - ( 1 - ( - )	
	760,042 760,742							40.6 (1.60)	
	1765.062					-		41.9 (1.65)	37.7
	766,542			87.6 (3.45)		-			
	770,042			88.9 (3.50)					
	770,742							43.2 (1.70)	
	772.742	82.6 (3.25)	83.8 (3.30)						
	776 242	83.8 (3.30)		90.2 (3.55)					
1	777,342	83.8 (3.30)					1 (1)		1- (-)
9 Hz	779,842				76.2	(3.00)	41.9 (1.65)		41.9 (1.65)
FINAI	CRACK								
LET	NOTHS	83.8 (3.30)	83.8 (3.30)	90.2 (3.55)	76.2	(3.00)	41.9 (1.65)	43.2 (1.70)	41.9 (1.70)
	RESI	DUAL STRENGTH	= 223,290 N	(50,200 lbs	.)				
		-	-		-				1

Panel 1	No. 2C		Second milli	ary Crack Length meters (inches)	ns		
Cycle Rate	No. of Cycles	# <del>1</del>					
9 Hz	741,642	38.1 (1.50)					
A	746,042						
	748,442						
	749,542						
	750,142	39.4 (1.55)					
	751.642						·
	752.842						
	753.742						
	755.142				-		
	760.042						
	760.742	40.6 (1.60)					
	765.062	41.9 (1.65)			-		
	766.542				-		
	770.042				-		
	770.742	43.2 (1.70)			-		
	772.742				-		
	776 242				-		
	777,342				-		
9 Hz	741,642 746,042 748,442 749,542 750,142 751,642 752,842 753,742 760,042 760,742 760,742 760,042 770,742 770,742 772,742 776,242 777,342 779,842				-		
	-						
FINAL	CRACK				1		
TET	GTH	43.2 (1.70)					
						· · · · · · · · · · · · · · · · · · ·	
							-
	-						

## PANEL #3C

MATERIAL: ALUMINUM-GRAPHITE

ADHESIVE: AF 126

ALUMINUM STRESS: 116 MN/m<sup>2</sup> (16.9 ksi)

MAXIMUM FATIGUE LOAD: 155,680N (35,000 lbf)

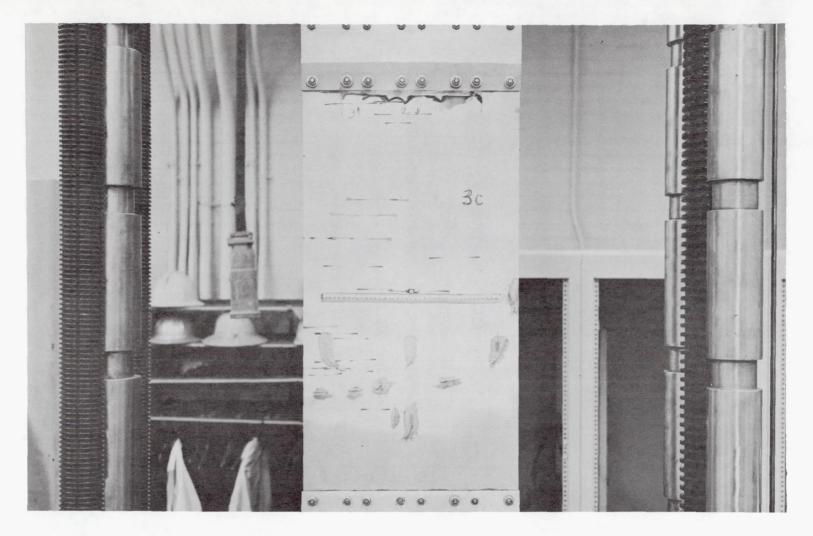


FIGURE F-9 OUTSIDE FACE SHEET AFTER CRACK GROWTH TEST - PANEL #3C



FIGURE F-10 INSIDE STIFFENER SHEET AFTER CRACK GROWTH TEST - PANEL #3C (LEFT OF CENTERLINE)



FIGURE F-11 INSIDE STIFFENER SHEET AFTER CRACK GROWTH TEST - PANEL #3C (RIGHT OF CENTERLINE)

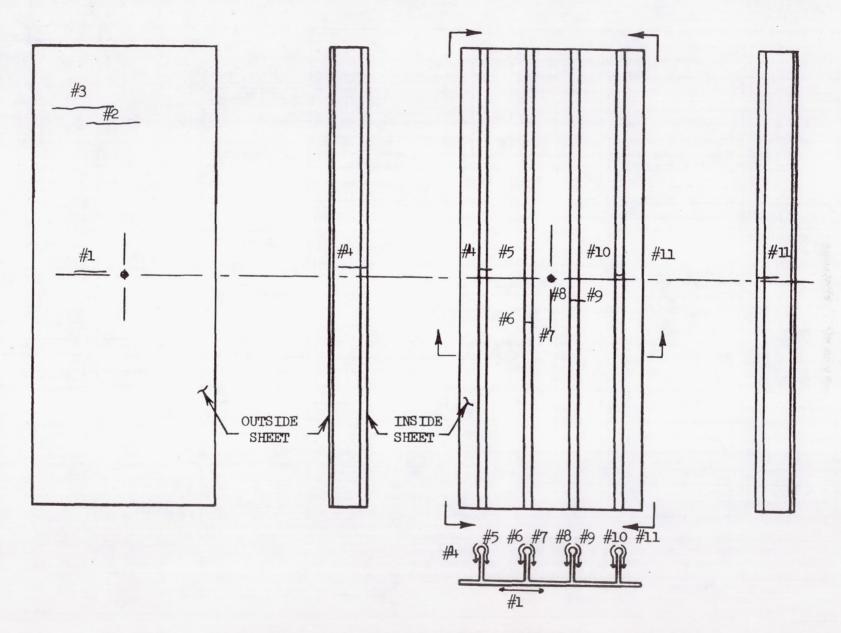


FIGURE F-12 SECONDARY CRACK LOCATIONS ALUMINUM-GRAPHITE PANEL #3C

Panel	No. 3C		Primary Cra			Secondary Crack Lengths millimeters (inches)
Cycle	No. of	L/H	R/H	I./H	R/H	
Rate	Cycles	Outside	Outside	Inside	Inside	
	26,456	6.4 ( .25)				
7 Hz	30,824	0.4 ( .2)	5.1 ( .20)	400		
-	30,024	7.6 ( .30)	201 (00)			
-	33,404	(.0 ( .30)		Initiation	3.8 ( .15)	
-	37,324			3.8 ( 15)	3.0 ( .15)	
	40.924	0 - 1 -5	( ) ( 05)	3.0 ( -12)		
-	41,024	8.9 ( .35)	6.4 ( .25)			
	45,824	-	7.6 ( .30)	F 1 / 001	6.4 ( .25)	
	46,224		0 0 / 05)	5.1 ( .20)	0.4 ( .62)	
-	49,224		8.9 ( .35)	( ), ( 05)	-	
	49,924	-		6.4 ( 25)		
	53,124	10.2 ( 40)	-		7.6 (.30)	
	53,624 56,724			7 ( / 20)	1.01.301	
	56,724		-	7.6 ( .30)	0.0 ( 05)	
	62,024				8.9 ( .35)	
	68,324	11.4 ( .45)	10.2 ( .40)	8.9 ( .35)		
	69,024		1		10.2 (.40)	
	79,424		11.4 ( .45)			
	79,724	12.7 ( .50)				
	80.524			10.2 ( .40)		
	90.174	14.0 ( .55)				
	90.224				11.4 ( .45)	
	90,424			11.4 (.45)		
	92.024				12.7 ( .50)	
	92,724		12.7 ( 50)			
	95.824			12.7 ( .50)		
	95,824	15.2 ( .60)				
	99,024				14.0 (.55)	
	103,324		14.0 ( .55)			
	104,024			14.0 ( .55)		
	107.424				15.2 ( .60)	
1	107.624	16.5 ( .65)				
7 Hz	107.924	1	15.2 ( .60)			

anel	No. 3C		Primary Cra			millim	ry Crack Lengths eters (inches)	
Cycle	No. of	L/H	R/H	L/H	R/H	#1		
Rate	Cycles	Outside	Outside	Inside	Inside			
7 Hz	113,824			15.2 ( .60)				
4	114,024				16.5 ( .65)			
	118,124	17.8 ( .70)						
	120,924		16.5 ( .65)					
	121,424			16.5 ( .65)				
	125,424				17.8 ( .70)			
	132.024	19.0 ( .75)	17.8 ( .70)		1			in the
	135.224			17.8 ( .70)				Same
	135.324				19.0 ( .75)			
	137,424	20.3 ( .80)			7			9
	139.324		19.0 ( .75)					
	140.324			19.0 ( 75)				
	144,324				20.3 ( 80)			3
	147,724	21.6 ( .85)						
	151,824		20.3 ( .80)					
	152,324			20.3 ( .80)				
	152,324 157,924				21.6 ( .85)			
	158.724	22.9 ( .90)						
	163,424		21.6 ( .85)					
	164.524			21.6 ( .85)	22.9 ( .90)			
	172,334			22.9 ( .90)				
	176.744	24.1 ( .95)						
	176,944		22.9 ( .90)					
	181,444		7-3		24.1 ( .95)			
	188,644			24.1 (.95)				
	193,644	25.4 (1.00)	24.1 ( 95)					
	194,044				25.4 (1.00)			
	206,193	26.7 (1.05)	25.4 (1.00)			8.9 ( .35)		
	209,293	27.9 (1.10)				, , , , ,		
	210,093		26.7 (1.05)					
	214,593	-		34.3 (1.35)	34.3 (1.35)			
7 Hz	1218,993	29.2 (1.15)						

Panel	No. 3C		Primary Crac				ry Crack Leng	)
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	I/H Inside	R/H Inside	#1	#5	#6
7 Hz	223,493		27.9 (1.10)					
1	223,993				1	19.0 ( .75)		
T	229,693					20.3 ( .80)		
	235,593		29.2 (1.15)					
-	241,193		->			21.6 ( .85)		
-	248,293		30.5 (1.20)					
	249,193		1000		4	THIS CRACK		
	254,993	52.1 (2.05)				JOINS L/H		
	256.493	7-0- (2.0)1			36.8 (1.45)	OUTSTDE PRI-		
	258 003		31.8 (1.25)	-		MARY CRACK		
-	265 003	53.3 (2.10)	31.0 (1.6.1)			21.6 ( .85)		
	267,193	7303 120201		THE RESIDENCE OF THE PARTY OF T		IS ADDED		
-	267,793	-				TO L/H		6.4 ( .25)
-	269.193	-				CRACK	10.2 ( .40)	
-	271.493	-						
-	272,193		17				12.7 ( .50)	
	072 703	-	33.0 (1.30)					
-	273,793	-	133.11 (1.31)	-				7.6 ( .30)
-	275,093		1					
	278.093							
$\vdash$	278.593	-						
-	278.993	54.6 (2.15)	1					
	270,993	100 (201)					14.0 ( .55)	
-	279,393 280,793	-	-					8.9 ( .35)
	282,393			The second secon			15.2 ( .60)	
-	282,893	<del>                                     </del>						
-	283.593		-					
77 77	283,993	1					the first transfer	10.2 ( 40)
7 Hz 6 Hz	285,393	55.9 (2.20)						
U ITZ	285.493	1107 (2020)						JAL I
1	286.093			-			16.5 ( .65)	
6 Hz	287,593			and the second s				17 4 ( 45)

Panel	No. 3C			Secondary millimet	Crack Lengers (inches	ths )	
Cycle Rate	No. of Cycles	. #7 Secondary	#8 Secondary	#9 Secondary			
7 Hz	223,493						
-	223,993						
	229,693						
	235,593					-	
	241,193						
	248,293						
	249,193						
	254,993						
	256,493						
	258,993						
	258,993 265,093						
	267,193		15.2 ( .60)				
	267,793						
	269,193	5.1 ( .20)	16.5 ( .65)	21.6 ( .85)			
	271,493	6.4 ( .25)	1	, , , , ,			
	272,193						
	273.793						
	273,793 273,993						
	275,093	8.9 ( .35)					
	278,093		17.8 ( .70)				
	278,593	10.2 (.40)					
	278,993			24.1 (.95)			
	279,393						
	279,393 280,793						
	282,393						
	1282.893	11.4 ( .45)					
Y	283,593		20.3 ( .80)				
7 Hz	283,993						
6 Hz	1285,393	I					
-	285.493			25.4 (1.00)			
*	286,093 287,593	I					
6 Hz	287,593						

Panel	No. 3C		Primary Crac	k Lengths (inches)		Secondar millim	ry Crack Leng	)
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#4	#5	#6
6 Hz	287,993							
	289.093		34.3 (1.35)					
	289,693							
	289,693 291,893							
	292,193							
	292,793						17.8 ( .70)	()
	293.593							12.7 (.50)
	204.193						19.0 ( .75)	
	294,193 295,593						19.0 ( .75)	
	297,193						00 0 / 001	71. 0 ( 55)
	298.793						20.3 ( .80)	14.0 ( .55)
	301.046	57.2 (2.25)						
	301,406							
	302,906					20.3 ( .80)		
	302,906 303,406		35.6 (1.40)					
	303,606							
	303,706						21.6 ( .85)	
	305,306				-		21.6 ( .85)	
	307,506						00 0 / 00)	-
	307.806						22.9 ( .90)	15.2 ( .60)
	307,806 308,306							15.2 ( .00)
	308,606						1 = 1 = ( 05)	-
	311.806						24.1 ( .95)	
	312,306							
	314,306							
	315.006	58.4 (2.30)	36.8 (1.45)		-			16.5 ( .65)
	315,506							10-2 (-02)
	315.806					21.6 ( .85)		
	315,806 315,906						1 1 / 1	-
	316,606						25.4 (1.00)	17.8 ( .70)
	318.006							17.8 ( .70)
6 Hz								

Panel	No. 3C			Secondary millimet	Crack Lengers (inches	gths s)	
Cycle Rate	No. of Cycles	#7	#8	#9			
6 Hz	287,993			26.7 (1.05)			
<b>A</b>	289,093						
	289,693		21.6 ( .85)				
	291,893			27.9 (1.10)			
	292,193	14.0 ( .55)					
	292.793						
	293.593						
	294,193 295,593		22.9 ( .90)				
	295,593						
	1297,193	15.2 ( .60)					
	298,793						
	301.046						
	301,406		24.1 (.95)				
	302,906 303,406						
	303,406						
	1303,606	16.5 ( .65)					
	303,706			29.2 (1.15)			
	305,306		-				
	307,506	17.8 ( .70)					
	1307,806						
	307,806 308,306						
	308,606			30.5 (1.20)			
	311.806						
	312,306	19.0 ( .75)					
	314,306		26.7 (1.05)				
	315,006						
	315,506						
	315,806						
	315,906			31.8 (1.25)			
	316,606				1411111		
1	318,006		-				
6 Hz	318,406		27.9 (1.10)				

Panel	No. 3C		Primary Cra	ck Lengths (inches)		Seconda:	ry Crack Leng	)
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	# <del>1</del> +	#5	#6
6 Hz	318,806		1 Table 18 1	35.6 (1.40)				
O IIZ	318,906			371				
1	320,706				39.4 (1.55)			
-	320,700							19.0 ( .75)
-	321,306 321,406 323,806					22.9 ( .90)		
-	1322 806	-						
-	1325 806	59.7 (2,35)	38.1 (1.50)				August Laboratoria	
-	327,006	1901 (60.17)	10.1				:	
-	328,406							
6 Hz	328,706			39.4 (1.55)				
5 Hz	328,906						27.9 (1.10)	
	329,106				40.6 (1.60)			
1	329,606					25.4 (1.00)		
-	333,106							
	222 706	61.0 (2.40)						001
	333.806	TOTAL CARL						20.3 ( .80)
	333,806 334,606	1						
	335,006			40.6 (1.60)				
	335,306							
	336,206		39.4 (1.55)					( ( 0-1
	336,806							21.6 ( .85)
	339.806				41.9 (1.65)			
	339,806 340,106					26.7 (1.05)		
	340,406							22.9 ( .90)
	340,406 340,806						29.2 (1.15)	
	343,106			41.9 (1.65)				
	344,406					-	30.5 (1.20)	-1 - (1
	345,406						·	24.1 ( .95)
	347,706					27.9 (1.10)		
	347.906	62.2 (2.45)	40.6 (1.60)				0 (2 05)	
V	348,606				-		31.8 (1.25)	
5 Hz	349,406							

Panel	No. 3C				Crack Lengers (inches			
Cycle Rate	No. of Cycles	#7	#8	#9				
6 Hz	318,806							
4	318,906	20.3 ( .80)						
	320.706							
	321,306 321,406 323,806							
	321,406							
	323,806		29.2 (1.15)					
	1325,806							
	327,006			33.0.(1.30)				
	328,406	22.9 ( .90)						
6 Hz	328,706							
5 Hz	328,906							
1	329,106							
	329,606	1						
	333,106		30.5 (1.20)					
	333,706							-
	333,806 334,606							
	334,606	-		34.3 (1.35)				3/
	335,006	<u> </u>						-
_	335,306	24.1 ( .95)						
	336,206		31.8 (1.25)					
	336,806					<u> </u>		
	339,806					-		
	340,106					-		-
	336,806 339,806 340,106 340,406 340,806	-				-		
	340,806	-		35.6 (1.40)				-
-	1343,106					-		
-	344,406							-
-	345,406	-				-		-
-	347,706					-	-	+
	347,906 348,606 349,406	-				-	-	+
5 Hz	1348,606	-		36.8 (1.45)				

Panel	No. 3C		Primary Cra millimeters	ck Lengths (inches)		Seconda:	ry Crack Leng	)
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#4	#5	#6
5 Hz	350,406							25.4 (1.00)
1	350,606		WARE THE RESERVE		43.2 (1.70)			
1	350,906							
	352,106							
	352,806							
	353,006			43.2 (1.70)				
	354.806							
	354,806 354,906					29.2 (1.15)		
	356,906							26.7 (1.05)
	358,706							
	361,306							
	363,406	63.5 (2.50)	41.9 (1.65)					
	364,306							
	366,206 366,606				11 (			
	366,606				44.4 (1.75)		<del></del>	
	367,006				OVERLAPS			
	368,006				#8 SECON-			
	368,906			44.4 (1.75)	DARY CRACK			
	369.406						33.0 (1.30)	
	370,706 371,306							
	371,306							07 0 /3 30
	371,606							27.9 (1.10)
	372,106					30.5 (1.20)	0 (2 05)	
	372,406						34.3 (1.35)	
	378,806			45.7 (1.80)				
	380,206	66.0 (2.60)	43.2 (1.70)				25 ( /2 ) 23	
	381,406			1 = (- 0-)			35.6 (1.40)	
	382,206			47.0 (1.85)				
	382,506 382,806			OVERLAPS		27 9 /7 05		
	382,806			#7 SECON-		31.8 (1.25)		00 0 /2 35
	383,006 383,206		44.4 (1.75)	DARY CRACK				29.2 (1.15)

Panel	No. 3C				Crack Lengths cers (inches)		
Cycle Rate	No. of Cycles	#7	#8	#9	#10		
5 Hz	350,406						
	350,606						
	350,906 352,106		33.0 (1.30)				
	352,106		331		10.2 ( .40)		
	352,806	26.7 (1.05)					
	353,006						
	353,006 354,806 354,906	27.9 (1.10)					
	354,906						(a)
	356,906						
	358,706			38.1 (1.50)			- 1
	361,306	29.2 (1.15)					
	363.406						
	364,306	30.5 (1.20)					
	366,206 366,606		34.3 (1.35)				
	366,606		OVERLAPS R/H				3
	367,006		INSIDE PRI-		11.4 (.45)		
	368,006		MARY CRACK	39.4 (1.55)			- Par
	368,906				-	 	
	369.406				-		
	370.706				14.0 ( .55)		
	370,706 371,306		36.8 (1.45)				
	371,606						
	372,106						
	372,406						
	378,806	31.8 (1.25)					
	380.206						
	381,406				-		
	382,206	OVERLAPS L/H					
	382,506	TNSTDE PRT-		40.6 (1.60)	-		
	382,806	MARY CRACK			-		
	383,006		20 2 /2 501	-			
5 Hz	383,206		38.1 (1.50)				

Panel	No. 3C		Primary Cra				ry Crack Leng eters (inches	)
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#4	#5	#6
5 Hz	383,606				47.0 (1.85)			
1112	384,006	67.3 (2.65)						
	384,806	01.3 (2.0)	44.4 (1.75)					
-	385,606		(=012)	48.3 (1.90)				
-	387,206							
-	389,206	67.3 (2.65)						
	393.806	01.07			48.3 (1.90)			
	303 006			49.5 (1.95)				
	393,906 394,706			1,000				
	394,906		45.7 (1.80)					
	395,406					33.0 (1.30)		
	400,806			of the second	49.5 (1.95)			
	401,206	68.6 (2.70)						
	401,306							
	402,306						36.8 (1.45)	
	403,306							30.5 (1.20)
	403,306					34.3 (1.35)		
	404,706							
	404,906			50.8 (2.00)				
	405.806							
	410,106							
	410,606	69.8 (2.75)	47.0 (1.85)					
	414.706			52.1 (2.05)	·			
	416,306							
	417,206				50.8 (2.00)			
	418,906	71.1 (2.80)						
	419,206		48.3 (1.90)					
	419,506						· .	
300	419,806			53.3 (2.10)				33.0 (1.30)
	421,206			FI. ( (0.35)				33.0 (1.30)
	424,806			54.6 (2.15)	FO 1 (0 OF)			<del> </del>
5 Hz	424,956				52.1 (2.05)			

Panel	No. 3C				crack Lengths ers (inches)	
Cycle Rate	No. of Cycles	#7	#8	#9	#10	
5 Hz	383,606 384,006 384,806					
	384,006				16.5 ( .65)	
	384,806					
	38 <b>5</b> ,606 38 <b>7</b> ,206	33.0 (1.30)				
	387,206				17.8 ( .70)	
	1389,206					
	393.806					
	393,906 394,706					
	394,706				19.0 ( .75)	
	394,906 395,406					
	395,406					
	400.806					
	401,206					
	401,306	34.3 (1.35)				
	402,306					
	403,306 403,506					
	403,506			41.9 (1.65)		
	404,706			41.9 (1.05)		
	404,906				21.6 ( .85)	•
	405,806	-			22.9 ( .90)	
	410,106	-			22.9 ( .90)	
	410,606	-				
	414,706 416,306	-			24.1 ( .95)	
	416,306	-			24.1 ( 97)	
	417,206 418,906	-			-	
	418,906	-				
-	419,206	125 6 (2 10)				
-	419,506	35.6 (1.40)			-	
	419,806	+				
	421,206 424,806	-			-	
5 Hz	424,000	+				

Panel	No. 3C		Primary Cra millimeters	ck Lengths (inches)		Seconda millim	ry Crack Leng eters (inches	ths )
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#4	#5	#6
5 Hz	425,606							
	428,006							
T	428,906	72.4 (2.85)						
_	431,606			55.9 (2.20)				Company of the Company
	432,206						38.1 (1.50)	
	432,206							
	433,106 433,606 434,206 434,806				53.3 (2.10)			
	433,606		49.5 (1.95)				all the same	
	434,206					35.6 (1.40)		
	434.806							
	435.706			57.2 (2.25)				
	439,606							
	439.706				54.6 (2.15)			
1	440,406 443,906							34.3 (1.35)
	443,906			58.4 (2.30)				
	1445,206	73.7 (2.90)						
	447,506							
	1447,506				55 2 72 22			
	1449,806				55.9 (2.20)			
	451,206							
	452,006						1 (2 55)	
	452,406						39.4 (1.55)	
	452,706		52.1 (2.05)					
	453,106	74.9 (2.95)						05 ( /2 10)
	453,906							35.6 (1.40)
	454,706			59.7 (2.35)				
	458.106							
	458,306						<del> </del>	
	462,606	76.2 (3.00)		61.0 (2.40)			/	
	463,606							
5 Hz					57.2 (2.25)			

Panel	No. 3C				y Crack Length ters (inches)	s	
Cycle Rate	No. of Cycles	#7	#8	#9	#10	#11	
5 Hz	425,606	,		43.2 (1.70)	25.4 (1.00)	at	
	428,006	36.8 (1.45)					
	428,906				26.7 (1.05)		
	431,606				1001		
	432,206						
	1432.806				27.9 (1.10)		
	433,106						
	433,606						
	434,206						
	433,106 433,606 434,206 434,806			44.4 (1.75)			
_	435.706						
	439,606		-	OVERLAPS #10	29.2 (1.15)		
	439,706			SECONDARY	OVERLAPS #9 SECONDARY		
	440,406			CRACK	SECONDARY		
	443,906				CRACK		1
	445,206						
	1445.606				30.5 (1.20)		
	447,506			45.7 (1.80)			
	449,806						
	451,206					14.0 ( .55)	
	452,006	38.1 (1.50)				•	
	452,406						
	452,706						
	453,106						
	453,906						
	454.706						
	458,106					15.2 (.60)	
	458,306			47.0 (1.85)			
	460.506						
	460,506 462,606						
	463,606				34.3 (1.35)		
5 Hz	464,106						

Panel 1	No. 3C		Primary Cra millimeters	ck Lengths (inches)			ry Crack Leng	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#4	#5	#6
5 Hz	464,306					36.8 (1.45)		
5 Hz	465,006							
5 Hz	465,006 465,606		53.3 (2.10)	-				
FINAL	CRACK							
	GTHS	76.2 (3.00)	53.3 (2.10)	61.0 (2.40)	57.2 (2.25)	36.8 (1.45)	39.4 (1.55)	35.6 (1.40)
	REST	DUAT. STRENGTH	= 364,736	W (82,000 lbs	)			

To. 3C	!		Secondary	crack Length ters (inches)	ıs		
No. of Cycles	#7	#8	#9	#10	#11		
464,306 465,006 465,606	39.4 (1.55)				17.8 ( .70)		
CRACK GTHS	39.4 (1.55)	38.1 (1.50)	47.0 (1.85)	34.3 (1.35)	17.8 ( .70)		
							2
			· ·				
	Cycles 464,306 465,006 465,606 CRACK	Cycles 464,306 465,006 465,606 39.4 (1.55)	Cycles 464,306 465,006 465,606 39.4 (1.55)	Cycles 464,306 465,006 465,606 39.4 (1.55)	Cycles 464,306 465,006 465,606 39.4 (1.55)	Cycles       464,306       17.8 ( .70)         465,006       17.8 ( .70)         465,606       39.4 (1.55)       38.1 (1.50)       47.0 (1.85)       34.3 (1.35)       17.8 ( .70)         CRACK       39.4 (1.55)       38.1 (1.50)       47.0 (1.85)       34.3 (1.35)       17.8 ( .70)	Cycles 464,306 17.8 (.70) 17.8 (.

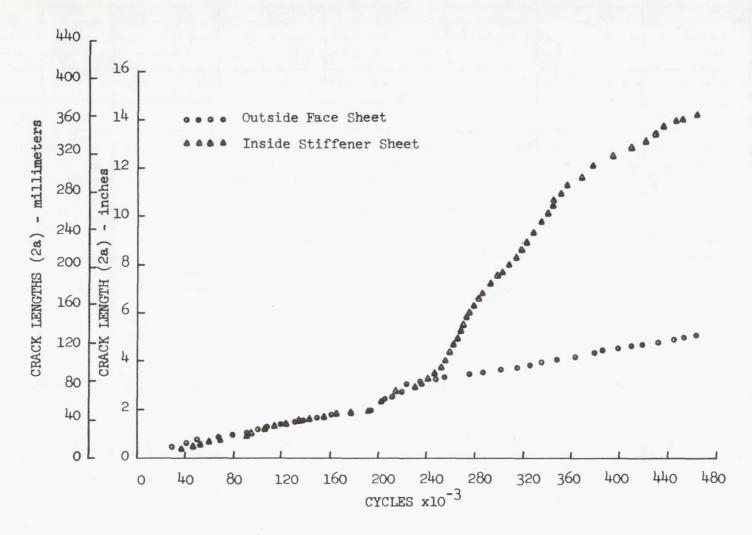


FIGURE F-13 CRACK GROWTH CURVE FOR ALUMINUM-GRAPHITE PANEL #3C (CRACK LENGTH INCLUDES SECONDARY CRACKS)

## PANEL #4C

MATERIAL: ALUMINUM-GRAPHITE

ADHESIVE: AF 126

ALUMINUM STRESS: 116 MN/m<sup>2</sup> (16.9 ksi)

MAXIMUM FATIGUE LOAD: 137,890N (31,000 lbf)

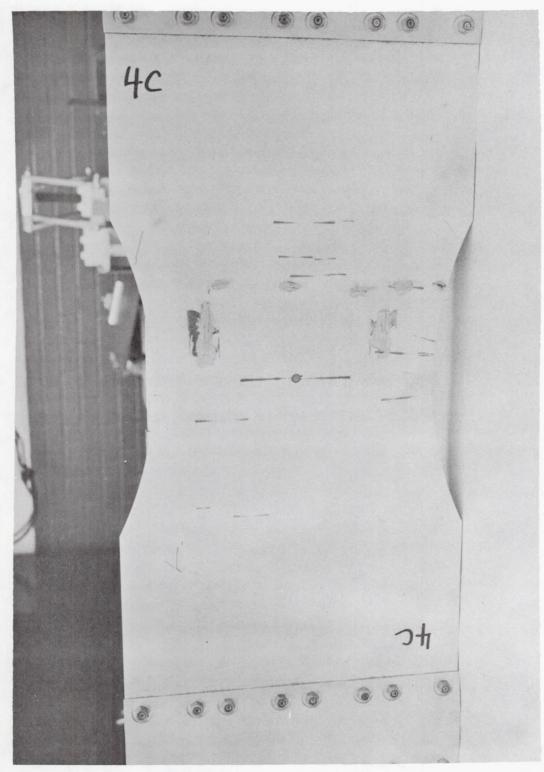


FIGURE F-14 OUTSIDE FACE SHEET AFTER CRACK GROWTH TEST - PANEL #+C

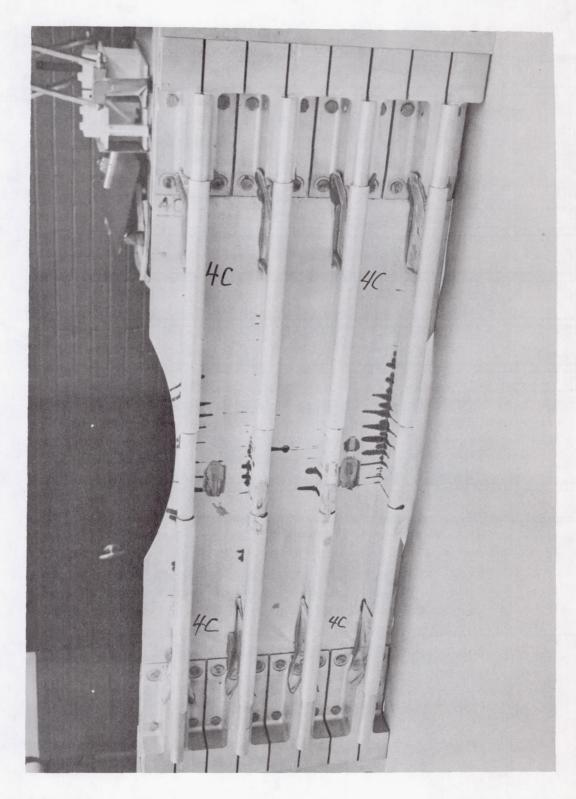


FIGURE F-15 INSIDE STIFFENER SHEET AFTER CRACK GROWTH TEST - PANEL #4C

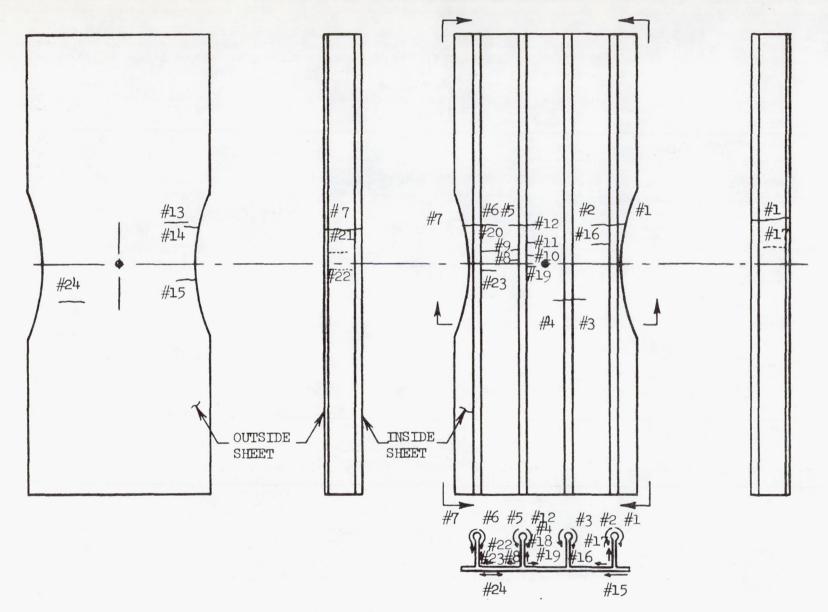


FIGURE F-16 SECONDARY CRACK LOCATIONS ALUMINUM-GRAPHITE PANEL #+C

Panel	No. 4C		Primary Crac millimeters			Secondar millime	ry Crack Leng	ths )
Cycle	No. of	L/H	R/H	I./H	R/H	#1	#2	#3
Rate	Cycles	Outside	Outside	Inside	Inside			
5 Hz	3,879				3.8 (.15)			
1	16,179				5.1 (.20)			
	10 070	3.8 (.15)						
	22,179 25,279 25,879 26,279 28,279 30,279 32,479	5.1 (.20)	3.8 (.15)		6.4 (.25)			• •
	25,279			3.8 (.15)				
	25,879		5.1 (.20)					
1	26,279				7.6 (.30)			
5 Hz	28,279	6.4 (.25)						
6 Hz	30,279			5.1 (.20)				
	32,479		6.4 (.25)					
	33,579				8.9 (.35)			
	41 979	7.6 (.30)						
	44.779	8.9 (.35)	7.6 (.30)					
	44,779 45,779 47,679 48,379			6.4 (.25)	10.2 (.40)			
Y	47,679		8.9 (.35)					7.6 (.30)
6 Hz	48,379						22.9 (.90)	
3 Hz	48.519						25.4 (1.00)	
	48,779							
	48,879	And the second			11.4 (.45)			
	49,079							10.2 (.40)
	49,279						27.9 (1.10)	
	49.929					36.8 (1.45)		
	50,079							11.4 (.45)
	50,279						20 5 /2 200	
	50,479					-0 - /0\	30.5 (1.20)	
	50,679					38.1 (1.50)		
	50,779	10.2 (.40)					02 0 (2 22)	
	51,379 51,679						31.8 (1.25)	20 5 / 50
	51,679							12.7 (.50)
	52,279		1000/10				21. 2 /2 25	14.0 (.55) 16.5 (.65)
1	54,179		10.2 (.40)			20 1 /2 55	34.3 (1.35)	10.5 (.65)
3 Hz	54,279					39.4 (1.55)		

	net	Crack Lengths ers (inches)		Secondary millimete	Cr	
-	R/H Outside	L/H Inside	R/H Inside	#1	#	#3
-		7.6 (.30)				
-						20 2 1 801
-						
1					25 6 11 LOT	
+			103 / 4 05			
+			1000) 107	10 6 (1 60)		
+				40.0 (1.00)		100 0 0 00
+					26 8 (1 1,E)	1
+					7	And the second s
-						1
						24.1.32
						1
1		1				(30 1) 27
		8.9 (35)				700.1
7				41.9 (L.65)		
7					102 1/1 00	- en - elles legale unitante de circumente preside principal de
7	-				- 1	
+	11.4 (.45)					
1						
T						
T				107 1) 9 61		
1	1				and the same of th	and the state of the state of the state of the state of
7	12.7 (.50)				The state of the s	(Ar r) 0 70
7				1		6104 LACAVI
7				44.4 (1.75)	20 1, (1 55)	
7		1	-		7	
		10.2 (.40)	15.2 (.60)			
1						
T		1311 / 1/ 1/ 1/ 1/ 1/ 1/ 1/ 1/ 1/ 1/ 1/ 1/				
102		772-1-1-1				
				And the first particular descriptions and the first particular descrip		

Panel	No. 4C			Secondar	ry Crack Leng eters (inches	ths )		
Cycle Rate	No. of Cycles	<del>//2</del> +	#5					
3 Hz	54,379							
	54,779	34.3 (1.35)						
	55,179		11.4 ( .45)					
	54,779 55,179 56,179 56,479 56,879							
	56,479		12.7 ( .50)					
	56,879							·
	57,079							
	57,079 57,379 57,479							
	57,479							-
	57,779							-
	57.879	35.6 (1.40)						
	58,179 59,079 59,579							
	59,079							-
Y	59,579		15.2 ( .60)				-	
3 Hz	59,679 63,203 63,703							-
4 Hz	63,203							
-	63,703							-
1	64,903				-		-	
4 Hz	65,103							-
5 Hz	65,803	(0 (- 1-)	-		+			-
A	.66,103	36.8 (1.45)	0 ()		-	-		-
	66,103 66,503 67,703		17.8 ( .70)		-	-		-
	67,703						-	-
	70,603	-	-		-	-		-
	73,303					-	-	-
	73,603	-			-		-	-
	73,703		-					
	73,903	100 7 (7 50)					1	-
	74,103	38.1 (1.50)	100 2 / 901		-			-
	74.903	-	20.3 ( .80)		-		-	-
	76,003		-					
5 Hz	76,103							

Panel No. 4C			Primary Crac		Secondary Crack Lengths millimeters (inches)			
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1	#2	#3
5 Hz	76,603							29.2 (1.15)
-	77;003					45.7 (1.80)		
-	79 103					1701 (2000)	40.6 (1.60)	
	78,103 80,003						40.0 (2.00)	
-	90,003							
	80,103		14.0 (.55)			-		
-	80,503		14.0 (.22)			-		
-	81,203				36 5 1 (5)	-	:	
-	82,603 82,703				16.5 (.65)		41.9 (1.65)	
-	82,703					-	41.9 (1.0)	30.5 (1.20)
	83,003					48.3 (1.90)		JUAN (IAEV
	83,403 85,103 87,203 87,703 88,203	31: 0 / 551				40-3 (1-90)		
-	05,103	14.0 (.55)			17.8 (.70)			
	87,203				11.0 (.10)	-		
-	87,703		35.0 ( (0)					
	88,203		15.2 (.60)			-	43.2 (1.70)	
	88,303					10 5 (1 05)	4302 (10/0)	
	88,703					49.5 (1.95)		31.8 (1.25
-	89,003 89,603					Edge of		2100 (100)
	89,603			20 5 / 50		Panel		
	89,703			12.7 (.50)		-		
	90,703							
	90,803							
	91,703							
	92,603	15.2 (.60)						- ( (- ) -
	94,403			14.0 (.55)				35.6 (1.40)
	97,703							
	97,703				19.0 (.75)			
	98,903						44.4 (1.75)	
	99,603							
	99,603 99,803		16.5 (.65)					
V	100,303					11.0 5 (2.05)		
5 Hz	101,303					49.5 (1.95)		38.1 (1.50)

Panel No. 4C		Secondary Crack Lengths millimeters (inches)								
Cycle Rate	No. of Cycles	#4	#5	#6	#7					
5 Hz	76,603					A				
*	77,003 78,103 80,003									
-	78,103									
	80,003			21.6 ( .85)						
	80,103				39.4 (1.55)					
	80,503									
	81,203		7 200 3	29.2 (1.15)						
	82,603									
6	82,703						100			
3	83,003									
	83,403									
	85,103									
	87,203									
7	87,703				41.9 (1.65)					
	88,203									
	88,303									
	88,703									
	89.003									
	89,603		1	36.8 (1.45)						
the Payment	89,703									
	90,703				43.2 (1.70)					
	90,803	40.6 (1.60)								
	81,203 82,603 82,703 83,403 85,103 87,203 87,703 88,203 88,303 88,703 89,603 89,603 90,703 90,803 91,703 92,603 94,403			38.1 (1.50)						
	192,603			1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1						
	94,403			39.4 (1.55)						
	97,703				44.4 (1.75)					
	98.603			2000						
	98,903									
	99,603	41.9 (1.65)								
	99,803			10000						
Y	98,903 99,603 99,803 100,303			40.6 (1.60)						
5 Hz	101,303									

Panel	No. 4C		Primary Crac millimeters			S	econda:	ry Crack Lengt eters (inches)	hs
Cycle	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1		#2	#3
5 Hz	103,003	16.5 (.65)				49.5	(1.95)		
6 Hz	104,903	10.) (.0)/							
O nz	105,703			15.2 (.60)	20.3 (.80)				
-	109,203		17.8 (.70)						
-	111,703		21.0 (010)					45.7 (1180)	
-	112 202	-							
-	112,303			16.5 (.65)					
-	113,403	17.8 (.70)							
	113,703	11.0 (.10)							
-	113,803				21.6 (.85)				
-	117.003		19.0 (.75)						
-	117.603								
-	118.003								
-	118,603								
	119,103							47.0 (1.85)	
_	123,603			17.8 (.70)	22.9 (.90)				
-	121,003	19.0 (.75)							
	124,003 124,703								39.4 (1.55
1	129,003								
	129,203		20.3 (.80)						1 1- 1-
	129,703								41.5 (1.65
	131,003							10 - ()	
	131,803							48.3 (1.90)	
	132,203								
	133,103	20.3 (.80)			01. 2 / 05				
	134,703 137,403			19.0 (.75)	24.1 (.95)				
	137,403				-		-		
	139,503		21.6 (.85)		1 (2 50)		-		
	142,003				25.4 (1.00)		-		
	142.203			00 0 ( 00)	-				
-	1/12,603			20.3 (.80)	-	ho s	(1105)	49.5 (1.95)	
6 Hz	143,303					44.7	11437		

Panel	No. 4C				Crack Lengths ers (inches)	
Cycle Rate	No. of Cycles	#4	#5	. #6	#7	
5 Hz	103 003					
6 Hz	104,903 105,703 109,203 111,703 112,303 112,603					
	105,703					
1	109,203					
-	111,703					
-	112,303			41.5 (1.65)		
-	112,603					
_	113,403			1		
	113,703				47.0 (1.85)	
	113,803					
	117.003					
	117,603			43.2 (1.70)		
	118.003				48.3 (1.90)	
	118,603 119,103 123,603	44.4 (1.75)				
	119,103					
	123,603					
	124,003 124,703					
	124,703					
	129,003 129,203			44.4 (1.75)		
	129,203					
	1.129.703				0 (2)	
	131,003				50.8 (2.00)	
-	131,803	1 0 1			EDGE OF	
-	132,203	45.7 (1.80)			PANEL	 
	133,103					
-	134,703 137,403			45.7 (1.80)		
-	137,403			42.7 (1.00)		
-	139,503			-		
-	142,003	1.7 0 (2 05)		-		
-	142,203	47.0 (1.85)				
6 Hz	142,603				50.8 (2.00)	

Panel	No. 4C		Primary Cramillimeters	ck Lengths (inches)			ry Crack Leng	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1	#2	#3
6 Hz	144,203					49.5 (1.95)		
O ILL	147,603	21.6 (.85)						
-	147,703	21.0 (.0))						44.4 (1.75)
-	148,803						50.8 (2.00)	
-	140,003						70.0 (2.00)	
_	149,703		22.9 (.90)					
-	149,722	00 0 / 00)	22.9 (.90)					
	152,222	22.9 (.90)						
-	154,722							
	157,522		01 7 ( 05)					
	157,682		24.1 (.95)	22.9 (.90)				
	158,722			22.9 (-90)				
	160,022							
	160,722							47.0 (1.85)
	160,822							TIOU
	161,222						52.1 (2.05)	
	161,722 161,822		-		07 0 /2 70)		JEOT (COU)	
	161.822		-		27.9 (1.10)		a a matter and a state of the s	
	165,122	24.1 (.95)						
	166,122 166,722		25.4 (1.00)					
	166,722							
	167,322							48.3 (1.90)
	167,722							40.3 (1.90)
	168,022							
	168,722							
	168,822 169,822	25.4 (1.00)						
	169,822				29.2 (1.15)			
	170.022			24.1 (.95)	OVERLAPS #4			
	170,422				SECONDARY		54.6 (2.15)	
	172.822		26.7 (1.05)		CRACK			
	173,522	26.7 (1.05)						
L.Y	175.622			25.4 (1.00)		1 - (		
6 Hz	176,122					49.5 (1.95)		

Panel	No. 4C		Secondary Crack Lengths millimeters (inches)											
Cycle Rate	No. of Cycles	# <del>1</del> +	#5	#6	#	7	<b>#1</b> 2	#13	#14					
6 Hz	144,203			47.0 (1.85)	50.8	(2.00)								
1	147,603 147,703 148,803 149,703													
-f	147,703													
	148,803			RED LESS										
	149,703			48.3 (1.90)										
_	149,722		41.9 (1.65)											
	152,222													
	154,722						43.2 (1.70)							
	157,522			50.8 (2.00)										
	157.682				-									
	157,682 158,722						11 1 ()							
	160,022 160,722 160,822 161,222						44.4 (1.75)	20.57 ( 50)	7.6 ( .30)					
	160,722		PARTY CONTRACTOR					12.7 ( .50)	1.0 ( .30)					
	160.822				-			-						
	161,222		43.2 (1.70)											
	161,722 161,822				-									
	161.822				-									
	165,122				-	· · ·								
	166,122 166,722				-			15.2 ( .60)	10.2 ( .40)					
	166,722			(5 -5)	-			15.2 ( .00)	10.2 ( .40					
	167,322			52.1 (2.05)	1									
	167,722				-									
	168,022	49.5 (1.95)												
	168,722		44.4 (1.75)		-									
	168.822				-			-						
	169,822	OVERLAPS R/H			-									
	170.022	INSIDE PRI-		-	-									
	170,422	MARY CRACK		53.3 (2.10)	-	-	-	-						
	172,822			-	-	-								
	173,522			-			-	-						
6 Hz	175,622			-	50 B	(2.00)	-	17.8 ( .70)	12.7 ( .50					

. .. .

Panel	No. 4C		Primary Cra				ry Crack Leng eters (inches	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	I/H Inside	R/H Inside	#1	#2	#3
6 Hz.	178,422					49.5 (1.95)		
	179,922		27.9 (1.10)					
1	180,222		-102 (-0-0)					
	180,322				30.5 (1.20)			
	182,022				2-07			
	182,422						55.9 (2.20)	50.8 (2.00)
	185,122			26.7 (1.05)				
	185,622							
	186 822	27.9 (1.10)						
	188 122	Clay (Ially)						
	188,122		29.2 (1.15)					
	188,822				31.8 (1.25)			
	190.022							
	193.722							
	103 822							52.1 (2.05)
	10/1 622	29.2 (1.15)						
	193,822 194,622 195,322						57.2 (2.25)	
	196,522							
	198,122							
	199-622				33.0 (1.30)			
	199,622			29.2 (1.15)				
	202,122	30.5 (1.20)	30.5 (1.20)					
	206,022							53.3 (2.10)
	207,322							
	207,922							
	208,222						64.8 (2.55)	
	209,222							
	210.522				34.3 (1.35)			
	210,922			30.5 (1.20)				
	212,122			OVERLAPS #12				
V	213,722	31.8 (1.25)	31.8 (1.25)			V		
6 Hz	215,322			CRACK		49.5 (1.95)	66.0 (2.60)	55.9 (2.20)

Pane <b>l</b>	No. 4C			Secondar millime	y Crack ters (i		s		
Cycle Rate	No. of Cycles	# <del>1</del> +	#5	#6	#	7	#12	#13	#14
6 Hz.	178,422			54.6 (2.15)	50.8	(2.00)			
	179,922								
1	180,222	50.8 (2.00)							
_	180,322								
-	182,022						45.7 (1.80)		
_	182,422		45.7 (1.80)	55.9 (2.20)					
	185,122			·					
	185,622						47.0 (1.85)		
	186.822								
	188,122 188,522		47.0 (1.85)						
	188,522								
	188,822								
	190,022							20.3 ( .80)	OVERLAPS
	193.722	52.1 (2.05)					48.3 (1.90)		#13
	193,822 194,622 195,322				-	-			SECONDARY
	194,622								CRACK
	195,322			The second second	-				(DATA
	1196,522		48.3 (1.90)		-				RECORDING
	198,122					-		22.9 ( .90)	DISCONTINUE
	199,622	1				-			1
	200,522					-			
	202,122								
	206,022								
	207,322			59.7 (2.35)		-			
	207,922	54.6 (2.15)							
	208,222								
	209,222		49.5 (1.95)			-			
	210.522				-				
	210,922			( ) ( ) ( ) ( ) ( )		-	OVERLAPS L/H		
	212,122			61.0 (2.40)	-		INSIDE PRI-		
	213,722					10.000	MARY CRACK		12.7 ( .50
6 Hz	215,322				150.8	(2.00)			12.7 ( .50

Panel	No. 4C		Primary Crac	ck Lengths (inches)			ry Crack Leng	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1	#2	#3
6 Hz	218,122					49.5 (1.95)		
The state of the s	218,191	33.0 (1.30)	33.0 (1.30)			1	67.3 (2.65)	
1	221,791	3310 (2007)	33 1					March Land
-	225,951	34.3 (1.35)	34.3 (1.35)					
-	227.691	JTOJ (LOJZ)	170) 150)//		35.6 (1.40)			
-	227,891							
-	228,091			33.0 (1.30)				
-	220,391	# # #						58.4 (2.30)
	229,391 230,891						69.8 (2.75)	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
_	231,591							
-	231.791							
	233,191							
	233,791		35.6 (1.40)	34.3 (1.35)				
	233,791 234,191							
	235,791				36.8 (1.45)		( 0 )	
	241,191				-		71.1 (2.80)	
	241,291							
1.	241.891							
	242,191	35.6 (1.40)						
	242,191 243,291							== = (0.05)
	243,791							59.7 (2.35)
	244,891							
	245,591	36.8 (1.45)	36.8 (1.45)					
	246,391							
	252,091	38.1 (1.50)	38.1 (1.50)					
	252,691							( (- ) -
	253,391				38.1 (1.50)		73.7 (2.90)	62.2 (2.45
1 1	258,191	39.4 (1.55)	39.4 (1.55)					
	263,191							63.5 (2.50
	263,191 263,991							
1	264.391	40.6 (1.60)				1 1		
6 Hz	264.491					49.5 (1.95)		

Cycle Rate 6 Hz	No. of Cycles	#4	11-		010 (11	nches)				
6 Hz	0		#5	. #6 	#	7	#12	#13	#1	L4
	218,122				50.8	(2.00)			12.7	(.50)
	218,191							25.4 (1.00)	4	
	221,791		50.8 (2.00)							
	225,951									
	227,691									
	225,951 227,691 227,891	55.9 (2.20)								
	228.091									
	229.391			•						
	229,391 230,891									
	231,591							27.9 (1.10)		
	231,791			64.8 (2.55)						
	233,191	57.2 (2.25)								
	233,791									
	234,191		52.1 (2.05)							
	235,791									
	1241.191									-
	241,291 241,891	58.4 (2.30)								
	241.891		53.3 (2.10)			-			-	
	242,191						(		-	
	242,191 243,291						52.1 (2.05)		-	
	243,791		-							
	244,891			66.0 (2.60)	·					
	245,591				-					
	246,391							30.5 (1.20)	-	
	252,091									
	252,691 253,391		54.6 (2.15)	(= - /2 /=)	· · ·					
	253,391		-	67.3 (2.65)	-	-				
	1 258,191							-		
	263,191 263,991				-	-				
	263,991	-			-			33.0 (1.30)		,
6 Hz	264,391 264,491	-	-		FO 0	(2.00)			12.7	( .50

Panel	No. 4c		Second milli	lary Crack Length imeters (inches)	ns		
Cycle Rate	No. of Cycles	#24					
6 Hz	218,122						
	218,191						
	221,791						
	225,951						
	227,691 227,891						
	227,891						
	228,091						
	229,391 230,891						
	230,891						
	231,591						· · · · · · · · · · · · · · · · · · ·
	231,791						
	233,191				-		
	233,791						
	234,191				<del></del>		
	235,791						
	241,191						-
	241,291 241,891				-	-	-
	241,891					-	
	242,191				-		-
	242,191 243,291 243,791 244,891					-	
	243,791				-		
	244,891			•		-	
	245,591° 246,391				-		-
	246.391						-
	252.091					-	-
	252,691				-	-	· · · · · · · · · · · · · · · · · · ·
	252,691 253,391				-	-	
	258,191	29.2 (1.15)			-	-	
	263,191 263,991				-	-	
1	263,991				-	-	
	264,391	31.8 (1.25)				-	
6 H	2 264.491	31.8 (1.25)				1	

Panel Panel	No. 4C		Primary Crac	k Lengths (inches)		Se	econdar millim	ry Crack Lengt eters (inches)	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1		#2	#3
6 Hz	264,591		40.6 (1.60)			49.5 (	(1.95)		
	266,191					4			
T	266,791								
_	267,591								
_	260,001								
	269,091 269,191				40.6 (1.60)				
	269,391								
	273.791			35.6 (1.40)					
	275,391	41.5 (1.65)	41.5 (1.65)						
	275,991								
	276,491				41.5 (1.65)				
	275,991 276,491 280,691								
	281,391		6						
	281,391 281,591				43.2 (1.70)				
	282,691	43.2 (1.70)	43.2 (1.70)						
	285.891				44.4 (1.75)				
	285,991 289,191								
	289,191				1- 0-1				
	290.091				45.7 (1.80)				
	290,091 291,791	44.4 (1.75)	44.4 (1.75)		1= (= 0=)				
	294,791				47.0 (1.85)	-	-		
	295,791								
	297,191					-			73.7 (2.90
	297,191 298,091		1			-			13.1 (2.90
	298,791	45.7 (1.80)	45.7 (1.80)			-			
	298,991 301,991				1.0 2 (2 00)		-		
	301,991				48.3 (1.90)		-	83.8 (3.30)	
	302,191					-	-	03.0 (3.30)	
	302,491	47.0 (1.85)	-						
	302,591 303,291		-			-			
6 Hz	303,291		-		-	40.5	(1.95)		74.9 (2.95)

P		el N	10. 4C			Secondary millimet	Crack ers (in	Length iches)	s			
	yc]	le	No. of Cycles	#4	#5	#6	#	7	#12	#13	#1	4
-	2	-	264,591				50.8	(2.00)			12.7 (	.50)
-	0	HZ	266,191						54.6 (2.15)			
-	-1	-	200,191			68.6 (2.70)						
-	-	-	266,791			69.8 (2.75)						
-	-	-	267,591		57.2 (2.25)	09.0 (2.1)						
-	-	-	269,091 269,191		21.6 (6.6)							
-	-		269,191	( ()		-						
-	_			63.5 (2.50)		1						
			273.791									
-			275,391						55.9 (2.20)		1	
L			275,991						77.9 (2.20)			
_			276,491		=0 1. (0.00)	-						
L			280,691	(1 0 1)	58.4 (2.30)		-				-	
		1	275,991 276,491 280,691 281,391 281,591	64.8 (2.55)			-				-	
			281,591		·							
			282,691								-	
			282,691 285,891									
1			285,991	66.0 (2.60)							-	
			285,991 289,191		59.7 (2.35)			· ·				
-			290,091									
			290,091 291,791 294,791									
F	_		294,791								-	
			295,791		61.0 (2.40)							
			207 101						59.7 (2.35)			
-	_	-	297,191 298,091									
-			298,791	1								
.  -		-	200, 191	1		76.2 (3.00)						
1	-	-	298,991 301,991			The state of the s			1			
-		-	302,191	-								
-		-	302,191	-								
1		-	302,491	68.6 (2.70)			1					
1		1	302.591	00.0 (2.10)		1	1	1				1
1	-	Hz	303,291	-		-	50.8	(2.00)			12.7	(.50)

. . .

Panel	No. 4C			Second mill:	lary Crack Leng imeters (inches	ths )		
Cycle Rate	No. of Cycles	#24						
6 Hz	264.591							
1	264,591 266,191							-
1	266,791							
	267,591						-	
	269.091						-	
	269,091 269,191						-	-
	269,391						-	-
	273,791							-
	275,391 275,991 276,491						-	-
	275,991						-	- in
	276.491					-		1
	280.691						-	
	281,391 281,591							
	281,591							
	282,691						-	1
	285.891						-	
	285,991 289,191							
	289,191							
	290,091 291,791							
	291,791							
	294,791						1	
	295,791						-	1
	297,191							
	298,091							1
	298,791							
	298,991 301,991							
	301,991		•					
	302,191							
	302,491							
	302,591 303,291	11 /5 55						
1	303,291	44.4 (1.75)						

Panel :	No. 4C		Primary Cra	ck Lengths (inches)	Secondary Crack Lengths millimeters (inches)			
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1	#2	#3
6 Hz	304,134		47.0 (1.85)			49.5 (1.95)		
FTNAL LEN	CRACK	47.0 (1.85)	47.0 (1.85)	35.6 (1.40)	48.3 (1.90)	49.5 (1.95)	83.8 (3.30)	
	RES	TDUAL STRENGI	H = 263,770	N (59,300 1)	)s.)			
				·				
	1							

Panel	No. 4C		Secondary Crack Lengths millimeters (inches)									
Cycle Rate	No. of Cycles	#4	#5	#6	#7	#12	#13	#14				
6 Hz	304,134		62.2 (2.45)	77.5 (3.05)	50.8 (2.00)		40.6 (1.60)	12.7 ( .50)				
FTNAL	CRACK	68,6 (2.70)	62.2 (2.45)	77.5 (3.05)	50.8 (2.00)	59.7 (2.35)	40.6 (1.60)	12.7 ( .50)				
					•							

Secondary Crack Lengths millimeters (inches) Panel No. 4C #24 Cycle No. of Cycles Rate 6 Hz 304,134 FINAL CRACK 44 4 (1.75)

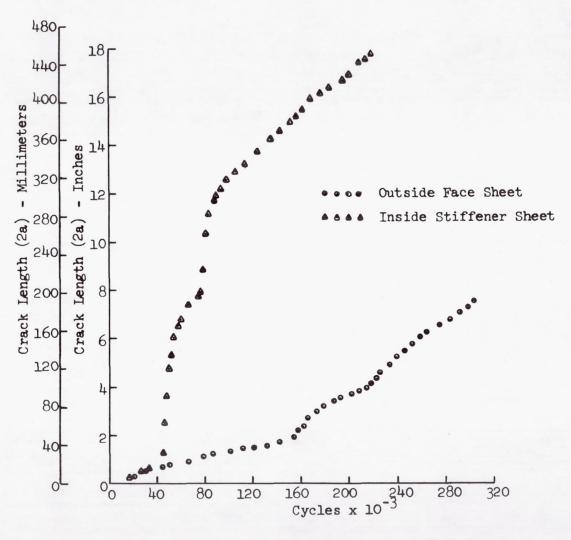


FIGURE F-17 CRACK GROWTH CURVE FOR ALUMINUM-GRAPHITE PANEL #4C (CRACK LENGTH INCLUDES SECONDARY CRACK)

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## PANEL #5C

MATERIAL: ALUMINUM-GRAPHITE

ADHESIVE: AF 126

ALUMINUM STRESS: 172 MN/m<sup>2</sup> (25 ksi)

MAXIMUM FATIGUE LOAD: 200,160N (45,000 lbf)

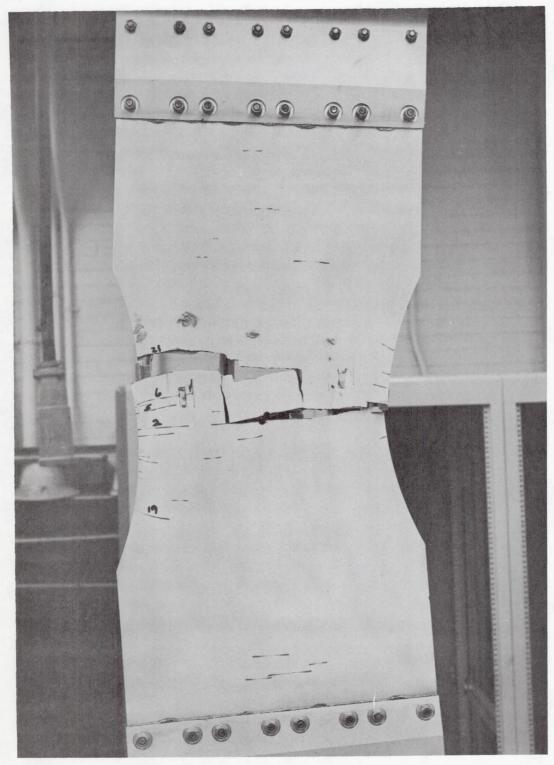


FIGURE F-18 OUTSIDE FACE SHEET AFTER RESIDUAL STRENGTH TEST - PANEL #5C

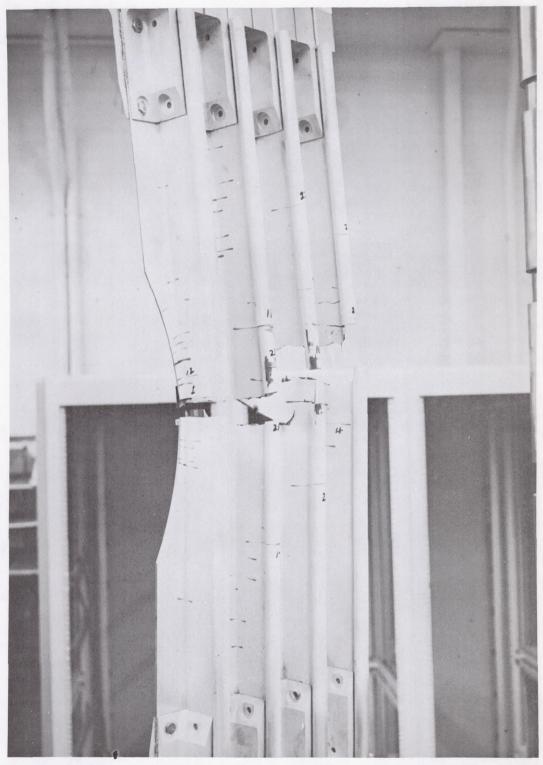


FIGURE F-19 INSIDE STIFFENER SHEET AFTER RESIDUAL STRENGTH TEST - PANEL #5C (LEFT OF CENTERLINE)

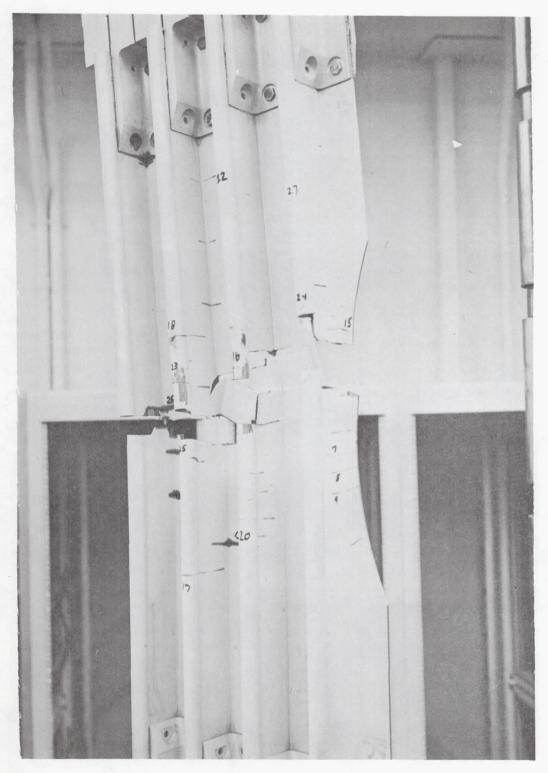


FIGURE F-20 INSIDE STIFFENER SHEET AFTER RESIDUAL STRENGTH TEST - PANEL #5C (RIGHT OF CENTERLINE)

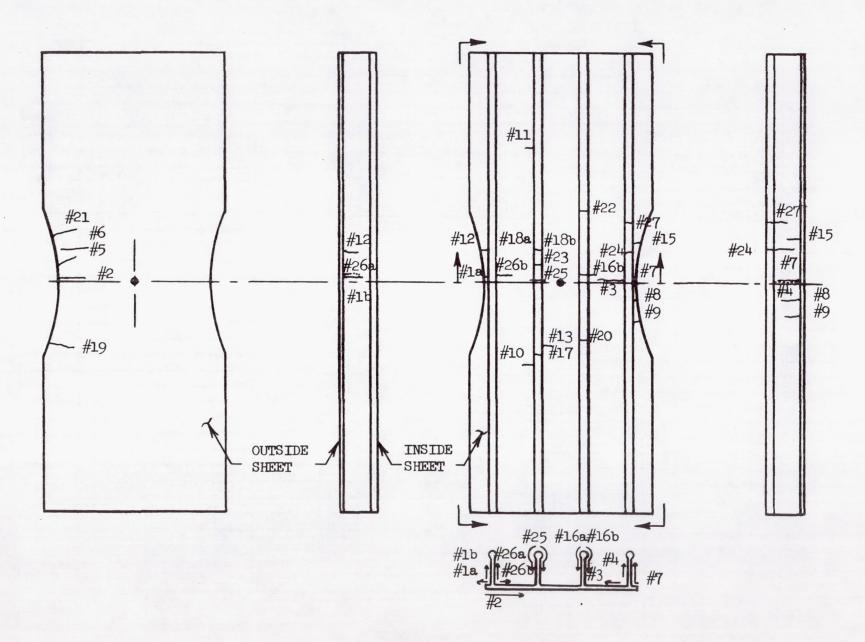


FIGURE F-21 SECONDARY CRACK LOCATIONS ALUMINUM-GRAPHITE PANEL #5C

Panel	No. 5C		Primary Cra	ck Lengths (inches)		Secondary Crack Lengths millimeters (inches)
Cycle Rate	No. of Cycles	L/H Outside	E/H Outside	L/H Inside	R/H Inside	
	-					
3 Hz	7,820	5.1 (.20)	5.1 ( .20)			
3 Hz	10,100		6.4 ( .25)		20/ 75)	
3 Hz	10,200				3.8 ( .15)	
4 Hz	11,600	6.4 ( .25)		- 0 ( -5)		
<b>A</b>	12,400			3.8 ( .15)		
	12,600		7.6 ( .30)		5 2 ( 00)	
	12,800				5.1 ( .20)	
	14,350	7.6 ( .30)			(1 ( 05)	
	16,500 17,300		0 0 / 251		6.4 ( .25)	
	17,300		8.9 ( .35)		7.6 ( .30)	
	19,300	0 - ( -5)			1.0 ( -30)	
	19,500	8.9 ( .35)		7 ( / 20)		
	21,200			7.6 ( .30)	0 0 / 25)	
	21,400				8.9 ( .35)	water and a second seco
	22,100	120 0 / 105	10.2 (.40)			
	22,300	10.2 ( .40)			10.2 (.40)	
	26,000			8.9 ( .35)	1005 10401	Applications in the contract of the contract o
-	26,200		1 1 ( 1.5)	0.9 ( .32)	-	The state of the s
	26,500	1 1 - 1	11.4 ( .45)		-	
	30,300	11.4 (.45)	-	10.2 ( .40)		
-	32,400			10.2 ( .40)	11.4 ( .45)	
	32,700	10 7 / 50	110 7 / 50		1104 (04)	
	37,500	12.7 ( .50)	12.7 ( .50)	11.4 ( .45)	12.7 ( .50)	
	39,500	1-1 - ( 55)	121.0 / 551	11.4 ( .47)	661	
	40,900	14.0 ( .55)	14.0 ( .55)			
	43,500	15.2 ( .60)	15.2 ( .60)	12.7 ( .50)		
	44,000	-		12.7 (_50)	14.0 ( .55)	
	44.800				14.0 ( .55)	
-	48,450		126 5 ( 65)		1)00	
	49,000	126 5 1 (5)	16.5 ( .65)			
	49,600	16.5 ( .65)	-	71.0 / 551		
4 Hz	50,200			14.0 (.55)		

Panel	No. 5C		Primary Cra millimeters	ck Lengths (inches)			ry Crack Leng eters (inches	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#la	#1b	#2
4 Hz	55,600	17.8 ( .70)	17.8 ( .70)					
1	55,600 56,800				16.5 ( .65)			
	59,300			15.2 ( .60)				
	63,000			16.5 ( .65)	17.8 ( .70)			·
	65,890	19.0 ( .75)	19.0 ( .75)					
	67,260	20.3 ( .80)	20.3 ( .80)					
	68,740			17.8 ( .70)	19.0 ( .75)			
	69,590					2.5 ( .10)	:	
	72,390 72,690				20.3 ( .80)	3.8 ( .15)	2.5 ( .10)	
	72,690	21.6 ( .85)	21.6 ( .85)					
	74.090			19.0 ( .75)				
	74,190 74,490					5.1 ( .20)	0 (>	
	74,490						3.8 ( .15)	
	75,890					6.4 ( .25)		
	77,290					OFF EDGE	5.1 ( .20)	
	78,090 78,890	22.9 ( .90)	22.9 ( .90)		-	OF PANEL		10.7./ 50
	78,890			1 000 0 / 000	07 ( / 05)			12.7 ( .50
	79,890	-	-	20.3 ( .80)	21.6 ( .85)			14.0 ( .55
	80,390	101 - ( 05)	-					14.0 ( .55
	80,590	24.1 ( .95)	101 2 ( 05)					-
	80,990		24.1 ( .95)				(1, ( 05)	
	81,790				20 2 / 22		6.4 ( .25)	-
	82,790				22.9 (.90)			75.01.60
-	83,890	1 \ />		-				15.2 ( .60
	84,590	25.4 (1.00)	105 1. (7. 00)					
	84,990		25.4 (1.00)	07 6 / 051			761 001	
-	85,190	-		21.6 ( .85)			7.6 ( .30)	16.5 ( .65
-	85,990	-		-	24.1 ( .95)		8.9 ( .35)	10.5 ( .05
	88,290				24.1 ( .92)		0.9 ( .35)	
-	89,090	-	-					17.8 ( .70
4 Hz	90,890		-			6.4 ( .25)		1.00

Fanel	No. 50			Secondary millimete	Crack Lengthers (inches)	ns		
Cycle Rate	No. of Cycles	#3	#24					
4 Hz	55,600 56,800							
A	56,800							
	59,300							
	63,000							
	65,890							
	67,260							-
	68,740							
	69,590							
	72.390							
	72,390					-		-
	74.090						-	-
	74,190 74,490						-	-
	74,490							-
	75,890			/			-	-
	77,290							
	78,090							-
	78,090 78,890							-
	79,890				•		-	
	80,390			· · ·		-		-
	80,590	1000				-	-	-
	80,990					-	-	-
	81,790						-	-
	82,790 83,890 ·						-	-
	83,890 .							-
	84,590 84,990						-	-
	84,990					-		-
	85,190					-		-
	85,990						-	-
	88,290					-		-
	89,090	2.5 ( .10)	2.5 ( .10)			-	-	-
L.V	90,890					-	-	-
4 Hz	91,390		3.8 ( .15)					

Panel	No. 5C		Primary Crack Lengths millimeters (inches)					ry Crack Leng eters (inches	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#3	la	#1b	.#2
3 Hz	91,490	26.7 (1.05)	26.7 (1.05)			6.4	(.25)		
	92,790			22.9 ( .90)					
	92,890				25.4 (1.00)				
	93,690								
	94,390								
	95,290								19.0 ( .75
	95,490	27.9 (1.10)	27.9 (1.10)						
	96,290								20.3 ( .80
	98,090								
	98,490							10.2 ( .40)	
	98,590			24.1 ( .95)	. 26.7 (1.05)				
	99,490								21.6 ( .85
	99,890	29.2 (1.15)	1						
	101.390		29.2 (1.15)		-				
	103,190								00 0 / 00
	104,090		-					12.7 ( .50)	22.9 ( .90
-	104,590				27.9 (1.10)			1601 101	
-	105,390			-	21.9.(1.10)				24.1 ( .95
	106,790 106,840	30.5 (1.20)	-	-	-				24.1 ( .9)
-	100,040	30.5 (1.20)	-						
-	107,290		<del> </del>	25.4 (1.00)					
-	108,390		30.5 (1.20)	22.4 (1.00)	-				
-	108,590		30.) (1.20)	<del> </del>				14.0 ( .55)	
-	109,290								
-	109,290		-		29.2 (1.15)				
-	110,890								
	110,990			1					
	111,680								25.4 (1.00
	112,590								
1									
3 Hz	112,790					6.4	( .25)	15.2 ( .60)	

Pane	el I	No. 5C			Secondary millimet	Crack Lengters (inches	ths )		
Cycl		No. of Cycles	#3	# <del>1</del> 1	#7				
3.1	Hz	91,490							
A		92,790							
		92,890							
		93,690		5.1 ( .20)					
		94,390	5.1 ( .20)						
		92,790 92,890 93,690 94,390 95,290 96,290							
		95,490					1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	11.	
		96,290			1				
		98,090		6.4 ( .25)					
		98,490							
		98,590							
		99,490							
		99,890							
		101.390							
		103,190		7.6 ( .30)					
		104,090							
		104,590							
		105,390 106,790 106,840	8.9 ( .35)	8.9 (.35)		·			
		106,790							
		106,840	-						
		107,290		10.2 ( .40)					
		107,490							
		108,390							
		108;590							
		109,290		11.4 ( .45)					
		109,760				·			
		110,890			5.1 ( .20)				-
		110,990	10.2 ( .40)						
		111,680					-		
		112.590			6.4 ( .25)				
I Y		112,790	-	12.7 ( .50)			-		
3 1	Hz_	1113,190					1		

.

Panel :	No. 5C		Primary Cra millimeters			Secondary Crack Lengths millimeters (inches)			
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#la	#1b	#2	
2 Hz	113,790	31.8 (1.25)	31.8 (1.25)			6.4 ( .25)			
	114,390	3200 (20-2)	3			1			
T	116,090			26.7 (1.05)					
-	116,290								
	116,290 116,790							26.7 (1.05	
_	117,390								
	117,390								
	117,790						·		
	117,990				30.5 (1.20)				
	118,090 119,290							07 0 /2 20	
	119,290							27.9 (1.10	
	119,790								
	120,090								
	120,290						-(-1		
	120,560	33.0 (1.30)	33.0 (1.30)				16.5 ( .65)		
	121,260		-	27.9 (1.10)					
	121,960 123,320							29.2 (1.15	
	123,320		-				<del> </del>	29.2 11.17	
	123,760						17.8 ( .70)		
	123,960 124,460			00 0 /2 75			11.0 ( .10]		
	124,460	1 (2 2 2 2 )	1 2 (2 25)	29.2 (1.15)	-				
	125,320	34.3 (1.35)	34.3 (1.35)		31.8 (1.25)				
	125,660				31.0 (1.2)				
	126,160	-						30.5 (1.20	
	126,160 126,880 126,960		-				-	30.7 (1.20	
	126,960		-		-				
-	1127.160		-	-	33.0 (1.30)				
	127,960 128,260 128,360			-	33.0 (1.30)		19.0 ( .75)		
	128,260	-	-				1300		
	128,360		105 ( /2 1:0)	-					
	128,460		35.6 (1.40)		-	6.4 ( .25)			

Pan	el 1	No. 5C			Secondary millimet	y Crack Lengtl ters (inches)	ns		
Cyc		No. of Cycles	#3	# <del>4</del> +	#7	#16a	#16b	#18a	#18b
2	Hz	113,790							
1		114,390				8.9 ( .35)	8.9 ( .35)		
		116,090	11.4 ( .45)						
-		116,290				10.2 ( .40)	10.2 ( .40)		
		116,790						10.2 ( .40)	7.6 ( .30)
		117,390						11.4 ( .45)	
		117,490							8.9 (.35)
		117,790				12.7 ( .50)	12.7 ( .50)		
-		117,990		15.2 ( .60)					
		118,090			7.6 ( .30)				
		119,290							
		119,790			8.9 ( .35)				
		120,090				14.0 ( .55)	14.0 ( .55)		
		120,290						12.7 ( .50)	10.2 ( .40)
		120,560	12.7 ( .50)			15.2 ( .60)	15.2 ( .60)		
		121,260							
		121.060		16.5 ( .65)					
		121,960							
		123,760			10.2 ( .40)	·			
		123,960							
		123,960		-					
		125,320					·	1	
	1	125,660		100					
		126,160				16.5 ( .65)	16.5 ( .65)	16.5 ( .65)	11.4 ( .45)
		126,880							
	1	126,960			11.4 ( .45)				
-		127,160				17.8 (70)			
-	1	127,960							
	1	128,260			12.7 ( .50)	19.0 ( .75)		17.8 ( .70)	
	1	128,360		17.8 ( .70)					
-	*	128,460							
2	Hz	129,260	-				17.8 ( .70)		

Panel	No. 5C		Primary Cra millimeters	ck Lengths (inches)		Seconda millin	ry Crack Leng eters (inches	ths )
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#la	#1b	#2
2 Hz	129,460					6.4 ( .25)		
	130,310							
T	130,360	35.6 (1.40)					·	31.8 (1.25)
	130,860	3,						
	130,860							
	131,960			·	34.3 (1.35)			
	132,060							
	132,560							33.0 (1.30)
	134.360		36.8 (1.45)					
	134,460 134,560 134,660			30.5 (1.20)				
	134.560							
	134,660						20.3 ( .80)	
	134,760							
	134.860							
	134.960							
	135,160 135,260							
	135,260							
	135,560							
	136,660							
	137,660	36.8 (1.45)					-	
	137,760			-	( /- ) ->			34.3 (1.35
	137,860			0 (	35.6 (1.40)		-	-
	138,060			31.8 (1.25)				
	138,260		-				21.6 ( .85)	
	138,460		-					
	138,760	-	-					-
	138,860	-	-				1.	
	139,760	1-0 - ()	1-0 - ()					
	139,860	38.1 (1.50)	38.1 (1.50)				-	- ( )
	139,960 140,560						-	35.6 (1.40
	140,560			-	-	6.4 ( .25)	1	

Panel	No. 5C				Crack Length ers (inches)	ns		
Cycle Rate	No. of Cycles	#3	#4	#7	#16a	#16b	#18a	#18b
2 Hz	129,460							12.7 ( .50
1	130,310							
T	130,360							
-	130,860	14.0 ( .55)						
	131,860	16.5 ( .65)						
	131,960							
	132,060		20.3 ( .80)					
	132,560							
	134.360							
	134.460		21.6 ( .85)	7.				
	134,560	19.0 ( .75)	12968					
	134,660							
	134,760			17.8 ( .70)	07 ( / 05)	00 0 / 00)		
	134,860				21.6 ( .85)	20.3 ( .80)	20.3 ( .80)	
	134,960						20.3 ( .80)	16.5 ( .6
	135,160						21.6 ( .85)	10.5 ( .0
	135,260	( 0-)	-				21.0 ( .0)	
	135,560	20.3 ( .80)			-			
-	136,660	21.6 ( .85)	-	<u> </u>				
-	137,660	1 1 1				-		
-	137,760	22.9 ( .90)	-					
-	137,860		1					
	138,260	-		19.0 ( .75)				
-	138,460	-	22.9 ( .90)	19.0 (-(3)				
-	138,760	-	26.9 ( .90)		22.9 ( .90)			
	138 860	-				21.6 ( .85)		
_	138,860	1	1				22.9 ( .90)	
	139,860							
	139,960			-14-14-14				
V	139,960	24.1 ( .95)		20.3 ( .80)				
2 Hz			24.1 ( .95)					

Panel	No. 5C		Primary Cra millimeters	ck Lengths (inches)	•	Secondary Crack Lengths millimeters (inches)			
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#la	#1b	#2	
2 Hz	141,660					6.4 ( .25)			
	141,760					1			
-	141,860		39.4 (1.55)						
	141,960		370 (2007)				22.9 ( .90)		
	142,260								
-	142,760								
	142,960								
	143,260				36.8 (1.45)				
	143,260 144,160			33.0 (1.30)				4	
	145,660	39.4 (1.55)						(A)	
	146.060	1						36.8 (1.45)	
	146,260								
	148,660	40.6 (1.60)	40.6 (1.60)						
	149,060						24.1 ( .95)	-	
	149,160								
	1149.260								
	150,560				38.1 (1.50)				
	151,160							-	
	152,760			34.3 (1.35)	39.4 (1.55)				
	153,060	41.5 (1.65)	41.5 (1.65)						
	153.342								
	154,972							38.1 (1.50)	
	156,342							-	
	154,972 156,342 156,442				40.6 (1.60)				
	156,542			35.6 (1.40)					
	157.142								
	157.342			-			27.9 (1.10)	-	
	157,742 158,242						-		
	158,242	43.2 (1.70)					-	-	
	158,442 161,042		43.2 (1.70)	10000	11.5 5 /5 /5				
	161,042	1 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3 3		36.8 (1.45)	41.5 (1.65)		100 0 (0 00)		
2 Hz	161,142					6.4 ( .25	29.2 (1.15)		

Panel No. 5C		Secondary Crack Lengths  millimeters (inches)						
Cycle Rate	No. of Cycles	#3	#1	#7	#16a	#16b	#18a	#18b
2 Hz	141,660	25.4 (1.00)						
	141,760				24.1 (.95)	22.9 (.90)		
1	141,860							
	141,960							
	142,260						24.1 ( .95)	17.8 ( .70
-	142,760	26.7 (1.05)	25.4 (1.00)					
-	142,960	20.1 (2.0)		21.6 ( .85)				
_	143,26Q			1			the state of the s	
-	144,160	27.9 (1.10)			25.4 (1.00)	24.1 ( .95)		
-	145,660	21.9 (1.40)						
-	146.060							
_	146,260	29.2 (1.15)						
-	148,660	29.6 (1.1)						
-	149,060			22.9 ( .90)				
-	149,000						25.4 (1.00)	19.0 ( .75
-	149,160.		-		26.7 (1.05)	26.7 (1.05)	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	The state of the s
-	149,260 150,560		-		2001 (3007)			
-	151,160		27.9 (1.10)					Same the state of the same of
-	152,760		16109					
		1						
	153,060	30.5 (1.20)			27.9 (1.10)	27.9 (1.10)	27.9 (1.10)	20.3 ( .80
-	153,342	30.2 (1.20)						
-	154,9/2	34.3 (1.35)	29.2 (1.15)					
-	1256 142	34.3 (1.37)	29.2 (1.1)					
-	156,442	-	-					
	156,542	-	-		29.2 (1.15)	29.2 (1.15)		
-	157,142		-	27.9 (1.10)	29.2 (1.1)	27.02 (202)	29.2 (1.15)	21.6 ( .85
	157,342			C109 (1010)	-		-/0	
	157,742 158,242	-	-	-				
	158,242				-			
	158,442	-	-	-				
-	161,042	1-( 0 (2 1-1)		00 0 (3 35)	20 5 (7 00)	30.5 (1.20)		22.9 ( .90
2 Hz	1161,142	36.8 (1.45)		29.2 (1.15)	130-5 (1-20)	2002 (100)		

Panel No. 5C		Secondary Crack Lengths millimeters (inches)							
Cycle	No. of Cycles	#26a	#26b					,	
2 Hz	141,660								
	141,760								
T	141.860								
-	141,960 142,260								
_	142,260								
	142,760								
	142,960								
	142,760 142,960 143,260 144,160			1					
	144,160								
	1145,660	1					-	·	
	146,060 146,260								
	146,260								
	148,660								
	149,060								
	149,160						-		
	149.260								
	150,560	11.4 ( .45)	10.2 (.40)				-		
	151,160				-		-		
	152,760								
	153,060	12.7 ( .50)						-	
	153,342	12.7 (.50)	11.4 ( .45)				-		
	154,972 156,342 156,442				· · · · · · · · · · · · · · · · · · ·				
	156,342								
	156,442						-		
	1156,542								
	157,142	The second second						-	
	157,342	-							
	157,742 158,242	15.2 (.60)	12.7 (.50)				-	-	
	158,242						-	-	
	158,442 161,042								
Y	161,042	1					-	-	
2 Hz	161,142								

Panel No. 5C		Primary Crack Lengths millimeters (inches)				Secondary Crack Lengths millimeters (inches)		
Cycle	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#la	#1b	#2
2 Hz	161,192					6.4 ( .25)		
1	161,242	-						
-	161,842	-						
-	162 2/12	44.4 (1.75)	44.4 (1.75)					
-	162,242 162,442 163,642 164,342	44.4 (1.1)	77.7 (101)					43.2 (1.70
-	162 642	45.7 (1.80)	-					
-	16/1 2/12	17.1 (1.00)						
-	1364 642	-			43.2 (1.70)		:	
	164,042	-		38.1 (1.50)			30.5 (1.20)	
-	164,642 164,742 164,842							
-	164.942							
-	165 Oli2							
_	165,042 165,742		45.7 (1.80)					
_	167 202		1701 12001					45.7 (1.80
_	167,342 167,842	47.0 (1.85)						
_	167 042	11.0 (1.02)		39.4 (1.55)				
-	168 3/12		1				31.8 (1.25)	-
_	167,942 168,342 168,642 168,742 169,142	1						
-	168 7/12							
_	160 142							
_	169.342		47.0 (1.85)					
_	177 542							47.0 (1.85
	171,542 171,742			40.6 (1.60)				
_	172,292							
	172,342							
_	172,542	1						
	172 7/12	1			OVERLAPS			-
	172,742 173,242				OVERLAPS			
	173.342				#16a			-
	173 142				SECONDARY			
-	173,342 173,442 173,842			41.5 (1.65)	CRACK			10 - 1
2 Hz	174.242					6.4 ( .25		148.3 (1.90

Panel No. 5C		Secondary Crack Lengths millimeters (inches)							
Cycle	No. of Cycles	#3	#1	#7	#16 <b>a</b>	#16b	#18a	#18b	
2 Hz	161,192		31.8 (1.25)				30.5 (1.20)		
	161.842								
	162,242 162,442 163,642								
	163,642	38.1 (1.50)	33.0 (1.30)						
	164,342 164,642 164,742 164,842								
	164,842			30.5 (1.20)	31.8 (1.25)	31.8 (1.25)	0 ()		
-	1164.942						31.8 (1.25)	24.1 ( .95)	
	165,042 165,742								
	167,342 167,842	(2.55)	al. a. (7, 25)						
	167,942 168,342	39.4 (1.55)	34.3 (1.35)						
-	1168.642		-	31.8 (1.25)	33.0 (1.30)	33.0 (1.30)	33.0 (1.30)	25.4 (1.00)	
	168,742 169,142								
	169,3 <sup>1</sup> 42 171,5 <sup>1</sup> 42 171,7 <sup>1</sup> 42 172,292								
	172,292		35.6 (1.40)						
	172,342	41.5 (1.65)		33.0 (1.30)					
	172,742				34.3 (1.35) OVERLAPS R/H	34.3 (1.35)	35.6 (1.40)	27.9 (1.10)	
	172,742 173,242 173,342 173,442 173,842	43.2 (1.70)			INSIDE PRI- MARY CRACK				
	173,442	45.2 (1.70)			I'M CINCK				
2 H	174,242			L			1		

Panel	No. 5C		•	Seconda millim	ry Crack Leng eters (inches	gths 3)	
Cycle Rate	No. of Cycles	#26a	#26b				
2 Hz	161,192						
F	161,242						
	161.842	16.5 ( .65)	14.0 ( .55)				
	161,842	10.71					
	162,442 163,642 164,342 164,642 164,742						
_	163.642						
	164.342		- Parking the second				
	164.642						
	164,742						
	17611 2010						-
	164,042 165,042 165,742 167,342 167,842 167,942 168,342						
	165.042	17.8 ( .70)	15.2 ( .60)				
	165,742						
	167,342						
	167,842						
	167,942	1970					
	168,342						
	168,642				·		
	168.742						
	168,642 168,742 169,142	20.3 ( .80)	16.5 ( .65)				
	169.342				-		
	171,542						
	169,342 171,542 171,742 172,292						
	172,292						
	172,342						
	172.542						
	172,742						 
4 6	172,742 173,242			· · · · · · · · · · · · · · · · · · ·			
1	1773.342	22.9 ( .90)	19.0 ( .75)				
	173.442						
V	173,442 173,842						
2 Hz	174,242						

Panel 1	No. 5C		Primary Cra	ck Lengths (inches)		Seconda millim	ry Crack Leng	ths )
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#la	#1b	#2
2 Hz 2 Hz	174,342 176,452	49.5 (1.95)	48.3 (1.90) 49.5 (1.95)		·	6.4 ( .25) 6.4 ( .25)	34.3 (1.35)	49.5 (1.95)
FINAL	CRACK	49.5 (1.95)	49.5 (1.95)	41.5 (1.65)	44.4 (1.75)	6.4 ( .25)	34.3 (1.35)	49.5 (1.95)
	RES	IDUAL STRENGT	H = 302,460	N (68,000 1)	os.)			

Secondary Crack Lengths Panel No. 5C millimeters (inches) #18a #18b #7 #16a #16b #4 #3 Cycle No. of Cycles Rate 2 Hz 174,342 2 Hz 176,452 34.3 (1.35) FINAL CRACK 35.6 (1.40) 34.3 (1.35) 34.3 (1.35) 34.3 (1.35) 35.6 (1.40) 27.9 (1.10) LENGTHS 43.2 (1.70)

Panel :	No. 50			Secondary millimet	Crack Length ers (inches)	S	
Cycle Rate	No. of Cycles	#26a	#26b				
2 Hz	174,342 176,452	24.1 ( .95)	21.6 ( .85)				
FINAL	CRACK	24.1 ( .95)	21.6 ( .85)				
					•		
				·			

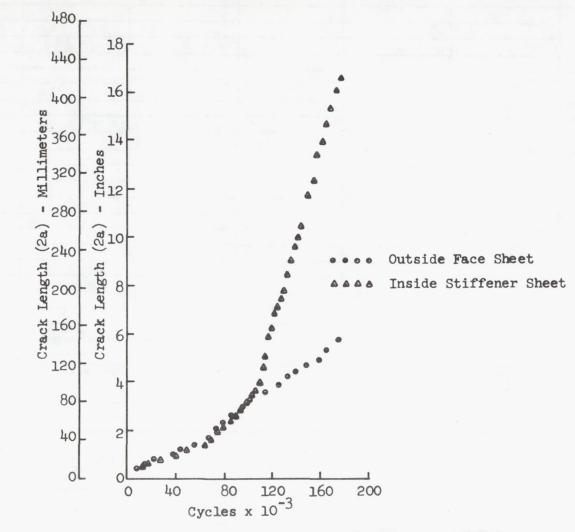


FIGURE F-22 CRACK GROWTH CURVE FOR ALUMINUM-GRAPHITE PANEL #5C (CRACK LENGTH INCLUDES SECONDARY CRACKS)

## PANEL #6C

MATERIALS: ALUMINUM-GRAPHITE

ADHESIVE: AF 126

ALUMINUM STRESS: 138 MN/m<sup>2</sup> (20 ksi)

MAXIMUM FATIGUE LOAD: 164,580N (37,000 lbf)

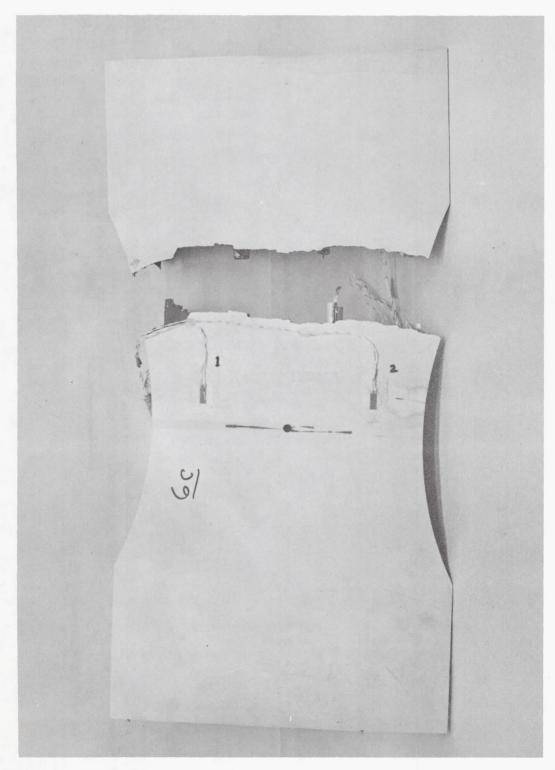


FIGURE F-23 OUTSIDE FACE SHEET AFTER RESIDUAL STRENGTH TEST - PANEL #6C

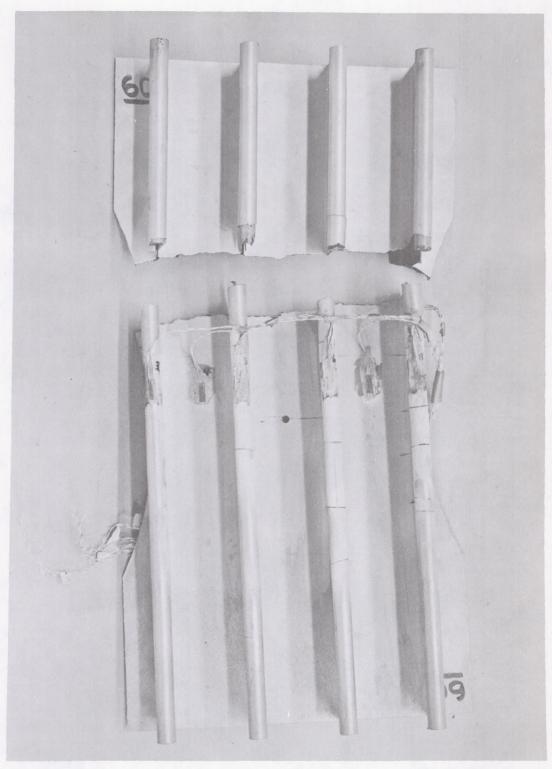


FIGURE F-24 INSIDE STIFFENER SHEET AFTER RESIDUAL STRENGTH TEST - PANEL #6C

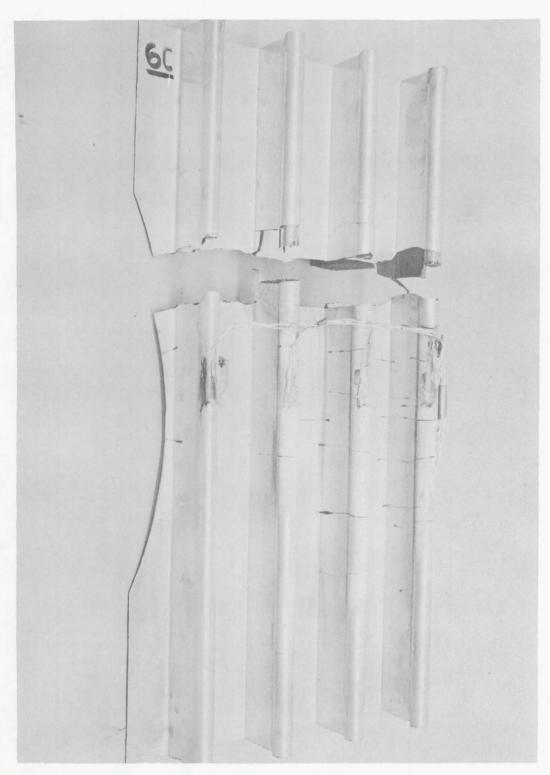


FIGURE F-25 INSIDE STIFFENER SHEET AFTER RESIDUAL STRENGTH TEST - PANEL #6C (LEFT OF CENTERLINE)

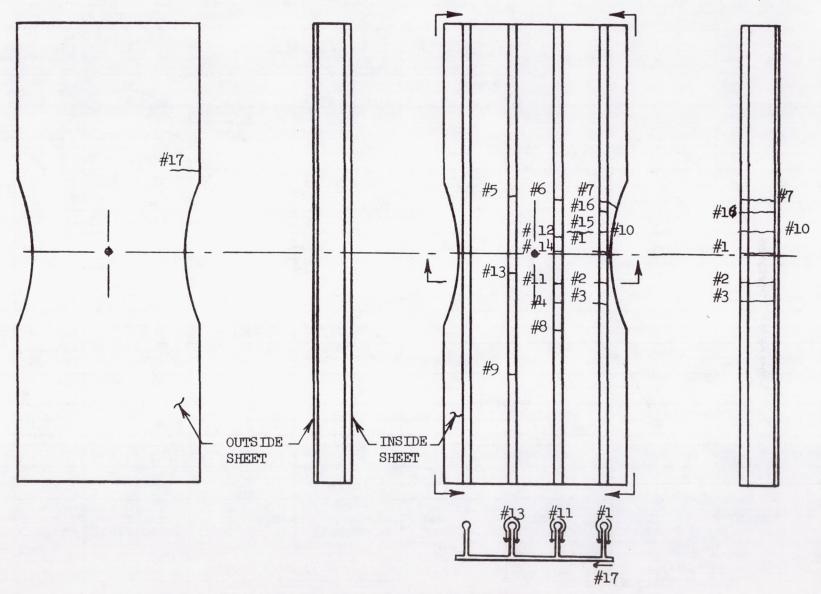


FIGURE F-26 SECONDARY CRACK LOCATIONS ALUMINUM-GRAPHITE PANEL #6C

Panel	No. 6C		Primary Cra			Secondary Crack Lengths millimeters (inches)	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside		
3Hz	7,400	3.8 (.15)					
5Hz	9,027	5.1 (.20)					
	9,577				3.8 (.15)		
	12,827	6.4 (.25)	3.8 (.15)				
	13,127			7	5.1 (.20)		
	17.927	7.6 (.30)		·			
	17,927 18,627		5.1 (.20)				
	18,927			5.1 (.20)	6.4 (.25)		
	19,927	8.9 (.35)					
	20.427				7.6 (.30)		
	24.627	10.2 (.40)	6.4 (.25)				
	24.727			6.4 (.25)			
	24,927				8.9 (.35)		
	25,927		7.6 (.30)				
	29.527		8.9 (.35)				
	29,727	11.4 (.45)					
	29,927 30.427			7.6 (.30)			
	30.427				10.2 (.40)		
	38.127	12.7 (.50)	10.2 (.40)				
	38,227			8.9 (.35)			
	1. 38. 327			-	11.4 (.45)		
		14.0 (.55)	11.4 (.45)	10.2 (.40)	12.7 (.50)		
	48,527	15.2 (.60)					
	48,927			11.4 (.45)	14.0 (.55)		7
	49.127		1.12.7 (.50)				
	53,727	16.5 (.65)	14.0 (.55)				
	53,927 54,127			12.7 (.50)			
	54,127				15.2 (.60)		
	60,927	17.8 (.70)	15.2 (.60)	14.0 (.55)			
	61,127				16.5 (.65)		
5 Hz	65.727	19.0 (.75)	16.5 (.65)				-

Panel	No. 6C			ry Crac						millime	ry Crac	inches	ths
Cycle	No. of	L/H	R/H		L/1		R/H		#1 L/	H	#1 R/I	H	
Rate	Cycles	Outside	Outsi	de	Ins	ide	Insi	de					
5Hz	66,127				15.2	(.60)							
	66,427						17.8	(.70)					
T	70,427		17.8	(.70)									
	75,227	20.3 (.80)											
	75.727				16.5	(.65)							
	75,727 75,927		19.0	(.75)				7 55					
	76,927 80,727	7 7 7 - 1					19.0	(.75)					
	80,727	21.6 (.85)			35.0	7 50							
	80,927			( 00)	17.8	(.70)							
	85,727	22 2 ( 22)	20.3	(.80)									
	87,927	22.9 (.90)			10.0	( 75)	~ ~	(.80)					
-	88,227		67 (	(.85)	19.0	(.75)	20.3	(00)					
	89,927		21.6	(.05)			700		10.2	(.40)	16.5	(.65)	
-	90,927	24.1 (.95)	22.9	(.90)					10.2	(.40)	10.2	1.021	
-	94,930	24.1 (.95)	62.9	(-90)	20 2	(80)	21.6	(.85)	12.7	(.50)	17.8	(.70)	
-	95,430 98,680		1		20.3	(.80) (.85)		1001		1000			
-	98,930								14.0	(.55)	20.3	(.80)	
-	99,130		24.1	(.95)									
	100,830						22.9	(.90)					
	102,930	25.4 (1.00	1)										
	103,930				22.9	(.90)							
	104,830		25.4	(1.00)									
	104,930								15.2	(.60)	22.9	(.90)	
	104,930	26.7 (1.05	)					/ ==(					
	107,730		-		24.1	(.95)	24.1	(.95)	36 =	1 (=)	01. 3	1 051	
	107,930		-	,					16.5	(.65)	24.1	(.95)	
	108,630		26.7	(1.05)							05 1	(2 00)	
-	110,930	05 0 /2 3/	1								25.4	(1.00)	
1	111,930	27.9 (1.10	1/		OF I	(1 00)	OF I	(1.00	1				
5Hz	112,630		07.0	(1.10)	25.4	(1.00)	25.4	(1.00					

Panel	No.6C			ary Cra					2	Secondar				
Cycle Rate	No. of Cycles	L/H Outside	R/I Outs:	Н	I./	Charles of the Control of the Contro	R/H Insi		#1 I	/н	#1	R/H	#11	L/H
5Hz	116,930										27.9	(1.10)		
	117,930				26.7	(1.05)			17.8	(.70)				
	119,430					()			20.3	(.80)				
-	119,630	29.2 (1.15	)29.2	(1.15)										
-	122.630	29.2 (1.1)	129.2	(101)									20.3	(.80)
-	123.330								21.6	(,85)	30.5	(1,20)		1111
	123,530				27.9	(1.10)	26.7	(1.05)		1000		-		
_	124,330	30.5 (1.20)	30.5	(1.20)		1		1			- :			
	126 120	300,7 (1000)	300	1-1-01	29.2	(1.15)								
	126,430	31.8 (1.25)			- Innglitude	-								
	130.830		31.8	(1.25)										
	130,930			-					22.9	(.90)				
	131,430				30.5	(1.20)	27.9	(1.10)						
	131,530												22.8	(.90
	131,630										33.0	(1.30)		
	134,130		33.0	(1.30)										
	137.830	33.0 (1.30	)											
	137,930				31.8	(1.25)	29.2	(1.15	)				24.1	(.95
	138,530								25.4	(1.00)	34.3	(1.35)		
	141,930		34.3	(1.35)										-
	143,430	34.3 (1.35	)		33.0	(1.30)								
	143,930						30.5 (	1.20)					26.7	(1.0
	144,430								26.7	(1.05)	36.8	(1.45)		
	147.830		35.6	(1.40)					-					
	148.130	35.6 (1.40	)										27.9	(1.10
	148,730			-		7			27.9	(1.10)				
	148,930				34.3	(1.35)								-
	153,330	-	36.8	(1.45)		7-7-1		/2 0						
	153,930				35.6	(1.40)	31.8	(1.25	)					
	156,530									/2 2 =	00 1	15 55		
	156,730	200 /212	100 -	/2 50			-		29.2	(1.15)	39,4	(1.55)		
5Hz	157,830	36.8 (1.45	)38.1	(1.50)	L									

Panel	No.6C			Second mill:	dary Crack Len imeters (inche	gths es)		
Cycle Rate	No. of Cycles	#13 L/H	#13 R/H					
5Hz	116,930							
	117,930							
7	119,430							
-	119.630							
1	119,630							
	123,330							1.
	123,530							
	123,330 123,530 124,330 126,430 130,730 130,830							
	126-430							
	130,730							
	130,830							100
	130,930							
	131,430							
	131,530							<u> </u>
	131,630							-
	134,130							
	137.830							15
	137,930		-					
	138,530						-	
	137,930 138,530 141,930 143,430							-
	143,430		+				-	-
	11+3,930	1						
	143,930 144,430							
	147,830	-						-
	148,130	1					-	
	148,730		-					
-	147,830 148,130 148,730 148,930						-	1
-	1 153,330	1	-				-	-
-	153,930	611 / 051	80 / 2	E)			-	1
	156.530	6.4 ( .25)	8.9 (.3	2/				
	156,730 157,830	-	-					
5Hz	1 157, 830	1						-

Panel	No.6C				ary Cra								ck Leng		
Cycle Rate	No. of Cycles	L/I Outs:		P/i		I,/ Ins	H ide	R/ Ins	H ide	#1	L/H	#1	R/H	#11	L/H
5Hz	161.430	38.1	(1.50)												
A	161,930									30.5	(1.20)	40.6	(1.60)	30.5	(1.20)
	164,530			39.4	(1.55)										
	167,730							33.0	(1.30)						
	167.930					36.8	(1.45)								
	168.130	39.4	(1.55)												
	173.730			40.6	(1.60)										
	173.930									33.0	(1.30)	43.2	(1.70)	34.3	(1.35)
	173,930 174,530					38.1	(1.50)	34.3	(1.35)						
	184.894	41.9	(1.65)	43.2	(1.70)	39.4	(1.55)	35.6	(1.40)	35.6	(1.40)	45.7	(1.80)	35.6	(1.40)
	193,594											47.0	(1.85)		
	194,894			44.4	(1.75)							Crack			
	196,394									36.8	(1.45)	Ends	at	36.8	(1.45)
	196,494	43.2	(1.70)									Edge	of		
	200,994	44.4	(1.75)									Speci	men		
	204.194		100	45.7	(1.80)										
	208, 194				,					38.1	(1.50)			38.1	(1.50)
	208,494	45.7	(1.80)												
	209.094			47.0	(1.85)										
	213.894		The Control							43.2	(1.70)				
	214.394														
	214.894														
1	216.294	1				44.4	(1.75)								
	216.894			48.3	(1.90)										
	217.194				( )			40.6	(1.60)				4	Overl	aps R/H
	217.894	47.0	(1.85)					Overl	(1.60) aps #11	L/H				Prima	ry
	219,894			49.5	(1.95)			Secon	dary					Crack	
	220,294							Grack						39.4	(1.55)
	220,894					47.0	(1.85)	41.9	(1.65)						
	222,894	1													
	223,194	48.3	(1.90)										1		
5Hz	223,694			50.8	(2.00)							47.0	(1.85)		

Panel	No. 6c			Secondary Crack Lengmillimeters (inche		
Cycle Rate	No. of Cycles	#13 L/н	#13 R/H			
5 Hz	161,430					
1	161,930	8.9 (.35)	11.4 (.45)			
T	164,530					
	167,730					
	167,930 168,130					
	168,130					
	173,730		1			
	173,930 174,530 184,894	11.4 (.45)	14.0 (.55)			
	174,530		1 (-)			100
	184,894	14.0 (.55)	16.5 (.65)			
	193,594 194,894	-	-			
	194,894					
	196,394	-	-			
	196,494	-	+			
	200,994	+	+			
	1204,194	1000 ( 55)	21.6 (.85)		les les	
	208,194 208,494 209,094	19.0 (.75)	121.0 (.05)			
	1208,494	+				
	213.894	-				
	213,094	21.6 (.85)				
	214,394	121.0 (.0)	22.9 (.90)			
	216 204	-	1			
	216,294	-				
-	217,194					
	217.894					
	219.894					
	220,294					
	220 804					-
	220, 894 222, 894	22.9 (.90)				
1	223.194					
5 Hz	223,694					

Panel	No.6c		Primary Cra millimeters				ry Crac			
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1 L/H	#1	R/H	#11	L/H
4Hz	226,294						47.0	(1.85)		
1	227,594									
7	227,894				143.2 (1.70)					
-	228.894				1				40.6	(1.60
	228.994									
	229.394					44.4 (1.75)				
	220 004									
	229,994	49.5 (1.95)						:		
	233,094		52.1 (2.05)							
	234.694									
	235 194				144.4 (1.75)					
	235,194	50.8 (2.00)								
	236,394			48.3 (1.90)						
	236.594									1- (-
	239.094						1		41.9	(1.65
	239.294				45.7 (1.80)		-			
	239,394						-			
	239.894						-			
4Hz	240,294		53.3 (2.10)			1 = 0 (3 0=)	1			
5Hz	241.094					47.0 (1.85)	-			
	241,594						-			
	243,494	52.1 (2.05)								
	243.894			49.5 (1.95)			-			
	244,194		4	Overlaps		ļ	-			
	244,594			#13 R/H	11.5 7 /2 001		-			
	245,094			Secondary	45.7 (1.80)	1.0 2 (7 00)	+			
	246,194			Crack		48.3 (1.90)	-		112 0	/1 70
	246.494		=1 ( (0 ==1		-		-		43.2	(1.70
	247,894		54.6 (2.15)				-		-	
	249, 394						-		-	
4Hz	249,694	53.3 (2.10)			-		47.0	(1.85)	-	
4HZ	251,194		55.9 (2.20)	L			1 71.0	(1.00)		

Panel	No.6C			Se	econdary millimet	Crack Len	gths s)		
Cycle Rate	No. of Cycles	#13 L/H	#13 R/H	#14	R/H				
4Hz	226,294		25.4 (1.00)						
	227.594			25.4	(1.00)				
	22 <b>7,5</b> 94 22 <b>7,</b> 894								
	228 804								
	228,994 229,394	24.1 (.95)							
	229.394								
	229,904		26.7 (1.05)						
	230,494								
	233.094								
	234,694			26.7	(1.05)				
	235, 194								
	235,594								
	236,394								
	236,594		27.9 (1.10)						
-	239.094							 	
	239, 204	(2 00)		-					
	239,394	25.4 (1.00)		27.9	(1.10)			 	
	239,894	-		12-1.9	(1.10)				
4Hz	240,294	-		-					
5Hz	241.094	26.7 (1.05)							
-	241,594	20.1 (1.0)							
-	243,494	+							
-	243.894	-	29.2 (1.15)	-					
_	244,594	-	Overlaps L/H	20.2	(1.15)				
-	244,594	-	Primary	27.2	120-27				
-	246.194		Crack						
-	5/16 1/3/1		1						
-	247.894								
-	2/10 30/1	27.9 (1.10)							
-	249,694								
4Hz	251,194								

No. 6C		Primary Cra	ack Lengths (inches)		Seconda millim	ry Crack Leng	ths )
No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1 L/H	#1 R/H	#11 L/H
253,694 253,894 254,194	El. 6 (2.15)		50.8 (2.00)			47.0 (1.85)	
254, 294	24.6 (2.12)				49.5 (1.95)	47.0 (1.85)	
Crack HS	54.6 (2.15)	55.9 (2.20)	50.8 (2.00)	45.7 (1.80)	49.5 (1.95)	47.0 (1.85)	43.2 (1.70)
	No. of Cycles 253,694 253,894 254,194 254,294 254,394 Crack	No. of L/H Cycles Outside 253,694 253,894 254,194 254,294 54.6 (2.15) 254, 394 Crack	millimeters  No. of L/H R/H Cycles Outside Outside  253,694  253,894  254,194  254,294 54.6 (2.15)  254, 394  Crack	No. of L/H R/H L/H Inside  253,694  254,194  254,294 54.6 (2.15)  Crack  HS 54.6 (2.15) 55.9 (2.20) 50.8 (2.00)	Mo. of L/H R/H L/H R/H Cycles Outside Outside Inside Inside Inside C53,694 54.6 (2.15) 55.9 (2.20) 50.8 (2.00) 45.7 (1.80)	No. of L/H R/H L/H R/H #1 L/H Cycles Outside Cutside Inside Inside  253,694  253,894  254,294 54.6 (2.15)  Crack HS 54.6 (2.15) 55.9 (2.20) 50.8 (2.00) 45.7 (1.80) 49.5 (1.95)	No. of   L/H   R/H   L/H   R/H   L/H   H1 L/H   H2 R/H

Panel	No. 6C			Secondary millime	Crack Length ters (inches)	ns	
Cycle Rate	No. of Cycles	#13 L/H	#13 R/H	#14 R/H	#15		
OHZ .	253,694 253,894 254,194 254,294 254,394		30.5 (1.20)	30.5 (1.20)	No Crack Growth Data Recorded		
FINAL							-
LENG	THS	27.9 (1.10)	30.5 (1.20)	30.5 (1.20)	76.2 (3.00)		
	,						

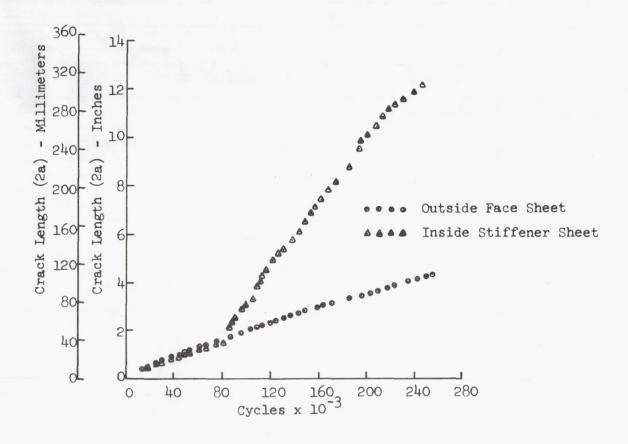


FIGURE F-27 CRACK GROWTH CURVE FOR ALUMINUM-GRAPHITE PANEL #6C (CRACK LENGTH INCLUDES SECONDARY CRACKS)

## PANEL #7C

MATERIALS: ALUMINUM-GRAPHITE

ADHESIVE: AF 126

ALUMINUM STRESS: 103 MN/m2 (15 ksi)

MAXIMUM FATIGUE LOAD: 124,540N (28,000 lbf)

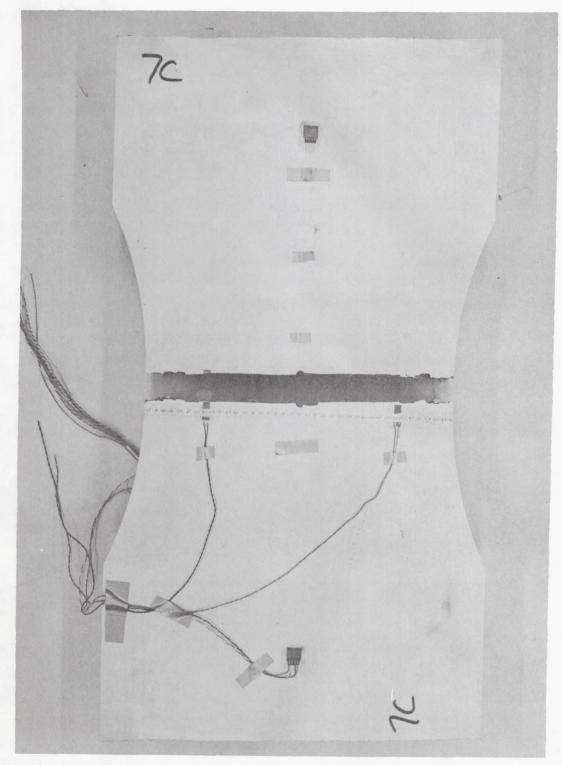


FIGURE F-28 OUTSIDE FACE SHEET AFTER RESIDUAL STRENGTH TEST - PANEL #7C

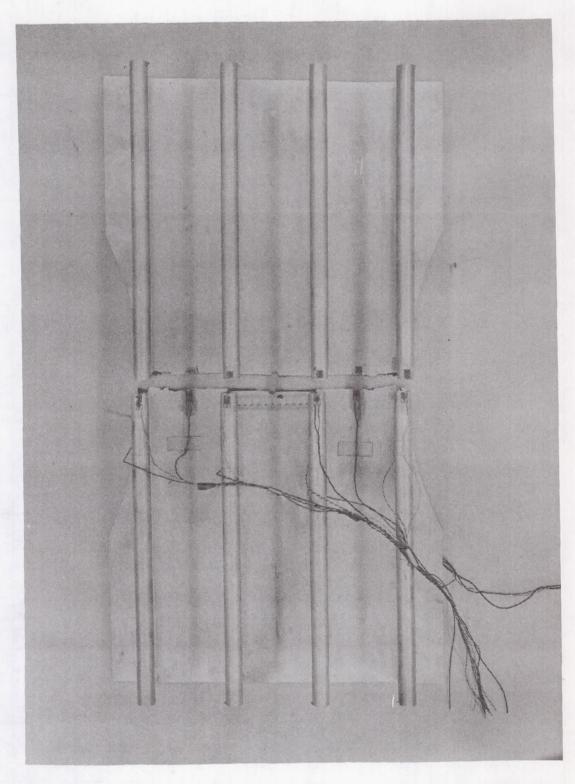


FIGURE F-29 INSIDE STIFFENER SHEET AFTER RESIDUAL STRENGTH TEST - PANEL #7C

Panel	No. 7C		Primary Cra			Secondary Crack Lengths millimeters (inches)
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H In <b>si</b> de	R/H Inside	
бнұ	32,000	3.8 ( .15)				
4	34,000				5.1 (.20)	
	40,000	5.1 (.20)				
	42,000				6.4 (.25)	
V	45,000	6.4 ( .25)				
6Hz	46,000				7.6 (.30)	
7Hz	50,556					
	55,056	7.6 (.30)				
	55.156			5.1 (.20)	8.9 ( .35)	
	60.556			6.4 (.25)		
	70.056		3.8 (.15)			
	72,256	8.9 (.35)				
	73,056			7.6 (.30)	10.2 (.40)	
	78.056		5.1 (.20)			
	78.456			8.9 (.35)		
	85.056	10.2 (.40)				
	85 556		64 (25)			
	86,056 92,256			10.2 (.40)	11.4 ( .45)	
	92,256		7.6 (.30)			
	98,056	11.4 ( .45)				
	99.056		8.9 (.35)			
	102,056			11.4 ( .45)		
	107,056				14.0 ( .55)	
	112,056	12.7 ( 50)	10.2 (.40)	12.7 (.50)		
	127,156				15.2 ( .60)	
	127.856	14.0 (.55)	11.4 (.45)			
	127,956			14.0 ( .55) 15.2 ( .60)		
	135,056			15.2 ( .60)		
	135,456				16.5 ( .65)	
	143.756	15.2 (.60)	12.7 (.50)			
Y	154,056				17.8 (.70)	
7Hz	159,216	16.5 ( .65)	14.0 (.55)	17.8 ( .70)		

Panel	No. 7C		Primary Cramillimeters			Secondary Crack Leng millimeters (inches	
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside		
7Hz	163,316		15.2 ( .60)				
	164,216			19.0 (.75)	19.0 ( .75)		
T	180,616				20.3 ( .80)		
	180,716	17.8 ( .70					
	190,936			20.3 (.80)			
	197.516			21.6 ( .85)	21.6 (.85)		
	205.716		16.5 ( .65)				- 14
	212,716			22.9 ( .90)	22.9 ( .90)		-
	220,816	20.3 ( .80	17.8 ( .70)	, ,			
	227,216				24.1 ( .95)		
	241.666	Committee of the last of the l	19.0 (.75)	25.4 (1.00)			-
	254,566			26.7 (1.05)			
	258, 366		-		26.7 (1.05)		
	258,866				-		in.
	266,166		20.3 (.80)	07 0 /2 70			
	267,566		-	27.9 (1.10) 29.2 (1.15)			
	277,166			L7. (1.1)	27.9 (1.10)		
-	280,666		1		21.9 (1.10)		
	281,066		21.6 (.85)				
	289, 266 295, 166		(1.0)		29.2 (1.15)		
	308.366		1	30.5 (1.20)			
	317.066		22.9 ( .90)	12.20			
			1009	31.8 (1.25)	30.5 (1.20)		
	317,566 337,566	26.7 (1.05	)	1			
	338 566	12.07		33.0 (1.30)	31.8 (1.25)		
	338,566 347,566		24.1 ( .95)				
	347,766				33.0 (1.30)		
	356.066						-
	356.566			34.3 (1.35)			-
1	371 066	20.2 (1.15	25.4 (1.00)				
7Hz	371.466			35.6 (1.40)			

Panel	No.7C	Primary Crack Lengths millimeters (inches)									ry Crack Lengths eters (inches)	
Cycle R <b>at</b> e	No. of Cycles	L/H Outside		R/H Outside			L/H Inside		H ide			
7Hz	380,271							34.3	(1.35)			
-	395.271	30.5	(1.20)									
	396,271		,	26.7	(1.05)		NAME OF THE OWNER, OWNE					
	403.271					36.8	(1.45)					
	409.871	31.8	(1.25)	27.9	(1.10)							
	431.271	33.0	(1,30)	29.2	(1.15)							
	434 450					38.1	(1.50)	35.6	(1.40)			
	449.271							36.8	(1.45)			
	452.771	34.3	(1.35)	30.5	(1.20)	39.4	(1.55)		(= ===			
	466,471	35.6	(1.40)			40.6	(1.60)	38.1	(1.50)			
	473.271			31.8	(1.25)							
	483.271	36.8	(1.45)									
	484 271					41.9	(1.65)					
	495 671			33.0	(1.30)	43.2	(1.70)					
	512,610	38.1 39.4	(1.50) $(1.55)$	34.3	(1.35)	111 11	(1.75)	40.6	(1.65)			
	522,419	39.4	(1.55)	25	73 1:01	45.7	(1.00)	41.9	(1.07)			
	534,119	40.6	(1.60)	35.6	(1.40)	1.77	(1.85)	112 0	(1.70)			
	534,619			20	/2 LEV	47.0	(1.05)	44.4	(1.75)			
	545,619	1 -	12 (=)	36.8	(1.45)	-		44.4	(1.13)			
	546,619	41.9	(1.65)	-		1.0 0	(2 00)	1.5 5	(1.80)			
-	562,951	-		00 -	/2 501	48.3	(1.90)	45.1	(1.00)			
	569,951	-		38.1	(1.50)			-				
	574,619	-	/= ==\	39.4	(1.55)	-		-	-			
	576,619	43.2	(1.70)	-		1	/3 051	1.7.6	(2 95)			
	577,119	1111	/2 55	1100	(1.60)	49.5	(1.95)	47.0	(1.85)			
	589,219	44.4	(1.75)	40.6	(1.60)	50.8	(2.00)	1),8 2	(1,90)			
	589,619	-		41.9	(1.65)	52.1	(2.05)	49.5	(1.95)			
	600,919	1.5 5	(1.80)		(1.70)		(2.0))	77.7	(1.77)	7.51		
_	619,619	45.7	(1.00)	43.2	(1.70)	53.3	(2.10)	-	-			-
-	620,619	1 = 6	/2 0=1	1.1. 1.	(1.75)		(2.15)		-			
7Hz	637,919 644,677	47.0	(1.85)	44.4	(1-75)	24.0	(215)	50.8	(2.00)			

Panel	No. 70				ry Cra						ry Crack Lereters (inche	
Cycle R <b>at</b> e	No. of Cycles	L/I Outs:	H ide	R/H Outsi	I	L/I Ins:	ŀ	R/I	H ide			
7Hz	654,477					55.9	(2.20)					
A	654,527	48.3	(1.90)									
	667,077			45.7	(1.80)							
	676,477	49.5	(1.95)	47.0	(1.85)							
	676,677					57.2	(2.25)					
	697.777	50.8	(2.00)	48.3	(1.90)							
	703.583							52.1	(2.05)	100		
	703,677					58.4	(2.30)					
	728.677			49.5	(1.95)							270
	729.177	52.1	(2.05)									- 4
	742.677	53.3	(2.10)									
	742.877			50.8	(2.00)							
	754,077			52.1	(2.05)							
	754,477	54.6	(2.15)									
	754,677					59.7	(2.35)					
	755,177							53.3	(2.10)			
	775,877	55.9	(2.20)	53.3	(2.10)							
	776,318					62.2	(2.45)		(2.20)			-
	789.718			54.6	(2.15)			57.2	(2.25)			
	807,318	57.2	(2.25)									
	808,318			55.9	(2.20)	63.5 64.8	(2.50)					
	823,418	58.4	(2.30)	57.2	(2.25)	64.8	(2.55)	58.4	(2.30)			
	839,318		(2.35)									
	851,218		(2.40)	58.4	(2.30)							-
	851,318 863,318					66.0	(2.60)	59.7	(2.35)			
	863,318					67.3	(2.65)	-				
	865.318			59.7	(2.35)							-
	872.318					68.6	(2.70)					
	1878.318	62.2	(2.45)	61.0	(2.40)							-
	880,318 881,118							61.0	(2.40)			-
	881,118	,		-		69.8	(2.75)	1	(0.1=)			+
7Hz	889,818	63.5	(2.50)			71.1	(2.80)	62.2	(2.45)			

Panel	No. 7C	Primary Crack Lengths millimeters (inches)									ry Crack Lengths eters (inches)
Cycle Rate	No. of Cycles	L/ Outs	H ide	R/i		L/ Ins		R/I	H ide		
7Hz	897,818			62.2							
1	911,118	64.8	(2.55)	63.5	(2.50)						
	911,818					72.4	(2.85)	63.5	(2.50)		
	929.818					73.7	(2.90)				
	934.818	66.0	(2.60)	64.8	(2.55)						
	941,908					74.9	(2.95)	66.0	(2.60)		
		67.3	(2.65)	66.0	(2.60)						
	962,118					76.2	(3.00)	67.3	(2.65)		
	963,918	68.6	(2.70)	67.3	(2.65)						
	968.818 977,218				( = == \ \		/a a=\	68-6	(2.70)		
	977,218	7 6		68.6	(2.70)	77.5	(3.05)	69.8	(2.75)		
	992,818	69.8	(2.75)	69.8	(2.75)				10.00		
	993,818							71.1	(2.80)		
	1,011,818	71.1	(2.80)	71.1	(2.80)				(= -1)		
	.012,718							73.7			-
7Hz	,020,520							74.9	(2.95)		
FINAL	CRACK										
	NGTHS	71.1	(2.80)	71.1	(2.80)	77.5	(3.05)	74.9	(2.95)		
RESID	UAL STREN	TH =	539,100	N (12	1.200 1	os.)					
	The second										
A CONTRACTOR	in the second									1	
N. 19						-		-			
								-			

## PANEL #8C

MATERIALS: ALUMINUM - GLASS

ADHESIVE: AF 126

ALUMINUM STRESS: 103 MN/m<sup>2</sup> (15 ksi)

MAXIMUM FATIGUE LOAD: 84,510N (19,000 lbf)

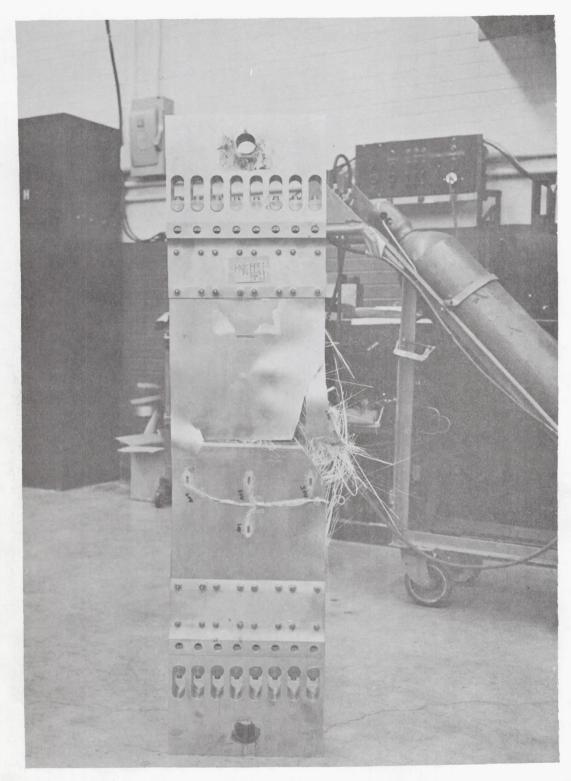


FIGURE F-30 OUTSIDE FACE SHEET AFTER RESIDUAL STRENGTH TEST - PANEL #8C

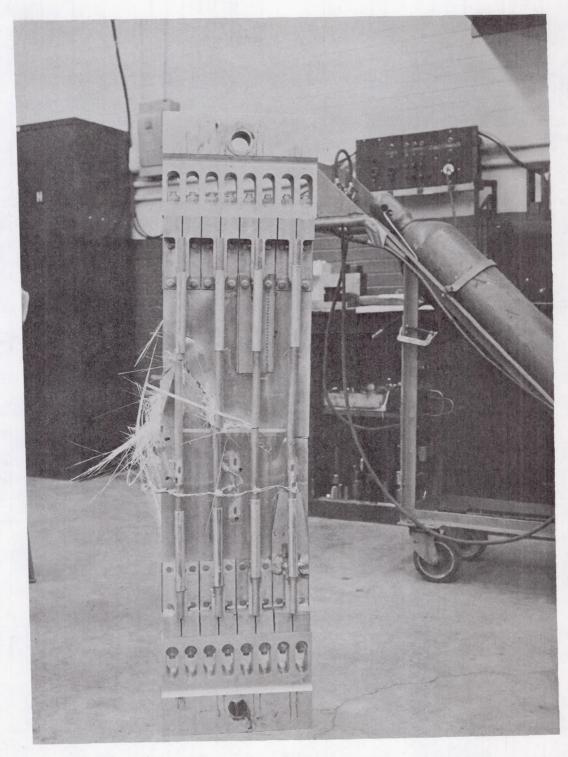


FIGURE F-31 INSIDE STIFFENER SHEET AFTER RESIDUAL STRENGTH TEST - PANEL #8C

Panel	No. 8C		Primary Cra		•	Secondary Crack Lengths millimeters (inches)
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	I/H Inside	R/H Inside	
4 Hz	20,000			3.2 ( .125	3.2 ( .125)	
1	21,000			3.8 ( .15)	3.8 ( .15)	
	21,700		5.1 ( .20)			
	22.000			5.1 ( .20)		
	22,200	3.2 ( .125)				
	22,200	3.2 ( .125) 3.8 ( .15)				
	23,900				5.1 ( .20)	
	25,100		6.4 ( .25)			
	25,700	5.1 (.20)				
	28,000			6.4 ( .25)		
	29,500				6.4 ( .25)	
	30,300	6.4 ( .25)				
	32,960			7.6 ( .30)		
	35,060				7.6 ( .30)	
	35,860		7.6 ( .30)			
	36,860 39,060	7.6 ( .30)				
	39,060		0 - ( -=>	8.9 ( .35)		
	40,060		8.9 ( .35)			
	40,460	80 / 351			8.9 ( .35)	
-	45,060	8.9 ( .35)		70.0 ( ).0		
+	45,360 46,860			10.2 ( .40)	70.0 ( 10)	
+	47,560		70.0 ( )(0)		10.2 ( .40)	
-	50,660		10.2 ( .40)	11.4 ( .45)		
+	51,360	10.2 ( .40)		11.4 ( .45)		
-	51,300	10.2 ( .40)			11.4 ( .45)	
+	53,360 57,360			12.7 ( .50)	11.4 (.45)	
+	57,560		11.4 ( .45)	10.1 ( .)0)		
+	59,360	11.4 ( .45)	11.4 ( .4)			
+-	61,060	11.7 ( .4)			10.7 ( 50)	
+	61,060 61,560		12.7 ( .50)		12.7 ( .50)	
4 Fiz	61,840			14.0 ( .55)		

Panel	No. 8c		Primary Crac			Crack Lengths ers (inches)
Cycle	No. of	L/H	R/H	L/H	R/H	
Rate	Cycles	Outside	Outside	Inside	Inside	
4 Hz	64,820				14.0 ( .55)	
112	67,360	12.7 ( .50)				
1	68,670	12.1 ( .)07		15.2 ( .60)		
-	69,260		14.0 ( .55)			·
	70.760				15.2 ( .60)	
V	70,760 73,860 73,960			16.5 ( .65)		
4 Hz	73,960	14.0 ( .55)				
4.5 H	77,030		15.2 ( .60)			:
1	77,060				16.5 ( .65)	
	79,680			17.8 ( .70)		
	81,360	15.2 ( .60)				
	82,300				17.8 ( .70)	
	83,310		16.5 ( .65)			 
	84,640			19.0 ( .75)		
*	88,110	16.5 ( .65)			( 70)	 
4.5 E	12 88,360				19.0 ( .75)	
4 Hz.	92,730		17.8 ( 70)	20.3 ( .80)		
-	93,170	0 /		20.3 ( .80)		 
	96,130	17.8 ( .70)			20.3 ( .80)	 
	98,150		70.0 / 75)	-	20.3 [ .00]	 
	101,830		19.0 ( .75)	21.6 ( .85)		`
	102,030			21.6 ( .85)		
	104.730	19.0 ( .75)			21.6 ( .85)	
-	105,030		20.3 ( .80)		21.01.021	
-	109,180		20.3 ( .80)	22.9 ( .90)	-	
-	109,530 112,730			6609 ( 090)	22.9 ( .90)	
	114,030	20.3 ( .80)				
-	117,130	20.3 ( .00)		24.1 ( .95)		
-	117,530	-	21.6 ( .85)			
-	1119.830	+	-1.0 ( .0)]		24.1 ( .95)	
4 Hz	122,930	21.6 ( .85)				

Panel 1	No.8c		Primary Cra millimeters			Secondary Crack Lengths millimeters (inches)
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	
4 Hz	126,830		22.9 ( .90)			
1	127,030			25.4 (1.00)		
-	128,530				25.4 (1.00)	
-	128,830	22.9 ( .90)				
-	134.630	22.9 ( .901		26.7 (1.05)		
-	134,030		24.1 ( .95)			
-	135,030		- to the state of		26.7 (1.05)	
4 Hz	137,730	24.1 ( .95)				
5 Hz	141,498			27.9 (1.10)		
The state of the s	141 308				27.9 (1.10)	
-	143,398 143,478		25.4 (1.00)			
	144,078	25.4 (1.00)				
	148.898			29.2 (1.15)		
	150.088		26.7 (1.05)			
	150,088 150,298				29.2 (1.15)	
	153.068	26.7 (1.05)				
	153,068 154,698			30.5 (1.20)		
	157.028				30.5 (1.20)	
	157.148		27.9 (1.10)			
	159.838	27.9 (1.10)				
	159,838 161,508			31.8 (1.25)		
	165.088				31.8 (1.25)	
	165,928	29.2 (1.15)	29.2 (1.15)			
	167,278	29.2 (1.15)				
	167,768			33.0 (1.30)		
	173,728					
	173,978		30.5 (1.20)			
	174,088			1 2 /2 25	33.0 (1.30)	
	174,578			34.3 (1.35)		
Y	179,079 179,719	0 /= 0=1	31.8 (1.25)			
5 Hz	179,719	31.8 (1.25)			21, 2 (3, 25)	
4.5H	z 187,009				34.3 (1.35)	

Panel	No. 8c		Primary Cra millimeters		•	Secondary Crack Lengt millimeters (inches)	ths
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside		
4.5 Hz	187,229			35.6 (1.40)			
	188,159		33.0 (1.30)				
	188.349	33.0 (1.30)					
	196.339		34.3 (1.35)				
	197,699	34.3 (1.35)					
	205.799	35.6 (1.40)					
	205,999 206,499 208,199		35.6 (1.40)				
-	206,499	8 2 2 2		36.8 (1.45)			
-	208,199		- ( ) ( )		36.8 (1.45)		
	213,599	-60 (= 1=)	36.8 (1.45)				
-	215,099	36.8 (1.45)					
	217,199			38.1 (1.50)			
-	219,499 220,129		38.1 (1.50)	20 1. /2 55			
). F 77-		20 7 (7 50)		39.4 (1.55)			
4.5 Hz	224,669	38.1 (1.50)		40.6 (1.60)			
4 HZ	229,129	· · · · · · · · · · · · · · · · · · ·	20 1. (2 55)	40.6 (1.60)			
-	229,219	39.4 (1.55)	39.4 (1.55)				
	233,499	39.4 (1.55)			20 7 /7 50		
-	234,919 235,919			41.9 (1.65)	38-1 (1-50)		
-	240,099		40.6 (1.60)	41.9 (1.05)			
-	240,499		40.0 (1.00)		39.4 (1.55)		
-	241.799	40.6 (1.60)			39.4 (1.33)		
-	2112 000	40.0 (1.00)		43.2 (1.70)			
1	243,999 245,649			100 140191	40.6 (1.60)		
	249.829			44.4 (1.75)	10.0 (1.00)		
1	249,919		41.9 (1.65)	1			
	253 000	41.9 (1.65)					
	254,969	11.5			41.9 (1.65)		
	256.199			45.7 (1.80)			
	257,699		43.2 (1.70)				
4 Hz		43.2 (1.70)					

Panel	No. 8c		Primary Cra			Secondary Crack Lengths millimeters (inches)
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	
4 Hz	262,479				43.2 (1.70)	
112	264,499		44.4 (1.75)			
1	266,969	1		47.0 (1.85)		
-	271,089	44.4 (1.75)				
-	274,749	1101 (20)			44.4 (1.75)	
-	278.029		45.7 (1.80)			
1	278.649			48.3 (1.90)		
-	287 049	45.7 (1.80)				
	283.869	7). (1.00)			45.7 (1.80)	
-	284.729			49.5 (1.95)		
	287.059		47.0 (1.85)			
	294,159	47.0 (1.85)				
1	294.349	1100			47.0 (1.85)	
-	294,519			50.8 (2.00)		
_	296,599		48.3 (1.90)			
	200 600				48.3 (1.90)	
1	299,609 299,729			52.1 (2.05)		
	303.519	48.3 (1.90)				
	307.479				49.5 (1.95)	
	208 510			53.3 (2.10)		
	308,519		49.5 (1.95)			
	312,449	49.5 (1.95)				
	316,079			54.6 (2.15)		
	316.399		50.8 (2.00)			
	317,019				50.8 (2.00)	
-	322.689			55.9 (2.20)		
1	325.659	50.8 (2.00)				
1	225 800				52.1 (2.05)	
	325,899 327,249		52.1 (2.05)			
-	330.029	1		57.2 (2.25)		
-	332,589	52.1 (2.05)				
4 Hz	335.259		53.3 (2.10)			

Panel	No. 8c		Primary Cra millimeters			Secondary Crack Lengths millimeters (inches)
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	
4 Hz	336,609			58.4 (2.30)		
	336,829				53.3 (2.10)	
	342,109	53.3 (2.10)				
	342,809	53.3 (2.10)		59.7 (2.35)		
	345,759		54.6 (2.15)			
	346.349			61:0 (2.40)		
	348,369				54.6 (2.15)	
	350,549 350,859	54.6 (2.15)				
	350,859			62,2 (2,45)		
	353,369 356,769		55.9 (2.20)			
	356.769				55.9 (2.20)	
	358.179			63.5 (2.50)		
	360,679	55.9 (2.20)	57.2 (2.25)			
	364,339			64.8 (2.55)		
	369,609	57.2 (2.25)				
	369,619 369,949		58.4 (2.30)		F7 0 10 05 \	
-	369,949			(6 0 10 60)	57.2 (2.25)	
-	370,389 376,189			66.0 (2.60) 67.3 (2.65)		
-	376,189	50 1 (0 20)		01.3 (2.07)		
	377,489	58.4 (2.30)	F0 7 (0 of)			
-	377,839		59.7 (2.35)		FQ 1, (0, 20)	
-	378,109	(0.05)	(7 0 (0 10)	(0 ( (0 70)	58.4 (2.30)	
	387,481	59.7 (2.35)	61.0 (2.40)	68.6 (2.70)	59.7 (2.35)	
	391,625		(0 0 /0 1/5)	69.8 (2.75)		
	391,625 394,365 394,595		62.2 (2.45)		61.0 (2.40)	
	1394,595	(7 0 (0 1:0)			01.0 (2.40)	
-	396,315	61.0 (2.40)		77 7 (0 90)		
-	396,095			71.1 (2.80) 72.4 (2.85)		
-	400,195		63.5 (2.50)	12.4 (2.0)		
-	402,165	-	03.7 (2.70)		62.2 (2.45)	
4 Hz	403,465	-		73.7 (2.90)	02.2 (2.7)	

Panel	No. 8c		Primary Cra	ck Lengths (inches)		Secondary Crack Lengths millimeters (inches)
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	
4 Hz	406,555	62.2 (2.45)				
1	412,905		E L SI		63.5 (2.50)	
	412,985	7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7	64.8 (2.55)			
	412,985 413,175			74.9 (2.95)		
_	413,225	63.5 (2.50)				
	420,475	63.5 (2.50)		76.2 (3.00)		
	421.835		66.0 (2.60)			
	422,385				64.8 (2.55)	
	427.685		67.3 (2.65)	77.5 (3.05)	67.3 (2.65)	
	443,210		68.6 (2.70)			
	443,235			78.7 (3.10)		
	443.240	67.3 (2.65)				
	443.285				68.6 (2.70)	
	1446.085	68.6 (2.70)				
	446,885 448,185				69.8 (2.75)	
	448,185		69.8 (2.75)			
	451,485				71.1 (2.80)	
	452,785		71.1 (2.80)			
	460,685		72.4 (2.85)			
	460.885	69.8 (2.75)			72.4 (2.85)	
	462,185	69.8 (2.75)				
	463,585			82.6 (3.25)		
	464,385			83.8 (3.30)		
	465,985 466,685 467,685			85.1 (3.35) 87.6 (3.45)		
	466.685			87.6 (3.45)		
	467,685			88.9 (3.50)		
	468,585			90.2 (3.55)		
	468,685	71.1 (2.80)		1 10 (-)		
	469,285 470,485			91.4 (3.60) 92.7 (3.65)		
	470,485	1		92.7 (3.65)	F0 F (0 00)	
	471.185	-	-	NI 0 (2 70)	73.7 (2.90)	
4 Hz	471,185 471,585			94.0 (3.70)	13.1 (2.90)	

Panel	No. 8c		Primary Cra			ry Crack Lengths eters (inches)
Cycle	No. of	L/H	R/H	T/H	R/H	
Rate	Cycles	Outside	Outside	Inside	Inside	
4 Hz	471,685	72.4 (2.85)				
	472,585			95.2 (3.75)		
-	474,485			95.2 (3.75) 96.5 (3.80)		
-	474,685	73.7 (2.90)		20.2 (3.00)		
_	475,285	13.1 (2.50)		97.8 (3.85)		
	1475 685		73.7 (2.90)	71.0 ().021		
_	475,685 476,885	İ	1001/5070/	99.1 (3.90)		
	477,685			77.1	74.9 (2.95)	i i i i i i i i i i i i i i i i i i i
_	1,70 185			100 3 (3 05)		
_	479,185 480,485			100.3 (3.95)		
	481.685			102.9 (4.05)		
	482 485		74.9 (2.95)			
	482,485 482,685			104.1 (4.10)		
	483,685	74.9 (2.95)				
	483.925			105.4 (4.15)		
	483,925 485,585				76.2 (3.00)	- 102°
	486,185 487,065 487,585				77.5 (3.05)	
	487,065			106.7.(4.20)		
	487,585		76.2 (3.00)			
	489,585			108.0 (4.25)		
	492,185			109.2 (4.30)		
	493,585			110.5 (4.35)		
	495,285			111.8 (4.40)		
	496.585			113.0 (4.45)		
	497,685			114.3 (4.50)		
	498.585	76.2 (3.00)	77.5 (3.05)			
	502.285	77.5 (3.05)				
	502,485 503,785			115.6 (4.55) 116.8 (4.60)		
	503.785			116.8 (4.60)		
	505.135		78.7 (3.10)			
	506,385	78.7 (3.10)	1			
4 Hz	506.785			118.1 (4.65)		

Panel	No.8c		Primary Cramillimeters			Secondary Crack Lengths millimeters (inches)
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	
4 Hz	508,885			119.4 (4.70)		
A	511,685			120.6 (4.75)		
	511,685 512,885				78.7 (3.10)	
-	515,135		80.0 (3.15)			· · ·
	519,585			121.9 (4.80)		
_	520,785	80.0 (3.15)				
1	521,285		81.3 (3.20)			
	522,826		and the second s	123.2 (4.85)		
	524.926	881.3 (3.20)				
_	526,426			124.5 (4.90)		
	526.626		82.6 (3.25)			
	527.426	82.6 (3.25)				
	520,826			125.7 (4.95)		
	529.826 532.826			127.0 (5.00)		
	534.226		83.8 (3.30)			
	534.526	83.8 (3.30)				
	534,526 535,826			128.3 (5.05)		
	542.650				82.6 (3.25)	
	542.760			129.5 (5.10)		
	543,170				83.8 (3.30) 85.1 (3.35)	
	544,610				85.1 (3.35)	
	545,370			4 1 1	86.4 (3.40)	
	545,470		85.1 (3.35) 86.4 (3.40)	7/4		
	545.570		86.4 (3.40)			
	548,370				87.6 (3.45)	
	548,770			130.8 (5.15)		
	550,270				88.9 (3.50)	
	551,470				90.2 (3.55)	
	551,670	85.1 (3.35)				
	552,670				91.4 (3.60)	
V	552,670 553,770	86.4 (3.40)			20 5 /0 /5	7
4 Hz					92.7 (3.65)	

No. 8c		Primary Cra millimeters	(inches)	•	Seconda:	ry Crack Leng eters (inches	ths )
No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside			
554,170 554,529	87.6 (3.45)		132.1 (5.20)				
CRACK	87.6 (3.45)	86.4 (3.40)	132.1 (5.20)	92.7 (3.65)			
RES	DUAL STRENGTH	= 453,700 N	(102,000 lbs	)			3
							5
	No. of Cycles 554,170 554,529 CRACK	No. of Cycles Outside  554,170  554,529 87.6 (3.45)  CRACK  THS 87.6 (3.45)	No. of L/H R/H Outside  554,170 554,529 87.6 (3.45)  CRACK FTHS 87.6 (3.45) 86.4 (3.40)	No. of Cycles Outside R/H L/H Inside Inside S54,170 132.1 (5.20) 87.6 (3.45) 87.6 (3.45) 86.4 (3.40) 132.1 (5.20)	No. of L/H Outside	No. 8C millimeters (inches) millim  No. of L/H R/H L/H R/H Inside  554,170	No. of L/H

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## PANEL #9C

MATERIALS: ALUMINUM - GLASS

ADHESIVE: AF 126

ALUMINUM STRESS: 103 MN/m2 (15 ksi)

MAXIMUM FATIGUE LOAD: 74,730N (16,800 lbf)

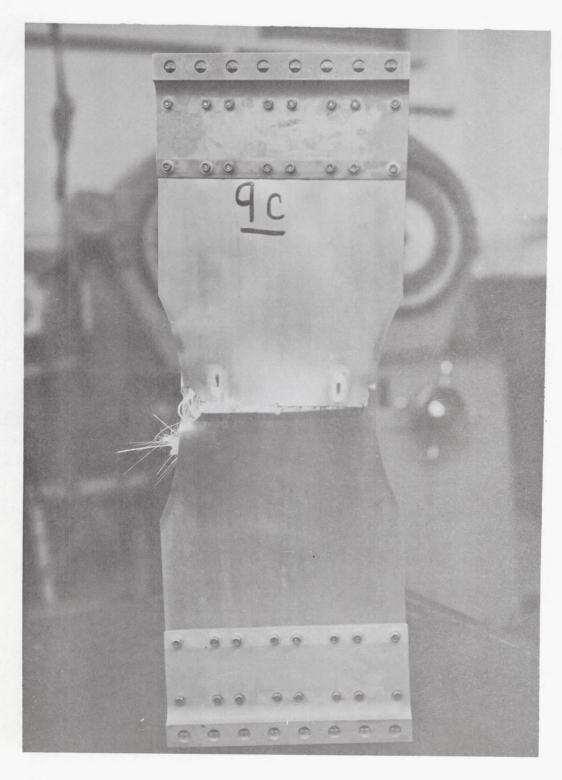


FIGURE F-32 OUTSIDE FACE SHEET AFTER RESIDUAL STRENGTH TEST - PANEL #9C

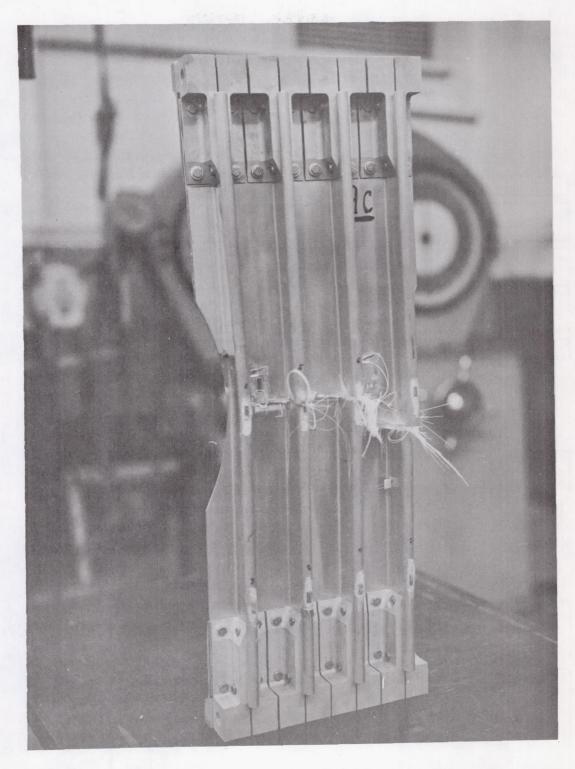


FIGURE F-33 INSIDE STIFFENER SHEET AFTER RESIDUAL STRENGTH TEST - PANEL #9C

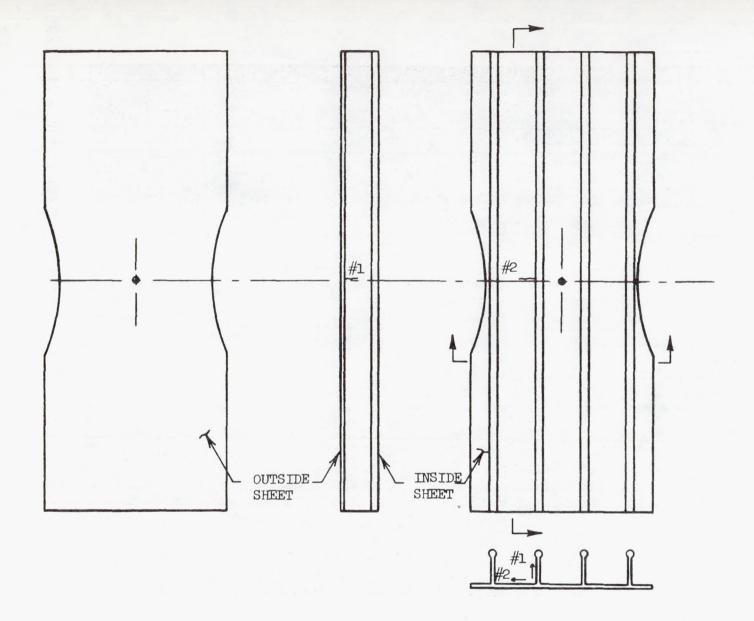


FIGURE F-34 SECONDARY CRACK LOCATIONS ALUMINUM-GLASS PANEL #9C

Panel	No. 90		Primary Crac			Secondary Crack Lengths millimeters (inches)
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	
10Hz	11,711					
4Hz	15,806		3.2 (.125)			
	21,711		5.1 (.20)			
Ŧ	22,061	3.2 (.125)				
	24.031	302 (022)	6.4 (.25)			
	24.961	5.1 (.20)				
100	29.991	6.4 ( .25)	7.6 ( .30)			
	34,711	7.6 ( .30)	7.6 ( .30) 8.9 ( .35)			
	37.111			3.2 (.125)	3.2 (.125)	
	40,011	8.9 ( .35)				
	40.511		10.2 (.40)			
	42,211			5.1 ( .20)	5.1 (.20)	
	45.311	10.2 (.40)	11.4 ( .45)			
	50.571	11.4 (.45)	12.7 ( 50)	6.4 ( 25)	6.4 (.25)	
	56,191	( )		6.4 ( 25) 7.6 ( 30)	7.6 ( .30)	
		12.7 (.50)	14.0 (55)			
		14.0 (.55)				
•	62.691			8.9 ( .35)	8.9 ( 35)	
	62,691		15.2 ( .60)		, ( 5,7)	
	68, 191	15.2 ( .60)				
	69,391				10.2 ( .40)	
	70,691		16.5 ( .65)			
	74.041	16.5 (.65)		10.2 (.40)		
	74,691		17.8 ( .70)			
	77,991	17.8 ( .70)			11.4 ( .45)	
	79,191			11.4 ( .45)		
	80.991		19.0 (.75)			
	83.391	19.0 ( .75)				
	85,291				12.7 ( .50)	
	186,391		20.3 ( .80)			
1	87,291			12.7 (.50)		
LH2	88,191	00 3 ( 80)				

Panel	No. 90		Primary Cra millimeters			Secondary Crack Lengths millimeters (inches)
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	
4Hz	91,391		21.6 ( .85)			
	91,791	-		14.0 (.55)		
1	92,691	-			14.0 ( .55)	
-	94,691	21.6 ( .85)				
-	95.891	21.0 ( .0)	22.9 ( .90)			
_	98.691		22.7	15.2 ( .60)	15.2 ( .60)	
_	100.691	22.9 ( .90)				
-	101.491	1	24.1 ( .95)			
_	102,201		1 211		16.5 ( .65)	
	104,691	24.1 ( .95)				
	105.691			16.5 ( 65)		
	106,591		25.4 (1.00)			
	109,591		26.7 (1.05)			
	110.191			17.8 (.70)	17.8 (.70)	
	111.591	25.4 (1.00)				
			27.9 (1.10)			
1	115,591			19.0 ( .75)		
-	117.691	26.7 (1.05)				
	118.091				19.0 (.75)	
	118,691		29.2 (1.15)			
	122,891			20.3 ( .80)		
	124,291				20.3 (.80)	
	124,691	27.9 (1.10)	30.5 (1.20)			
	28,291			21.6 (.85)		
	128,591	29.2 (1.15)	31.8 (1.25)		07 ( / 05)	
	129,291	100 5 /2 50	1 22 0 /1 20		21.6 ( .85)	
	135,191	30.5 (1.20)	33.0 (1.30)	00.0 / 00		
	135,591	-		22.9 (.90)		
	135,891	0 (====	10 0 (5 55)	01. 7. / 071	22.9 ( .90)	
	42.087	31.8 (1.25)	34.3 (1.35)	24.1 (.95)	24.1 (.95)	
	145,537		-	25.4 (1.00)		
4Hz	147.457			50.4 (1.00)		

Panel	No. 90		Primary Cramillimeter	ack Lengths s (inches)		Secondary Crack Lengths millimeters (inches)
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	
4Hz	149,027		35.6 (1.40			
1	152,865	34.3 (1.35	5)			
	154,987		36.8 (1.45	27.9 (1.10)	27.9 (1.10)	
	156,677	35.6 (1.40				
	159.387				29.2 (1.15)	
	160,987		38.1 (1.50			
	164,637	36.8 (1.4)	2		1 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	
	165,587				30.5 (1.20)	
1	167,987		39.4 (1.55	02 0 (2 05	31.8 (1.25)	
4Hz	169,663	38.1 (1.50		31.8 (1.25	Name and Address of the Owner, where the Party of the Owner, where the Party of the Owner, where the Owner, which is the Owner, which is the Owner, where the Owner, which is the Owner, where the Owner, which is the Owner, where the Owner, which is the Owner, whic	
5Hz	175,702		-	100 0 /2 00	the same of the sa	
	177,102		1	33.0 (1.30	4	
5Hz	178,634 182,454	1 (5 5	40.6 (1.60	2	+	
10Hz	182,454	39.4 (1.5	2	+	34.3 (1.35)	
	187.734			34.3 (1.35	the same of the sa	
	191,334		41.9 (1.65		4	
	191,824	11.6 12.6		2	-	
-	193,034	40.6 (1.6	0)	1	35.6 (1.40)	
-	198,234 201,234	-	+	35.6 (1.40	1 37.0 (1-40)	
	201,234	-	43.2 (1.70		4	
-	206,034			4		
-	206,234		24	1	36.8 (1.45)	
-	208, 734 217, 934	44.4 (1.7	5	+	10.0	
-	218,534	144.4 (I.e.)	4	36.8 (1.45		
-	210,534	-	-	7.0 (1.1)	38.1 (1.50)	
	219,634 224,434	1		38.1 (1.50		
	230,634		44.4 (1.7			
-	231.234		11.	39.4 (1.55		
-	237, 534		45.7 (1.8			
	015 034	45.7 (1.8				
10Hz	246,034	14701 (100	1		39.4 (1.55)	

Cycle No. co Rate Cycle 10Hz 249, 249,6 258, 259, 260,	234 234 234 234 234 734 47 334	L/H	(1.85)	R/I Outsi		L/I Ins:	ide	R/I Inst	i de		
249,6 258,2 259,2 260,2 270,6	534 234 734 47 334 534 48	.0	(1.85)	47.0	(1.85)	40.6	1- 1-1	R/H Inside			
249,6 258,2 259,2 260,2 270,6	534 234 734 47 334 534 48	.0	(1.85)	47.0	(1.85)		(1.60)				
258, 259, 260, 269, 270,	234 734 47 334 534 48	.0	(1.85)		_						
259, 260, 269, 270,	734 47 334 48	.0	(1 85)			41.9	(1.65)	40.6	(1.60)		
260. 269. 270.	334 48		LLOUI								
269, 270,	534 48			48.3	(1.90)						
270,6	534	.3	(1.90)								
	1.11			49.5	(1.95)						
1213.	934					48.3	(1.90)				
279.				50.8	(2.00)						
279.		.5	(1.95)					-			
281	434						(	41.9	(1.65)		
286,	334					49.5	(1.95)				
288.	234			52.1	(2.05)					-	
291.	834 50	8.0	(2.00)								
294	034					50.8	(2.00)			 -	
299,	534					52.1	(2.05)		(2.50)		
300	434							43.2	(1.70)		
304	434			53.3	(2.10)			-		 	
304	934 52	2.1	(2.05)					-			
306,	734			-1	10>	53.3	(2.10)	-			
				54.6	(2.15)			11.1. 1	(2 75)		
313.				-		-1	()	44.4	(1.75)		
318.	134		10 -5		(0.00)	54.6	(2.15)	-			
318,		3.3		55.9	(2.20)	-		-			
329,		1.6	(2.15)	57.2	(2.25)		(0.00)	-			
330, 338,	534			-		55.9 57.2	(2.20)	-			
338,	934			FO 1	(0.00)	21.2	(2.2)	-			-
339,	934		10.00	58.4	(2.30)	-		-			
340.		5.9	(2,20)	1 == =	(0.05)	-		-			
350.	852			59.7	(2.35)	58.4	(2 20)	50.8	(2.00)		
354, 10Hz 356.	652 5'	7.0	(2.25)	1		50.4	12.30	70.0	12.00		

Panel	No. 90		Primary Cra			Secondary Crack Lengths millimeters (inches)
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	
10Hz	362,852				52.1 (2.05)	
	363,552			59.7 (2.35)		
	364,252		61.0 (2.40)			
	369,152			61.0 (2.40)		
	369.252	58.4 (2.30)				
	375,652		62.2 (2.45			
	376,652				53.3 (2.10)	
	377,852			62.2 (2.45)		
	386,052	59.7 (2.35)		13 13 13 13		
	387,152		63.5 (2.50)			
	391,652		11 0 1	63.5 (2.50)		
	393,252		64.8 (2.55	)	FF 0 (0 00)	
	393,552			-	55.9 (2.20)	
	393,652	61.0 (2.40)		10 0 55		
	397,152		1. (.	64.8 (2.55)	-	
	401,552		66.0 (2.60	)	57.2 (2.25)	
				66.0 (2.60)		
	404,952	( (- )-)		100.0 (2.00)	-	
_	409,052	62.2 (2.45)		67.3 (2.65)		
-	414,352		67.3 (2.65			
	417,852	63.5 (2.50)	107-3 (2-65	4	-	
	1417.052	03.7 (2.30)		68.6 (2.70)		
	423,352		-	100.0 (2.10)	59.7 (2.35)	
	425, 152		68.6 (2.70	)		
	430,852	64.8 (2.55)				
-	430,852	04.0 (2.55,		69.8 (2.75)		
	11.35,352			1	61.0 (2.40)	
	435,852 436,252		69.8 (2.75			
	438.052		71.1 (2.80			
	438.952	66.0 (2.60)		The Real Property of		
10Hz	July 152		72.4 (2.85			

Panel	No. 90		Primary Cra millimeters				Secondary Crack Lengths millimeters (inches)	
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside			
LOHz	449,252				62.2 (	2.45)		
1	452,052			72.4 (2.85)				
	454,052	67.3 (2.65						
	458,552		73.7 (2.90)					
	461,252				63.5 (	2.50)		
	462.052			73.7 (2.90)				
	468,152	68.6 (2.70		1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1				
	468,552				64.8 (	2.55)		
	469.152		74.9 (2.95)					
	471,552 475,452			74.9 (2.95)				
	475,452				66.0 (2	2.60)		
	478,552		76.2 (3.00)					
	478,652	69.8 (2.75)						
	481,252			76.2 (3.00)				
	482.252				67.3 (2	2.65)		
	485,152				68.6 (2	2.70)		
	485,352		77.5 (3.05)					
	491,952	71.1 (2.80)	)					
	492.452		78.7 (3.10)					
	500.252		80.0 (3.15)					
	505,752	72.4 (2.85)						
	506,652		81.3 (3.20)					
	508,852				69.8 (2	2.75)		
	513,252		82.6 (3.25)					
	520,152	73.7 (2.90)						
	520.852		83.8 (3.30)					
	536.252		85.1 (3.35)					
	536,452	74.9 (2.95)						
	540,252				71.1 (2	2.80)		
	541.852		86.4 (3.40)		`			
	545,452	76.2 (3.00)			-	0 - 11		
10Hz	547,952				72.4 (2	2.85)		

Panel	No. 9C		Primary Cra					ry Crack Leng	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1		#2	
10Hz	551,153		88.9 (3.50)		73.7 (2.9	90)			
4Hz	555,653	77.5 (3.05)	90.2 (3.55)						
	557,053				74.9 (2.9	95)			
	557,153			81.3 (3.20)					
	557,853			82.6 (3.25)					
	558,253			83.8 (3.30)	76.2 (3.0	00)			
	559,153			85.1 (3.35)					
	559,853			86.4 (3.40)					
	560,353	78.7 (3.10)							
	560,453			87.6 (3.45)					
	561,153			88.9 (3.50)					
	562,253				80.0 (3.	15)			
	562,753		91.4 (3.60)	91.4 (3.60)					
	563,053				82.6 (3.1				
	563,353				83.8 (3.	30)			
	563,553			92.7 (3.65)		-	7 101		
	564,153 564,653				85.1 (3	35) 10.2	(.40)		
	564,653			94.0 (3.70)	06)				
	565,353				86.4 (3.1		1 1 = 1		
	565,453					11.4	( .45)		
	565,553				87.6 (3.	45)			
	565.653			95.2 (3.75)		-		22 / / / 5	
	565,753				100 0 70	FAV		11.4 ( .45)	
	565,653 565,753 566,353	80.0 (3.15)	4	-	88.9 (3.	201		10.7 / 50	
	566,653			1- 1- 0-	-	-		12.7 ( .50)	
	566,753		-	96.5 (3.80)	)	==\			
	567,053			-	90.2 (3.	55)	/ 501		
	567,153			-	107 1 /6	12.7	( .50)		
	567,753			- 0 (- 0-	91.4 (3.	60)			
	567.953			97.8 (3.85	00 7 /2	CEV			
-	568,853	-	10 5 10 10		92.7 (3.	071		14.0 (.55)	
4Hz	569.253		92.7 (3.65					14.0 (.33)	

Panel	No. 90		Primary									k Length	ns
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside		L/I Insi		R/I		#1		#2		
Hz	570,153				100.3	(3.95)							
1	- Contract of the last of the								14.0	( .55)			
1	570,253 570,453						94.0	(3.70)					
	570,953				101.6	(4.00)					- 9		
	571.253						95.2	(3.75)					
	571.653				102.9	(4.05)							
	572,253				104.1	(4.10)			15.2	(.60)			
	572,953				Joins				Joins	L/H	15.2	(.60)	
	573,253				#1		97.8	(3.85)	Pri	mary			
	573.653		94.0 (	3.70)	Secon	dary			Crack				
	573-953	81.3 (3.20)			Crack				-				
	574.853						99.1	(3.90)			-/-	7-7-1	
Y	576,653										16.5	( .65)	
4Hz	576.953							(4.00)					
5Hz	577.739							(4.05)	-				
A							104.1	(4.10)				/ ==>	
	579,239 579,439								-		17.8	(.70)	
	580.439							(4.15)	-				
	581,239							(4.20)	-			-	
	582 139						108.0	(4.25)				( ==\	
	582,139 582,939										19.0	(.75)	
	583.939						109.2	(4.30)					
	584 939		95.2 (	3.75)									
	585 030						110.5	(4.35)	-			-	
	585,039 585,639	82.6 (3.25)											
	586.439						111.8	(4.40)			00 0	1 001	
	586,939					-					20.3	( .80)	
	587,739		96.5 (	3.80)				(1 )				-	
	588,439					-	114.3	(4.50)				-	
		83.8 (3.30)						71 ==1				-	
Y	591,039			0.1	7 01: -	(1. 76)	115.6	(4.55)	75.0	1 605			
5Hz	591.539		97.8 (	3.85)	104.1	(4.10)		-	15.2	(.60)	Name and Address of the Owner, where the Owner, which the		

Panel	No. 90		Primary Cra millimeters			Seconda millim	ry Crack Lengths eters (inches)
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1	#2
5Hz	592,339			104.1 (4.10		15.2 (.60)	21.6 ( .85)
	593,139				116.8 (4.60)		
	594,839	85.1 (3.35)			1	T	
	595,139				118.1 (4.65)		
	596,239						22.9 ( .90)
	597.239		99.1 (3.90)				
	599,239				119.4 (4.70)		
	600,439						24.1 (.95)
	601.339	86.4 (3.40)					148
	601,439				120.6 (4.75)		
	601,939		100.3 (3.95)				32
	603,139				121.9 (4.80)		
	604,039						25.4 (1.00)
	606.539				123.2 (4.85)		
	607.139				1		26.7 (1.05)
	609,639				124.5 (4.90)		
	610.739		101.6 (4.00)		-		
-	610,939 611,139	007 / 15 1.00			-	-	27.9 (1.10)
	611,139	87.6 (3.45)			1305 7 () 05		
	614,039				125.7 (4.95)		00 0 71 151
	616,439	00 0 /2 50	100 0 // 05		-		29.2 (1.15)
	617,139	00.9 (3.50)	102.9 (4.05)		1100 0 15 001		
-	617,339				127.0 (5.00)		
-	620,139				1200 0 (5 05)		30.5 (1.20)
-	620,639		701. 7 (1. 70)		128.3 (5.05)		
-	620,939	20 0 /= 5=1	104.1 (4.10)		+		
-		90.2 (3.55)					
-	624.139				129.5 (5.10)		23 9 (3.05)
-	624,339		705 1 (1. 75)		+		31.8 (1,25)
-	624.839	107 1 /0 (0)	105.4 (4.15)		+		
5Hz	625,139	91.4 (3.60)		104 1 14 10	130.8 (5.15)	15.2 ( .60)	

Panel 1	No. 9C		Primary Cra	(inches)		Secondary Crack Lengths millimeters (inches)			
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1	#2		
5 Hz	628,439		106.7 (4.20)	104.1 (4.10)		15.2 ( .60)			
4	628,539	92.7 (3.65)		1		1	33.0 (1.30)		
5 Hz	628,939			104.1 (4.10)	132.1 (5.20	)15.2 ( .60)	33.0 (1.30)		
FINAL	CRACK	-0.5 (0.65)	70( 7 (), 00)	701. 7 (1. 10)	120 1 (5 20)	15.2 ( 60)	33.0 (1.30)		
LEN	CTH	92.7 (3.65)	106.7 (4.20)	104.1 (4.10)	132.1 (5.20)	15.2 ( 60)	33.0 (1.30)		
RESID	VAL STREN	GTH = 286,900	N (64,500 1b	5)					
	+	-	-	1	-				

## PANEL #10C

MATERIALS: ALUMINUM - GLASS

ADHESIVE: AF 126

ALUMINUM STRESS: 138 MN/m<sup>2</sup> (20 ksi)

MAXIMUM FATIGUE LOAD: 100,080N (22,500 lbf)

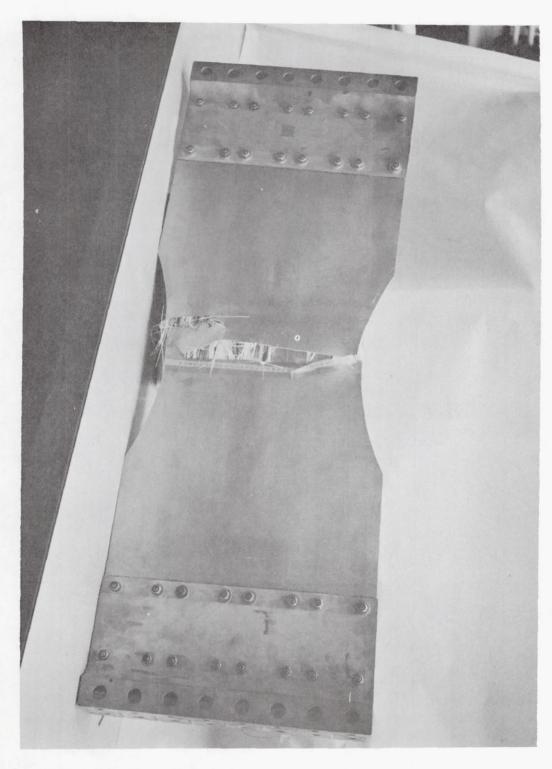


FIGURE F-35 OUTSIDE FACE SHEET AFTER RESIDUAL STRENGTH TEST - PANEL #10C

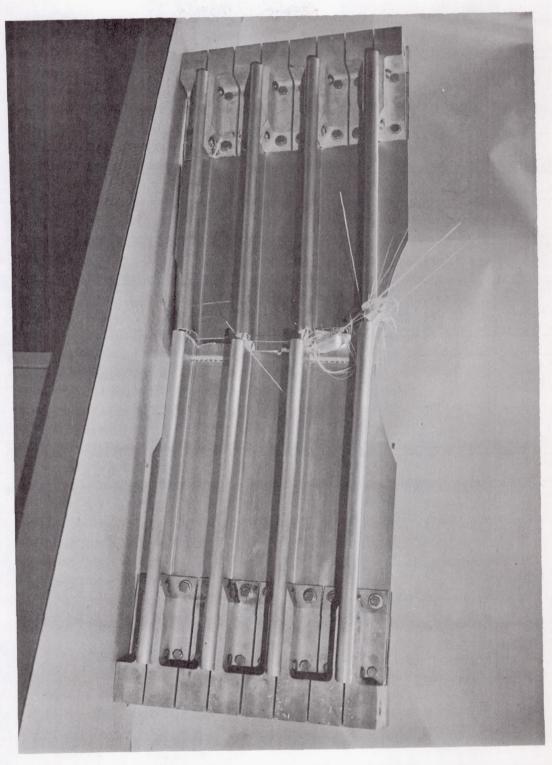


FIGURE F-36 INSIDE STIFFENER SHEET AFTER RESIDUAL STRENGTH TEST - PANEL #10C

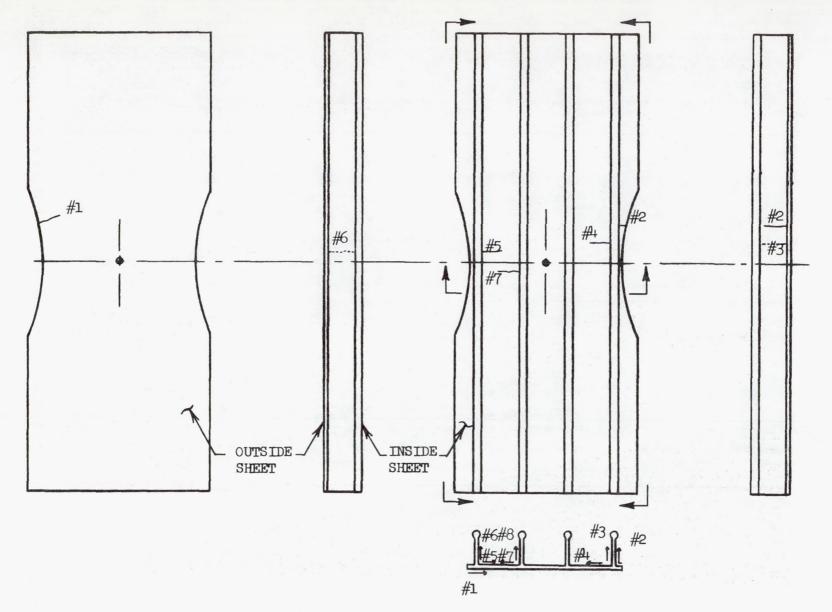


FIGURE F-37 SECONDARY CRACK LOCATIONS ALUMINUM-GLASS PANEL #10C

Panel	No. 10C		Primary Crac			Secondary Crack Lengths millimeters (inches)
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	
5.5Hz	10,800		3.2 (.125)			
	12,000	3.2 (.125)				
	12,600		5.1 ( .20)			
	15,700	5.1 ( .20)				
	15,900		6.4 ( .25)			
	16,000				3.2 (.125)	
Y	16,300			3.2 (.125)		
5.5Hz	18.800				5.1 ( .20)	
10Hz	22,000		7.6 ( .30)			
	22,300	6.4 ( .25)				
	22,800			5.1 (.20)		
	25,100				6.4 ( .25)	
	27,500	7.6 ( .30)				
	28,000			6.4 ( .25)		
	28,700		8.9 ( .35)			
	30,000				7.6 ( .30)	
	32,300	8.9 ( .35)				
	33,900		10.2 ( .40)			
	34,400			7.6 ( .30)		*
	35,100				8.9 ( .35)	
	140,1400			0 ( ==\	10.2 (.40)	
	40,700			8.9 ( .35)		
	41,500	10.2 ( .40)	11.4 ( .45)			
	46,100	11.4 ( .45)				
	46,500		12.7 ( .50)			
	47,200		-	10.2 (.40)	11.4 ( .45)	
	50,100	12.7 ( .50)	-		30.5 ( 50)	
	51,400				12.7 ( .50)	
	53,100		14.0 ( .55)			
	56.100	14.0 ( .55)				
	56,200	-	/ (a)	11.4 ( .45)		
10Hz	58,200		15.2 ( .60)			

Panel	No. 100		Primary Cra			Secondary Crack Lengths millimeters (inches)
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	
10Hz	59,600				14.0 ( .55)	
1	61,400			12.7 ( .50)		
T	61,700	15.2 ( .60)				
	64,400				15.2 ( .60)	
	65.100		16.5 ( .65)			
	67,000			14.0 ( .55)		
	68,500	16.5 ( .65)				
	70,800				16.5 ( .65)	
	71,100		17.8 ( .70)			
	74,200			15.2 ( .60)		
	74.800	17.8 ( .70)				
	76,500		19.0 ( .75)			
	77,000				17.8 ( .70)	
	80.800			16.5 ( .65)		
	80,900	19.0 ( .75)				
	82 400				19.0 ( .75)	
	85,900		20.3 ( .80)			
	86,100	20.3 ( .80)				
	87,200			17.8 ( .70)		
	88,400		21.6 ( .85)			
	88,800				20.3 ( .80)	
	94,100			19.0 ( .75)		
	94,200		22.9 ( .90)			
	94,500	21.6 ( .85)				
	94,900				21.6 ( .85)	
	100,200			20.3 ( .80)		
	100,400	22.9 ( .90)				
	100,800		24.1 ( .95)			
	101,000				22.9 ( .90)	
	106,000			21.6 ( .85)		
T.	106,400	24.1 ( .95)	25.4 (1.00)		-	
10Hz	107,400				24.1 ( .95)	

Panel	No. 10C		Primary Cra			Secondary Crack Lengths millimeters (inches)
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	
10Hz	112,200	25.4 (1.00)				
LOMB	112,300		26.7 (1.05)			
T	112,600			22.9 ( .90)		
	113,400				25.4 (1.00)	
	117 600	26.7 (1.05)				
	117,600			24.1 ( .95)		
	119,000				26.7 (1.05)	
	120,700	27.9 (1.10)	27.9 (1.10)			
	124,100		29.2 (1.15)			
	125,200			25.4 (1.00)		
	125,500				27.9 (1.10)	
		29.2 (1.15)				
	131,400			26.7 (1.05)		
	131,800				29.2 (1.15)	
	132,200		30.5 (1.20)			
		30.5 (1.20)				
	137,000			27.9 (1.10)	100 5 /1 00	
	137,500		1 2 2 7 0 7	-	30.5 (1.20)	
	140,600	31.8 (1.25)	31.8 (1.25)	(2.25)	-	
	143,100	-	-	29.2 (1.15)	102 0 (2 05)	
-	143,700		1 2 2 (2 25)	-	31.8 (1.25)	
	143,900	12 2 (2 55)	33.0 (1.30)	-	-	
-	147,800	33.0 (1.30)	-	00 5 (2 00)		
-	148,000	-	34.3 (1.35)	30.5 (1.20)	-	
	149,700	-	34.3 (1.32)	-	33.0 (1.30)	
-	150,000	-		31.8 (1.25)	33.0 (1.30)	
-	154,100	34.3 (1.35)		11.0 (1.55)		
-	156,100	24.3 (T.32)	-	-	34.3 (1.35)	
	158,653			33.0 (1.30)	DT. D (T. D)	
-	1160 143	35.6 (1.40)	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	33.0 (1.30)		
10Hz	160,453		35.6 (1.40)			

Panel	No. 100		Primary Cra millimeters			Secondary Crack Lengths millimeters (inches)
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	
10Hz	162,253				35.6 (1.40)	
4	163,053			34.3 (1.45)		
	165,053	36.8 (1.45)	36.8 (1.45)			
	170,303		38.1 (1.50)			
	170,653	38.1 (1.50)				
	170,753			35.6 (1.40)		
	176,953	39.4 (1.55)				
	177.753		39.4 (1.55)			
	178.453			38.1 (1.50)	40.6 (1.60)	
	178.853	40.6 (1.60) 41.9 (1.65)	40.6 (1.60)			
	182,753	41.9 (1.65)				
	186,553		41.9 (1.65)			
	188,953	43.2 (1.70)				
	190,353			39.4 (1.55)		
	194,153		43.2 (1.70)			
Ť	194,653	44.4 (1.75)				
10 Hz	196.353			40.6 (1.60)		
8Hz	199,403				41.9 (1.65)	
8Hz	204,603			41.9 (1.65)	143.2 (1.70)	
8Hz	207,309	47.0 (1.85)	45.7 (1.80)	43.2 (1.70)	44.4 (1.75)	
10Hz	213,709			44.4 (1.75)		
1	214.309		47.0 (1.85)			
	216,309	50.8 (2.00)			45.7 (1.80)	
	218,309		49.5 (1.95)	45.7 (1.80)		
	219,409				47.0 (1.85)	
	220,309	52.1 (2.05)				
	223,309		50.8 (2.00)	47.0 (1.85)	48.3 (1.90)	
	224,309	53.3 (2.10)				
	225,309		52.1 (2.05)			
	226,209			48.3 (1.90)	49.5 (1.95)	
Ý	227,909	54.6 (2.15)	50.0.70.35		-	
10Hz	229,209		53.3 (2.10)			

Panel	No. 10C		Primary Cra				Crack Lengths
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	-	
10Hz	231,109				50.8 (2.00)		
LOUZ	233,809		54.6 (2.15)				
T	234,509	55.9 (2.20)					
1	234,609	77.7		49.5 (1.95)			
	236,609			1207	52.1 (2.05)		
	237, 309	57.2 (2.25)			7=-1=-071		
	238,009		55.9 (2.20)				
	238,809			50.8 (2.00)			
	240, 309		57.2 (2.25)	10.00			
	240,409				53.3 (2.10)		3
	240,709	58.4 (2.30)					
	244,409		58.4 (2.30)				
	244,709			52.1 (2.05)			76
	244,909	59.7 (2.35)					· ·
	245,309				54.6 (2.15)	A THE STATE OF THE	100
	247,909			53.3 (2.10)			
			59.7 (2.35)				
	248,909	61.0 (2.40)					
	249,109				55.9 (2.20)		
	251,609			54.6 (2.15)			
	251,909		61.0 (2.40)				
	252,309	62.2 (2.45)					
	252,909				57.2 (2.25)		
	254,109	63.5 (2.50)					
	255,809		62.2 (2.45)				
-	256,409			55.9 (2.20)			
-	258,209				58.4 (2.30)		
	258,409	64.8 (2.55)					
-	259,809		63.5 (2.50)				
1	260.509	66.0 (2.60)					
OT	260,909			57.2 (2.25)	F0 7 /0 05		
OHz	102,109				59.7 (2.35)		

Panel	No.10C		Primary Cra				ry Crack Leng	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1	#2	#3
10Hz	263,809 264,909			58.4 (2.30)				
1	264,909		64.8 (2.55)					
	266,209		04.0 (2.)		61.0 (2.40)			
	266,209	67.3 (2.65)						
	267,709	01.5 (2.0)		59.7 (2.35)				
	268.109		66.0 (2.60)					
	269.209	68.6 (2.70)						
	269,509		67.3 (2.65)					
	269,909				62.2 (2.45)			
*	271.009	69.8 (2.75)						
10Hz	273,009			61.0 (2.40)				
5Hz	273,189					29.2 (1.15)	25.4 (1.00)	35.6 (1.40)
4	274.109		68.6 (2.70)					
	274,409	71.1 (2.80)						
	275,909				63.5 (2.50)			
	276,609			62.2 (2.45)				
	278.009		69.8 (2.75)					
	280.009						-	
	280,309			63.5 (2.50)	10 0 10 55			
	280,709				64.8 (2.55)	22 0 (2 22)	1 - (2 - 20)	
	281,709		71.1 (2.80)		-	31.8 (1.25)	27.9 (1.10)	
	282,109	73.7 (2.90)			-	-		
	283,709			64.8 (2.55)	10000			
	284,109				66.0 (2.60)			
	284,409	74.9 (2.95)				-		
	284,809		72.4 (2.85)					
	285,309		-			-	-	20 7 (2 50)
	287,309		== = (= ==>			-		38.1 (1.50)
	287,609	76.2 (3.00)	73.7 (2.90)	(( 0 (0 (0)		-		
1	287.809		-	66.0 (2.60)	-	34.3 (1.35)		-
-	289,609		-	-	67.3 (2.65)	34.3 (1.33)		
5Hz	289,709				101.3 (2.03)			

Panel	No.100			Secondary millimet	Crack Lengters (inches	ths )		
Cycle Rate	No. of Cycles	#4	#5	#6				
10Hz	263,809							
4	264,909							
	266,209							
	266,609							
	267,709							
	268.109							
	269.209							
	269,509							
	269,909							
*	271.009							
10Hz	273.009					1		
5Hz	273.189	35.6 (1.40)	40.6 (1.60)	41.9 (1.65)				
4	274,109			Crack				
	274,409 275,909 276,609			Stopped				
	275,909			in Bulb				
	276,609			Radius				
	278,009			1				
	280,009	38.1 (1.50)						
	280,309					-		
	280,709							
	281,709 282,109		43.2 (1.70)					
	282,109							
	283,709							
	284,109					1		
	284,409							
	284,809					-		
	285,309	40.6 (1.60)			-	-		-
	287,309							
	287,609				-	-		-
	287,809 289,609					-		-
Ÿ	289,609			11 0 13 65	-	-	-	-
5Hz	289,709			41.9 (1.65				

Panel	No. 10C		Primary Cra				ry Crack Leng		
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1	#2	#3	
5Hz	289,809		74.9 (2.95)						
1	290,109		1.02 (-0)21				30.5 (1.20)		
	290,209								
	290.309								
	292,009	77.5 (3.05)							
	292,209	11.0		67.3 (2.65)					
	294,209		76.2 (3.00)						
	294.909				68.6 (2.70)				
	296.509			68.6 (2.70)					
	296,509 296,709		77.5 (3.05)						
	296,909					36.8 (1.45)		40.6	1.60)
	297,309						33.0 (1.30)		
	299,309	78.7 (3.10)			69.8 (2.75)				
	300,009		78.7 (3.10)						
	300,309							41.9 (	(1.65)
	301,309	80.0 (3.15)	80.0 (3.15)					Crack	
	301,909			69.8 (2.75)				Stoppe	ed
•	303,409							in Bul	
	303,709	81.3 (3.20)						Radius	3
	304,109				71.1 (2.80)				4
	304,309		81.3 (3.20)						
	304,609	-Overlaps #1				39.4 (1.55)			
	305,309	Secondary				Overlaps L/H			
		Crack		71.1 (2.80)		Primary			
	307,309	82.6 (3.25)				Crack			
	307,509		82.6 (3.25)						
	309,309								
	309,909				72.4 (2.85)				
	310,209		83.8 (3.30)						
		83.8 (3.30)							
*	311,309					41.9 (1.65)	35.6 (1.40)	15	7
5Hz	311,500			72.4 (2.85)				41.9	(1.65)

Panel No. 100		Secondary Crack Lengths millimeters (inches)								
Cycle Rate	No. of Cycles	#24	#5	#6						
5Hz	289,809			41.0	(1.65)					
	290,109			1.1.0	(1.0)		+		-	
T	290,209	43.2 (1.70)					+		-	
	290,309		45.7 (1.80)							
	292,009		1200/							
	202 200						1			
	292,209 294,209									
	294.909								-	
	296.509									
	296,709									7
	296,909		48.3 (1.90)							
	297,309									
	299.309	45.7 (1.80)								-
	300,009		7 4 4							
	300,309 301,309									
	301,309									
-	301.909									
-	303,409		50.8 (2.00)							
-	303,709 304,109									
-	304,109									
-	304,309									
-	304,609	10 - 1-								
-	305,309	48.3 (1.90)								
-	307,009 307,309									
-	307,309									
-	307,509									
-	309,309		53.3 (2.10)							
-	309,909									
-	310,209									
	310,709									
5Hz	311,309			117 0	(1.65)		-			

Panel No. 100			Primary Cra		Secondary Crack Lengths millimeters (inches)				
Cycle R <b>at</b> e	No. of Cycles 312,409	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1	#2	#3	
5Hz								41.9	(1.65)
SHZ	312,609				1	<del>                                     </del>		-	1
1	315,309				73.7 (2.90)			1	1
-	315,709				13.1 (2.50)			-	-
_	216, 100	86.4 (3.40)	86.4 (3.40)		<del>                                     </del>		-	1	
	316,109 316,309	33.107	33.1.7						
	316,809				1				1
	317,309					44.4 (1.75)			
	317.809		and the same of						
	317,909		87.6 (3.45)						
	318.009				81.3 (3.20)				
	318,109			81.3 (3.20)					
	318,309				82.6 (3.25)				
	318,509 318,609			82.6 (3.25)					
	318,609	87.6 (3.45)							
	318,709				83.8 (3.30)				
	318,909			83.8 (3.30)			1		
	319,009				85.1 (3.35)			-	-
	319,209				86.4 (3.40)			-	
	319,309			85.1 (3.35)	0= ( /- 1=)				
	319,709			86.4 (3.40)	87.6 (3.45)				
	320,209			87.6 (3.45)	88.9 (3.50)				
	320,509			22	90.2 (3.55)				
_	320,609			88,9 (3.50)	-			-	
	320,709		88.9 (3.50)				00 1 71 50	-	
	321,009			22 2 (2 55)	10 (0)		38.1 (1.50)	-	
	321,109			90.2 (3.55)	91.4 (3.60)			-	-
-	321,409			07 1 (0 (0)	92.7 (3.65)			-	
-	321,609			91.4 (3.60)	10 (0.50)			-	-
-	322,009			92.7 (3.65)	94.0 (3.70)			-	-
5Hz	322,309			96.1 (3.05)	95.2 (3.75)			111 0	(1.65)

Panel	No. 100	Secondary Crack Lengths millimeters (inches)								
Cycle Rate	No. of Cycles 312,409	#4	#5	#6		#7	#8			
				41.9	(1.65)		10.2 ( .40)		+	
1	312,609	50.8 (2.00)				1	1 2002 ( 0.10)		-	
T	315,309	24.0 (2.00)				<del>                                     </del>	1		+	
1	315,709		55.9 (2.20)		-				+	
	316,109		11.9 (2.20)		-	+	1		+	
	316,309			1	-	6.4 ( .25)	+		+	
	316,809				+	10.11.2)	11.4 ( .45)		+	
	317,309						11.4 ( .4)		1	
	317,809					7.6 (.30)			+	
	317,909					1.0 ( .50)	1			
	318,009								+	
	318,109								+	
	318,109 318,309								1	
	318,509								1	
	318,609								1	
	318.709								1	
	318.909									
	319.009	53.3 (2.10)								
	319.209									
	319.309									
	319.709									
	320,209									
	320,509								1	
	320,609								1	
	320,709									
	321,009									
	321,109									
	321,409			3 3						
	321,609									
	322,009									
	322,309				1					
Hz	322,509			41.9	(1.65)					

Panel	No. 100		Primary Cra				y Crack Letters (inch		
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1	#2	#3	
5Hz	322,709	88.9 (3.50)						41.9	(1.65)
)III	322,909	00.9 (3.20)		94.0 (3.70)					
1	323,009			13.1.7	96.5 (3.80)				
-	323,083			95.2 (3.75)		47.0 (1.85)			
	324.083			96.5 (3.80)					
	324,983			97.8 (3.85)					
	325.083								
	325.183				97.8 (3.85)				
	325.583	90.2 (3.55)							
	325,683			99.1 (3.90)					
	326.333		91.4 (3.60)						
	326, 383				100.3 (3.95)				
	326,783			101.6 (4.00)	101.6 (4.00)				
	327.083			102.9 (4.05)					
	327,273		92.7 (3.65)	Overlaps #8					
	327,273 327,283			Secondary	102.9 (4.05)				
	1327.683	91.4 (3.60)		Crack					
	327,583 327,883 328,183			104.1 (4.10)					
	327,883				104.1 (4.10)				
	328,183			105.4 (4.15)					
	328,383								-
	328,683								
	328,883				105.4 (4.15)				
	329.083		94.0 (3.70)						
	329.383				106.7 (4.20)			_	
	320 583							-	
	330.183	92.7 (3.65)						-	
	330.383			106.7 (4.20)	108.0 (4.25)			-	-
	330,483				-	49.5 (1.95)		-	
	331,483			-	100 0 //- 00			-	-
-	331,683				109.2 (4.30)	1		117.0	(1.65
5Hz	332,483				110.5 (4.35)			41.9	(1.0)

Panel	No. 100			Se m	condary	Crack Lengthers (inches)	8	
Cycle Rate	No. of Cycles	#4	#5	#6		#7	#8	
5Hz	322,709			47 0	(1.65)			
1	322,909			71.9	11.0)/			 
1								 
	323.083	55.9 (2.20)				12.7 ( .50		 
	324,083							
	324.983							
	325.083		61.0 (2.40)		-	- Overlaps		
	325.183		Overlaps			#5		12 12 12 1
	325,583 325,683 326,333		#7			Secondary		
	325.683		Secondary			Crack		4
	326,333		Crack					
	1326.383							
	326,783 327,083							
	327,083					-	-Overlaps	the second
-	1327.273						L/H Primary	
	327,283						Crack	
1.	327,683 327,583							
	327,583							
-	327,883							
-	328,183							
	328,383		-			14.0 ( .55)		
-	328,683						17.8 ( .70)	
	328,883							
-	329,083		-					
-	329,383							
-	329,583		63.5 (2.50)					
-	330.183	150 1. (2. 22)						
-	330,383	58.4 (2.30)						
+	330,483 331,483							A
-	131,483					15.2 ( .60)	20.3 ( .80)	
5Hz	331,683		-	1.2	(1.65)			

Panel 1	No.10C		Primary Cra millimeters			Seconda:	ry Crack Leng	ths )
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1	#2	#3
5Hz	332,683		95.2 (3.75)					41.9 (1.65)
	333,283				111.8 (4.40)			
	333,483			108.0 (4.25)				
	333,783		96.5 (3.80)					
	334.283				113.0 (4.45)			
	334,383 334,583	94.0 (3.70)						
	334,583			109.2 (4.30)	114.3 (4.50)			
	335,183		97.8 (3.85)					
	335.883	95.2 (3.75)			118.1 (4.65)			
	336,083			770 5 (), 25)				
	336.383		99.1 (3.90)	110.5 (4.35)				
	337,283	96.5 (3.80)	99.1 (3.90)		119.4 (4.70)			
	338,583	90.7 (3.00)	100.3 (3.95)		119.4 (4.10)			
	330,003		100.5 (5.97)	111.8 (4.40)				
	340,083 340,583		101.6 (4.00)	1110				
	341.183		101.0 (1.007		120.6 (4.75)			*
5Hz	341,983	97.8 (3.85)	102.9 (4.05)					41.9 (1.65)
FINAL	CRACK							
LE	IGTH	97.8 (3.85)	102.9 (4.05)	111.8 (4.40)	120.6 (4.75)	49.5 (1.95)	31.8 (1.50)	41.9 (1.65
	DESTRIAT	STRENGTH =	276,900 N	(62,250 lbs				
	NISOLDUAL		210,700 41	(32,12)0 220				

Panel	No. 10C			Se	condary millimet	Crack Length cers (inches)	ns		
Cycle Rate	No. of Cycles	# <del>P</del> 4	#5	#6		#7	#8		
5Hz	332,683 333,283 333,483 333,783 334,283 334,283 334,583			41.9	(1.65)				
1	333,283								
	333,483								
	333,783								
	331: 283								
	3314 383								
	334,583								
-	335,183								
-	335.882								
-	335,183 335,882 336,083 336,389						-		
-	336,383					-			
-	337,283						-		
-	337,283 338,583 338,883 340,083 340,583 341,183								-
-	350.003					<del> </del>	1		
-	340,003					<del> </del>			-
-	3/11 182			-					
5Hz :-	341.983			41.9	(1.65)				
	341,303								
FTNAT.	CRACK								
LEN	CTHS	58.4 (2.30)	63.5 (2.50)	41.9	(1.65)	15.2 ( .60)	20.3 ( .80)		
			1-27-						
	-					-		-	
	-								-
	-			-					-
	+								

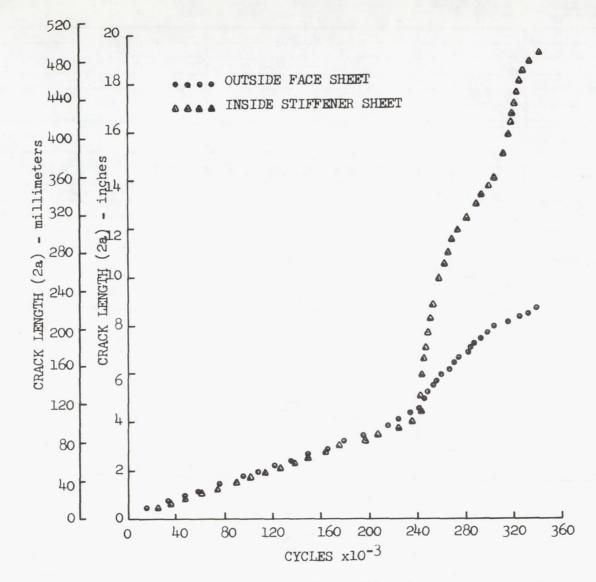


FIGURE F-38 CRACK GROWTH CURVE FOR ALUMINUM-GLASS PANEL #10C (Crack Length Includes Secondary Cracks)

## PANEL #11C

MATERIALS: ALUMINUM - GLASS

ADHESIVE: AF 126

ALUMINUM STRESS: 172 MN/m<sup>2</sup> (25 ksi)

MAXIMUM FATIGUE LOAD: 124,540N (28,000 1bf)

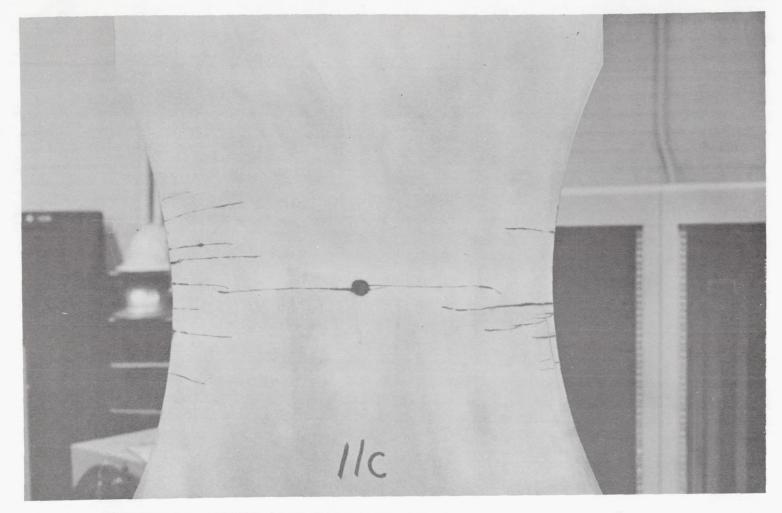


FIGURE F-39 OUTSIDE FACE SHEET AFTER CRACK GROWTH TEST - PANEL #11C

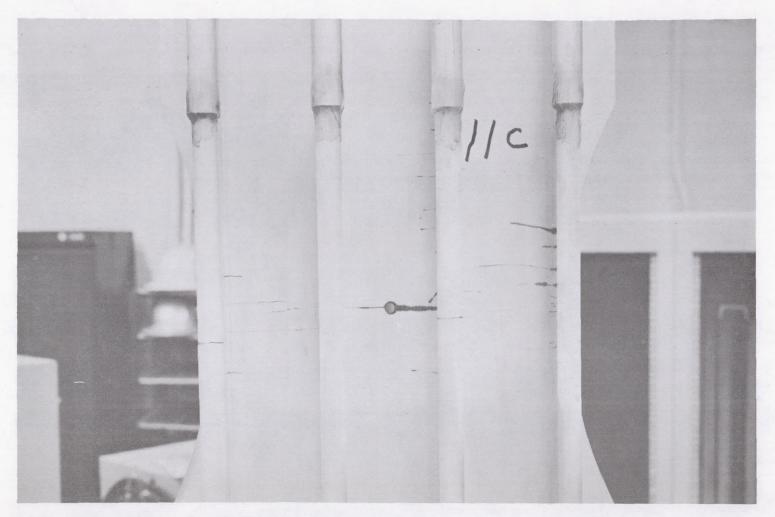


FIGURE F-40 INSIDE STIFFENER SHEET AFTER CRACK GROWTH TEST - PANEL #11C

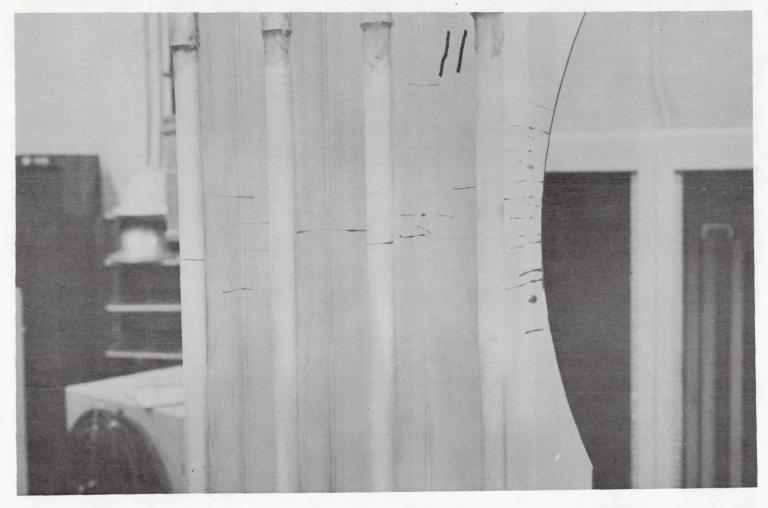


FIGURE F-41 INSIDE STIFFENER SHEET AFTER CRACK GROWTH TEST - PANEL #11C (RIGHT OF CENTERLINE)

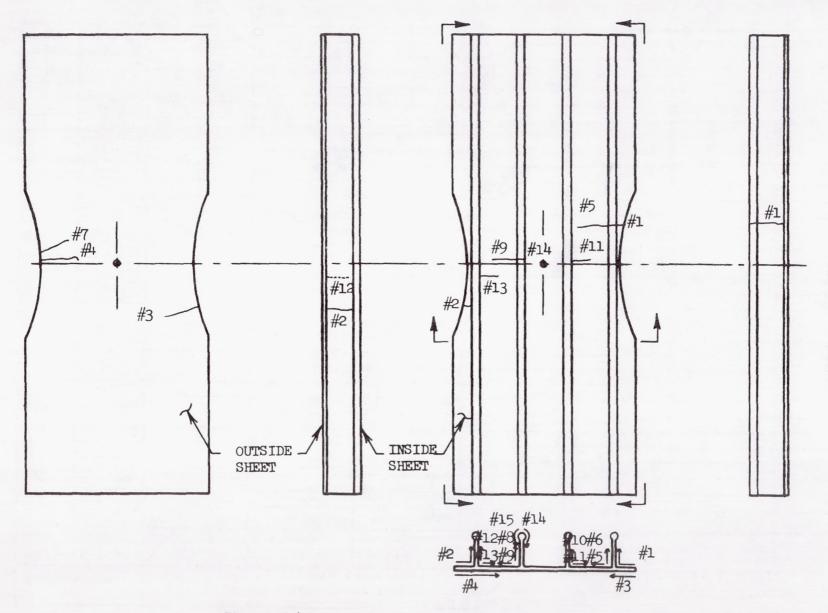


FIGURE F-42 SECONDARY CRACK LOCATIONS ALUMINUM-GLASS PANEL #11C

Panel	No.11C		Primary Crac	ck Lengths (inches)		Secondary millimet	Crack Lengths
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside		
9 Hz	13,200			3.2 (.125)			
9	14,700			5.1 ( .20)			
-	15,500	6.4 ( .25)					
-	16,500		6.4 ( .25)				
1	17,000			6.4 ( .25)			
	18,900	7.6 ( .30)					
	19.500		7.6 ( .30)				
	19,800			7.6 ( .30)			
	20,100				5.1 (.20)		
	21,200				6.4 (.25)	-	
	21,400	8.9 ( .35)					
	21,600		8.9 ( .35)	0 ( 0 = 1			
	22,000			8.9 ( .35)			
	22,600	10.2 ( .40)			- ( ( )		
	22.800 24,400		20 0 7 101		7.6 ( .30)		
	24,400		10.2 ( .40)	10.2 (.40)			
	24,900			10.2 ( .40)	8.9 ( .35)		
	25,600	1 1 ( 1.5)			0.9 ( .)		•
	26,300	11.4 ( .45)		11.4 ( .45)			
	27,100		22 1 ( 1.5)	11.4 ( .45)			
-	27,900	1 1	11.4 ( .45)				
	28,300	12.7 ( .50)			10.2 ( .40)		
	28,400		5 5 ( 50)		10.2 ( 40)		
	29,400		12.7 ( .50)	10.7 / 50)			
	30,200	14.0 ( .55)	-	12.7 ( .50)			
-	31,300	14.0 ( .))	14.0 ( .55)		11.4 ( .45)		
-	32,300	-	17.0 ( .)))	14.0 ( .55)			
-	33,400	-	-	-1.0	12.7 ( .50)		
-	34,100		15.2 ( .60)				
-	35,200		1		A 2		
9 HZ	35,400		1	15 2 ( 60)			

Pan	el 1	No. 11C		Primary Cra			Secondary Crack Lengths millimeters (inches)
Cyc. Rat		No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	
9	Hz	36,800	16.5 ( .65)				
		37,600				14.0 ( .55)	
7		38,100			16.5 ( .65)		
		38,400		16.5 ( .65)			
		39,400	17.8 ( .70)				
		40,000			17.8 ( .70)	15.2 ( .60)	
		41.400		17.8 ( .70)			
		42.500	19.0 ( .75)				
		42.700				16.5 ( .65)	
		43.000			19.0 ( .75)		
		45,300			20.3 ( 80)		
		45,800				17.8 ( .70)	
		46.000	20.3 ( .80)	19.0 ( .75)			
		47.000				19.0 ( .75)	
		47.200			21.6 ( .85)		
		48,400	21.6 ( .85)				
		48,800		20.3 ( .80)			
•		49,500			22.9 ( .90)		
		50,000		21.6 ( .85)			
		50,300				20.3 ( .80)	
		50,800	22.9 ( .90)				
		51,100		22.9 ( .90)			
		52,900				21.6 ( .85)	
		53,200	24.1 ( .95)				
		54,200		24.1 ( .95)			
_	-	55,400			25.4 (1.00)		
		56,100				22.9 ( .90)	
	_	57,100		25.4 (1.00)			
		57,500	25.4 (1.00)				
	_	58,600			26.7 (1.05)		
	-	59,000 59,400		26.7 (1.05)		01: 1 / 05	
9 ]	Hz	59,400				24.1 ( .95)	

THE RESERVE OF THE PERSON NAMED IN	No. 11C		millimeters	ck Lengths (inches)		Secondary Crack Lengths millimeters (inches)
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	
9 Hz	59,900	26.7 (1.05)				
1	60,600		27.9 (1.10)			
T	60,700			27.9 (1.10)		
-	61 200				25.4 (1.00)	
1	61,200 61,500	27.9 (1.10)				
	63,200		29.2 (1.15)			
	63,600				26.7 (1.05)	
	64,000			29.2 (1.15)		
	65,600		30.5 (1.20)			
	66.000	29.2 (1.15)				
	66,100			30.5 (1.20)	07 0 /1 10	
	66,900				27.9 (1.10)	
	68,300	30.5 (1.20)	31.8 (1.25)	0 (2 05)		
	68,500			31.8 (1.25)	00 0 /3 35	
	69,800				29.2 (1.15)	
Y	71,300		33.0 (1.30)	(2.25)	00 5 (3 00)	
9 H				33.0 (1.30)	30.5 (1.20)	
5 H		31.8 (1.25)	120 (2.05)			
-	74,991		34.3 (1.35)	0 0 (3 05)	22 9 /2 05	
	75,491			34.3 (1.35)	31.8 (1.25)	
	77,091		35.6 (1.40)			
	77,291	33.0 (1.30)				
	80.591	34.3 (1.35)				
	81,091 81,991	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1		35.6 (1.40)	33.0 (1.30)	
	81,991	35.6 (1.40)	1 - ( 0 ( - 1 - 1			
	82,091		36.8 (1.45)		34.3 (1.35)	
	82,391	-	-	56.0 (5.1.5)	34.3 (1.35)	
Y	83,391		0 - 1 - 1	36.8 (1.45)		
5 H	1 0 0 971	1 0 ( 1 1 -1	38.1 (1.50)			
6 H	z 85,491	36.8 (1.45)	-		35.6 (1.40)	
6 H	85,691 z 87,791	-		38.1 (1.50)	37.0 (1.40)	

Panel	No.11C		Primary Cra				ry Crack Lengths eters (inches)
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1	
6 Hz	87,891		39.4 (1.55)				
1	89,991				36.8 (1.45)		
1	90.291	38.1 (1.50)			10.0 (1.4)		
	90.991	1001	40.6 (1.60)				
	92,491	39.4 (1.55)	1000				
	93 691	37.1		39.4 (1.55)			
	93,691 93,791				38.1 (1.50)		
	94,691		41.9 (1.65)				
	95,291	40.6 (1.60)					
	97,091			40.6 (1.60)			
	97,391		43.2 (1.70)				
	98,691				39.4 (1.55)		
-	98,891	41.9 (1.65)					
-	100,991		44.4 (1.75)				
-	101.891	43.2 (1.70)					
-	102,391			1.2 6 (2 (5)	40.6 (1.60)		
	102,991		hr 7 (1 00)	41.9 (1.65)			
-	104,991		45.7 (1.80)	10 0 (7 70)			
+-	106,291	44.4 (1.75)		43.2 (1.70)			
-	107,491	77.7 (1.12)			41.9 (1.65)		
	108,691		47.0 (1.85)	44 4 (1 75)	71.9 (1.0)		
1	111,091		11.0	11.17	43.2 (1.70)		
1	111,191			45.7 (1.80)	13.2 (1.10)		
6 Hz		45.7 (1.80)					
5 Hz	112.626		48.3 (1.90)				
1	115,626	47.0 (1.85)			44.4 (1.75)		
	116,626			47.0 (1.85)			
	1117,226		49.5 (1.95)				
	119.226					1.3 (.05)	
-1	119,626		50.8 (2.00)				
5 Hz	120,126				45.7 (1.80)		

Panel	No. 11C		Primary Cra millimeters	ck Lengths (inches)			ry Crack Leng eters (inches	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1	#2	#3
5 Hz	121,426							10.2 ( .40)
) 112	121,626						1.3 ( .05)	
1	122,526			48.3 (1.90)				
-	122,726	49.5 (1.95)						
	123,326	,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,	52.1 (2.05)					
-	123,726				47.0 (1.85)			
	126,226	50.8 (2.00)						12.7 ( .50
	126,326			49.5 (1.95)				
	126.626		53.3 (2.10)					
	127.126						2.5 ( .10)	
	128,226							
	129,226				48.3 (1.90)			
	129,926	52.1 (2.05)	54.6 (2.15)					-
	131.526			50.8 (2.00)	49.5 (1.95)			
	131,526	53.3 (2.10)						
	135,026		55.9 (2.20)					-
	135.826			52.1 (2.05)		25 ( /2 1.0)		
	136,226 136,326					35.6 (1.40)		-
	136,326		57.2 (2.25)	-				-
	136,426	54.6 (2.15)		-	50 0 (0 00)		5.1 ( .20)	-
	136,826			-	50.8 (2.00)		5.1 ( .20)	17.8 ( .70
	138,526			-				17.01.10
	139,926	55.9 (2.20)	58.4 (2.30)	-	7 7 (0 05)			
	140,226			1	52.1 (2.05)			-
	140,626			53.3 (2.10)				-
	140,726	-		-	-	38.1 (1.50)		1
	142,226		(0 0=)	-		0.1 (1.)0)		
	143,026	57.2 (2.25)	59.7 (2.35)	-	-			
	143,5% 143.7%	-		+	-			20.3 ( .80
	143.726	-	-	-	-			
.5 Hz	143,826 144.026		-	-	53.3 (2.10)			

Panel	No. 11C			Second milli	ary Crack Lengmeters (inches	gths s)		
Cycle Rate	No. of Cycles	<i>#</i> <del>1</del> 4						•
5 <b>Hz</b>	121,426							
1	121,626							
	122,526							
_	122.726							
	123.326							
	123 726							
	126.226							
	126,226 126,626 127,126 128,226		and the same of					
	126,626							
	127,126							
	128,226	10.2 ( .40)						
	129,226							
	129,926							
	131.526							
	134.226							
	135,026							
	135.826							-
	136,226						-	
	136,326						-	
	136,426				-		+	-
-	136.826	20.3 ( .80)		+			+	-
	138,5%						-	
	139,926							
	140,226						-	
	140,626							
	140.726	22.9 ( .90)					-	-
-	142,226					-	-	
-	143,026	65 1 /6			-	-	-	-
-	143,526	25.4 (1.00)			-	-	-	-
-	143,726			-			+	-
5 Hz	144,026						-	-

Panel	No.11C		Primary Cra				ry Crack Leng eters (inches	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1	#2	#3
5 Hz	144,926							
	145,126	58.4 (2.30)	61.0 (2.40)					
1	146,426						7.6 ( .30)	
	146,826							
	147.326	59.7 (2.35)	62.2 (2.45)					
	148.426							
	149.026				54.6 (2.15)			
		61.0 (2.40)	63.5 (2.50)					
	151 026							22.9 ( .90)
	151,026							
	151,726					40.6 (1.60)		
	152,426						10.2 ( .40)	
	152,526		64.8 (2.55)					
	153.826	62.2 (2.45)						
	154,826		66.0 (2.60)					
	155.126							
	155.226							
	155,426			55.9 (2.20)				
	155 826							
	156,926	63.5 (2.50)			55.9 (2.20)			
	157,626		67.3 (2.65)					
	157,926							
	158.326	64.8 (2.55)						
	158.526							25.4 (1.00)
	158,626				57.2 (2.25)			
				57.2 (2.25)				
	159,326							
	160,326							
	160,626							
	161.026	-	68.9 (2.70)	-				
-	161,226		(- 0 ()		-			
5 Hz	1162,826	66.0 (2.60)	169.8 (2.75)					

Panel	No. 11C			Secondary millimet	Crack Length ers (inches)	8		
Cycle Rate	No. of Cycles	#4	#5	#6	#8	#9		
5 Hz	144,926 145,126		7.6 ( .30)					
1	145,126					(-)		
7	146,426							
+	146,826	27.9 (1.10)					- E	
-	147.326	21.9 (1.10)						
+	148,426		10.2 (.40)		The second			
	140 026							
	149,026							
	151 026							
	151,126	33.0 (1.30)						
	151.726		12.7 ( .50)					-
	152,426							
	152.526							-
	153.826	35.6 (1.40)						-
	154.8%							-
	155,126	35.6 (1.40)						-
	155,2% 155,4% 155,8%			7.6 ( .30)				+
	155,426				-			-
	155,826		17.8 ( .70)					-
	1156,926		-		-			+
	157,626		-		<b>—</b>			+
	157,926 158,326				10.2 (.40)			+
	158.326				-			+
	158,526							+
	158,526 158,626				-			+
	1159.126				133 11 / 115	-		-
	159,326				11.4 ( .45)	36 5 1 (5)		+
	160,326					16.5 ( .65)		+
	160,626 161,026	38.1 (1.50)			-			+
	161,026		4 0 1		-			+
_	1161,226	140.6 (1.60)	20.3 (80)	10.2 ( .40)	-	-		-
5 Hz	162,826							

Panel	No. 11C		Primary Crac			ry Crack Leng		
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1	#2	#3
5 Hz	163,426	67.3 (2.65)						
	163,626	01 (2.0)						
T	163,826							27.9 (1.10)
-	163,926				58.4 (2.30)			
	164,026				70.1 (2)01			
	164,226							
_	164.726			58.4 (2.30)				
	164.826					45.7 (1.80)		
_	165 226						12.7 ( .50)	
1	165,226 165,326							
	165.426	68.6 (2.70)	71.1 (2.80)					
	165,726							
	166,126							
	167,926	69.8 (2.75)	72.4 (2.85)					
	168,226							
	168,426				59.7 (2.35)			
	168,426 168,726					48.3 (1.90)		
	169,026		73.7 (2.90)			Crack		
	169.326			59.7 (2.35)		Stopped		
	169.426					in Bulb		
	169.726					Radius		
	169,726		74.9 (2.95)			A		
	170,526	71.1 (2.80)						
	170.826							30.5 (1.20)
	171,326				61.0 (2.40)			
	172.726							
	172,726					-	-	
	173,026	72.4 (2.85)	76.2 (3.00)					22 0 11 201
	173,526	Overlap #4				-	-	33.0 (1.30)
	173.726	Secondary	77.5 (3.05)					
1	174.026	Crack		(1.0.70.10)		11.0 2 (2 60)	-	
5 Hz	174.126			61.0 (2.40)		[48.3 (1.90)		

Panel	No. 11C	Secondary Crack Lengths millimeters (inches)									
Cycle Rate	No. of Cycles	#4	#5	#6	#8	#9	#10	#1.1			
5 Hz	163,426										
-	163,626					19.0 ( .75)		+			
	163.826	43.2 (1.70)				129.0 ( . 1)					
	163,826 163,926	130-1-1				<b>†</b>		-			
	164,026				12.7 ( .50)			1			
	164.226		22.9 ( .90)								
	164.726										
	164,726 164,826										
	165,226							A. C. C. C.			
	165,326					21.6 ( .85)					
	165,426										
	165,726				14.0 ( .55)						
	166,126	45.7 (1.80)									
-	167.926										
-	168,226 168,426		25.4 (1.00)	12.7 ( .50)							
-	168,426										
-	168,726 169,026					-		-			
-	1169,026		-								
-	169,326	-				06 7 (3 05)		-			
-	169,426				15.2 ( .60)	26.7 (1.05)		-			
-	169,726				17.2 ( .60)	-					
-	170,226 170,526 170,826					-		-			
-	170.83	48.3 (1.90)	27.9 (1.10)		-						
	171,326	10.5 (2.)0)	2.1.9 (1.10)			-		-			
-	172,726					-		6.4 ( .25			
	172.826						6.4 ( .25)	0.7 ( .2)			
	173 026	Overlaps					0.1 (.2)	1			
	173.526	L/H Primary	30.5 (1.20)					-			
1	173,726	Crack	10.7 (1.20)					1			
1	174.026					31.8 (1. 25)		<del>                                     </del>			
Hz	174 126					1					

Panel	No. 11C		Secondary Crack Lengths millimeters (inches)								
Cycle Rate	No. of Cycles	#12	#12 #13								
5 Hz	163,426										
A	163,626										
	163.826							1			
	163,926					1	1	1			
	163,926 164,026										
	164.226							1			
	164.726							1			
	164,726 164,826 165,226							1			
	165,226										
	165.326										
	165,426 165,726						1				
	165.726										
	166.126										
	166,126 167,926 168,226										
	168,226										
	168.426										
	168.726										
	169.026										
	169.326										
	169.426										
	169.726										
	170,226										
	170 EX						1				
	170.826							1			
	171.326										
	170,8% 171,3% 172,7% 172,8% 172,8% 173,5%		6.4 ( .25)								
	172.826	6.4 ( .25)					T				
	173 026										
	173,526										
	13,120			112			T				
	174.026				49-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1						
Hz	174.126							T			

Panel	No. 11C		Primary Crack Lengths millimeters (inches)					Secondary Crack Lengths millimeters (inches)			
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1		#2	#3		
5 Hz	175,626		78.7 (3.10)			48.3 (1	.90)				
A	176,126	74.9 (2.95)	(3								
7	176,426	1.17							35.6 (1.40		
	176,626								12.0 (2.10		
	176 726										
	176,726 176,826										
	177.526										
	177,626				62.2 (2.45)						
	177.726		80.0 (3.15)								
	178.326										
	178.526	76.2 (3.00									
	178.626			62.2 (2.45)							
	178,726 178,826		81.3 (3.20)		350						
	178,826										
	178,926										
	179.526										
	179,626 180,326										
	180,326										
	180,426				63.5 (2.50)						
	181,026	77.5 (3.05)	82.6 (3.25)	63.5 (2.50)							
-	181,326				-	-					
	181,426										
	181.526 181,726	4	Overlaps #3						38.1 (1.50		
_	181,726		Secondary	64.8 (2.55)		-			Overlaps		
-	181,826		Crack 83.8 (	B.30)	-	-			R/H Primar		
	182,026		-	-	-				Crack		
	182,226	78.7 (3.10)		-					+		
-	182,426 183,026	-			-	-			-		
-	183,026			(( 0 10 (0)	-	-			+		
-	183,126	-		66.0 (2.60)	+	1			+		
5 Hz	183,526	-			+	48.3 (1	ani		-		

Panel	No. 11C	Secondary Crack Lengths millimeters (inches)									
Cycle Rate	No. of Cycles	<del>//1</del> +	#5	#6	#8	#9	#10	和			
5 Hz	175,626										
1	176,126										
	176,426		33.0 (1.30)								
	176 606		1 33.0 (2007)				8.9 ( .35)				
	176 726							11.4 ( .45)			
	176 826	53.3 (2.10)									
	177 526	7,10,1 (=0.40)			17.8 ( .70)	34.3 (1.35)					
	177 626	53.3 (2.10)									
	177 726				1						
	178 326							16.5 ( .65)			
	178 526										
	178,526 178,626										
	178,726										
-	178,826	<b>†</b>									
	178.926	55.9 (2.20)					10.2 ( .40)	10.2 ( .40)			
	179.526	1									
	179,626		35.6 (1.40)		20.3 ( .80)						
	180,326										
	180,426										
	181.026										
	181.326	58.4 (2.30)					12.7 ( .50)				
	181,426	1						21.6 ( .85)			
	181,526							1.00			
	181,726	1									
	181,826										
	182.026				1						
	182.226					39.4 (1.55)					
	182.426				21.6 ( .85)						
	183,026							24.11 ( .95)			
	183,126										
	183 526				1		15.2 ( .60)				
5 Hz	183,526					41.9 (1.65)					

Panel	No. 11C	Secondary Crack Lengths millimeters (inches)								
Cycle Rate	No. of Cycles	#12 Secondary	#13 Secondary	#14 Secondary	#15 Secondary					
5 Hz	175,626			1.3 ( .05)						
1	176,126									
T	176,426									
	176,626	8.9 ( .35)								
	176,726	1 2 1 321	11.4 ( .45)							
-	176.826		1							
	177,526									
	177,626									
	177.726									
	178,326		16.5 ( .65)							
	178.526									
	178.626									
	178.726									
	178.826			3.8 ( .15)						
	178,926	10.2 ( .40)								
	179,526				5.1 ( .20)					
	179.626									
	180,326				6.4 ( .25)					
	180,426			6.4 ( .25)						
	181,026									
	181.326	12.7 ( .50)								
	181.426		21.6 ( .85)							
	181.526									
	181.726									
	181,826									
	182,026			7.6 ( .30)						
	180 000				7.6 ( .30)					
	182,226									
	183,026			8.9 ( .35)						
	183,126									
	183,526	15.2 ( .60)								
Hz	184,026									

Panel	No. 11C	Primary Crack Lengths millimeters (inches)					Secondary Crack Lengths  millimeters (inches)  #1 #2 #3			
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	Secon	l nd <b>ar</b> y	#2 Secondary	#3 Secondary	
5 Hz	184.326				64.8 (2.55)	48.3	(1.90)			
	184.726		7-1-1		1 1 2 2 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7				1	
7	185.026		86.4 (3.40)							
	185.226	80.0 (3.15)	150.	67.3 (2.65)						
	185.426		87.6 (3.45)							
	185,626			#14						
	185,726			Secondary						
	185,926			Crack						
	186.026								40.6 (1.60	
	186,126		88.9 (3.50)				a day			
	186,226									
	186,326	81.3 (3.20)								
	186,426									
	187,026									
	187,326								1	
	187,426					-				
	187.626				(( 0 /0 /0)	-			-	
	188,126				66.0 (2.60)				-	
_	188,726					-			-	
-	189,226						-		-	
-	189,726								-	
	189,926					-	-		•	
-	191.026				(7 2 (0 (5)	-			-	
-	191,826				67.3 (2.65)	-			+	
-	192,126									
-	192,826	00 0 (0 00)	07 1/0 (0)		-	-			+	
-	193,226	83.8 (3.30)	91.4(3.60)			-	-		-	
-	193,726					-			43.2 (1.70	
-	193,826								43.2 (1.10	
-	194,126								-	
5 Hz		85.1 (3.35)				118 3	(1.90)		+	

Panel	No. 110		Secondary Crack Lengths millimeters (inches)										
Cycle Rate	No. of Cycles	#14	#5	#6	#8	#9	#10	#11					
5 Hz	184,326												
1	184,726		40.6 (1.60)		24.1 ( .95)								
1	185,026												
	185,226												
_	185,426												
	185 626						16.5 ( .65)	29.2 (1.15)					
	185,726	63.5 (2.50)											
	185,926				25.4 (1.00)								
	186,026												
	186,126	1											
	186,226				26.7 (1.05)								
	186,226 186,326												
	186,426			4	Overlaps #15								
	1187 026		43.2 (1.70)		Secondary	47.0 (1.85)							
	187,326				Crack	Overlaps	17.8 ( .70)						
	187,326 187,426 187,626 188,126 188,726	4	Overlaps			#13		31.8 (1.25)					
	187,626		#11			Secondary		Overlap #5					
	188.126		Secondary			Crack		Secondary					
	188,726		Crack					Crack					
	189,226						20.3 ( .80)						
	189,226	A STATE OF THE PARTY OF THE PAR						34.3 (1.35)					
	189,926						21.6 ( .85)	·					
	191 026		1				22.9 ( .90)						
	191,026	1											
	102 126					54.6 (2.15)		36.8 (1.45)					
	192,126 192,826						24.1 ( .95)						
	193,226												
	193,726					57.2 (2.25)							
	193.826												
	194.126	68.6 (2.70)											
1	194,426	100.0 (5.10)					25.4 (1.00)						
5 Hz	194.526												

Panel	No. 110				Crack Lengths ters (inches)		
Cycle Rate		#12	#13	#14	#15		
5 Hz	184,326						
1	184,726				11.4 ( .45)		
1	185,026						
-	185,226			10.2 ( .40)			
	185,426			Overlaps			
1	185.626	16.5 ( .65)	29.2 (1.15)	L/H Primary			
	185.726	2017		Crack			
	185,926						
	186,026						
	186,126						
	186,126 186,226				14.0 ( .55)		
	186,326					1,17	
	186,426				15.2 ( .60)		
	187,026	4	Overlaps #9		Overlaps		
	187,326	17.8 ( .70)	Secondary		#8		
	187,426 187,626		Crack		Secondary		
	187,626			11.4 ( .45)	Crack		
	188.126						
	188,726			12.7 ( .50)			
	189,226	20.3 ( .80)					
	189,726		34.3 (1.35)				
	189,926	21.6 ( .85)					
	191,026	22.9 ( .90)					
	191,826						
	192,126		36.8 (1.45)				
100	192,826	24.1 ( .95)					
	193,226		1 1 2				
	193,726						
	193.826						 
	194.126	25.4 (1.00)					
Ą	194,426	25.4 (1.00)			-		
5 Hz	194,526						

Panel	No. 11C		Primary Cra		Secondary Crack Lengths millimeters (inches) #1 #2 #3			
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1	#2	#3
5 Hz	194,826		92.7 (3.65)			48.3 (1.90)		
*	196,026				68.6 (2.70)	1		
	197,026							
	197,026 197,226							
	197,426							
	199,326				69.8 (2.75)			
-	199,526					-		
-	199,826		94.0 (3.70)					
-	201,2%	86.4 (3.40)	94.0 (3.10)			<del>                                     </del>	-	
_	201,226 201,626	30.1 (3.10)						
	203,426							
	203,526				71.1 (2.80)			
	205.526					1		
5 Hz	206,476	87.6 (3.45)	95.2 (3.75)			48.3 (1.90)	48.3 (1.90)	2
Final	Crack							
	eths	87.6 (3.45)	95.2 (3.75)	67.3 (2.65)	71.1 (2.80)	48.3 (1.90)	48.3 (1.90)	43.2 (1.70)
	Residua	1 strength =	257.500N	(57,900 lbs)				

Panel	No. 11C	Secondary Crack Lengths millimeters (inches)									
Cycle Rate	No. of Cycles	#4	#5	#6	#8	#9	#10	#11			
5 Hz	194,826										
	196,026										
	197,026						26.7 (1.05)				
	197,226					59.7 (2.35)					
	197,426 199,326							39.4 (1.55			
	199,326										
	199,526						27.9 (1.10)				
	199,826					64.8 (2.55)					
	201,226										
	201,326										
	201,626						29.2 (1.15) 30.5 (1.20)				
	203,426						30.5 (1.20)				
	203,526						27 9 (7 05)				
F 77-	205,526			1.2 0 (2 (5)			31.8 (1.25)				
5 Hz	206,476			41.9 (1.65)							
Final	crack	68.6 (2.70)	43.2 (1.70)	41.9 (1.65)	26.7 (1.05)	64.8 (2.55)	31.8 (1.25)	39.4 (1.55			
Len	oths										
1 -											
						17					

Panel	No. 110		Secondary Crack Lengths millimeters (inches)								
Cycle Rate	No. of Cycles	#12	#12 #13	#14	#15						
5 Hz	194,826										
	196,026										
	197.026	26.7 (1.05)									
	197,226 197,426 199,326						1				
	197.426		39.4 (1.55)								
	199,326										
	1199.526	27.9 (1.10)									
	199.826										
	201,226 201,326										
	201,326										
-	201,626	29.2 (1.15)									
-	203,426	30.5 (1.20)									
-	203,526										
-	205,526	31.8 (1.25)	-								
5 Hz	206,476		-			-	+				
Final	Crack	-	-				+				
Leng		31.8 (1.25)	39.4 (1.55)	12.7 ( .50)	15 2 / 60)		+	<del>                                     </del>			
		132.0 (2.2)	77.4 (1.7)	1.01	17.2 (.00)		+	<del> </del>			
							+	<del> </del>			
							+	1			
							1	<del>                                     </del>			
							1	1			
			7 9		73.7						
	1										

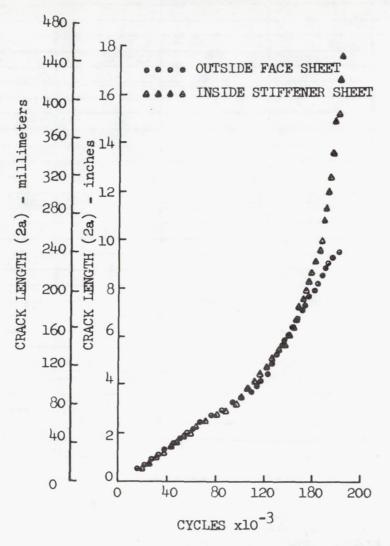


FIGURE F-43 CRACK GROWTH CURVE FOR ALUMINUM-GLASS PANEL #11C (Crack Length Includes Secondary Cracks)

## PANEL #12C

MATERIALS: ALUMINUM - GRAPHITE

ADHESIVE: EA-927R

ALUMINUM STRESS: 172 MN/m<sup>2</sup> (25 ksi)

MAXIMUM FATIGUE LOAD: 200,160N (45,000 lbf)

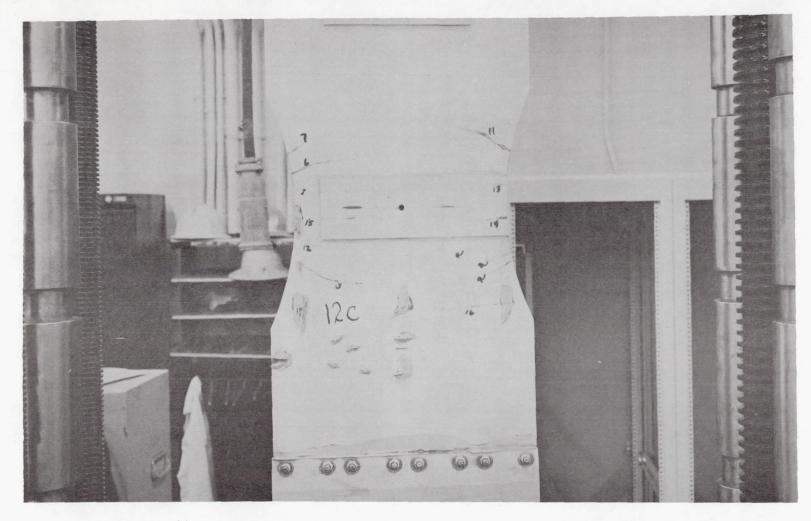


FIGURE F-44 OUTSIDE FACE SHEET AFTER CRACK GROWTH TEST - PANEL #12C

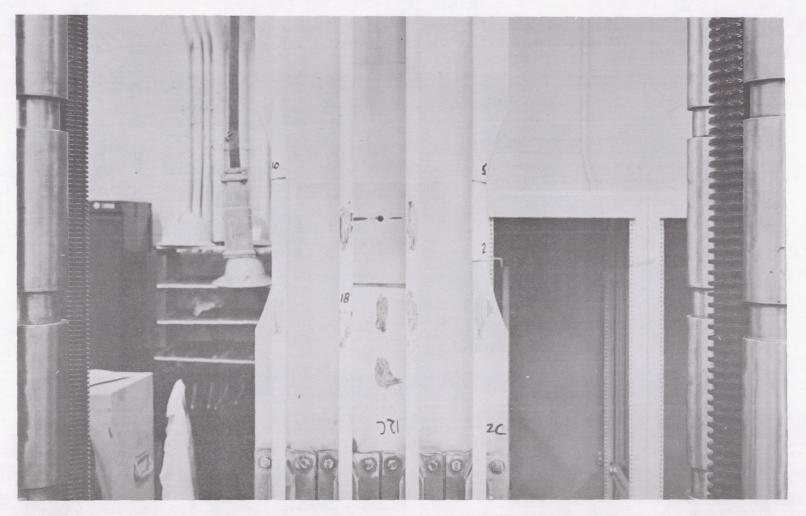


FIGURE F-45 INSIDE STIFFENER SHEET AFTER CRACK GROWTH TEST - PANEL #12C



FIGURE F-46 INSIDE STIFFENER SHEET AFTER CRACK GROWTH TEST - PANEL #12C (RIGHT OF CENTERLINE)



FIGURE F-47 INSIDE STIFFENER SHEET AFTER CRACK GROWTH TEST - PANEL #12C (LEFT OF CENTERLINE)

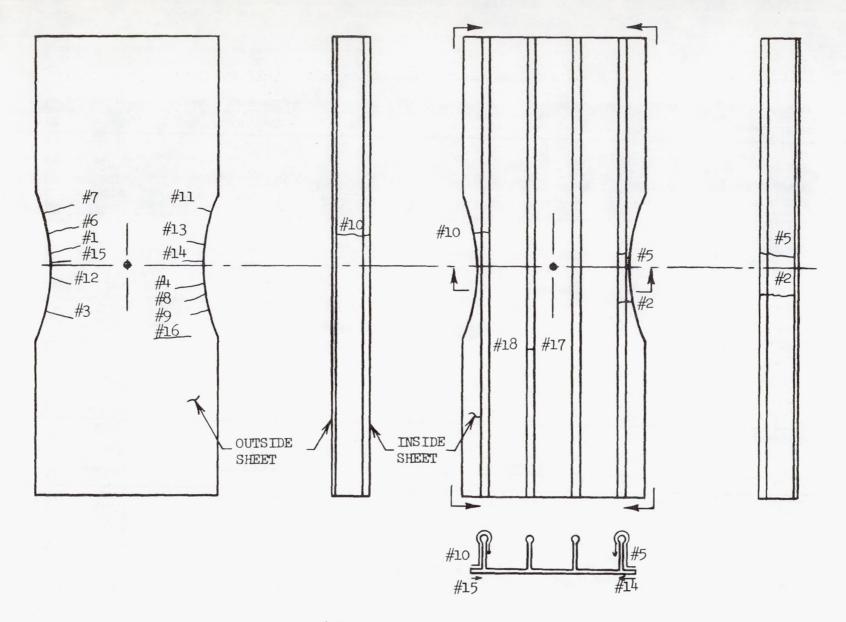


FIGURE F-48 SECONDARY CRACK LOCATIONS ALUMINUM-GRAPHITE PANEL #12C

Panel	No. 12C		Primary Cra millimeters				ry Crack Lengt eters (inches)	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	L/H Edge Outside	R/H Edge Inside	
4 Hz	6,500	5.1 ( .20)	6.4 ( .25)					
<b>A</b>	8,200	6.4 ( .25)	7.6 ( .30)					
	10,600	7.6 ( .30)	8.9 ( .35)	3.8 ( .15)				
76	13,100	8.9 ( .35)						
	13,300		10.2 (.40)					
	13,700			5.1 ( .20)	5.1 (.20)			
	15,000	10.2 ( .40)			6.4 ( .25)			
	15,400	`		6.4 ( .25)				
	18,200	11.4 ( .45)	11.4 ( .45)					
-	18,400				7.6 ( .30)			
	21.000			7.6 ( .30)				
	21,500	12.7 ( .50)	10.7.7.50\		0 0 7 251			
	21,700	71. 0 / 55\	12.7 ( .50)		8.9 ( .35)			
-	24,700	14.0 ( .55)		9.0 / 25)				
-	24,900	75.0 / (0)	71. 0 / 55)	8.9 ( .35)		-		
-	28,000	15.2 ( .60)	14.0 ( .55)					
-	31,400		15.2 ( .60)	10.2 ( .40)	10.2 ( .40)			
-	37,000	16.5 ( .65)	16.5 ( .65)	10.2 ( .40)	10.2 ( .40)			
	37,300	10.7 ( .07)	10.) ( .0)	11.4 ( .45)				
-	37,500			11.4 ( .4)	11.4 ( .45)			
1	42,000	17.8 ( .70)		12.7 ( .50)	1			
-	44,000	11.0 ( .10)	17.8 ( .70)	14.1	12.7 ( .50)			
	45,200	19.0 ( .75)	1.01.00		12.01			
	45,600			14.0 ( .55)				
	47,100		19.0 ( .75)	1				
	49.800				14.0 ( .55)			
-	50.300	20.3 ( .80)						
4 Hz.	50.450		20.3 ( .80)					
3 Hz	52,210			15.2 ( .60)		10.2 ( .60)	12.7 ( .50)	
	55,000	21.6 ( .85)					15.2 ( .60)	
3 Hz	55,100			16.5 ( 65)				

Panel	No. 12C		Primary Cra millimeters			Secondar millime	ry Crack Lengt	hs
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	L/H Edge Outside	R/H Edge Outside	L/H Edge Inside
3 .Hz	55,400					11.4 ( .45)		
1	55,600		21.6 ( .85)					
	57,200					12.7 ( .50)		
	57,600	22.9 ( .90)	22.9 ( .90)					
	57.800				15.2 ( .60)			
	59,200					14.0 ( .55)		
	59,300							
	62.600			17.8 ( .70)				
	62,800	24.1 ( .95)				15.2 ( .60)		
	63,000							
	63,600				16.5 ( .65)			
	65,100		24.7 ( .95)			16.5 ( .65)		
	67,800	0-1-1-1		and a second second				
	67,900			19.0 ( .75)	17.8 ( .70)			
	68,700					17.8 ( .70)		
	69 000			The state of the s			2.5 ( .10)	
	69,200	25.4 (1.00)	25.4 (1.00)					
	72,600	26.7 (1.05)						
	73,000			20.3 ( .80)				
	73,500		26.7 (1.05)					
	73,700					644 4 4		
	73,800					19.0 ( .75)		
	75,000						5.1 ( .20)	
	75,200							3.8 ( .1
	75,400							
	76,400	27.9 (1.10)						
	76,700							5.1 ( .2
	79,800		27.9 (1.10)	FORT			7.6 ( .30)	
1	80,300					21.6 ( .85)	and the state of t	
1	80.500							
3 Hz	80.800				19.0 ( .75)			
4 Hz	81,300					22.9 ( .90)		

Panel	No. 12C		Secondary Crack Lengths millimeters (inches)								
Cycle Rate	No. of Cycles	R/H Edge Inside									
3 Hz	55,400 55,600										
1	55,600										
	57,200										
	57,600 57,800	17.8 ( .70)									
	57,800										
	59,200										
	59,300	19.0 ( .75)									
	62,600										
	62,800										
	63,000	20.3 ( .80)									
	63,600										
	65,100										
	67.800	21.6 ( .85)									
	67,900				-						
	68,700 69,000				-						
	69,000										
	69,200	7 11 12 12 12 12 12 12 12 12 12 12 12 12			-						
	72,600 73,000	22.9 ( .90)									
	73,000				-						
	73,500										
	73,700	24.1 ( .95)			-						
	73,800					-					
	75,000										
	75,200										
	75,400	25.4 (1.00)			-						
	76,400				-						
	76.700				-	-					
	79.800										
	80.300				-	-					
	80,300	26.7 (1.05)			-						
3 Hz	80.800				-		-				
4 Hz	81.300					<u> </u>					

• 12C		millimeters	ck Lengths (inches)		Secondary Crack Lengths millimeters (inches)			
o. of cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	L/H Edge Outside	R/H Edge Out <b>si</b> de	L/H Edge Inside	
82,500							6.4 ( .25	
82,600			21.6 ( .85)					
83,500	29.2 (1.15)							
83,800						10.2 ( .40)		
83,850							7.6 ( .30	
							8.9 ( .35	
87,100						12.7 ( .50)		
							10.2 ( .40	
90,300	30.5 (1.20)	29.2 (1.15)						
91,800								
92.700								
92.750			22.9 ( 90)					
92,900							15.2 ( .60	
93,500				20.3 ( .80)				
94,800						15.2 ( .60)		
95,000		30.5 (1.20)						
95,200					25,4 (1.00)			
manufacture of the section of the se	21.8 (1.25)							
the Contract of the Contract o								
			24.1 ( .95)				10 0 / 55	
							19.0 ( .75	
						0 ()		
04.749	33.0 (1.30)	31.8 (1.25)		07 ( 05)	27.9 (1.10)	17.8 ( .70)		
		-	05 1 (5 55)	21.6 ( .85)			00 0 / 00	
			25.4 (1.00)			( 0 )	20.3 ( .80	
109.337	34.3 (1.35)	33.0 (1.30)				20.3 ( .80)		
111,749	35.6 (1.40)	34.3 (1.35)		00.0 /				
				22.9 ( .90)				
			26.7 (1.05)					
116,249							22.9 ( .90	
	87,800 90,300 91,800 92,700 92,750 92,900 93,500 94,800 95,000	87,100 87,800 90,300 91,800 92,700 92,750 92,900 93,500 94,800 95,000 95,000 95,400 95,400 91,8 (1.25) 97,900 03,000 03,400 03,400 03,400 04,749 09,329 09,329 09,337 34,3 (1.35) 11,749 35,6 (1.40) 11,849 12,249 16,249	87,100   87,800   90,300   30.5 (1.20)   29.2 (1.15)   91,800   92,700   92,750   92,900   93,500   94,800   95,000   95,000   95,400   21.8 (1.25)   97,900   03,000   03,400   03,700   04,749   33.0 (1.30)   31.8 (1.25)   06,649   09,329   09,337   34.3 (1.35)   33.0 (1.30)   31,749   35.6 (1.40)   34.3 (1.35)   11,749   35.6 (1.40)   34.3 (1.35)   11,849   12,249   15,749   16,249   16,249   16,249   16,249   16,249   16,249   16,249   16,249   16,249   16,249   16,249   10.30   1.30	87.100   87.800   90.300   30.5 (1.20)   29.2 (1.15)   91.800   92.700   92.750   92.900   93.500   94.800   95.000   95.000   95.000   97.900   97.900   93.700   94.11 (.95)   97.900   93.700	87,100 87,800 90,300 91,800 92,700 92,700 92,900 93,500 94,800 95,000 95,200 96,400 91,800 97,900 03,000 03,400 03,700 04,749 03,700 04,749 06,649 09,329 09,329 09,329 09,329 09,329 09,329 09,329 09,329 11,749 35,6 (1,40) 34,3 (1,35) 11,749 35,6 (1,40) 34,3 (1,35) 11,749 16,249 26,7 (1,05)	87,100   87,800   90,300   30.5 (1.20)   29.2 (1.15)   91,800   92,700   92,750   92,900   93,500   94,800   95,000   30.5 (1.20)   95,400   21.8 (1.25)   97,900   93,500   94,800   94,800   95,400   21.8 (1.25)   97,900   21.8 (1.25)   97,900   21.8 (1.25)   97,900   21.8 (1.25)   27,9 (1.10)	87,100 87,800 90,300 30.5 (1.20) 29.2 (1.15) 91.800 92,700 92,700 92,900 93.500 95,200 95,200 95,400 21.8 (1.25) 97,900 03,700 04,700 03,700 04,740 33.0 (1.30) 31.8 (1.25) 90,329 09,337 34.3 (1.35) 33.0 (1.30) 11,749 35.6 (1.40) 34.3 (1.35) 11,749 35.6 (1.40) 34.3 (1.35) 11,749 35.6 (1.40) 34.3 (1.35) 11,749 15,749	

Panel	No. 12C		Second milli	ary Crack Leng meters (inches	gths s)		
Rate	No. of Cycles	R/H Edge Inside					
4 Hz	82,500	27.9 (1.10)					
	82,600						
	83,500						
	83.800						
	83,850 85,500						
	85,500						<u> </u>
	1 87,100						
	87,800 90,300 91,800						
	90,300						
	91,800						
	92,700	29.2 (1.15)					
	92,750						
	92,900		 				-
	93,500						
	94,800		 				
	95,000 95,200						
	95,200		 			-	-
-	95,400						-
		30.5 (1.20)	 				
	103,000		 			-	-
-	103,400		 			-	
	103,700	31.8 (1.25)	 				-
	104,749						
	106,649		 				
	109,329	33.0 (1.30)					
	109,337						
	111,749						-
	111.849					·	-
	112,249	34.3 (1.35)					-
	112,249 115,749 116,249	25 ( /2 (0)					
1	116,249	35.6 (1.40)					-
4 Hz	117,249						

Panel	No. 12C		Primary Cra	ck Lengths (inches)		Secondary Crack Lengths millimeters (inches)			
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	L/H Edge Outside	R/H Edge Outside	L/H Edge Inside	
4 .Hz	117,849	36.8 (1.45)	35.6 (1.40)						
4	118,749								
	120,049				24.1 ( .95)			24.1 ( .95	
	121.149			27.9 (1.10)					
	123,949								
	124.349							25.4 (1.00	
	126.949			29.2 (1.15)	25.4 (1.00)			1200	
	129,049	38.1 (1.50)	36.8 (1.45)				25.4 (1.00)		
	130.849							26.7 (1.05	
	131.049								
	133.649			30.5 (1.20)	26.7 (1.05)				
	133,749	39.4 (1.55)	38.1 (1.50)			36.8 (1.45)			
	133.849 134,549							27.9 (1.10	
	134,549								
	136,749					38.1 (1.50)			
	137,349	40.6 (1.60)	39.4 (1.55)						
	137.749						27.9 (1.10)		
	140,749	41.9 (1.65)						30.5 (1.20	
	141,349			31.8 (1.25)	27.9 (1.10)				
	141.749								
	143,549					39.4 (1.55)	29.2 (1.15)		
	144,049		40.6 (1.60)						
	145,749							31.8 (1.25	
-	146,449			33.0 (1.30)	30.5 (1.20)				
	146,549								
-	148,249	43.2 (1.70)	41.9 (1.65)				30.5 (1.20)		
-	148,749					41.9 (1.65)			
-	149,649			0 0 1				33.0 (1.30	
	152,049			34.3 (1.35)	31.8 (1.25)				
100	152,349								
77	153.049	44.4 (1.75)	43.2 (1.70)				31.8 (1.25)		
4 Hz	155,885				34.3 (1.35)				

Panel	No.12C			ndary Crack Le limeters (inch		_
Cycle Rate	No. of Cycles	R/H Edge Inside				
4 Hz	117,849					
1	118,749	36.8 (1.45)				
	120,049					
_	121,149					
	121,149 123,949 124,349 126,949 129,049 130,849 131,049 133,649 133,749	38.1 (1.50)				
	124,349					ļ:
	126.949					-
	129.049					-
	130.849					
	131,049	39.4 (1.55)				1
	133,649	The second second second				-
	133,749					-
	133,849					-
	134.549	40.6 (1.60)			 	-
	136,749				 	
	137,349				 	-
	137,749		-		 	-
	140,749				 	1
	141,349				 	-
	141,749	41.9 (1.65)			 	1
	143,549					+
	144,049				 	-
	1145,749					
	146,449	\ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \			-	1
	146,549	43.2 (1.70)			 	-
	148,249				1	-
	148.749					
	134,549 136,749 137,349 137,749 140,749 141,349 141,749 143,549 144,049 145,749 146,449 146,549 148,749 148,749 148,749 152,049 152,049 153,049 153,049					-
-	152,049	11. 1. /2 75				
	152,349	44.4 (1.75)			-	
4 Hz	153,049					1

Panel :	No. 12C		Primary Cra-			Secondary Crack Lengths millimeters (inches)			
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	L/H Edge Outside	R/H Edge Outside	L/H Edge Inside	
4 Hz.	157,185	45.7 (1.80)	44.4 (1.75)				33.0 (1.30)		
A	157.685					43.2 (1.70)			
		47.0 (1.85)	45.7 (1.80)						
	163.285				Free 1 20		35.6 (1.40)		
	164,685								
	164,885							38.1 (1.50	
	168.885	49.5 (1.95)	48.3 (1.90)						
	169.885						38.1 (1.50)		
	170,785					48.3 (1.90)			
	173.885	50.8 (2.00)	49.5 (1.95)			49.5 (1.95)	39.4 (1.55)		
	176.685							39.4 (1.5	
	178,285	52.1 (2.05)	50.8 (2.00)						
	179,085						41.9 (1.65)		
	179,385	53.3 (2.10)							
	179.885		52.1 (2.05)			50.8 (2.00)			
	183.885	54.6 (2.15)	53.3 (2.10)			52.1 (2.05)	43.2 (1.70)		
	184,585				36.8 (1.45)			41.9 (1.65	
	186,585						44.4 (1.75)		
		55.9 (2.20)	54.6 (2.15)			53.3 (2.10)			
	187.885				38.1 (1.50)			43.2 (1.7	
	188,285			38.1 (1.50)					
	195,385		-	39.4 (1.55)	40.6 (1.60)				
	198,285		55.9 (2.20)						
	198,685						45.7 (1.80)		
	199,985	1 -0 1 (=				54.6 (2.15)			
	203,185	58.4 (2.30)	57.2 (2.25)						
	204,085				41.9 (1.65)				
	204,485	F0 F /0 0F	50 1. (0.00)	40.6 (1.60)					
		59.7 (2.35)	58.4 (2.30)						
-	207.985					57.2 (2.25)			
4 Hz	208,185					21.6 (6.62)			

Panel	No. 120		Secondary Crack Lengths millimeters (inches)								
Cycle Rate	No. of Cycles	R/H Edge Inside									
4 Hz	157,185										
-	157,685 161,885 163,285										
	161,885										
	163,285										
	164,685 164,885	45.7 (1.80)									
	164,885										
	168.885										
	169.885										
	170,785 173,885 176,685										
	173,885				-						
	176,685				-						
	178,285					-					
	179.085				-						
	179,385 179,885										
	179,885										
	183,885 184,585 186,585 186,885 187,885										
	184,585					-					
	186,585				-	-					
	186,885				-	-					
	187,885				-	-					
	188,285				-						
	195,385				-	-					
	198,285				-		<del> </del>				
	198,685					-					
	199,985 203,185					-					
	203,185				-	-					
	204,085				-		-				
	204,485				-	-	-				
	207,785	57.2 (2.25)			-						
	207.985	57.2 (2.25) 58.4 (2.30) 59.7 (2.35)			+	-					
	208,185	59.7 (2.35)			-	-					
4 Hz	1208,485	61.0 (2.40)									

Panel	No. 12C		Primary Cra			Seconda millim	ry Crack Lengt eters (inches)	hs
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	L/H Edge Outside	R/H Edge Outside	L/H Edge Inside
4 Hz	208,585						49.5 (1.95)	
4	208,885						49.7 (1.97)	
	209,085				43.2 (1.70)			
	209,585				13.2 (2.10)			
	210,785							
	211,485			41.9 (1.65)				
	212,435			120) (200)				
	212,885	61.0 (2.40)					-	
	213.385							
	213,685						50.8 (2.00)	
	214.785					58.4 (2.30)	72.00/	
	215.185		59.7 (2.35)					
	216.885							
	217.685				44.4 (1.75)			
1	218.885			43.2 (1.70)				
	219,085	62.2 (2.45)	61.0 (2.40)			OVERLAPS L/H		
	219,385	OVERLAPS L/H				OUTSIDE PRI-		
		OUTSIDE EDGE				MARY CRACK		
	222.985	SECONDARY	62.2 (2.45)					
	223,185	CRACK						
	223,485						53.3 (2.10)	
	223,585				45.7 (1.80)			
	226,085							
	226,185			44.4 (1.75)				
	228,132							
	228,532		63.5 (2.50)			63.5 (2.50)		
	228,632	63.5 (2.50)						
	228,832						54.6 (2.15)	
	229,432							
	231.832							
*	232,832		64.8 (2.55)	45.7 (1.80)				
4 Hz	233.032					64.8 (2.55)		

Panel	No. 12C		Secondary Crack Lengths millimeters (inches)								
Cycle Rate	No. of Cycles	R/H Edge Inside									
4 Hz	208,585										
A	208,885	62.2 (2.45)									
	200 085										
	209.585	63.5 (2.50) 64.8 (2.55) 66.0 (2.60)									
	210.785	64.8 (2.55)									
	211.485										
	212.435	66.0 (2.60)									
	212.885	67.3 (2.65)									
	213,385	67.3 (2.65)									
	213,685					-	-				
	214,785	68.6 (2.70)				-	-				
	215,185						-				
	216,885	69.8 (2.75)			-						
	217,685	71.1 (2.80)									
-	218,885	71.1 (2.80)									
	219.085										
	219,385	72.4 (2.85)			-		-				
-	220,685	72.4 (2.05)			-		-				
-	222,985	73.7 (2.90)			-		-				
-	223,185	73.7 (2.90)			-	-	-				
-	223,485	-					-				
-	223,585				-	-	-				
-	226,085	74.9 (2.95)				-	-				
-	226,105				-	-	-				
-	228,132	76.2 (3.00)			-	-	-				
-	228,532	-			-						
-	228,632			-	-		1.				
-	228,832	77 5 (2.05)			-		-				
-	229,432	77.5 (3.05) 78.7 (3.10)				1	1				
-	231.832	10.7 (3.10)			-	-	-				
4 Hz	232.832				-	-	-				

Panel	No. 12C		Primary Crac				ry Crack Lengueters (inches	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	L/H Edge Outside	R/H Edge Outside	L/H Edge Inside
4 Hz	233,732	64.8 (2.55)						
4	234,432							
	235,832	+	OVERLAPS R/H				55.9 (2.20)	
	237,432		OUTSIDE EDGE		48.3 (1.90)		OVERLAPS R/H	
	238,032		SECONDARY		1		OUTSIDE PRI-	
	238.432		CRACK				MARY CRACK	52.1 (2.05)
	238.532			47.0 (1.85)				
	238.832							53.3 (2.10)
	239.282							54.6 (2.15)
	239,332					66.0 (2.60)		
	239.832		67.3 (2.65)					
	240,032							55.9 (2.20)
	240,632							57.2 (2.25)
	240.932							
	241,282 241,932					a design of the		58.4 (2.30) 59.7 (2.35)
	241,932							59.7 (2.35)
	242.582							61.0 (2.40
	243.032							62.2 (2.45
	243,332 243,432						57.2 (2.25)	
	243,432	66.0 (2.60)						
	243,532				49.5 (1.95)			63.5 (2.50
	244,332							64.8 (2.55
	244,832							
	245,132							66.0 (2.60
	245,332			48.3 (1.90)				`
	245,632							67.3 (2.65
	246,132							68.6 (2.70
	246,632		68.6 (2.70)					, , ,
	246,832							69.8 (2.75
-	247.582							71.1 (2.80
77-	247,732					67.3 (2.65)	FO 1. /0 == V	
4 Hz	248,032						58.4 (2.30)	

Panel	No. 12C		Secondary millimet	Crack Lengths ers (inches)	
Cycle	No. of Cycles	R/H Edge Inside			
∔ Hz	233,732	80.0 (3.15)			
	234,432	80.0 (3.15)			
T	235,832	The second second second second			
	237,432				
	238.032	86.3 (3.20)			
	238,432				
	238,532 238,832 239,282				
	238.832				
	239,282				
	239,332 239,832 240,032				
	239.832				 
	240,032				
	1010 622				 
	240.932	82.6 (3.25)			
	241.282				
	241.932				 
	242,582				 
	243,032			-	
	243,332				
	243,432			-	 
	243.532				
	244,332			-	 
	244,832	83.8 (3.30)			
	245,132				 
	245,132 245,332				
	245,632			-	
	245,632 246,132				
	246,632			-	 
	246.832				
	247,582 247,732			-	
4 Hz	247,732				

Panel	No. 12C		Primary Cra				ry Crack Leng eters (inches	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	L/H Edge Outside	R/H Edge Outside	L/H Edge Inside
4 Hz	248,332							72.4 (2.85)
1	248,832							
	249,082		War of the Control of					73.7 (2.90)
	249,632				50.8 (2.00)			= (====
	250,632			10 5 (2 05)				76.2 (3.00)
-	250,932			49.5 (1.95)				77 5 (2.05)
-	251,332 251,432							77.5 (3.05)
-	252 322							78.7 (3.10)
4 Hz	252,332 252,993	67.3 (2.65)	69.8 (2.75)			68.6 (2.70)	59.7 (2.35)	
FTNAL LEN	CRACK CTHS	67.3 (2.65)	69.8 (2.75)	49.5 (1.95)	50.8 (2.00)	68.6 (2.70)	59.7 (2.35)	78,7 (3,10)
	RESI	DUAL STRENGTH	= 253,500	N (57,000	1bs.)			
					-			

Panel	No. 12C		Secondary Crack Lengths millimeters (inches)								
Cycle Rate	No. of Cycles	R/H Edge Inside									
4 Hz	248,332	<b>85.1</b> (3.35) 86.4 (3.40)									
4	248.832	85.1 (3.35)									
	249.082										
	249.632										
	250,632			-							
	250,932										
-	251,332	0( ) (0 ) 0)									
	251,432	86.4 (3.40)		-							
1 77-	252,332			-							
4 Hz	252,993										
	-										
							Edition 1				
FINAL	CRACK			*							
	GTH	86.4 (3.40)									
				-							
				-							
				-							
	-										
					-						
	-										
				1							
	-			1							
	1										

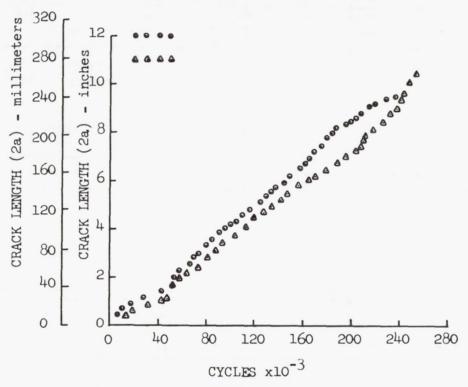


FIGURE F-49 CRACK GROWTH CURVE FOR ALUMINUM-GRAPHITE PANEL #12C (Crack Length Includes Secondary Cracks)

## PANEL #13C

MATERIALS: ALUMINUM-GRAPHITE

ADHESIVE: EA-927R

ALUMINUM STRESS: 103 MN/m<sup>2</sup> (15 ksi)

MAXIMUM FATIGUE LOAD: 124,540N (28,000 lbf)

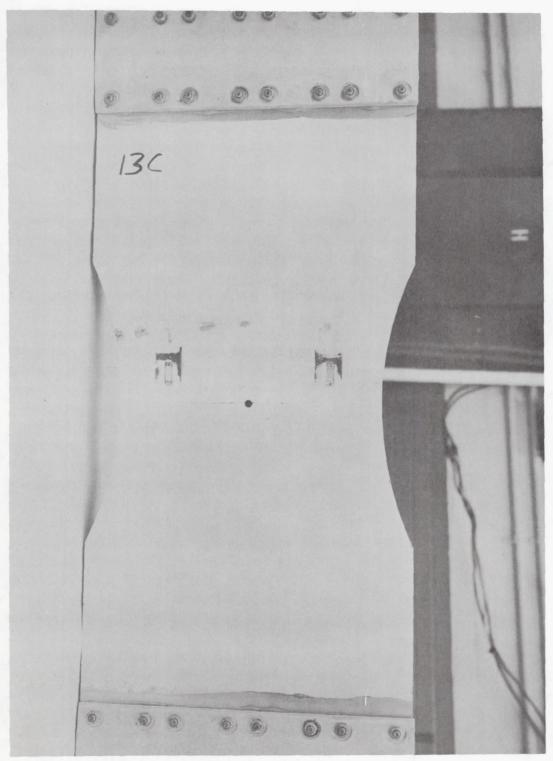


FIGURE F-50 OUTSIDE FACE SHEET AFTER CRACK GROWTH TEST - PANEL #13C

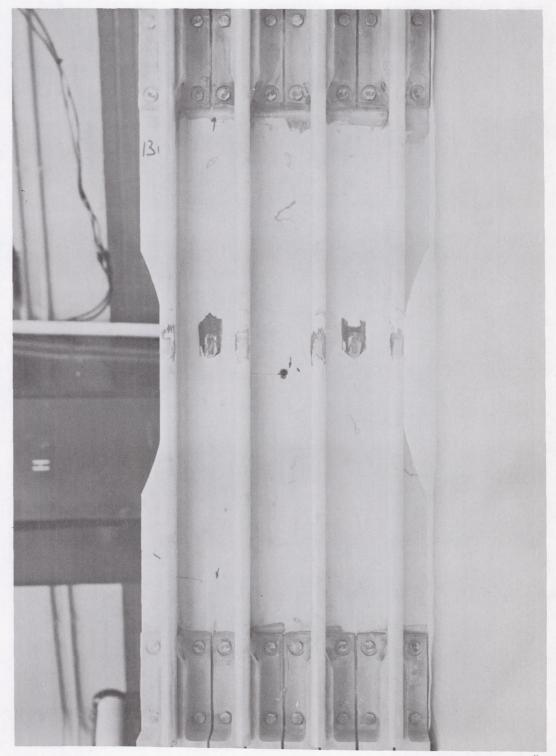


FIGURE F-51 INSIDE STIFFENER SHEET AFTER CRACK GROWTH TEST - PANEL #13C

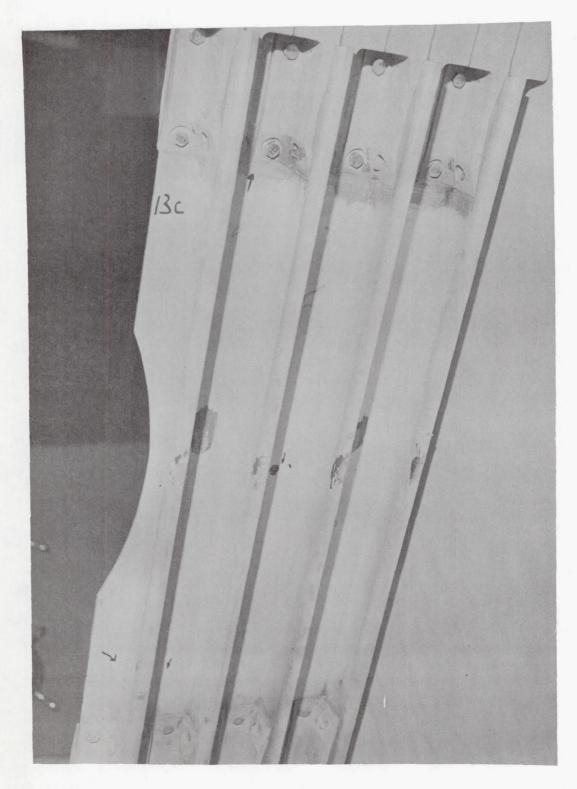


FIGURE F-52 INSIDE STIFFENER SHEET AFTER CRACK GROWTH TEST - PANEL #13C (LEFT OF CENTERLINE)

Panel	Panel No. 13C		Primary Crac millimeters	k Lengths (inches)		Secondary millimete	Crack Lengths rs (inches)
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside		
4Hz	24,700	5.1 ( .20) 6.4 ( .25)					
A	28,800	6.4 ( .25)					
	32,300		3.8 ( .15)				
	34,450		5.1 ( .20)				
	36,200	7.6 ( .30)					
	44,300	0 - ( 057)	6.4 ( .25)				
1	45,700	8.9 ( .35)	7 ( / 20)				
4Hz	54,300	10.2 ( .40)	7.6 ( .30) 8.9 ( .35)				
5Hz	69,660	1 1 ( 1.5)	8.9 ( .35)				
1	69,660	11.4 ( .45)					4.5
-	79,260	12.11.001	10.2 ( .40)				
5Hz	81,160	14.0 ( .55)	10.2 ( .40)				
6Hz	92,460		11.4 ( .45)				
1	93,560 103,160 104,460	15.2 ( .60)	11.4				
-	1104.460	17.2	12.7 ( .50)				1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
-	114,760	16.5 ( .65)					
6Hz	115,360		14.0 ( .55)				
8Hz	127,550		15.2 ( .60)				
	135,150	17.8 ( .70)					
	152,250		16.5 ( .65)				
	158,250	19.0 ( .75)		5.1 ( .20)			
	170,650		17.8 ( .70)				
	170,650 170,950	20.3 ( .80)					
	175,150	21.6 ( .85)	19.0 ( .75)				
	186,764		20 0 ( 00)	7.6 ( -30)			
	196,350	22.9 ( .90)	20,3 ( .80)				
	207,850	24.1 ( .95)	-	0 = ( 25)			
-	208,250		21.6 ( .85)	8.9 ( .35)			
-	1217,650	+	21.6 ( .85)	10.2 ( 40)			
8Hz	224,750	25.4 (1.00)	-	1000 (40)			

Panel	No. 13C		Primary Cra			Secondary Crack Lengths millimeters (inches)
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	
8Hz	233,550	26.7 (1.05)				
-	236,150		22.9 ( .90)			
	253,091	27.9 (1.10)				
	256,491		24.1 ( .95)			
	256,691			11.4 ( .45)		
	269.691	29.2 (1.15)	25.4 (1,00)			
	282,491			12.7 ( .50)		
	283,891	30.5 (1.20)				
	294.591		26.7 (1.05)			
	295,691	31.8 (1.25)				
	308,391	33.0 (1.30)	27.9 (1.10)			
	320,991	34.3 (1.35)				
	333,064		29.2 (1.15)	14.0 ( .55)	6.4 ( .25)	
1 1	348,791 348,891		30.5 (1.20)			
	348,891	35.6 (1.40)				
	361,791	36.8 (1.45)		1507 (0)	7 ( / 20)	
-	369 821		21 0 (1 05)	15.2 ( .60)	7.6 ( .30)	
-	375,601		31.8 (1.25)		9 0 / 35	
	381,521	20 3 /3 50			8.9 ( .35)	
-	414,521	38.1 (1.50)	00 0 (1 00)			
-	414,621	(2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	33.0 (1.30)	26 5 1 (5)		
	432.761	39.4 (1.55)	34.3 (1.35)	16.5 ( .65)		
-	449,121	40.6 (1.60)	( (- \ -)			
-	458,021	12 0 /2 (5)	35.6 (1.40) 36.8 (1.45)			
	495,821	41.9 (1.65)	38.1 (1.50)	17.8 ( .70)		
-	516,798	43.2 (1.70)	39.4 (1.55)	17.8 ( .70)		
-	535,098	111, 1, (1, 75)	39.4 (1.22)			
-	551,898	44.4 (1.75)	10 6 12 (0)			
-	562,798		40.6 (1.60)			
-	578.898	45.7 (1.80)	41.9 (1.05)			
8Hz	592,798	47.0 (1.85)	43.2 (1.70)			

Panel	No. 13C		Primary Cra millimeters			Secondary Crack Lengths millimeters (inches)	
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside		
8Hz	615,398		44.4 (1.75)	19.0 ( .75)			
	629,998	48.3 (1.90)					
	1630.098		45.7 (1.80)				
	649,380	49.5 (1.95)	47.0 (1.85)				
	671,380	49.5 (1.95) 50.8 (2.00)	48.3 (1.90)				
	1686.380		49.5 (1.95)				
	689,380	52.1 (2.05)					
	701.380		50.8 (2.00)				
	708,940	53.3 (2.10)		20.3 ( .80)			
	720,180		52.1 (2.05)				
	1732 080		53.3 (2.10)				
	734.680	54.6 (2.15)	54.6 (2.15)				
	750.280	55.9 (2.20)		21.6 ( .85)			
1	761,480	55.9 (2.20)	55.9 (2.20)				
	783,180		57.2 (2.25)				
	791.980	57.2 (2.25)					
	803.080	58.4 (2.30) 59.7 (2.35) 61.0 (2.40)	58.4 (2.30) 59.7 (2.35)				
	820,280	59.7 (2.35)	59.7 (2.35)				
	841,580	61.0 (2.40)	61.0 (2.40)				
	1859,080		62.2 (2.45)				
		62.2 (2.45)					
	1863,160			22.9 ( .90)			
	872.780	63.5 (2.50)	63.5 (2.50)				
	1875,780	63.5 (2.50)					
	885,780		64.8 (2.55)				
		64.8 (2.55)					
	903,980		66.0 (2.60)				
	914,880		67.3 (2.65)				
	1924,780	66.0 (2.60)	The second second				
	1931.580		68.6 (2.70)				
	942,580	67.3 (2.65)	-				
8Hz	950.032		69.8 (2.75)				

Panel	No. 13C		Primary Cra			Secondary Crack Lengths millimeters (inches)	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside		
8Hz	964,532	68.6 (2.70)					
1	965,032		71.1 (2.80)				
T	968,732	69.8 (2.75)					
	987,832	71.1 (2.80)	72.4 (2.85)				
	1,000,032		73.7 (2.90)				
	1.005.345			24.1 ( .95)			
	005.346	72.4 (2.85)					
	1.019.532			25.4 (1.00)	10.2 ( .40)		
	1,020,032		74.9 (2.95)				
	1.030.032	73.7 (2.90)					
	1.037.422	74.9 (2.95)	76.2 (3.00)				
	1,065,032	76.2 (3.00)	77.5 (3.05)				
	1,073,473		78.7 (3.10)	26.7 (1.05)			
	1,083,873	77.5 (3.05)					
	1,092,879		80.0 (3.15)				
_	1,116,223		81.3 (3.20)				
	1.123.543	78.7 (3.10)	00 ( (2.05)				
	1,143,473	80.0 (3.15)	82.6 (3.25)				
_		81.3 (3.20)	00 0 (0 00)				
	170,406	0 ( ()	83.8 (3.30)				
-	1,184,906	82.6 (3:25)	85.1 (3.35)		70.7 ( 50)		
-	1,206,956				12.7 ( .50)		
	1,215,456	83.8 (3.30)	87.6 (3.45)		14.0 ( .55)		
-	,217,236	0= = (= ==)	00 0 (0 50)		14.0 ( .55)		
	1,227,756	85.1 (3.35)	88.9 (3.50)		15.2 ( .60)		
-	1,229,936	96 1, 12 1,01	90.2 (3.55)		1).2 ( .00)		
	240,906	86.4 (3.40)					
-	256 266	87.6 (3.45) 88.9 (3.50)	91.4 (3.60) 92.7 (3.65)				
-	200,200	90.2 (3.55)	94.0 (3.70)				-
-		91.4 (3.60)	95.2 (3.75)				
8Hz	1.326.806	92.7 (3.65)	96.5 (3.80)				

anel	nel No. 13C Primary Crac millimeters				Secondary Commillimeters	rack Lengths (inches)	
ycle ate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside		
BHz	1,343,608	94.0 (3.70)	97.8 (3.85)				
A	1,363,008	95.2 (3.75)	99.1 (3.90)				
	1,372,408	96.5 (3.80)	100.3 (3.95)				
	1,407,908	100.3 (3.95)	101.6 (4.00)	27.9 (1.10)			
	1,415,908	101.6 (4.00)	104.1 (4.10)				
	1,422,408	102.9 (4.05) 104.1 (4.10)	105.4 (4.15)				
-	1.430.308	104.1 (4.10)	106.7 (4,20)	(2.25)			
	1.430.408	105.4 (4.15)		29.2 (1.15)			
-	1,439,408	105.4 (4.15)	108.0 (4.25)				
-	1,452,408	106.7 (4.20)	109.2 (4.30)				
-	1,462,408	108.0 (4.25)	110.5 (4.35)				
_	473,408	109.2 (4.30)	111.0 (4.40)				
-	488,408	110.5 (4.35)	113.0 (4.45)		+		
-		111.8 (4.40)				<del></del>	
-	509,408	113.0 (4.45) 114.3 (4.50)	118 1 (4.60)				
+	502 400	115 6 (), 55)	120.6 (4.0)				
_	523,400	115.6 (4.55) 116.8 (4.60)	121 9 (4 80)				
+	532,400	118.1 (4.65)	123 2 (4.85)				
-	5/16 108	119.4 (4.70)	A+ Edge				
-	553.408	120 6 (4.75)	of Panel				
1	558,408	120.6 (4.75) 121.9 (4.80)					
Hz	853,408	At Edge		36.8 (1.45)			
	1	of Panel					
INAL	CRACK LEN	GTH 121.9(4.8	0) 123.2(4.85	36.8 (1.45)	15.2 ( .60)		
	Residual	Strength =	371.850 N	(83,600 lbs			

## PANEL #14C

MATERIALS: ALUMINUM - GRAPHITE

ADHESIVE: EA-927R

ALUMINUM STRESS: 138 MN/m<sup>2</sup> (20 ksi)

MAXIMUM FATIGUE LOAD: 168,690N (37,925 lbf)

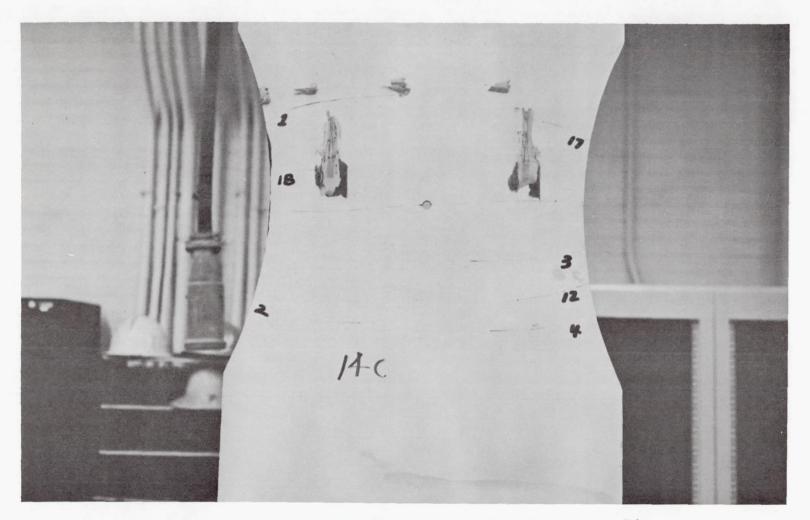


FIGURE F-53 OUTSIDE FACE SHEET AFTER CRACK GROWTH TEST - PANEL #14C

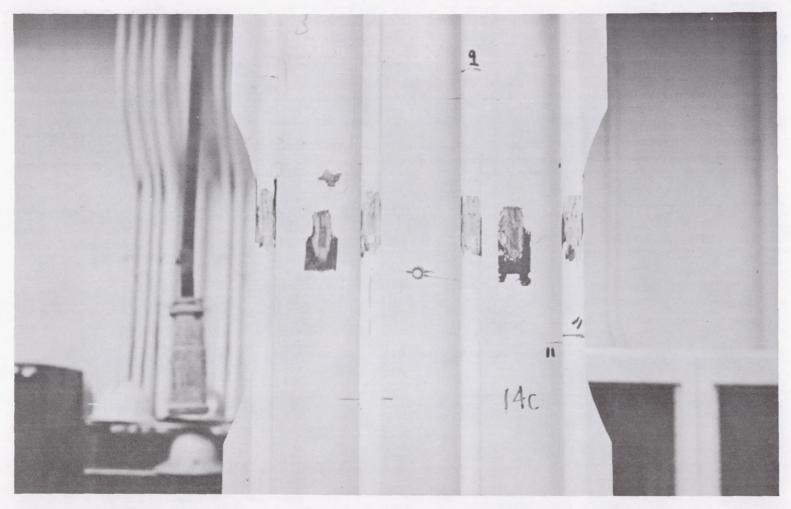


FIGURE F-54 INSIDE STIFFENER SHEET AFTER CRACK GROWTH TEST - PANEL #14C

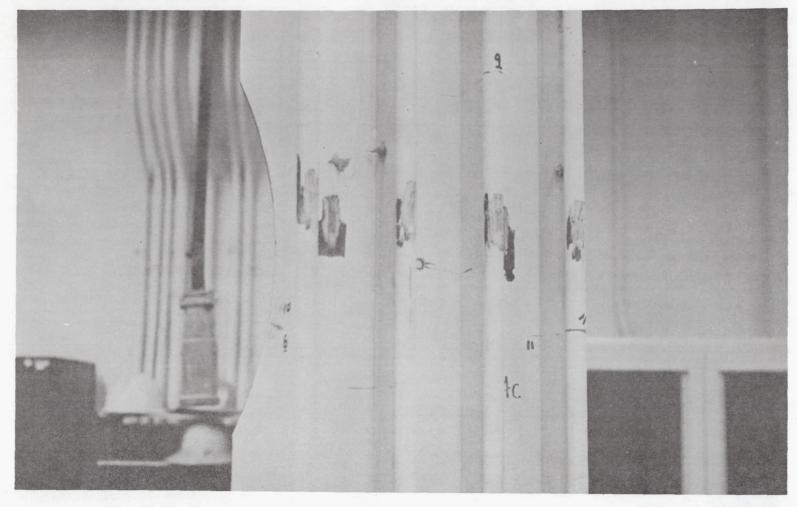


FIGURE F-55 INSIDE STIFFENER SHEET AFTER CRACK GROWTH TEST - PANEL #14C (LEFT OF CENTERLINE)

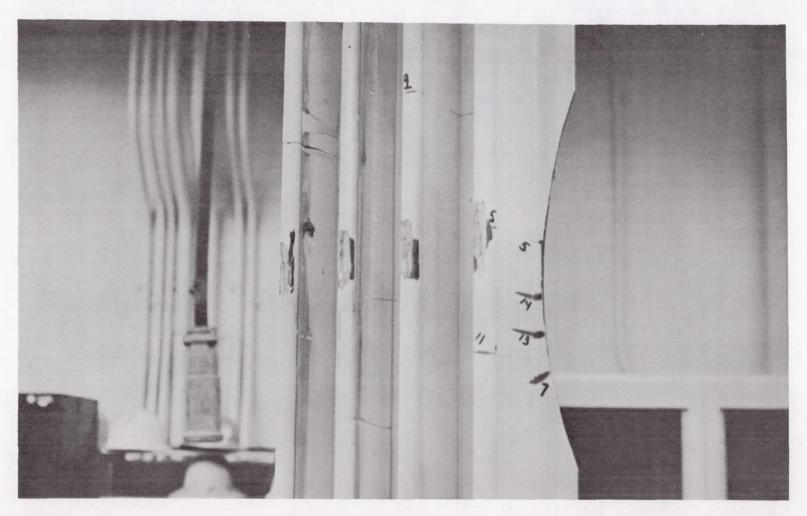


FIGURE F-56 INSIDE STIFFENER SHEET AFTER CRACK GROWTH TEST - PANEL #14C (RIGHT OF CENTERLINE)

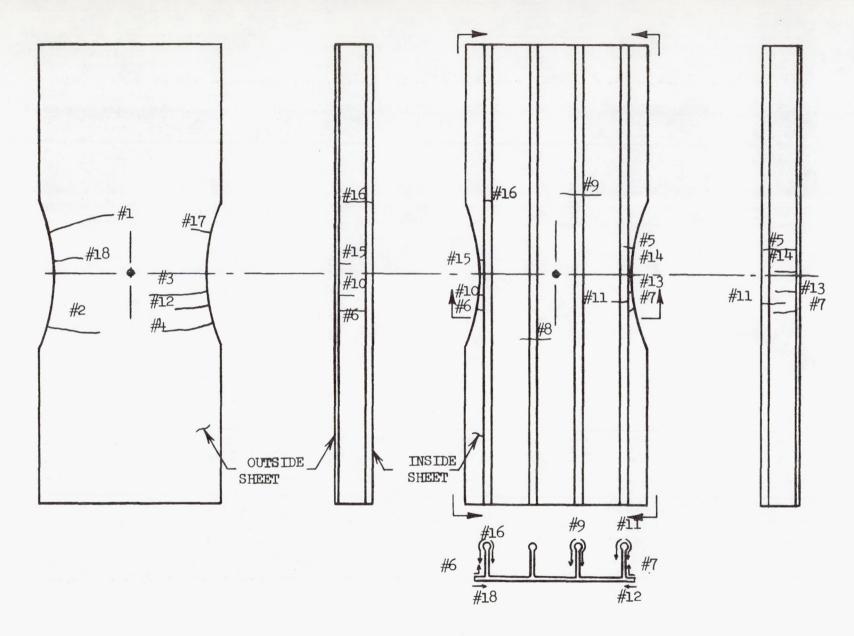


FIGURE F-57 SECONDARY CRACK LOCATIONS ALUMINUM GRAPHITE PANEL #14C

Panel	<b>No.</b> 14C		Primary Crac	ck Lengths (inches)		y Crack Leng ters (inches	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside		
4 Hz	10,000		5.1 ( .20)				
1	13,000	5.1 ( .20)				<u> </u>	
	14,000		6.4 ( .25)				
	17,600	6.4 ( .25)	7.6 ( .30)				
	23,000	7.6 ( .30)	8.9 ( .35)				
	27,000			5.1 ( .20)			
	27,800	8.9 ( .35)				1.0	
	28,500		10.2 ( .40)				
	33,700			6.4 ( .25)			
	35,100	10.2 ( 40)	11.5 ( .45)				
	39,000	11.5 ( .45)				 	
4 Hz	39,953		*	- x	5.1 ( .20)	 	
5 Hz	41,333		12.7 ( .50)				
	41.733			7.6 ( .30)		· · · · · · · · · · · · · · · · · · ·	· · · · · · · · · · · · · · · · · · ·
	45.783	12.7 ( .50)				 	
	46,353		14.0 ( .55)				
	52.553	14.0 ( .55)			6.4 ( .25)		
	52.653	1		8.9 ( .35)			
	53,153		15.2 ( .60)			 	1
	59,953	15.2 ( .60)	16.5 ( .65)	10.2 ( .40)		 	
	63.853	16.5 ( .65)					
	65.353	l			7.6 ( .30)		
	66.753		17.8 ( .70)				
	69,053	17.8 ( .70)					
	71,153	<u> </u>	19.0 ( .75)				
	75,353	19.0 ( .75)	:			 	
	76,653			11.5 ( .45)		 	
LL	78,453		20.3 ( .80)		8.9 ( .35)		
	82,553	20.3 ( .80)	21.6 ( .85)				
	82.653			12.7 ( .50)			
	89,753		22.9 ( .90)			 	
5 Hz.	89.953	21.6 ( .85)		L			<u> </u>

Panel 1	No. 14C		Primary Cra			Seconda millim	ry Crack Leng eters (inches	ths )
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	<i>#</i> 3	#5	#6
5 Hz	90,153			14.0 ( .55)	10.2 ( .40)			
	100,953	22.9 ( .90)						
	101,053		24.1 ( .95)					
	105,253			15.2 ( .60)	11.5 ( .45)		4	
	107,353	24.1 ( .95)						
	111,953		25,4 (1.00)					
	114.953	25.4 (1.00)						
	121.853		26.7 (1.05)					
	122,853	26.7 (1.05)						
	126,253			16.5 ( .65)	12.7 ( .50)			
	126.453		27.9 (1.10)					·
	134,253	27.9 (1.10)	29.2 (1.15)					
	134,953				14.0 ( .55)			
	135,753			17.8 ( .70)				
	137,253	29.2 (1.15)						
	144,647	L	30.5 (1.20)					
	145.347	30.5 (1.20)						
	150.047				15.2 ( .60)			
	151.047	<u> </u>		19.0 ( .75)			·	
	151,247	31.8 (1.25)	02 0 /2 05					
	155,047	(2 22)	31.8 (1.25)					
<del></del>	158,047	33.0 (1.30)						
	160,047	1 0 (3 0E)	33.0 (1.30)					
<b></b>		34.3 (1.35)		20.3 ( .80)				
$\vdash$	169,047		01 0 (5 05)	20.3 ( .80)				
<u> </u>	172,247	<b></b>	34.3 (1.35)			12.7 ( .50)		
<b></b>	180,347			07 6 / 051	36 5 / (5)		07.0 (7.70)	F2 ( / 22)
	180,390	05 ( (2 1/2)	05 ( /2 1/0)	21.6 ( .85)	16.5 ( .65)		27.9 (1.10)	7.6 ( .30)
<del></del>		35.6 (1.40)	35.6 (1.40)					
<del> </del>	185.947	36.8 (1.45)	36.8 (1.45)				00 0 (2.35)	<del></del>
5 Hz	186,047 193,047					16.5 ( .65)	29.2 (1.15)	-8.9 ( .35)

Panel	No. 14C		Primary Cramillimeters			Secondary Crack Lengths millimeters (inches)		
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#3	<b>#</b> 5	#6
5 Hz	193,547			22.9 ( .90)			·	
4	194,647	38.1 (1.50)						
	195,047		38.1 (1.50)					
	200,047						30.5 (1.20)	_
	207,747	39.4 (1.55)	39.4 (1.55)					
	208.547			24.1 ( .95)				
	208.647		ar and a second			21.6 ( .85)		
	209,747						31.8 (1.25)	
	209.847							10-2 ( .40)
	215.647	40.6 (1.60)						
	216,647							12.7 ( .50)
	217,847			25.4 (1.00)				
	224,047		40.6 (1.60)	;				
	229.947	41.9 (1.65)	41.9 (1.65)					
	230,647						33.0 (1.30)	14.0 ( .55)
	231,047					25.4 (1.00)		
	234,300				17.8 ( .70)			<b>.15.</b> 2 ( <b>.6</b> 0
	235,300	43.2 (1.70)	43.2 (1.70)					
	238,100			26.7 (1.05)				
	243,100	44.4 (1.75)	44.4 (1.75)					
	243,500					30.5 (1.20)		
	244,300				19.0 ( .75)			
	245,300			27.9 (1.10)				
	246,300						34.3 (1.35)	
	246,800							17.8 ( .70
	254,700	45.7 (1.80)	45.7 (1.80)					
	261,100							19.0 ( .75
	261,300			30.5 (1.20)				
	262,900					35.6 (1.40)		
	264,300				20.3 ( .80)		35.6 (1.40)	
	267.300	47.0 (1.85)	47.0 (1.85)					
5 Hz	270,300			31.8 (1.25)				

Panel	No. 14C		Primary Cra				ry Crack Leng eters (inches	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#3	#5	#6
5 Hz	271,300	48.3 (1.90)	48.3 (1.90)					
	274,800				21.6 ( .85)			
	275,800			33.0 (1.30)				
		49.5 (1.95)	49.5 (1.95)			***************************************		
	283,100	1 2227						20.3 ( .80)
	283,300			34.3 (1.35)	22.9 ( .90)		36.8 (1.45)	
	284 500					39.4 (1.55)		
	288,300		50.8 (2.00)					
	290,100	50.8 (2.00)						
	291,600		52.1 (2.05)	35.6 (1.40)	24.1 ( .95)			22.9 ( .90)
	294,300	52.1 (2.05)						
	298,700		53.3 (2.10)					
	299,500	53.3 (2.10)						
	305,600	54.6 (2.15)						
	306,500			36.8 (1.45)				
	306,600						39.4 (1.55)	
	306,800				25.4 (1.00)			
	307,300							25.4 (1.00)
	308,700		54.6 (2.15)					
	312,300	55.9 (2.20)						
	316,200		55.9 (2.20)					
	316,800					48.3 (1.90)		
	317,900			38.1 (1.50)				
	320,163	57.2 (2.25)		39.4 (1.55)	26.7 (1.05)			
	323,163							26.7 (1.05)
	323.963	58.4 (2.30)	57.2 (2.25)					
	326,163	59.7 (2.35)						
	326,263			40.6 (1.60)				
	328,163		58.4 (2.30)					
	330.163						40.6 (1.60)	
		61.0 (2.40)			· · · · · · · · · · · · · · · · · · ·			
5 Hz	331, 163					50.8 (2.00)		

Panel :	No. 14C	4	Primary Cramillimeters	ck Lengths (inches)			ry Crack Leng eters (inches	
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#3	#5	#6
5 Hz	333,963		59.7 (2.35)					
1	334,963			41.9 (1.65)	, .			
	335,463				27.9 (1.10)	The second second		
	338.163	62.2 (2.45)					a grand	
	340,163						43.2 (1.70)	
	340,663		61.0 (2.40)					
	344,163	63.5 (2.50)	62.2 (2.45)		29.2 (1.15)			
	344.663						44.4 (1.75)	
	350,163	64.8 (2.55)	63.5 (2.50)				CRACK	
	351 .163			43.2 (1.70)			STOPPED	29.2 (1.15)
	352,163				30.5 (1.20)		IN BULB	
	357,163	66.0 (2.60)	64.8 (2.55)				. RADIUS	
	357.663				31.8 (1.25)			
	361 282	67.3 (2.65)	OVERLAPS #3	44.4 (1.75)	33.0 (1.30)	57.2 (2.25)	<u> </u>	<u> </u>
	367,963	68.6 (2.70)	SECONDARY			OVERLAPS R/H		<u> </u>
	371.463		CRACK		34.3 (1.35)			
	374.563	69.8 (2.75)				MARY CRACK		<u> </u>
	376.663				35.6 (1.40)			
	377,163			45.7 (1.80)				30.5 (1.20)
	378.163	71.1 (2.80)	68.6 (2.70)		·			
	382,163		69.8 (2.75)					
	383,163	72.4 (2.85)						
	385.163					64.8 (2.55)	<u> </u>	
	387,763		71.1 (2.80)					
	390,163	73.7 (2.90)						<u> </u>
	392,163			47.0 (1.85)				<u> </u>
	393,163	74.9 (2.95)				ļ		<del> </del>
	394.663			<u> </u>	36.8 (1.45)			
	398,963		72.4 (2.85)				<b> </b>	<b>↓</b>
	400, 363		<u> </u>		38.1 (1.50)			<b></b>
		76.2 (3.00)	<u> </u>	L		<b></b>	11 1 12	<del> </del>
5 HZ	401.563	1	73.7 (2.90)	I	<u> </u>	<u> </u>	44.4 (1.75)	<u> </u>

Panel	No. 14C		Primary Cra			36.8 (1.4) 72.4 (2.85) 38.1 (1.5)		
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#3	#5	#6
5 Hz	407.763	77.5 (3.05)	74.9 (2.95)				44.4 (1.75)	
	408,176			48.3 (1.90)	39.4 (1.55)	68.6 (2.70)		35.6 (1.40)
	412,176				40.6 (1.60)		•	
		78.7 (3.10)						
	414,976				41.9 (1.65)			
	415.776		76.2 (3.00)					
	420,176			49.5 (1.95)				
1.5	423,576		77.5 (3.05)					
	423,726	80.0 (3.15)						
	424.176				43.2 (1.70)			
	424,676							36.8 (1.45)
	425,576					72.4 (2.85)		
	428,176		78.7 (3.10)					
	430,676	81.3 (3.20)						
	431.176				44.4 (1.75)			
	432.176			50.8 (2.00)				
	436,176				45.7 (1.80)			38.1 (1.50)
	437,176		80.0 (3.15)					
	437.676	82.6 (3.25)						
	441,576	83.8 (3.30)	81.3 (3.20)					
	443.076							
	443,351			52.1 (2.05)				39.4 (1.55)
		85.1 (3.35)						
	446.676							,
	446.976		82.6 (3.25)					•
	447,176					77.5 (3.05)		
	448,276				47.0 (1.85)			
	450,576	86.4 (3.40)						
	453,076		83.8 (3.30)					
	453,476	87.6 (3.45)						
	454,176 455,176							
5 Hz	455,176			53.3 (2.10)			44.4 (1.75)	

Panel	No. 14C	Secondary Crack Lengths millimeters (inches)								
Cycle Rate	No. of Cycles	#11	#18							
5 Hz	407,763							L		
	408,176									
	412,176									
	412,576									
	414.076									
	415.776									
	415,776 420,176 423,576					<b> </b>				
	423,576									
	423,726					<b></b>		<u> </u>		
	424,176							<del></del>		
	424.676							<del></del>		
	425,576					ļ		<del></del>		
	428,176							<del></del>		
	430,676 431,176					<del></del>				
	431,176					ļ				
	432,176					ļ				
	436,176			· · · · · · · · · · · · · · · · · · ·		<del> </del>				
	437,176					<del> </del>	<del> </del>			
	437,676					<del></del>	<del></del>			
	441,576 443,076		31.07 55			<del> </del>		<u> </u>		
	443,076		14.0 ( .55)			<del>                                     </del>		<u> </u>		
	1443,351					<del> </del>	<del> </del>			
	443,351 446,476 446,976 446,976 447,176 448,276		15 0 / 60\			1	<del> </del>	<del> </del>		
	1446,076		15.2 ( .60)			<del> </del>	<del> </del>	<del> </del>		
	449,976	10. 1. (2 FE)				<del> </del>		<del> </del>		
	1446 - 176	44.4.(1.75)				<del> </del>		<u> </u>		
	1440,510						1	<del> </del>		
	1450.570					<del> </del>		1		
<del></del>	453,076 453,476	<b> </b>	,							
	1453,476	<del> </del>	17.8 ( .70)	<del> </del>						
5 -	454,176 455,1 <b>7</b> 6	<u> </u>	-1.0 ( .)0/			1				
5 Hz	1455,176	<del></del>	L		<del></del>					

Panel .	No. 14C		Primary Cra millimeters			Seconda millim	ry Crack Leng eters (inches	ths )
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#3	#5	#6
5 Hz	456,176				48.3 (1.90)		44.4 (1.75)	
	459,176						4	:
	459,576		85.1 (3.35)					
	465,076	88.9 (3.50)						
	465,176		86.4 (3.40)					
	465.376				49.5 (1.95)			
	466.476	<u> </u>						40.6 (1.60)
	469.376	90.2 (3.55)	87.6 (3.45)	54.6 (2.15)		80.0 (3.15)		
	471,176							
	474.876	91.4 (3.60)						
	477,476							
	478,576		88.9 (3.50)					
	479.176	92.7 (3.65)					'	
	479.776				50.8 (2.00)		·	
	480,676			55.9 (2.20)				
	484.176							
	485.476	94.0 (3.70)	90.2 (3.55)					
	487,676				52.1 (2.05)			
	488,176							
	1488,376		91.4 (3.60)		·			
		95.2 (3.75)						
	494,476	OVERLAPS #18	92.7 (3.65)					
	494,776				F 48 .			
	499.368	CRACK	94.0 (3.70)	57.2 (2.25)		83.8 (3.30)		41.9 (1.65)
	505.268	97.8 (3.85)	95.2 (3.75)	58.4 (2.30)	53.3 (2.10)			•
	510,368	99.1 (3.90)				86.4 (3.40)		
	512,368		96.5 (3.80)					
		100.3 (3.95)						
	517,768		97.8 (3.85)					
	517.868			59.7 (2.35)	54.6 (2.15)			44 4 (1.75)
	519.368							
5 Hz	523,268	101.6 (4.00)	99.1 (3.90)				44.4 (1.75)	

Panel	No. 14C	Secondary Crack Lengths millimeters (inches)									
Cycle Rate	No. of Cycles	#11	#18								
5 Hz	456,176		·								
- i	450 176	45.7 (1.80)			Α						
<del></del>	459,576	<del>-1</del> 7.1 (1.00)	1								
	465,076										
	465,176		19.0 ( .75)								
	465,376						<b>]</b>				
	466.476										
	469.376	47.0 (1.85)					<u> </u>				
	471,176 474,876		20.3 ( .80)					<u> </u>			
	474,876	ļ	( ( 0=)								
	477,476		21.6 ( .85)								
	478,576				ļ						
	479,176	<u> </u>			<del> </del>						
	479,776 486,676										
	1 480,676		24.1 ( .95)		<del> </del>						
	484,176 485,476		Z-4. Z ( . 37)		<del>                                     </del>						
	487.676										
	488.176		24.5 (1.00)								
	488.376										
	1,02,076	-	OVERIAPS L/H				<u> </u>	ļ			
	492,976 494,476	<del>                                     </del>	OUTSIDE PRI-								
<del></del>	494,776	49.5 (1.95)	MARY CRACK					<b></b>			
<b></b>	499.368	50.8 (2.00)					<b>↓</b>				
	505,268		30.5 (1.20)								
	510,368						<del> </del>	<b>}</b>			
	512,368							<del> </del>			
	516.368				<u> </u>		<del> </del>				
	517.768						<del></del>	+			
	517,868	53.3 (2.10)	× /					<del> </del>			
Y	519.368	<u> </u>	31.8 (1.25)				<del> </del>	<del>                                     </del>			
5 Hz	523.268	3 [	<u> </u>	<u> </u>			<del></del>				

Panel :	No. 14C		Primary Cra	ck Lengths (inches)		Secondary Crack Lengths millimeters (inches)			
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	<i>#</i> 3	#5	#6	
5 Hz	526,568	102.5 (4.05)					44.4 (1.75)		
A	526,868						Ā		
	527,068 527,368				55.9 (2.20)				
	527,368		70.05	61.0 (2.40)		00 - /		45.7 (1.80)	
5 胚	530,897		100.3 (3.95)	62.2 (2.45)	57.2 (2.25)	88.9 (3.50)	44.4 (1.75)		
	CRACK								
TE	GTHS	102.5 (4.05)	100.3 (3.95)	62.2 (2.h5)	57.2 (2.25)	88.9 (3.50)	<u>հի ի (1.75)</u>	45.7 (1.80)	
	RESI	DUAL STRENGTH	= 279,300	N (62,800	lbs.)				
	Ī								

Panel	No. 14C	Secondary Crack Lengths millimeters (inches)							
Cycle Rate	No. of Cycles	#11	#11 #18						
5 Hz	526,568 526,868 527,068 527,368 530, <b>89</b> 7				·				
1	526,868		33.0 (1.30)						
	527,068								
Y	527,368	54.6 (2.15)	ol				<del></del>		
5 Hz	530,897		34.3 (1.35)					-	
<del></del>									
				<del></del>					
FINAL	CRACK GTHS		-1 A /4 AF1		i				
LEI	GTHS	54.6 (2.15)	34.3 (1.35)						
·									
	<b></b>								
	<del> </del>							·	
	<b>†</b>					- 1 A			
	2.3			·					
		:			<b></b>				
<u></u> .	ļ								
	<del> </del>		<b> </b>	<del> </del>		•		3	
	<del> </del>								
<del></del>							•		
								<b></b>	
								<b></b>	
		<u> </u>			<del> </del>			<del> </del>	
-	<del></del>	<u> </u>				-		<b> </b>	
	<del> </del>		<del> </del>						
	<del>                                     </del>		<del>                                     </del>						
								ļ	
					<u> </u>	<u> </u>	<u> </u>	<u> </u>	

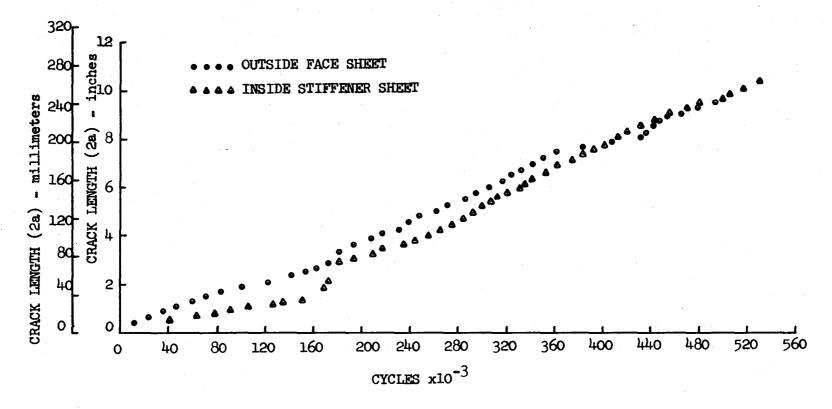


FIGURE F-58 CRACK GROWTH CURVE FOR ALUMINUM-GRAPHITE PANEL #14C (Crack Length Includes Secondary Cracks)

### PANEL #15C

MATERIALS: ALUMINUM-GRAPHITE

ADHESTVE: EA-927R

ALUMINUM STRESS: 116 MN/m<sup>2</sup> (16.9 ksi)

MAXIMUM FATIGUE LOAD: 144,560N (32,500 lbf)

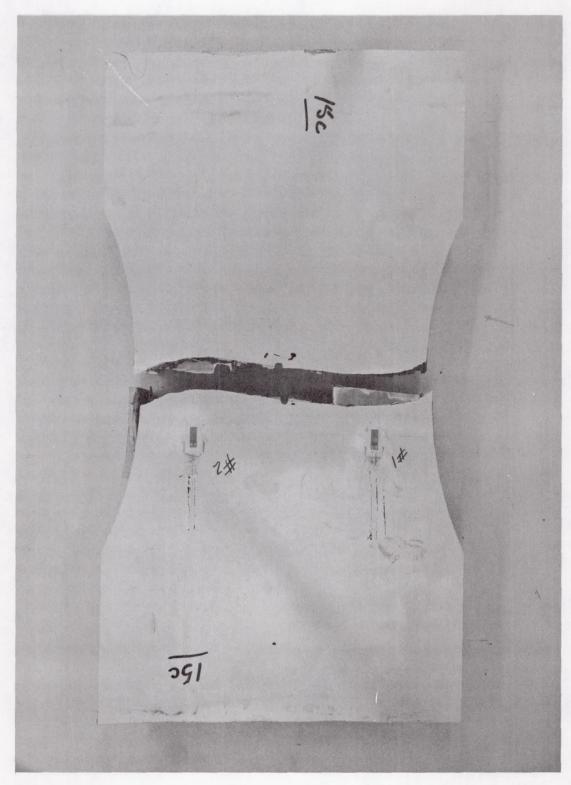


FIGURE F-59 OUTSIDE FACE SHEET AFTER RESIDUAL STRENGTH TEST - PANEL #15C

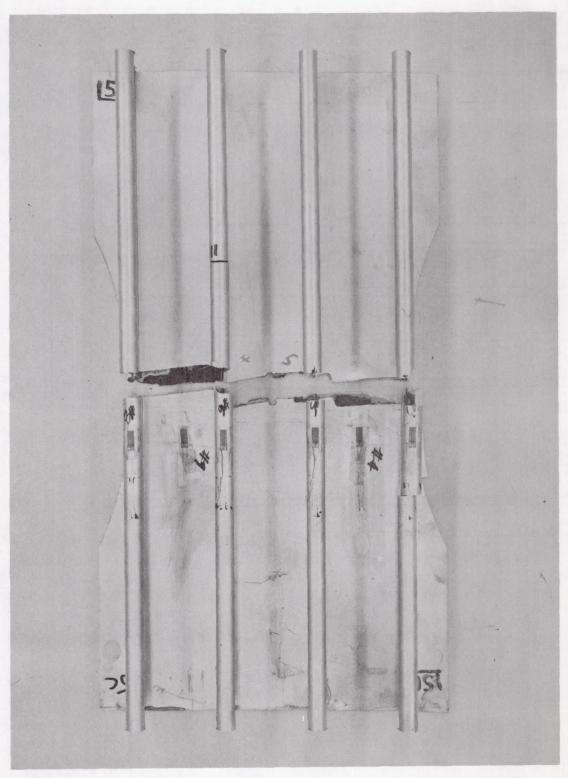


FIGURE F-60 INSIDE STIFFENER SHEET AFTER RESIDUAL STRENGTH TEST - PANEL #15C

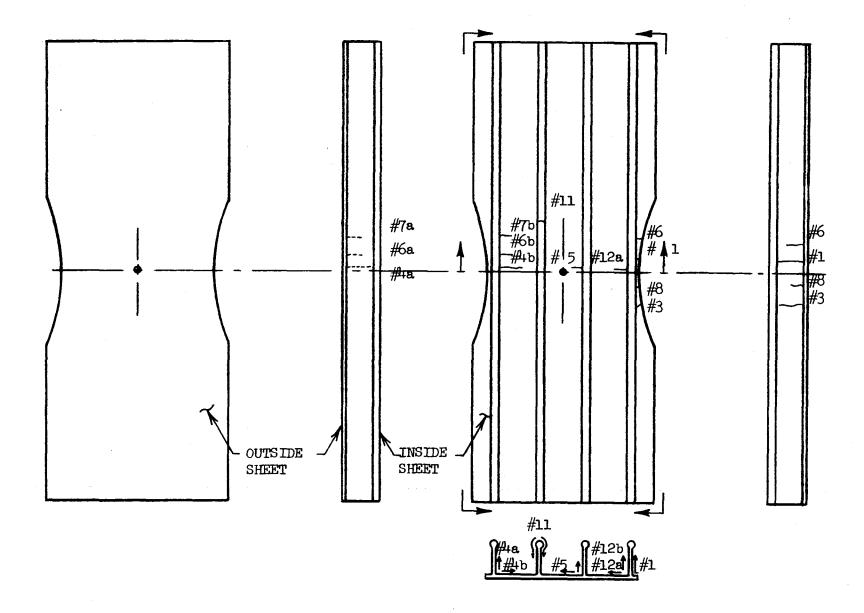


FIGURE F-61 SECONDARY CRACK LOCATIONS ALUMINUM-GRAPHITE PANEL #15C

Panel	Panel No. 15C Primary Crac			k Lengths (inches)			ry Crack Leng eters (inches	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside			
5Hz	33,009	3.8 ( .15)	3.8 ( .15)					
. 1	34,009		6.0	3.8 ( .15)	5.1 ( .20)			
<del>-                                    </del>	36,809	5.1 ( .20)	Ne 2 2					
	37.569			5.1 ( .20)			<u> </u>	
	38,169		5.1 ( .20)					
	41,489				6.4 ( .25)			
	42.020	6.4 ( .25)						
	44,309		6.4 ( .25)			<del></del>		
	47,739			6.4 ( 25)				
	49,289		7.6 ( .30)		7.6 ( .30)		<u> </u>	
	51.089			7 ( ( 20)	1.0 1 .307			
	55.519		2 ( 25)	7.6 ( .30)				
	56,179		8.9 ( .35)					
	62,609		10.2 ( .40)					
	65,469		10.2 ( .40)	8.9 ( .35)	10.2 ( .40)			
<b> </b>	70,659	11.4 ( .45)		0.9 ()	10.2 (	4,5,		
			11.4 ( .45)					
	72,609	12.7 ( .50)	12.7 ( .50)	10.2 ( .40)	11.4 ( .45)			
	114,150		12.7 ( .50)	10.2 1 - 74/	12.7 ( .50)		ورياد داري المواد	1.11
<b> </b>	123,060 124,170	14.0 ( .55)	14.0 ( .55)	11.4 ( .45)				
<b></b>	129,240		14.0 ( .221	12.7 ( .50)	14.0 ( .55)			
	134.130		15.2 ( .60)	1201 ( 323)	m to the second			
	142,320		16.5 ( .65)					
1	150,550		1		15.2 ( .60)			
<del> </del>	157,450			14.0 ( .55)			1.77	<b></b>
<del>                                      </del>	158.857				200		<u> </u>	<u> </u>
1-1-	161,557		17.8 ( 70)					
1	165,957		1					<b></b>
	166.157	<del>- 121 - 12</del>			16.5 ( .65)			
	178,857	7		15 2 ( 60)			<del> </del>	
5Hz	181.157	<b>20.</b> 3 ( .80)		L	<u> </u>		<u> </u>	

Panel	No. 15C		Primary Crack Lengths Secondary Crack Lengths millimeters (inches) millimeters (inches)					
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	<i>#</i> 1	#\a.	#\b
5Hz	181,257		19.0 ( .75)			÷		
5Hz	181,657				17.8 ( .70)		and the second	
6Hz	189,787							
J.I.L	189,789			16.5 ( .65)				
1	198,357		20.3 ( .80)					
	200,257				19.0 ( .75)			
	204.657	22.9 ( .90)						
	213,657	24.1 ( .95)	21.6 ( .85)					
	213,757				20.3 ( .80)			
	213.857			17.8 ( .70)				
	222,627		22.9 ( .90)					
	222,757	25.4 (1.00)						
	234,457	26.7 (1.05)						· · · · · · · · · · · · · · · · · · ·
	234,557		24.1 ( .95)					
	235,557			19.0 ( .75)	21.6 ( .85)			
	238.857				` `	15.2 ( .60)	16.5 ( .65)	17.8 ( .70)
	249,257		25.4 (1.00)					
	252,357	27.9 (1.10)			<u></u>			
	252,957			20.3 ( .80)	22.9 ( .90)			
	255,311	29.2 (1.15)	26.7 (1.05)			17.8 ( .70)	17.8 ( .70)	
	257,311				24.1 ( .95)			
	258,311				<u></u>			19.0 ( .75)
	263,881					19.0 ( .75)		
	263.890	30.5 (1.20)	27.9 (1.10)					
	273,311			21.6 ( .85)	25.4 (1.00)	20.3 ( .80)	19.0 ( .75)	21.6 ( .85)
	287.511			22.9 ( .90)	26.7 (1.05)			
	287.711					ļ.,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,	21.6 ( .85)	22.9 ( .90)
	287.811			<u> </u>	<u> </u>	21.6 ( .85)		
	288,311	33.0 (1.30)	30.5 (1.20)	L	<b></b>			
	297,211	34.3 (1.35)	31.8 (1.25)	<u> </u>				
	299,311				57.2 (2.25)			
6HZ	304.111			<u></u>	1	22.9 ( 90)	<u> L</u>	

Panel	<b>No.</b> 15C		Primary Cra millimeters	ck Lengths (inches)			ry Crack Leng eters (inches	
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1	#4a.	#4b
бHz	304,811			24.1 ( .95)				25.4 (1.00)
OHZ	307,175	35.6 (1.40)					and the second	
7	310,875	3213 (=131)	33.0 (1.30)				tur y	and the second
	316,175	36.8 (1.45)	33-1		58.4 (2.30)	24.1 ( .95)	25.4 (1.00)	26.7 (1.05)
	325,675	38.1 (1.50)	34.3 (1.35)					
	325,875	JO. 2 (2. 7. 7. 7. 7. 7. 7. 7. 7. 7. 7. 7. 7. 7.	7 • 7 • 1	24.5 (1.00)	59.7 (2.35)			
	326.075			er ut			27.9 (1.10)	
	335.675	39.4 (1.55)	35.6 (1.40)		* * * * * * * * * * * * * * * * * * * *			
	336,175			26.7 (1.05)	61.0 (2.40)			
	349 175	40.6 (1.60)	36.8 (1.45)		63.5 (2.50)	26.7 (1.05)		30.5 (1.20)
	358 175					<u> </u>	29.2 (1.15)	
	358 275			• •	64.8 (2.55)			
	359,175	41.9 (1.65)	38.1 (1.50)					
	268,175							33.0 (1.30)
	368 875			27.9 (1.10)			* * * * * * * * * * * * * * * * * * * *	<u> </u>
	368,925		39.4 (1.55)					
	368,935					27.9 (1.10)	30.5 (1.20)	
	370,975	43.2 (1.70)				entry transfer	Associated the second	<del></del>
	379,675		40.6 (1.60)			the second of the second		
	380,075	44.4 (1.75)				<u> </u>		
	380,575			29.2 (1.15)	67.3 (2.65)		20 0 /2 00	25 ( /2 ):0
$\Gamma \Gamma$	381,175						33.0 (1.30)	35.6 (1.40
	392,175	1						
	393.875	45.7 (1.80)	41.9 (1.65)			and the second second	the second second second	
	394,175				68.6 (2.70)	1 00 0 71 1EV	1 31 35	10 6 17 60
6Hz	399.467			<u> </u>	<b></b>	29.2 (1.15)	34.3 (1.35)	40.6 (1.60
5Hz	402,937		43.2 (1.70)	30.5 (1.20)	<b></b>		<del> </del>	
	408,967		44.4 (1.75)	1				e se esta a la
	400.067			<del> </del>	100 00 ==		<del> </del>	12 0 /1 70
100	418,467		<u></u>	1	69.8 (2.75)	ļ	35.6 (1.40)	43.2 (1.70
	419,167		1,		<del> </del>	<del> </del>	35.6 (1.40)	
5Hz	1122,367		45.7 (1.80)		1			<u> </u>

Panel	No. 150			Secondar millime	y Crack Lengt ters (inches)	hs		
Cycle Rate	No. of Cycles	#12 <b>a</b>	#12b					
бнz	304,811							
	307,175							
-	310,875							
<del>-  </del>	316,175							
	325,675							
	325,875						<u> </u>	
	326.075							
	335.675				<u> </u>			
	336,175							<u> </u>
	349,175				<u> </u>	<b></b>	ļ	<u> </u>
	358 <b>.1</b> 75				<del> </del>		<u> </u>	
	358,275				<u> </u>	<u> </u>	<del> </del>	
	359,175 368,175 368,875				<u> </u>		<u> </u>	
	368,175					<del></del>	<b></b>	
	368,875			<u></u>		<del> </del>	<del> </del>	<b></b>
	368,925				<u> </u>		<del> </del>	<u> </u>
	368,935 370,975				<del>                                     </del>		<del> </del>	<u> </u>
	370.975				<del></del>		<del> </del>	<del> </del>
	379,675			· · · · · · · · · · · · · · · · · · ·	<del> </del>			
	380,075				<u> </u>			
	380,575	L		<del></del>	<del> </del>		<del> </del>	
	381,175	<u> </u>			<del></del>	<u> </u>		<del> </del>
	392,175	8.9 ( .35)	10.2 ( .40)		<del></del>	<del>- </del>	<del> </del>	
	393,875		ļ					<del> </del>
	1394,175	10.2 ( .40)			<del>                                     </del>			<del> </del>
6Hz					<del></del>	+	<del> </del>	
5Hz	402,937				+		<del> </del>	
	408,967	<b></b>	10 5 / 52		<del> </del>		<del>                                     </del>	<del>                                     </del>
	409,967 418,467	<del>  , _ , _ , _ , _ , _ , _ , _ , _ , </del>	12.7 ( .50)			<del></del>	<del> </del>	<u> </u>
	418,467	11.4 ( .45)			+		<del> </del>	<del>                                     </del>
	419,167		<u> </u>		+	<del> </del>		1
OHZ	1422,367	<u></u>	L	<u> </u>		<del></del>	<u> </u>	<u> </u>

Panel	No. 15C		Primary Cra millimeters			1 (2.80) 36.8 (1.45) 4.4 (2.85) 7 (2.90) 38.1 (1.50) 48.3 (2.95)		
Cycle Rate	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1	#Ha.	#4b
5Hz	422,467	48.3 (1.90)		e de la companya de		ar .		
112	429,967							
-1	430,667			31.8 (1.25)	71.1 (2.80)			
	431,267						36.8 (1.45)	44.4 (1.75)
	1131 317	49.5 (1.95)					, , , , , ,	
	1130 1167	1202 (=0)21	47.0 (1.85)					
	439,467 440,967					30.5 (1.20)		4
	443,767			33.0 (1.30)	72.4 (2.85)			
	444,467	50.8 (2.00)						
	454,467	Section 1	48.3 (1.90)					
	455,267	<b>52.1 (2.05)</b>						
	455,467		-	34.3 (1.35)	73.7 (2.90)			
	156 267				1			47.0 (1.85)
	456,467						38.1 (1.50)	
	460.467		49.5 (1.95)		<u> </u>			
	461,267							48.3 (1.90
	T461,567			35.6 (1.40)				
	461,767		<u> </u>	l	74.9 (2.95)			<del></del>
	469,467		<u> </u>	36.8 (1.45)	76.2 (3.00)			49.5 (1.95
	1470,467	53.3 (2.10)			Crack			
	476,867			<u> </u>	Stopped	<u> </u>		
	478,467	54.6 (2.15)	50.8 (2.00)		in Bulb			
	479,267				Radius		39.4 (1.55)	50.8 (2.00)
	483,765						<del></del>	
	489,465	55.9 (2.20)	52.1 (2.05)			<b></b>		
	490.965				<u> </u>	1 22 0 72 06	<b>↓</b>	FO 7 70 05
	491,765				<u> </u>	31.8 (1.25)	100 ( 10 (0)	52.1 (2.05
	491,965			20 2 /2 =23		<del> </del>	40.6 (1.60)	
	492,065		<b></b>	38.1 (1.50)	1.		<b> </b>	
		57.2 (2.25)		<u> </u>	<u> </u>	<del> </del>	<b></b>	
<u> </u>	501,765		53.3 (2.10)		ļ	<del> </del>		
5Hz	1502,565	58.4 (2.30)		<u> </u>	<u> </u>	<u> </u>	1	

Panel	No. 15C	Secondary Crack Lengths millimeters (inches)						
Cycle Rate	No. of Cycles	#12 <b>a</b>	#12b					
5Hz	422,467							
	429,967	14.0 ( .55)	15.2 ( .60)					•
	430,667							
	431.267							
	434,317						<u> </u>	
	440.967	15.2 ( .60)	16.5 ( .65)			<u> </u>	<u></u>	
	443.767 444,467				<u> </u>		ļ	
	444,467	<u> </u>						<del></del>
	454,467					<u> </u>	ļ	
	455,267						<del>                                      </del>	<del> </del>
	455,467	16.5 ( .65)	19.0 ( .75)				<u> </u>	<u> </u>
	456,267							<del> </del>
ļ	456,467							
<b></b>	460,467							
<del>                                     </del>	461,267	ļ				<del> </del>	<u> </u>	
<del>-   -</del>	461,567 461,767							<del></del>
<del>-  </del>	469,467			<del></del>			<u> </u>	
┝╾┾╼╼	470,467	19.0 ( .75)	20.3 ( .80)				<u> </u>	
<del></del>	476.867		20.5 ( .00)			<u> </u>		
<del>  </del>	478,467	20.3 ( .00)	21.6 ( .85)					
<del>  </del>	479,267	<del> </del>	<u> </u>				<u> </u>	
<del> </del>	483.765	21.6 ( .85)	22.9 ( .90)				<u> </u>	
	489,465	21.0 ( .02)	22.9 ( . )0/	<del></del>				
<del></del>	490,965	<del>                                     </del>	25.4 (1.00)			<u> </u>		
<del>                                     </del>	491.765			· · · · · · · · · · · · · · · · · · ·	1			
<del>                                     </del>	491.965	<del>                                     </del>						
<del>                                     </del>	492,065	<del>                                     </del>		· · · · · · · · · · · · · · · · · · ·				
	493.965							
	501,765							
5Hz	502.565							<u> </u>

Panel	No. 15C		Primary Cra	ck Lengths (inches)		Secondary Crack Lengths millimeters (inches)		
Cycle R <b>at</b> e	No. of Cycles	L/H Outside	R/H Outside	L/H Inside	R/H Inside	#1	# <b>4a</b>	#\b
5Hz	505,265			39.4 (1.55)				
5Hz	506,465			- 39.7 · ( 1.0 / ) / .				53.3 (2.10
FINAL	CRACK							
LENG	THS	58.4 (2.30)	53.3 (2.10)	39.4 (1.55)	76.2 (3.00)	31.8 (1.25)	40.6 (1.60)	53.3 (2.10
	RESIDUAL	STRENGTH =	351,400N	(79,000 lbs)				
·								
			1					
	ļ			<b>}</b>				
				· · · · · · · · · · · · · · · · · · ·				
	ļ							
<del></del>	<del> </del>							
		<u> </u>				ļ		
<del></del>	<del> </del>		<b> </b>	ļ	ļ			

Panel	No. 150	Secondary Crack Lengths millimeters (inches)							
Cycle Rate	No. of Cycles	#12a	#12b						
5 Hz	505,265								
5 Hz	506,465								
FINAT.	CRACK				<del></del>	<u> </u>	<b></b>	<del> </del>	
LENG	THS	21.6 (.85)	25.4 (1.00)						
						<u> </u>			
						<u> </u>	ļ	<b>}</b>	
	ļ						<u> </u>	<b></b>	
						<b></b>	<del>}</del>	<del></del>	
<del></del>		<del></del>				<del> </del>	<del></del>	<b></b>	
<del></del>						<del> </del>		<u> </u>	
	<del> </del>		<del></del>			<u> </u>	<del> </del>	<b>†</b>	
			<del> </del>			<u> </u>	<u> </u>		
		<u> </u>							
		<del></del>							
						<u> </u>			
					<b></b>	<b></b>			
	<b></b>	<del></del>			<b></b>	<b></b>			
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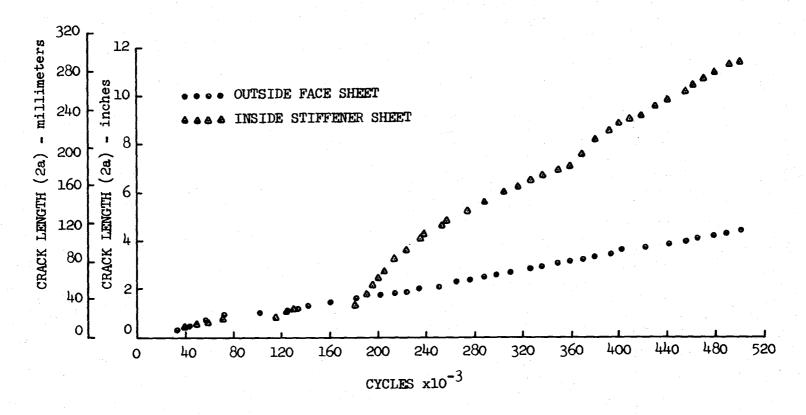


FIGURE F-62 CRACK GROWTH CURVE FOR ALUMINUM-GRAPHITE PANEL #15C (Crack Length Includes Secondary Cracks)

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